

FINITE ELEMENT SOLUTION OF THE WARPING TORSION USING THE  
EQUIVALENT BEAM-COLUMN ANALOGY IN MULTI-SPAN BEAMS

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## ABSTRACT

### FINITE ELEMENT SOLUTION OF THE WARPING TORSION USING THE EQUIVALENT BEAM-COLUMN ANALOGY IN MULTI-SPAN BEAMS

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A small, powerful, versatile and straightforward software application will be introduced to analyze problems of warping torsion by means of the analogy between the straight beam-column under mixed transverse and axial loading and the straight beam under restrained warping. This software, originally developed to analyze multi-span beam-columns immersed on elastic foundations, resorts to a high order finite element. Validation against a number of already solved cases will be provided, including, but not limited to the cases in the AISC-Design Guide 9. Additionally, the present work will provide a brief account on the first order general torsion theory, a discussion on the roles analogies play in sciences and engineering, an account on the equivalent beam-column analogy as special case of the equivalent beam on elastic foundation, some details of the aforesaid high-order-finite element, and a contribution of a novel convergence study regarding the axial tensile parameter.

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## LIST OF SYMBOLS

3DOF RE	rod element with 3 degrees of freedom,
6-DOF BC	beam column with 6 degrees of freedom, also 6-DOF BE
6-DOF BE	beam element with 6 degrees of freedom,
9-DOF BC	beam column with 9 degrees of freedom, also 9-DOF BE
9-DOF BE	beam element with 9 degrees of freedom,
$a$	$L/2$ , semi-length of finite element, or $\sqrt{EC_w/GJ} = 1/\lambda$ according to the AISC DG 9 nomenclature for TWB, in length units,
$A$	cross section area
AISC	American Institute of Steel Construction,
AISC-DG-9	Steel Design Guide Series No. 9,
$b$	a characteristic dimension of a TWB cross-section, width or height. The convex contact width in a Winkler problem, in,
$B$	bimoment, an abstraction in TWB theory, force·length <sup>2</sup> ,
[B]	strain matrix defined by the generic curvature, it is the 2 <sup>nd</sup> derivative of the shape function matrix [N],
BEF	beam on elastic foundation,
BC	beam-column,
BMCOL, BMCOLG	original name of beam-column software renamed BMTORSWP to undertake restrained warping problems,
BMTORSW	after beam-torsion-warping, 32-bit-software application,
BMTORSWP	beam-torsion-warping with Portland adaptation for 64 bits,
$C_w, Cd$	warping constant for the cross-section, length <sup>6</sup> ,
$d$	a characteristic dimension of a TWB cross-section, width or height,
$C_e$	dimensionless factor in the 9-DOF- BC matrix
DOF	degree of freedom,
DG-9	Steel Design Guide Series No. 9,
$E$	modulus of elasticity of material, force/length <sup>2</sup> ,
EBEF	equivalent beam on elastic foundation,
{g}	generic transversal field of displacements,
$G$	shear modulus of elasticity of material, force/length <sup>2</sup> ,
$I$	cross section inertia, length <sup>4</sup> ,
$I_f$	moment of inertia of one I-beam flange around the cross section minor axis,
$J$	torsional constant for the cross-section, length <sup>4</sup> ,
$JBW$	Half band width in a matrix,
$k_o$	soil normal modulus, a parameter for the BC, force/length <sup>3</sup> ,
$k$	soil normal modulus multiplied by the convex contact width $b$ , force/length <sup>2</sup> ,
$l$	$\pi/\lambda$ characteristic wave length for a EBEF used in rail trucks,
$L$	member length,
$Le$	maximum element length = $2\sqrt{EC_w/GJ}$ , in length units,
$M$	bending moment,

$N, NI, NJ$	node, near node, far node, where $I < J$ ,
$N$	axial force always in the direction of the original straight BC,
$[N], [N,x], [N,xx]$	shape function matrix and 2 derivatives along the axial coordinate $x$ in the finite element formulation,
$N_e$	minimum number of elements in a given span of length $L$ , an integer $> (L/Le) = 0.5L\sqrt{(GJ/EC_w)}$ in the TWB,
$p$	applied distortional load in force/length units,
$q$	transverse loading in the BC, force/length,
$q_i$	degree of freedom with the specific number $i$ ,
$\{q\}$	vector of specific (reference) displacements along the DOFs in finite element formulation containing all the $q_i$ ,
$Q$	applied shear in the BEF under mixed loading, force units,
$Q_v$	vertical shearing force, always perpendicular to the original straight BEF,
$Q_n$	shearing force normal to the deformed BEF elastic line,
$r$	a distance from the bar axis in a cylindrical bar,
$R$	radius of a cylindrical bar, length,
$T_w, T_2$	resisting moment due to restrained warping of the cross-section, force-length,
$T_t, T_1$	resisting moment of unrestrained cross-section, force times length,
TWB	thin-walled beam or bar,
TWM	thin-walled member,
$u, u_s$	displacement along the contour, lateral in the case of an I-beam flange, length,
$u, v$	generic axial and transverse displacements in the 9DOF element,
$u_r$	displacement perpendicular to the contour, length,
$V_f$	the shear occurring in each flange of an I-beam, force,
$w$	distributed constant load in lb/in or force/length units,
$W$	vector with powers of varying distributed loads in lb/in, a measure of distortion in box girder analysis,
$x$	$a\xi$ , member longitudinal coordinate,
$y, y', y'', y''', y''''$	longitudinal coordinate, and four derivatives,
$z$	longitudinal coordinate, length,
$\alpha_i$	generalized displacements, coefficients of the polynomial expressing the field of the generic displacements used in the finite element formulation,
$\delta$	shell thickness, length,
$\varepsilon$	a unit axial tensile strain, dimensionless,
$\theta, d\theta/dz, d^3\theta/dz^3$	angle, mostly torsional, and 1 <sup>st</sup> and 3 <sup>rd</sup> derivatives with respect to $z$ ,
$\kappa$	deformational stiffness of the girder per unit length in force units,
$\lambda$	a characteristic BEF constant = $(k / 4E \cdot I)^{1/4}$ , 1/in, a characteristic dimensionless BC constant = $L(N / E \cdot I)^{1/2}$ , a characteristic dimensionless TWB constant = $L(GJ/E \cdot C_w)^{1/2}$ ,

$\rho, \rho_1$	initial and final radius of curvature of cross section contour, length,
$\sigma_y$	generalized distortional load per unit length in box girder analysis axial stress, force/length <sup>2</sup> ,
$\tau$	shearing stresses, force/length <sup>2</sup>

# CHAPTER 1

## INTRODUCTION

St. Venant torsion theory does not suffice to predict the behavior of thin-walled members with open cross sections, commonly used in steel industry, subjected to unavoidable torsional loads. Thin-walled members behave according to Vlasov's treatise on 1<sup>st</sup> order general torsion and must be checked for restrained warping according to the AISC. Calculations of the warping effect are tedious and complex, even if using AISC design aids; and more so when designing multi-span beams under mixed torsion.

### 1.1 The Restrained Warping Problem

A cross section of a straight bar will rotate freely when undergoing two equal and opposite torques at its end nodes. The twist per unit length will be constant along the bar; this is known as uniform torsion, which can be studied separately from flexure and their effects superposed. However in the XX century there was a new advent in mechanics of materials: The first order general theory of torsional flexure. It is a relatively modern advance in mechanic of materials that combines the study of both uniform and non uniform torsion (Gjelsvik, 1981 and Rhodes, 1984.) See Figure 1.

Thin-walled bars under torsional loading with node restraints are mostly susceptible to develop axial stresses. In addition, applied loads are particularly transmissible quite a distance along the length of the bar, flouting St. Venant principle as shown in Figure 2 and Figure 3. First order general torsion analysis involves even more complicated studies of cross section properties, and loading effect allocation.

In fact, the analysis gets more complicated in multi-span thin-walled members with open cross sections, and the number of calculations required for a comprehensive scrutiny of these torsional effects is paramount.

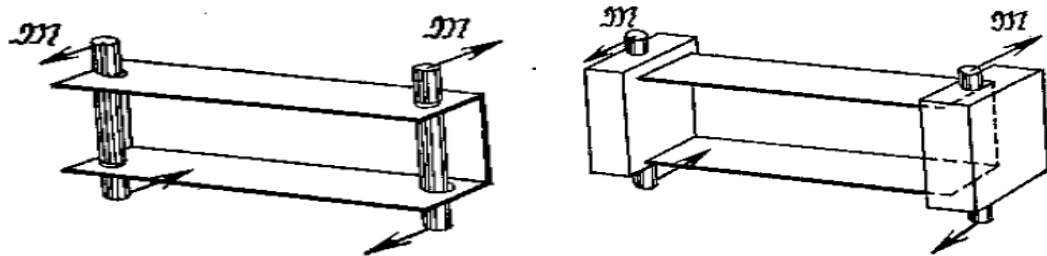


Figure 1. Free (Left) and Restrained (Right) Warping

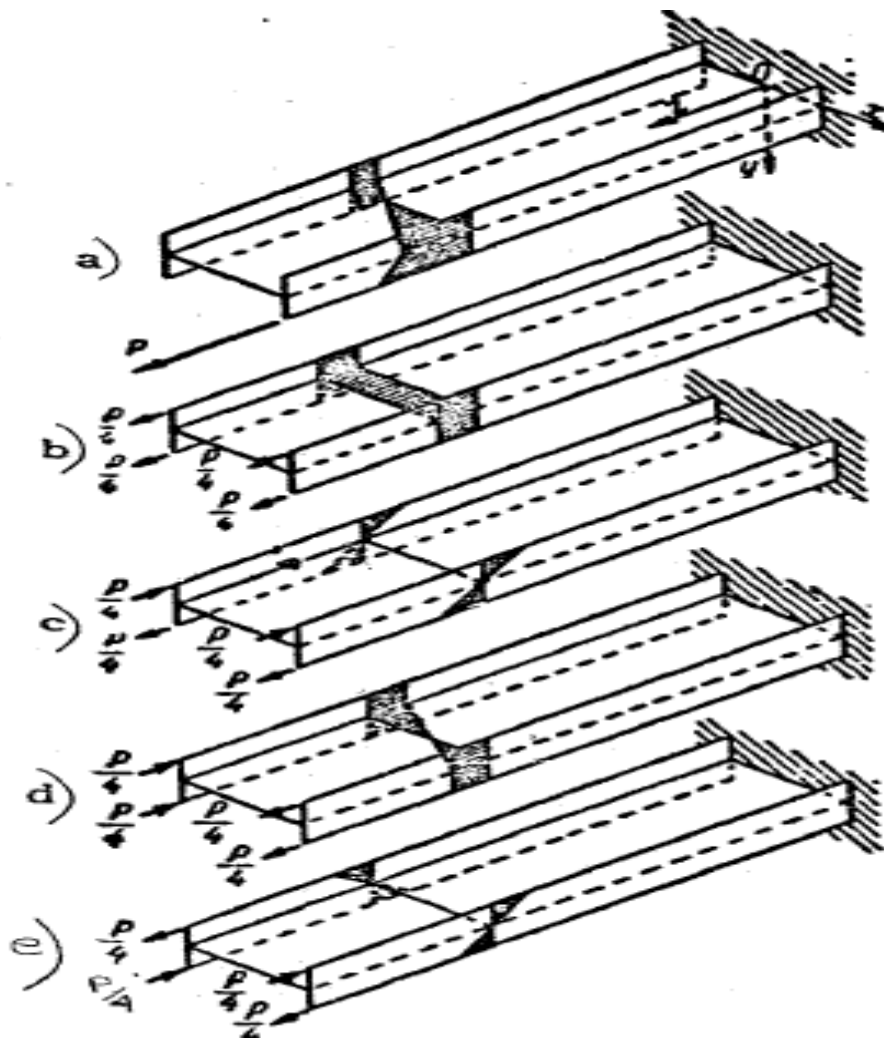


Figure 2. Load Location Exerts Influence along the Beam



Vlasov (1961) is credited as the author of the seminal work on modern 1<sup>st</sup> order general torsion theory. He found that, due to their peculiar geometry, thin-walled bars should be studied like a particular case of shells, whose geometry influences the acting shear and its propagation. For example even a centered axial resultant load applied to a member could produce transverse lateral bending with a zone of non uniform distribution of stresses. This effect is important around the cross section minor axis for thin-walled elements like the I-beam in Figure 3 taken from Feodosiev (1972).

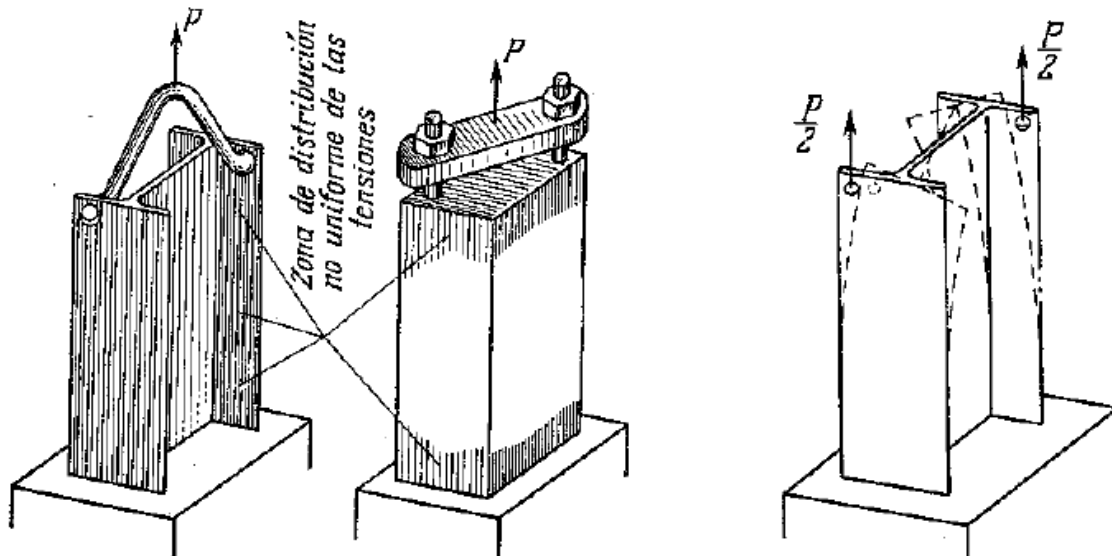


Figure 3. Restrained Warping due to a Centered Axial Tensile Load

The uniform or Saint Venant torsion Equation (1) is shown below; where  $G \cdot J$  is the cross section torsional rigidity and  $\theta$  is the angular twist around the longitudinal axis. The resisting moment for an unrestrained cross-section is:

$$T_t = G \cdot J \frac{d\theta}{dz} \quad (1)$$

For a better understanding of the non-uniform torsion, it is useful to consider the particular case of a structural steel W shape whose lateral flanges undergo opposite curvatures due to a pair of lateral opposite shearing forces in the flanges. In the lateral

direction, these forces are in equilibrium and make a force couple with a moment arm  $h$ , the distance between flanges, which is a torque  $T_w$  around the longitudinal axis  $z$  as expressed in Equation (2):

$$T_w = V_f h . \quad (2)$$

Moreover, the shear in each flange  $V_f$  may be derived by considering the beam lateral displacement  $u$ , parallel to the original major axis. Due to the fact that the beam cross section is symmetric and its rotation is relatively small, the lateral displacement  $u = (h/2) \theta$  and its first derivative  $u' = (h/2) \theta'$  are shown in Figure 4 and Equation (3):

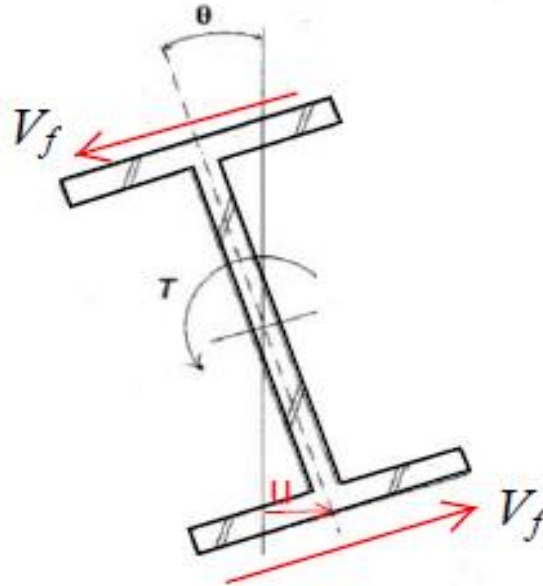


Figure 4. Lateral Transverse Displacement  $u$  due to Torsion

$$\frac{du}{dz} = \frac{h}{2} \frac{d\theta}{dz} . \quad (3)$$

Due to the opposite lateral displacement of each flange, the system of lateral bending moments  $B$  and shears  $V_f$  in each flange produced by the lateral curvatures in each flanges is in equilibrium because those forces are opposite and equal in magnitude.

Therefore, the system is statically equivalent to zero. Equations (4) and (5) are shaped using the curvature formula and the appropriate sign convention. Shear and normal stresses shown in Figure 5 will appear in the member cross section. See Ugural (1987) and Timoshenko (1978).

$$B = -EI_f \frac{d^2 u}{dz^2} = -EI_f \frac{h}{2} \frac{d^2 \theta}{dz^2}, \quad (4)$$

$$V_f = \frac{dB}{dz} = -EI_f \frac{d^3 u}{dz^3} = -EI_f \frac{h}{2} \frac{d^3 \theta}{dz^3}, \quad (5)$$

where  $I_f$  is the moment of inertia of one flange around the cross section minor axis. Hence, Equation (2) will be transformed into Equation (6), and the total torque will be that of Equation (7):

$$T_w = -EI_f \frac{h^2}{2} \frac{d^3 \theta}{dz^3}, \quad (6)$$

$$T = T_t + T_w = G \cdot J \frac{d\theta}{dz} - E \cdot I_f \frac{h^2}{2} \frac{d^3 \theta}{dz^3}. \quad (7)$$

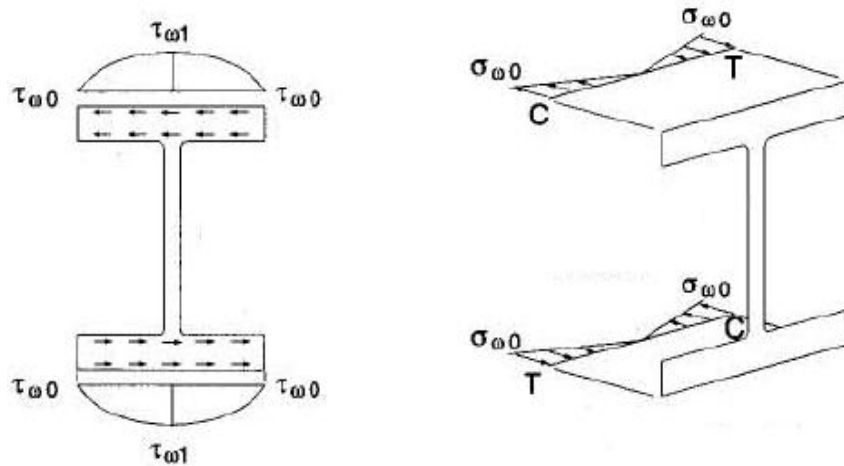


Figure 5. Shear and Normal Stresses due to Restrained Warping from DG-9

## 1.2 Significance of this Work

It is desirable, but not always possible, to avoid torsion: in slender structures, as steel structures, most structural members under torsion are not free to warp. In addition,

even when the ends are free to warp, internal restraint of warping is produced by torsional loading variations along the span (Bresler, 1963). In fact, structures undergo torsion quite frequently and open profiles are prone to suffer restrained warping stresses.

Open profiles are common in structural steel, crane girders, airplanes, naval structures, land vehicle with long frames including trucks and buses, box-girder bridges, and shear walls in high rise buildings including towers or elevator shafts. An I fact, restrained warping normal stresses might be critical design components.

Current practice calls for protective measures against the adverse effects of non-uniform torsion in case of members with modest resistance to restrained warping (AISC-DG-9). In thin-walled beam analysis, end restraints or changes in the internal torque along the length produce normal and shear stresses due to restrained warping (Figure 5). The problem of bars under mixed axial and transverse loads is a real predicament in structural engineering; particularly in the case of multi-span bars under mixed torsion.

### **1.3 Aims of this Work**

The following is an account of the tasks that will be undertaken in the first two chapters of this work: First, a straightforward software application able to solve multi-span TWB originally designed to analyze multi-span BEF subjected to three-parameter-foundations plus the geometric parameter of a tensile load will be selected. Second, a brief account on the first order general torsion theory and some of its major contributions will be provided. Third, the roles analogies play in engineering will be discussed including an account on the equivalent beam-column analogy. Fourth, some details of the high-order-finite element built in the software (Figure 6) will also be presented and discussed.

Afterwards, a convergence study of the size of the finite element to be used in the software application will be undertaken for the first time regarding two aspects: First, the effect of the geometric parameter of a tensile load will be studied for the first time, and second, the finite element stiffness matrix with 6 DOF used by the software will be studied. Then, a maximum finite element size will be recommended and used. Previous studies were made on the soil parameter, not in the geometric one, and a finite element stiffness matrix with 4 DOF was used as an upper limit for accuracy, instead of the original finite element stiffness matrix with 6 DOF.

Already analyzed single-span and multi-span beams in mixed torsion under distributed and concentrated torques with diverse boundary conditions will be reanalyzed to validate this software. The examples will be taken from different sources, including all the examples of restrained warping torsion contained in the AISC- Design Guide 9. Numerical data and curves will be provided. A performance experiment will be undertaken and analyzed with the software. Finally, user friendly adjustments to interact with the software will be implemented, and conclusions and recommendations will be presented. This study is not concerned with the analysis of box girders. However, one of the three soil parameters handled by the software would suffice to analyze box girders problems.

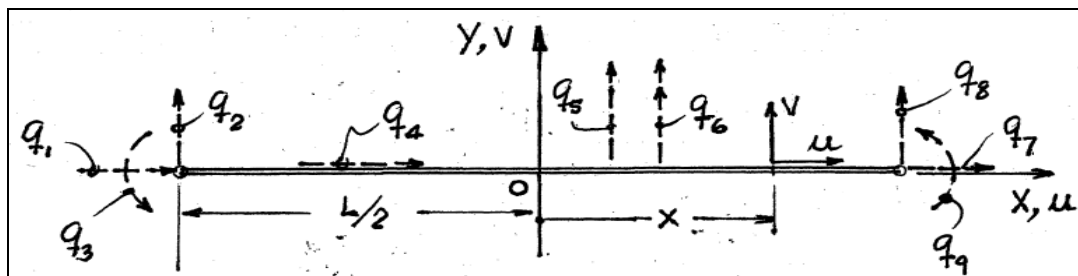


Figure 6. The 9-DOF-Beam Element by Deschapelles

## CHAPTER 2

### REVIEW OF RELEVANT TOPICS AND LITERATURE

The torsional flexure theory is only an approximate theory but supposes a great advance for engineering. Shear strains due to lateral flexure are neglected; just as the Navier hypothesis neglects shear strains in traditional bending theory, but shear strains due to torsion are indeed included (Gjelsvik, 1981).

#### 2.1 Significance of Analogies, Metaphors and Equivalences

It is convenient to take into consideration that analogy is the only one of those introductory terms accepted in science language without a debate. The other terms could be considered rhetoric language, i.e., the word metaphor (the pursuit of the ability to use language or representations to generate the intended results) was borrowed from rhetoric.

The simple engineering practice known since Kirchoff times of denoting any complex number  $r + ix$  by a single symbol expressing in one vector both the real and the imaginary fields, (Kron, 1939) might be considered as an analogy, a metaphor or an equivalence. In many fields, including software engineering, metaphor refers to a somewhat sophisticated system resemblance, but positivists reject the use of rhetoric language in research<sup>1</sup> contrary to constructionists.

##### 2.1.1 Concepts on Analogies, Metaphors and Equivalences

Analogy is a way of inferring properties of one lesser-known item from a similar better-known item on the basis of an acknowledged similarity between both of them. Modeling is the simplest statement or intellectual depiction of a process, concept,

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<sup>1</sup>According to dictionaries, constructionists find in the word rhetoric a useful meaning, because they construe or interpret. Positivist use the word rhetoric in the sense of unnecessary, because they believe in positivism, a thinking system founded by Comte, concerned with positive facts that rejects speculation.

operation or system. The modeling of an item or domain is the first step to identify analogies or equivalences. Thus, analogies and modeling are intertwined.

Analogies have been proven historically effective by making possible the findings and achievements in complex problems. Insights into two analogous systems are obtained by studying the easiest to conceive; thus improvements in the technique of solving problems in one mastered field favor the findings in the novel analogous field (Hetényi, 1966).

Analogies can be improved when basic assumptions are refined and redefined. For example, behavior (of beings) and computer output are not straightforward analogous for Cisek (1999), who thinks that behavior is more analogous to “a control process where actions are performed in order to affect perceptions...” Herbsleb (1999) states that metaphors, as opposed to straightforward analogies, help to recognize new or complex domains as if they were already understood.

Metaphors set up implicit conceptual foundations for the lesser-known domain, hence enlightening important unperceived characteristics but neglecting options and aspects of objects under scrutiny. Herbsleb (1999) considers analogies as a subgroup of metaphors. Conversely, Deschappelles (1987) conceives the bimoment as a metaphor included in the beam-column analogy. In fact, the use of this rhetoric language in research is always useful but not standard throughout interdisciplinary borders.

Analogies are not only useful but are also the core of the discipline of analog computer simulation. Mathematical models are analogous representations of reality or of other models that can be scrutinized not only by classical analytical and numerical

methods or digital computers, but also by means of analog computer simulation (James, 1971, Carlson, 1967).

### 2.1.2 Examples of Analogies

A good example of the practicality of analogies in engineering is the significant work by Wilby (1963) on the use of electronic analogy for the analysis of elastic shells. Another paradigm is the work of Kron (1939) on the analogy between electrical and mechanical networks (3D frames), which provides a complete algorithm in matrix form solving 3D frames: “Whatever the advantages of steel frames offer for the construction of buildings, analogous advantages are offered by the use of tensor spinor<sup>2</sup>.”

According to a work by Mindlin and Salvadori that appears in Hetényi (1966), in Kron analogy, currents represent stresses and voltage drops represent strains in the electric circuit. The equations of equilibrium in nodes are analogue to Kirchhoff’s current law<sup>3</sup>. The compatibility conditions are analogue to Kirchhoff’s voltage law<sup>4</sup>. Finally, Hooke’s law is analogue to “Ohm’s law  $I = Y \cdot E$ , so that the elastic constants are represented by lumped admittances  $Y$ .”

Kelvin’s fluid-flow analogy was very useful in picturing Saint Venant shear stress flux in terms of streamlines of an ideal fluid circulatory flow around the vortex inside a container tube with analogous cross section contour. Prandtl’s analogy was very useful in studying the torsion problem (Den Hartog, 1952). Weibel’s analogy helped to study the stress concentration at cross section fillets. Jacobsen’s analogy assisted to study the torsion in a shaft of variable diameter, etc. (Timoshenko, 1953).

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<sup>2</sup> Spinor is a quantity resembling a tensor used to represent the spins of fermions (particles that obey the exclusion principle and Fermi-Dirac statistics whose spins are half an odd integer: 1/2, 3/2, 5/2, (Webster).

<sup>3</sup> The summing up of all currents that enter a node is zero at every node.

<sup>4</sup> The sum of voltages drops  $E$  around every closed loop or mesh in the circuit must equal zero.



Still nowadays some British consider that epoch-making moment distribution method of flexure structural analysis by Cross (1930, 1932, and 1954) and Grinter (1932, 1933) is a particular case of the systematic relaxation of constraints advanced later by Southwell (1935).

Samuelsson and Zienkiewicz (2006) consider that the Framework Method by Hrennikoff (1941) and the Lattice Analogy Method by McHenry (1942, 1943) are analogies which solve linear elasticity problems by modeling a continuum elastic media with “elastic lattices of suitably defined discrete components.” Because the one who first coined the term finite element was Clough (1960) in an epoch-making paper, historically, Hrennikoff and McHenry may well be considered the grand-parents of the finite element method. The theory of open thin-walled beams shares with the finite element method the condition of latecomers in mechanics. And now, thanks to the development of digital computers, key advances in finite element method have taken place since the middle of the XX century.

However, the finite element genesis may well be dated back to the work of Lord Rayleigh in 1870s, later generalized by W. Ritz in 1900s. That is, to the times of the formulation of the variational Rayleigh-Ritz principle and the weighted-residual method by B. G. Galerkin (1915). In addition, the general theory of 3-D systems to solve the problem of a “self-contained space truss with the form of a closed polyhedron” was initiated by Möbius and independently developed by engineers like Föppl who published his works in 1892 calling these structures as lattice structures (Timoshenko, 1953).

The column analogy is used to solve many problems of mechanics. Magnitudes and directions can be modeled graphically like it is done with the old polygon of forces

and the funicular polygon—known since Varignon’s times (in 1720s). The conjugated beam is an analogy to calculate beam deflections from the bending moments created by a made up  $M/EI$  loading. Mohr’s circles are analogies that relate certain physic properties with the circumference geometry.

In page 144 of his book, Vlasov (1961) first proposed the use of the beam-column analogy by stating that the equation of a beam under mixed torsion is “analogous to the equation of the theory of transverse bending of an initially extended beam.” He also listed the matching analogous terms between his law of sectorial areas relevant to torsional bending and the law of plane sections applicable to flexural bending as shown in Table 1.

Table 1. Flexure and Warping Analogies

Bending in the plane $Oyz$ (law of plane sections, Figure 87a)	Restrained torsion (law of sectorial areas, Figure 87b)
$J_x = \int_F y^2 dF$	$J_\omega = \int_F \omega^2 dF$
$S_x = \int_F y dF$	$S_\omega = \int_F \omega dF$
$\eta = \eta(z)$	$\theta = \theta(z)$
$\eta' = \frac{d\eta}{dz}$	$\theta' = \frac{d\theta}{dz}$
$M_x = -EJ_x \eta''$	$B = -EJ_\omega \theta''$
$Q_y = M'_x = -EJ_x \eta'''$	$H_\omega = B' = -EJ_\omega \theta'''$
$\sigma_x = \frac{M_x(z)}{J_x} v(s)$	$\sigma_\omega = \frac{B(z)}{J_\omega} \omega(s)$
$\tau_x = -\frac{Q_y(z) S_x(s)}{J_x \delta(s)}$	$\tau_\omega = -\frac{H_\omega(z) S_\omega(s)}{J_\omega \delta(s)}$
$q_y = q_y(z)$	$m = m(z)$

In addition, the solution of multi-span bars by means of the 3-bimoments-equation, an analog version of the 3-moments-equation or Clapeyron’s equation was first proposed by Vlasov (1961) in page 145. Nevertheless, Medwadowski (1985) states that

Kollbrunner and Basler were the ones who proposed the 3 warping moment method in 1965 to solve the first order general torsion problem.

Boothby (1984) and Medwadowski (1985) works will be discussed later with more detail. They both acknowledged the beam-column-analogy to solve general torsion problems resorting to what they called a Bimoment Distribution Method and a Warping Moment Distribution Procedure respectively. These methods rely on principles similar to the moment distribution method of flexure structural analysis by Cross (1930, 1932, and 1954) and Grinter (1932-33).

In fact, the theory of beams on Winkler elastic foundations is an analogy between a continuous and a discrete model originally formulated to describe the discrete railroad track behavior. Hetényi (1961) states that this analogy is proficient if at least 4 separate elastic supports are provided to cover the characteristic wave length  $l = \pi / \lambda$ ; where  $\lambda = (k / 4E \cdot I)^{1/4}$ , and  $k$  (lbs/in<sup>2</sup>) is the equivalent continuous normal modulus of the foundation  $k_0$  times its contact width  $b$ .

The analysis of beam grillages or grid-works was developed as a consequence of the analysis of the railroad tracks, where the rails and the cross-ties intertwine similarly. An important application of beam grillages is found in ship structures. So beam grillages were analyzed by Muckle and Timoshenko (Heténi, 1966).

Vierendeel girders may be analyzed as analog grid-works. Beams unrestrained against deflection but restrained against angular rotations may be regarded as a special case of EBEF. In addition, Hetényi (1966) states: "...the effect of distributed rotational elastic restraint" is analogous to "an equivalent axial tensile force."

Other examples of the “Winkler problem” or BEF analogies are the floor-beam elastic interaction and the wall of a circular cylindrical tank with radius  $r$  and thickness  $t$  with a vertical axis of symmetry under axially symmetric loading and support (Lightfoot, 1961). Wright (1968) used the BEF analogy to analyze distorted box girders. In the BEF method, solutions are obtained either analytically or numerically by the Fourier series or by moment distribution methods as in Lightfoot (1961).

Hsu (1995) introduced a finite element formulation in the BEF analogy, so creating the EBEF analogy, and applied to the solution of multi-span beams. Spans are considered flanked by supports or internal restraints, namely diaphragms that might provide or not shear and/or warping restraints as shown in Table 2.

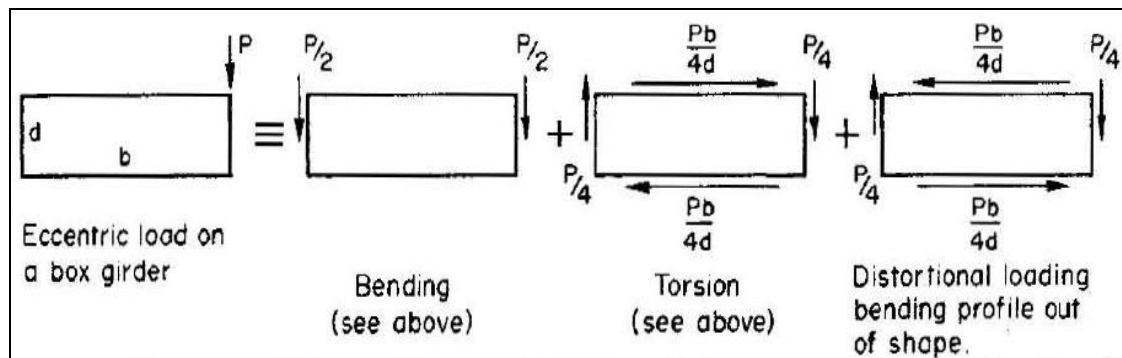

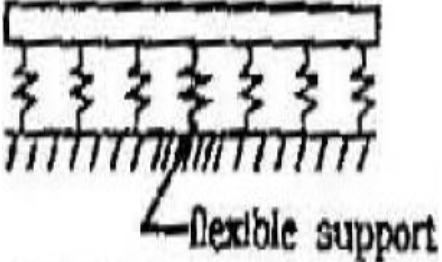




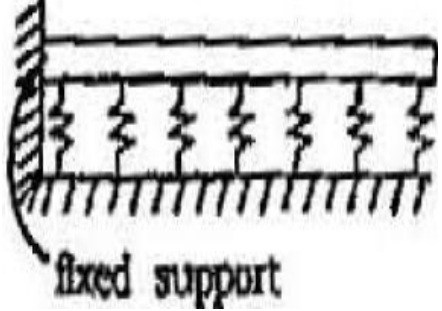


Figure 7. Distortion in Box Girders

It is important to notice that the fundamental differential equation of curved box girders, as closed TWB, is not analogous to the beam-column under mixed axial and transverse loading. Box girders undergo distortions that put cross section profiles out of shape due to the force system shown in Figure 7.

The differential equation of the beam on elastic foundation is  $E \cdot I \cdot y'''' + k \cdot y = q$  (Miranda, 1966), where  $E$ ,  $I$ ,  $y$ ,  $k$ , and  $q$  are the modulus of elasticity, the cross section inertia, the deflection, the soil modulus, and the intensity of the distributed transverse load at that point respectively.

Table 2. Internal Diaphragms in Box Girders and Analog Supports

Internal Diaphragm				
Deformation		Possible Configuration		Support in BEF
Allowed				
Shear	Warping			
Yes	Yes		truss type	 flexible support
No	Yes		frame type	
			thin plate	 simple support
No	No		solid plate, stiffened	 fixed support

The box girder (closed TWB) governing differential equation has the form  $E \cdot Cd \cdot W'''' - \kappa \cdot W = \rho$ ; therefore, it is analogous to the BEF differential equation according to Wright (1968). Where  $W$  is a measure of distortion,  $Cd$  is a cross section property related to warping,  $\kappa$  is a measure of the deformational stiffness of a unit length of the box cell,  $\rho$  is the applied generalized distortional load per unit length,  $E$  is the Young's modulus of the box girder (Wright, 1968).

## 2.2 Contrasts between Saint Venant and Vlasov Torsions

When a member is subjected to opposite torques at its end nodes, there is a stress state in which only tangential stresses appear over the faces of the cross section: the free warping, (Feodosiev, 1972), also called Saint Venant's uniform torsion. Each cross section at ends is a free surface. Member longitudinal fibers are not restricted to elongate or contract as shown in Figure 8, and shear stresses are generated as in Figure 14.

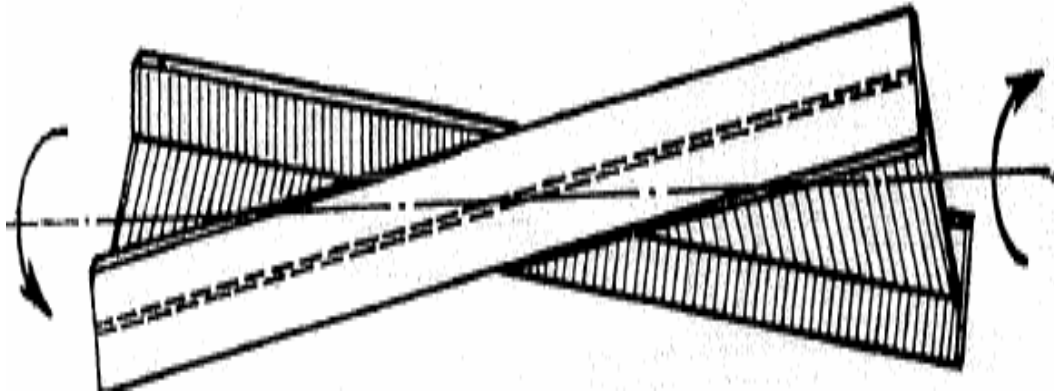


Figure 8. Free Warping or Uniform Torsion

Conversely, in the case of restrained warping, at least one of the end planes of the element is not free to warp<sup>5</sup> (Vlasov, 1961). The longitudinal fibers do elongate while the boundary restraints prevent the section rotation; thus normal and shear stresses appear as shown in Figure 9 (Rhodes, 1987), Figure 10 and Figure 15 (Hoogenboom, 2005).

St. Venant theory could be applied both to uniform and non uniform torsion, since the cross sections are free to deform out of its plane during torsion, a phenomenon referred to as the free warping of the cross section, which is a crucial point considering the later development of the general torsion theory according to Gjelsvik (1981).

<sup>5</sup> It is understood that it will suffice that any cross section, not necessarily the end planes, is not free to warp. i.e., a beam with two free ends with an intermediate fork subjected to a concentrated torque at any point suffers warping. This problem could be modeled as a two span beam.

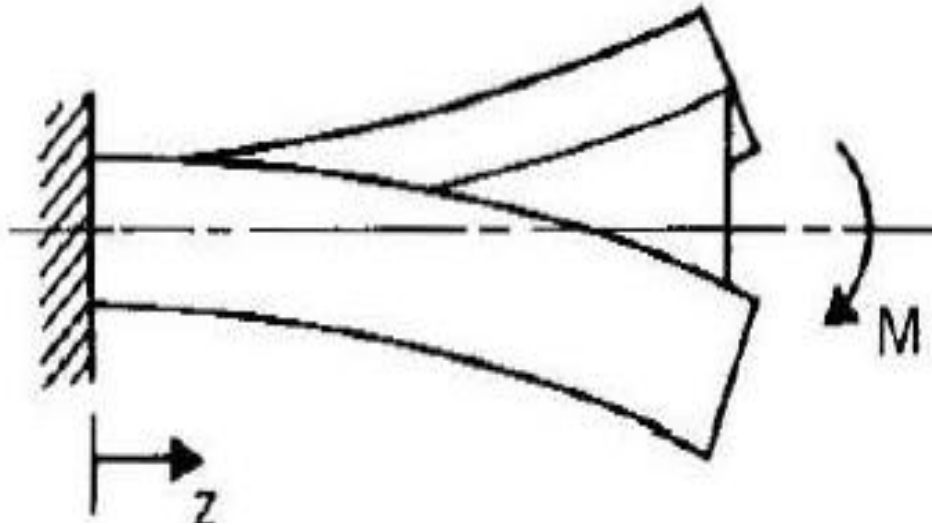


Figure 9. Warped Cantilevered Beam with Torque at Far End

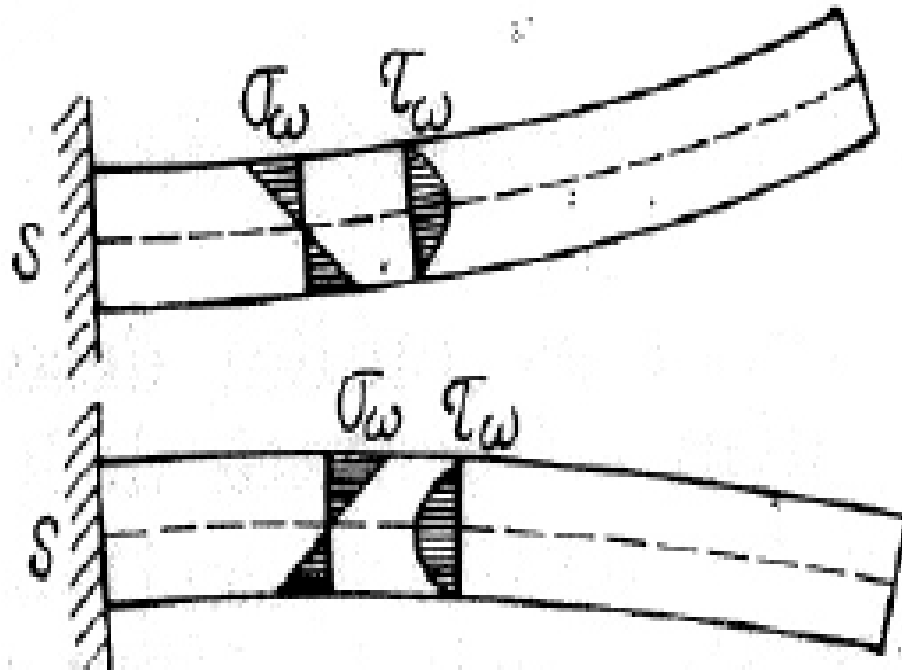


Figure 10. Shear and Axial Stresses in Cantilevered Beam under a Far- End Torque

### 2.3 Definition of Shear Center

The first shear center sound study is attributed to Maillart. The shear center axis is a line parallel to the beam generator (Figure 11). When the external transverse load passes through the shear center, pure transverse bending is produced. That is, a bending

deformation for which the cross sections will remain plane and only suffer transverse vertical or lateral displacements as it is shown in right side of Figure 12.

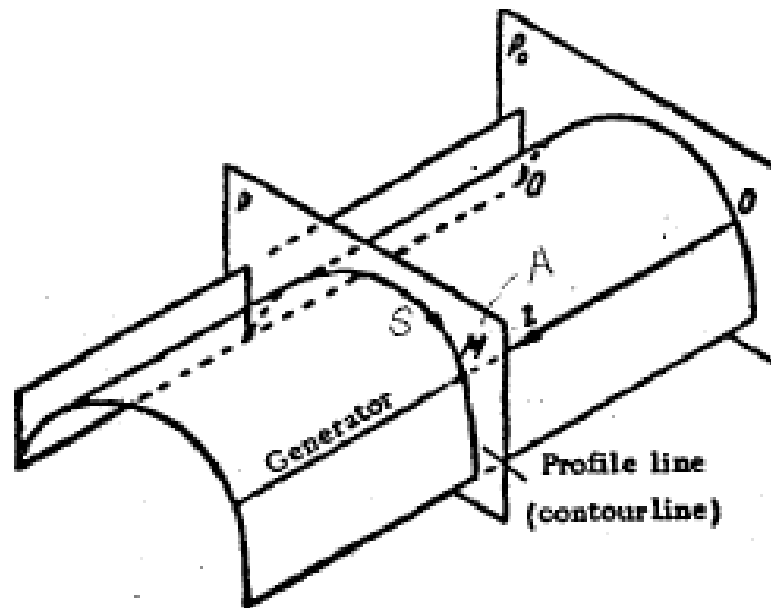


Figure 11. Generator Definition for Vlasov

If the resultant transverse load does not pass through the shear center, the beam will suffer restrained warping due to the presence of cross section restraints or changes in the torsional loading. Balancing stresses of flexural torsion (Figure 10) determined by the law of sectorial areas will materialize in the bar cross sections in addition to the pure bending stresses as shown in Figure 12 from Lundquist (1937).

As a rule of thumb, it is useful to know that: First, the shear center is located in the symmetry axis in symmetric cross-sections. Second, the centroid and shear center coincide in double symmetric cross-sections. Third, the shear center lies in the axis of symmetry but not necessarily coincides with the cross-section centroid in singly symmetric cross-sections (AISC-DG 9, 1997). Fourth, the centroid and shear center coincide in profiles composed of plate elements whose centerlines intersect at a common point such as structural T or L (Heins, 1963).



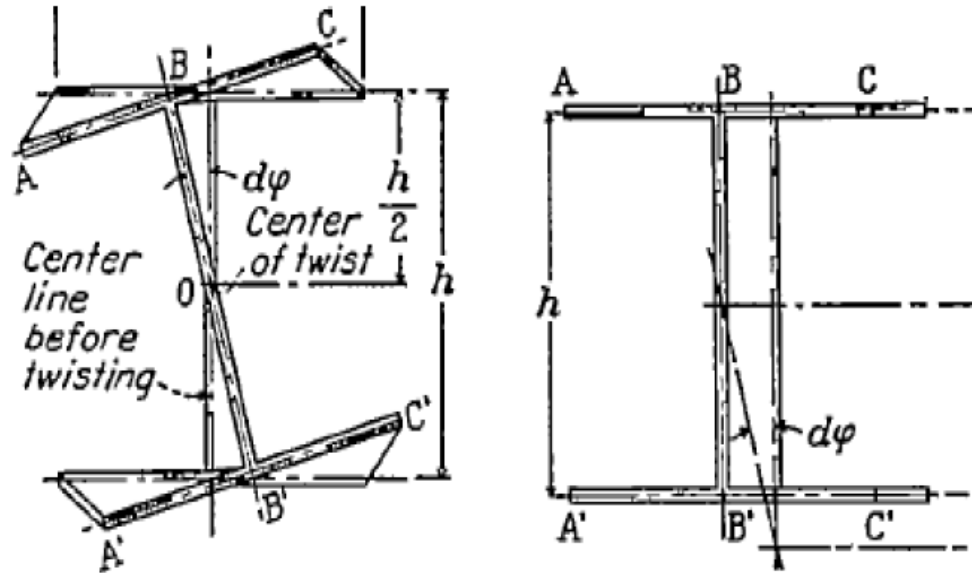


Figure 12. Bending and Bending with Torsion Responses

#### 2.4 Reclassification of Structural Elements as per TWB-Theory

Before Vlasov, traditional structural engineering classified bodies into three groups: First, massive bodies with three similar dimensions. Second, plates and shells whose thickness is small compared with the other two sharing the same order of magnitude. Third, beam-columns with two dimensions of the same order of magnitude, which are small compared with the third dimension or length. Vlasov introduced a new fourth class, thin-walled bars (TWBs) having the shape of long prismatic shells and a thickness much smaller than a representative dimension of the length of its cross section contour. In addition, thin-walled bars are subdivided according to whether their cross section contours are open or closed.

TWBs have cross-sections comprised by single long shells or combination of plates with proportions that satisfy Equation (8). The component plates of these shells are rigidly jointed along their contact lines; so that, no plate is free to move with respect to its neighbor at any point of the joint.

According to Vlasov (1961), if  $\delta$  is the shell thickness,  $d$  is any other characteristic dimension of the cross-section (its width or height) and  $L$ , its length, the proportions must satisfy:

$$\frac{d}{L} \leq 0.10 \dots \vee / \wedge \dots \frac{\delta}{L} \leq 0.10 \quad (8)$$

However, this somewhat arbitrary but reasonable standard is not a rule of thumb and could be misleading. Particular conditions like loading and mechanic characteristics should be more carefully analyzed (Vlasov, 1961).

## 2.5 Evolution of Approaches to the Restrained Warping Problem

Some rudiments of restrained warping were identified even before Timoshenko (1934), Wagner (1929), Goodier (1942), or Vlasov (1961.) The Euler-Bernoulli hypothesis stating that plane cross sections remain plane before and after bending was known not to be always accurate. The restrained warping in bars was already noticed by scientists like Young, who lived between 1773- 1829, the early stages of the primitive restrained warping theory.

Young delivered an approach for the twist of a cylindrical bar of radius  $R$  with a rectangular element  $abcd$  on its surface at a distance  $r$  of the bar axis (Figure 13). After twisting  $bc$  with respect to  $ad$  for a unit angle  $\theta'$  (angle of twist per unit length  $ab = cd$ ) around the bar axis, the parallelepiped  $ab_1c_1d$  is created. The shearing strains  $\gamma$  in  $ab$  and  $cd$  are  $r\theta'$ ; thus their corresponding shearing stresses  $\tau = Gr\theta'$  produce a torque around the bar axis expressed in Equation (9). However, if the unit distance  $ab$  (between the each cross section segment  $ad$  and  $bc$ ) is not allowed to shorten, the fiber  $ab$  experiences a unit axial tensile strain equal to  $\varepsilon = [(1 + r^2 \theta'^2)^{1/2} - 1] \approx \frac{1}{2} (r^2 \theta'^2)$  that is, a higher order small magnitude. From Timoshenko (1953) with notation adjusted.

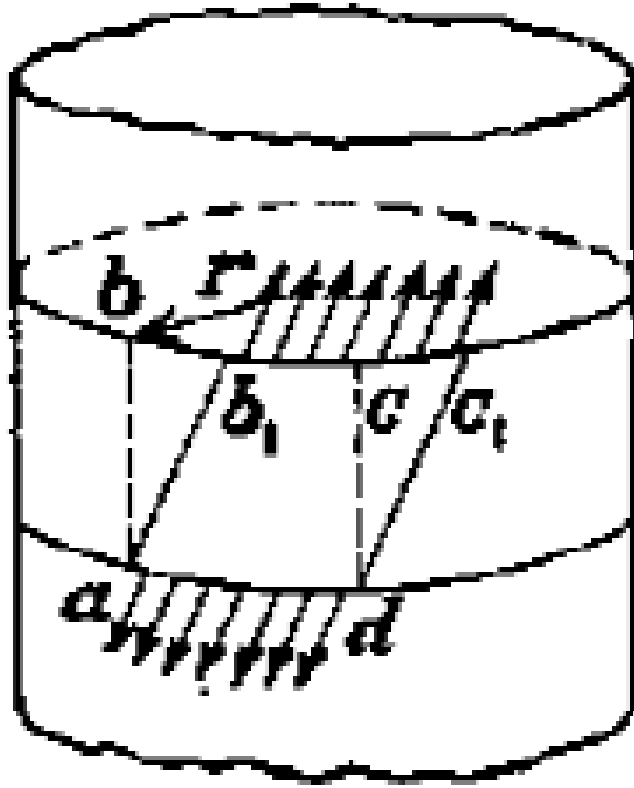


Figure 13. Young's Approach

The generated tensile stresses  $Er^2\theta'^2/2$  have a tangential component that produces an additional torque about the bar axis that, applying minor deformation theory, is  $\sin(r\theta'/l) \sim r\theta'/l$ ; thus the expression in Equation (10) is shaped. Young found that the total expression for torque consists of two terms: One proportional to the angle of twist per unit length  $\theta'$  corresponding to

$$T_1 = \int_0^R Gr\theta' \cdot r \cdot 2\pi r dr = \frac{\pi}{2} R^4 G\theta' = GJ\theta', \quad (9)$$

and another proportional to  $(\theta')^3$  multiplied by a constant expressed in length<sup>6</sup> related to the fact that axial elongation was prevented or restrained

$$T_2 = \frac{1}{2} \int_0^R Er^2\theta'^2 \cdot r\theta' \cdot 2\pi r^2 dr = \frac{1}{6} \pi R^6 \theta'^3 E, \quad (10)$$

where the axial stress projection in the transversal direction  $T_1$  is now known as the circulatory or Saint Venant torsion and  $T_2$  is the torsion produced by the longitudinal stresses preventing changes of distances between cross sections, now known as Vlasov torsion. In 1939, Biot proved that Young method provides the correct answer for circular shafts according to Timoshenko (1953).

In 1853, Saint Venant presented the first sound analytical study of free warping before the French Academy Committee solving the torsion problem of unrestrained cross sections for a variety of prismatic bars, including the elliptic shaft (Timoshenko, 1953). Free warping theory created a pathway for the restrained warping theory development.

At the end of the XIX century Bredt formulated his equations on torsion, one of them named by August Föppol the first Bredt equation. Bredt also proposed a 2<sup>nd</sup> equation for thin-walled hollow sections with constant thickness.

The work of Bredt in 1896 on tubes and the membrane analogy further developed by Prandtl in 1903 made it possible to apply the St. Venant torsion theory to the thin-walled engineering shapes of those times (Gjelsvik, 1981).

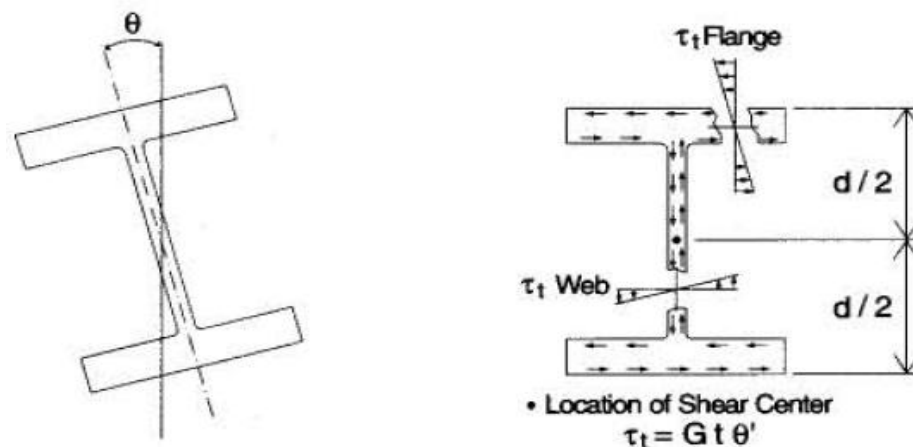


Figure 14. Shear Flux in Saint Venant Torsion

The early stages of the 1<sup>st</sup> order modern torsion theory for thin-walled bars with open profiles dates back to 1905-06, when Timoshenko published an analysis for I-section bars finding a previously unpredicted torsional resistance different from St. Venant's. Goodier also developed separately general theories on torsion at Cornell University (Ojalvo, 1990, and Gjelsvik, 1981).

The general torsion equation was first deduced for I beams of bisymmetrical cross section and the shear center was already discussed by Timoshenko in the Bulletin of the Institute of Engineers of Ways of Communications, St. Petersburg in 1913 (Pettersson, 1955, Timoshenko and Goodier, 1951). However, R. Maillart is credited with the finding of the shear center. In 1921<sup>6</sup>, Maillart published a work on a cross section point he had found through which the shear must pass to avoid the cross section twisting. He also provided the method to locate this point in open profiles (Ojalvo, 1990, and Gjelsvik, 1981).

Wagner (1929) presented the general torsion differential equation of thin-walled bars with open profiles with arbitrary shapes approaching the issue of combined bending and twisting<sup>7</sup> that influenced subsequent works including Vlasov's. Ojalvo (1981, 1982 and 1990) has criticized the application of Wagner's multifilament model to buckling studies.

Ojalvo considers this model inconsistent with the single filament model used in current general torsion theory; therefore, the Wagner hypothesis would be unnecessary for a general theory of rods. See also Falgoust (2004).

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<sup>6</sup> Ojalvo, 1990, mistakenly states that it was in 1924, but both Timoshenko, 1953, and Gvelski, 1981, state that it was in 1921

<sup>7</sup> Referring to restrained warping, the translator of the N .A. C. A. Technical Memorandum No. 807 S. Reiss used either combined bending and twisting or bending accompanying the torsion.

In addition to his previous studies in TWB, Timoshenko (1945) analyzed the bending of thin-walled members of open cross section demonstrating the identity of shear center and the center of twist by using Maxwell's reciprocal theorem.

In the 1930s, Vlasov (1961) developed the general theory of thin-walled bars, from his earlier engineering theory on orthotropic cylindrical and prismatic shells. This work is credited as the seminal contribution to the 1<sup>st</sup> order general torsion theory that put under the scope the restrained warping phenomenon. Vlasov's work was compiled in a book and first published in Russian, Moscow, 1940 (Timoshenko, 1953).

In order to solve problems of multi-span beams under mixed torsion, Pettersson (1955) advanced a wide-ranging of torsion cases to be solved by means of the Hardy Cross method of successive approximations to find the bending moments acting on the flanges of I-beams, which later was named as bimoment by Vlasov (1961). Possibly due to the year of publication, he did not mention either Vlasov or the moment-bimoment analogy. On the other hand, Pettersson did acknowledge the contributions of Wagner (1929) on the governing differential equation of the general torsion theory and the compilation of solutions of torsional fundamental cases by Bornscheuer in 1953.

In a more detailed analysis, Dabrowski (1960) found that few essential terms were missing in Vlasov's equations, which could have conducted to errors. His work on Curved Thin-Walled Girders published in 1968 is frequently quoted by many torsion theorists and practitioners as Heins (1975).

Ligthfoot (1961) and Gere (1963) used a comprehensive approach of the moment distribution method to undertake the problem of beam-columns under mixed axial and transverse loads. Stiffness coefficients and carry over factors for uniform beams axially

loaded in compression or tension and with their far end fixed or hinged can be found in these works, as well as influence lines for fixed end moments.

Heins (1963) and Seaburg prepared a manual to be used as design data for the Bethlehem Steel Corporation. This manual provides charts of torsional function curves, tables of torsional properties, case charts listings for various torsional loading and end conditions, case charts of torsional function curves with examples on general torsion, etc. All this material has been widely used by practitioners and even assimilated for other publications as the AISC-DG-9 by the very Seaburg (1997) and Charles J. Carter.

Wright (1968) with his BEF and Hsu (1995) with his EBEF provided a rational approach to box girder design. Box girders are thin walled beams with closed cross sections, and even that they are prone to suffer restrained warping; their governing differential equation is different from that of thin walled beams with open cross sections.

Boothby (1984) postulated the Bending Moment-Bimoment Analogy or Bending-Warping Analogy; that is, a bimoment distribution method for multi-span bars analogous to classical moment distribution methods of flexural analysis. He neither mentioned Vlasov's beam-column analogy or the book by Vlasov (1961).

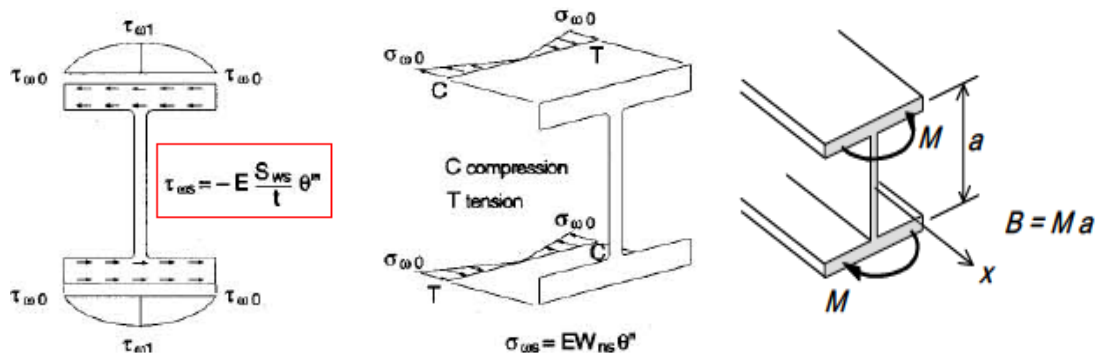


Figure 15. Bimoment and Related Stresses

Medwadowski (1985) states, quoting Bleich (1952), that “...the governing differential equation” of the beam-column “is identical to the equation governing the problem of beams in axial tension and bending.” Medwadowski (1985) redistributed bimoments in multi-span beams by using beam-column formulations proper of classical moment distribution methods similar to the ones developed by Lightfoot (1961) and Gere (1963). In fact, Åkesson (1987) discussed on the paper by Medwadowski that the real beam in mixed torsion rather be represented by an analog beam-column in mixed bending and tension for the sake of clarity, accuracy and efficacy.

## 2.6 Some Features Introduced by the Restrained Warping Theory

It has been discussed that thin-walled members with open cross sections are prone to warp. A restrained I-beam under torsion undergoes opposite lateral curvatures with balanced opposite lateral shears and moments (Figure 10). A system of generalized forces statically equivalent to zero, coined by Vlasov as bimoment  $B$ , is developed. It is equal to the flexural moment around the minor axis in one flange times the separation between flanges. As a result, warping axial strains and stresses are developed along the member.

The Vlasov hypotheses of general torsion 1<sup>st</sup> order theory are: First, the cross section contour  $s$  does not bend or elongate; then, it does not suffer changes in curvature as expressed in Equation (11), or changes in length as expressed in Equation (12), thus, the middle line projection of the thin-walled beam in the cross section surface  $xOy$  does not change (See Figure 16(a)). And second, the middle surface of the thin-walled bar does not suffer shear deformations as denoted in Equation (13) and Figure 16(b).



$$\frac{1}{\rho_1} - \frac{1}{\rho} = 0, \quad (11)$$

$$\frac{u_r}{\rho} + \frac{\partial u_s}{\partial s} = 0; \quad (12)$$

$$\gamma_1 + \gamma_2 \approx 0. \quad (13)$$

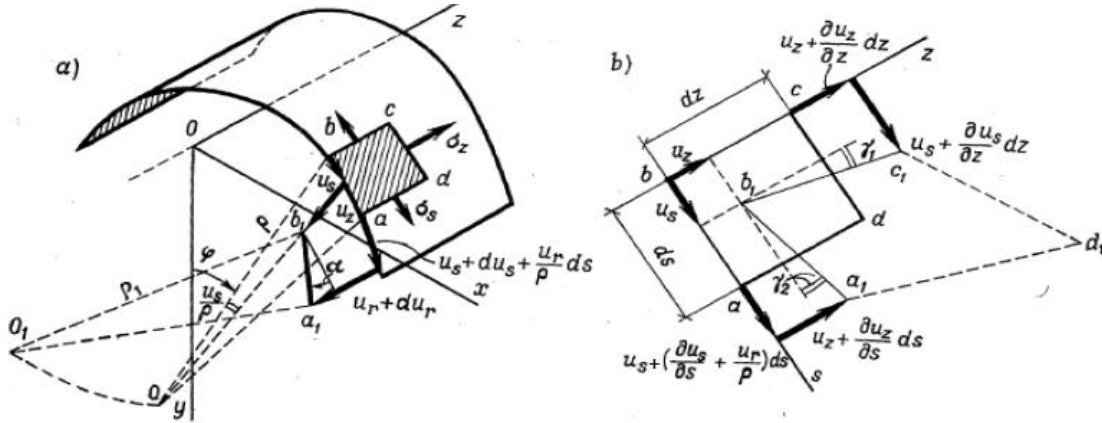


Figure 16. Differential Element in a Thin-Walled Beam

Equations (11) and (13) explain the cross section deformation out of its plane. The cross sections after the deformations will not remain plane but will obey the rule of sectorial areas according to Rekach (1978).

One of the consequences of this phenomenon is that the flanges of a thin-walled beam subjected to a centered axial tensile force  $P$  applied eccentrically as it is shown in Figure 17 would deform independently from one another. And each transverse lateral moment acting over each flange would spread over the entire length of the beam. The Saint Venant's principle does not hold true in this case (Feodosiev, 1972).

Figure 17 from Chaudhary (1982) illustrates a way in which the warping function could manifest; the example of an axial centered resultant force producing bimoments will be presented. If  $b$  and  $d$  are the section width and depth, respectively, Figure 17(e)

shows equal and opposite lateral bending moments  $Nb/4$  acting at each flange producing opposite lateral curvatures in each flange, which are statically equivalent to zero. The product of each opposite bending moment times their distance is called bimoment  $Nbd/4$ .

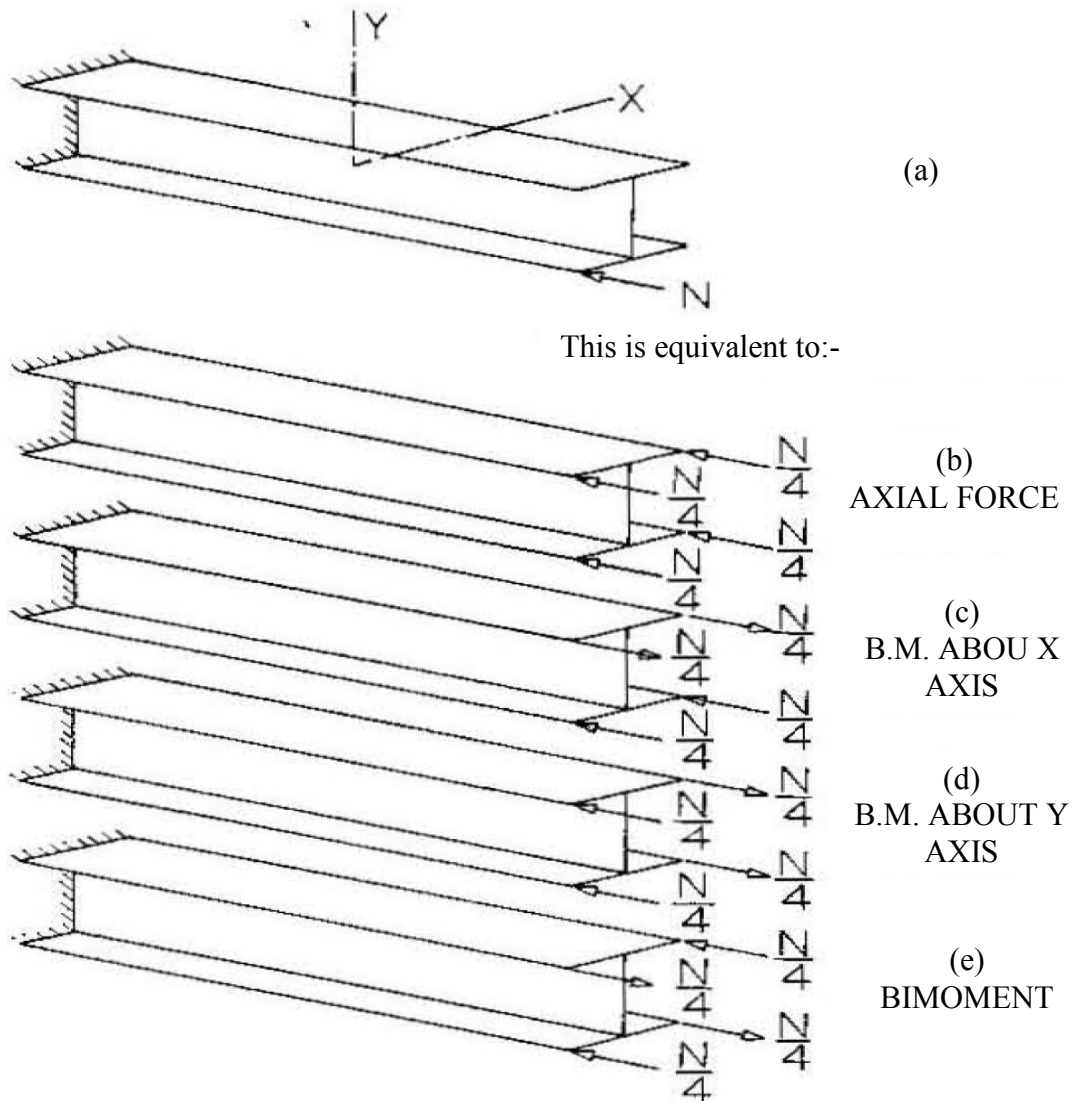


Figure 17. Centered Axial Load Producing Restrained Warping

According to Vlasov (1961), the same result could have been obtained by multiplying the axial force by the warping function  $(-N) \cdot (-bd/4) = Nbd/4$ . The summation of forces is statically equivalent to a centered axial force  $N$  plus a bending moment  $Nd/2$  around mayor axis plus a bending moment  $Nb/2$  around minor axis, plus

two opposite and equal moments producing opposite lateral curvatures in the flanges (Chaudbary, 1982).

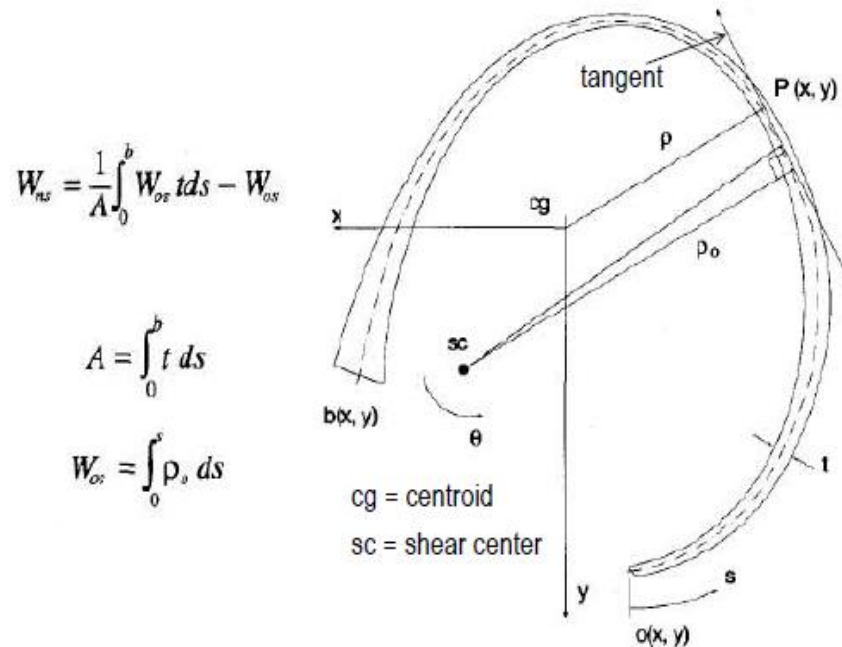


Figure 18. Definition of Warping Function

## 2.7 The Beam-Column Analogy

In Chapter II, Section 9 of his book, Vlasov (1961) restated the impeccable analogy between the three-term-differential-equation of 2<sup>nd</sup> order analysis of transversely and axially loaded bar and the differential equation of 1<sup>st</sup> order analysis of a beam under mixed St. Venant and Vlasov torsion.

Boothby (1984) produced a work on the application of the moment distribution method to torsional analysis of multi-span TWB with open cross sections. He renamed the bimoment as warping moment: The kind of warping that produces bending moment.

Medwadowski (1985) also resorted to the moment distribution method for the torsional analysis of multi-span TWB with open cross sections. His equations are presented in matrix form, and his transfer matrix helps to express the deformed configuration of each span of the beam once the MDM has converged.

Both Boothby (1984) and Medwadowski (1985) prepared separately formulas and tables with stiffness, carry over factors, fixed-end bimoments for various loading and boundary conditions based on the formulas of mixed torsion. Medwadowski (1985) also presented the analytic stiffness matrix for the beam under mixed torsion that will be used in this paper to recommend a size of the finite element to be used in the software.

Boothby (1984) only made references to works on combined bending and torsion without even including Vlasov's. Conversely, Medwadowski (1985) made a number of references to seminal works on beam-column and on moment distribution methods. Boothby (1984), for example, stated that he had to compute the fixed moments on a beam under a concentrated load by dividing the beam at the point of load application into two parts, as a result he solves simultaneously 4 equations with 4 unknowns for each span; while both Lighthfoot (1961) and Gere (1963) had already solved the problem through more straightforward methods for a beam-column under mixed tension and transverse load.

Åkesson (1987) stated that, works on the moment distribution method to torsional analysis of multi-span TWB with open cross sections were advanced by others including Pettersson (1955). He proposes to model and solve an analogue beam-column in mixed bending and tension. And Deschappelles (1987) acknowledged Medwadowski's effort to approach the statical redundancy in thin-walled bars under restrained warping and advised to better follow the philosophy of the finite element method.

### **2.7.1 The Beam-Column Differential Equation**

If an infinitely small element bounded by two cross sections a distance  $dz$  apart is cut out of the BC, the equilibrium of moments leads to the Equation (14) from Hetényi (1961) illustrated in Figure 19; where  $Q_v$  is the vertical shearing force, as shown in

Figure 19. The normal shear  $Q_n$  acting in the plane of the section normal to the deflection line can also be obtained using Figure 19b with the results of Equations (15) and (16), where it could be noticed that the rate of change of curvature is almost exclusively responsible for the normal shear.

$$(M + dM) - M + Ndy - Q_v dz = 0 \Rightarrow \frac{dM}{dz} + N \frac{dy}{dz} - Q_v = 0; \quad (14)$$

$$Q_n = Q_v \cos \theta - N \sin \theta, \quad (15)$$

$$Q_n = \cos \theta \left( Q_v - N \frac{\sin \theta}{\cos \theta} \right) = \cos \theta (Q_v - N \tan \theta) \Rightarrow \quad (16)$$

$$Q_n = \cos \theta \left( Q_v - N \frac{dy}{dz} \right) = \cos \theta \frac{dM}{dz} \approx \frac{dM}{dz}, \forall \theta \ll 1$$

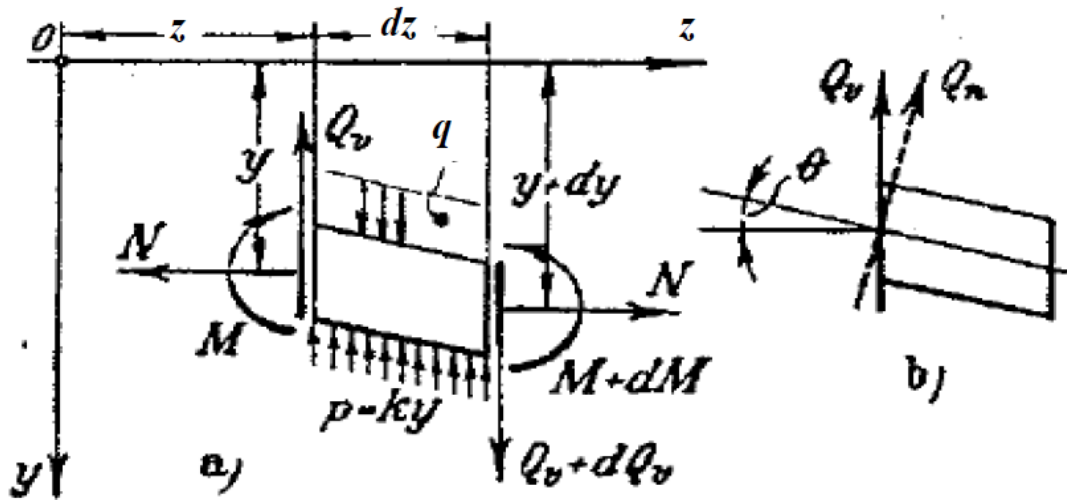


Figure 19. Beam Column under Mixed Axial and Transverse Loading

In the following derivations  $Q_v$  from Equation (16) will be used. Replacing  $M = -E \cdot I \cdot y''$  (bottom fibers are in positive location and under a tensile positive stress  $\sigma_y = \varepsilon \cdot y$  combined with an upwardly negative curvature). Then he differentiates with respect to the axial coordinate, making  $Q_v' = ky$ , thus obtaining the differential equation of the elastic line of a straight bar under mixed axial and transverse loading:

$$\begin{aligned}
& +EI \frac{d^4 y}{dz^4} - N \cdot \frac{d^2 y}{dz^2} = -ky + q \Rightarrow \\
& EI \frac{d^3 y}{dz^3} - N \frac{dy}{dz} = Q_v \Leftrightarrow ky \wedge q = 0,
\end{aligned}
\tag{17}$$

where  $E$  and  $I$  are the modulus of elasticity and the cross section inertia of the member, and  $k$  is the soil modulus. The 2<sup>nd</sup> of Equations (17) corresponds to the integration of the first one with both the modulus of the foundation  $k$  and the uniformly distributed load  $q$  equal to 0. Figure 20 illustrates a different convention for a BC with longitudinal axis along axis  $z$ , and where  $ky$  and  $q(z)$  are also replaced by a unique  $k(y,z)$ .

$$Q_v = N \frac{dy}{dz} - EI \frac{d^3 y}{dz^3} . \tag{18}$$

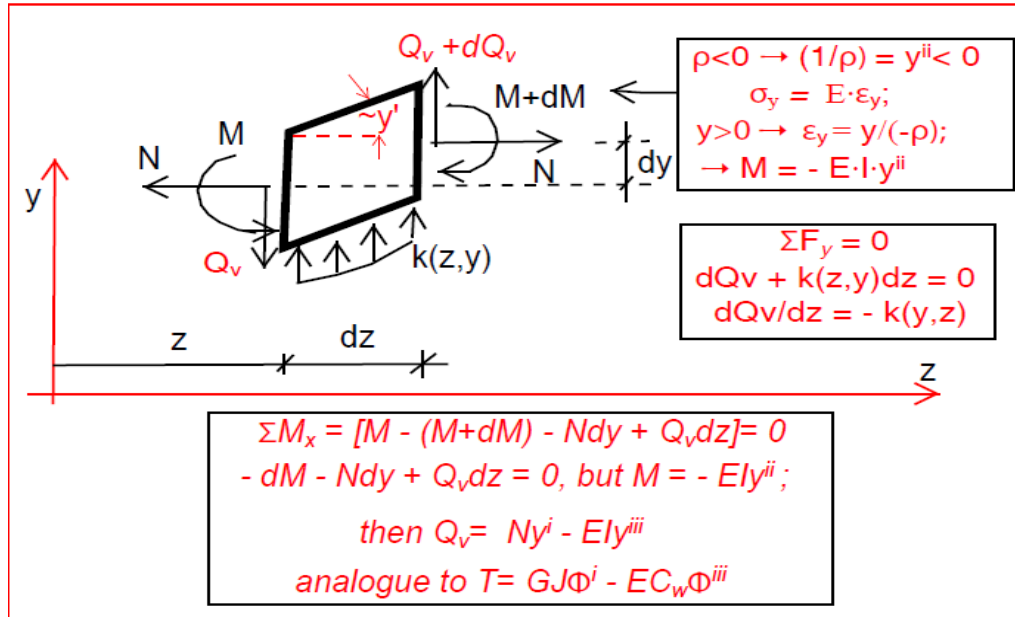


Figure 20. Another BC Sign Convention with Similar Results

### 2.7.2 Differential Equation of a Torsional TWB

The governing differential equation of a thin-walled bar under restrained warping has any of the forms shown in Equations (19), where  $z$  is considered the longitudinal axis,  $E$  is the modulus of elasticity of the material,  $G$  is the shear modulus of elasticity,  $C_w$  is the cross section warping constant,  $J$  is the Saint Venant torsional constant,  $\theta$  is the cross

section twist angle,  $t$  is a uniformly distributed torque,  $t_j$  is the highest applied torque at right support<sup>8</sup> in the case of a linearly varying torque, and  $T$  is the applied torque in the cross section under study. See Ugural (1987) Heins (1963) and Figure 26 from the AISC-Design Guide 9, 1997.

$$\begin{aligned}
 T &= G \cdot J \frac{d\phi}{dz} - EC_w \cdot \frac{d^3\phi}{dz^3}, \\
 E \cdot C_w \frac{d^4\phi}{dz^4} - G \cdot J \frac{d^2\phi}{dz^2} &= t \Leftrightarrow \frac{dT}{dz} = -t \\
 EC_w \frac{d^4\phi}{dz^4} - G \cdot J \frac{d^2\phi}{dz^2} &= t_j \frac{x}{l} \Leftrightarrow \frac{dT}{dz} = t_j \frac{x}{l}
 \end{aligned} \tag{19}$$

### 2.7.3 The Analogue Beam-Column

The analogue beam-column is a deformed straight bar vertically loaded and pulled by a horizontal force  $GJ$  as illustrated in Figure 21, Figure 22, Figure 23, which obeys Equations (19). Its longitudinal and transverse loadings are not tangent and normal to the elastic line respectively: they are measured in their original directions before the bar is deformed.

That is why the BC in Figure 22 remains in equilibrium. When the beam-column Equation (16) is formulated with forces acting normally and along the deformed elastic line, the analogy with the TWB is lost as it could be seen by comparing Equations (19). On the other hand, Equations (19) governing open TWB, are perfectly analogue to Equations (17) governing BC. In this sense, when using the small deflection theory in the BC straight bar, care should be taken so that the axial and transverse loads remain in their original directions before and after bending.

<sup>8</sup> The AISC uses  $t$  without subscript.

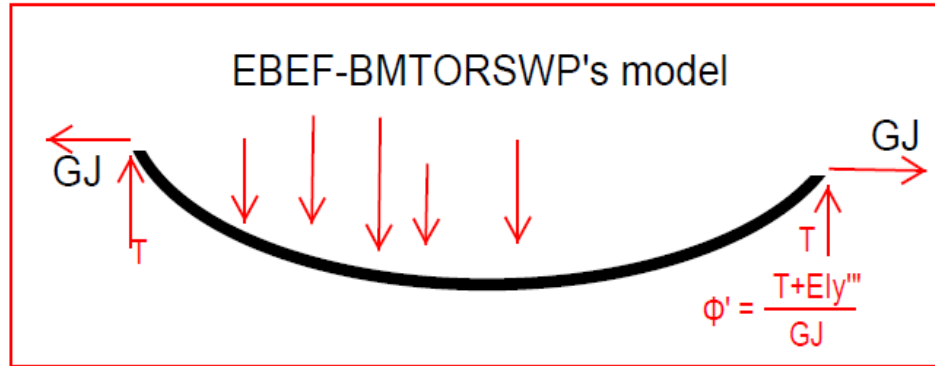


Figure 21. Analogue Model in Deformed Bar

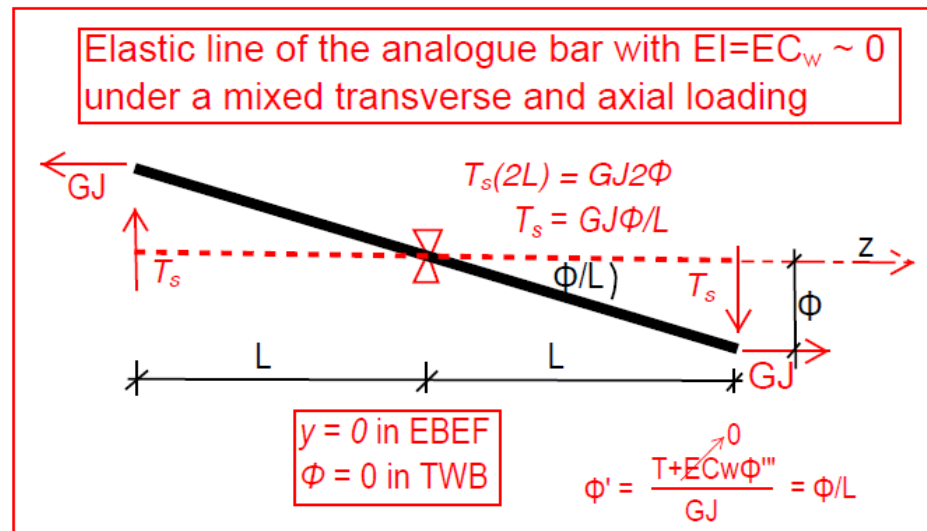


Figure 22. BC Model in Equilibrium

It is important to clarify statements by Åkesson (1987): “For a real beam in pure St. Venant torsion ( $E \cdot C_w = 0$ ), the analogue beam-column degenerates into a transversely loaded cable stretched by a force  $GJ$ ” (symbols have been adapted to those in AISC-DG-9). This statement should be understood under the light of Åkesson’s model shown in

Figure 23 and never interpreted too literally as in Figure 24. Where it could be noticed that the concurring longitudinal forces  $G \cdot J$  are incapable to provide opposition to the couple of forces formed by the transverse loads  $T_s$  and their arm  $2L$ . Because this transverse loads happen to be applied at the end nodes of the beam, and do not deform according to a commonly transversely loaded cable stretched by a tensional force.



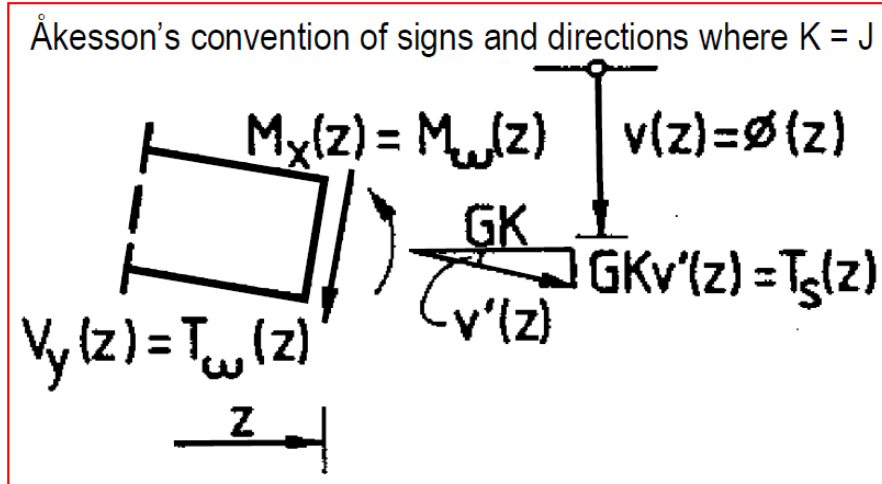


Figure 23. Equilibrium in Åkesson's Model

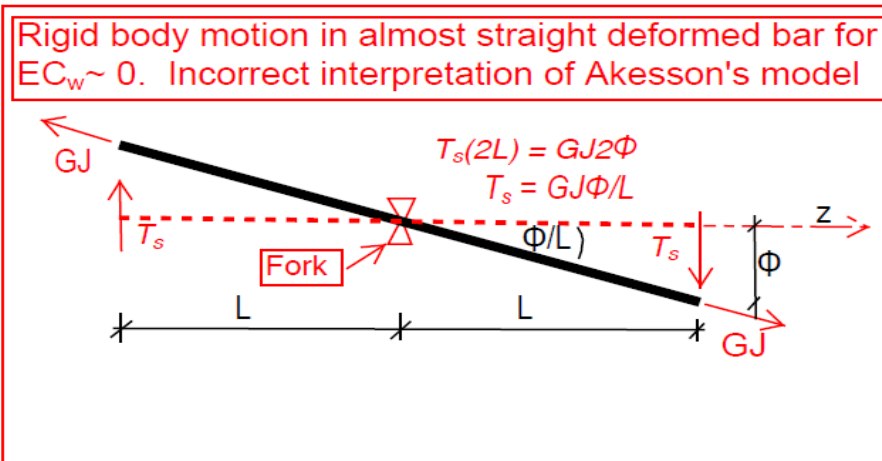


Figure 24. Incorrect Overly Literal Interpretation of Åkesson's Model

#### 2.7.4 Boundary Conditions and Equivalence of Symbols

The following are equivalences between the real TWB undergoing a twist angle  $\theta$  and the BC undergoing a transverse displacement  $y$  (Table 3 and Table 4):

Table 3. Analogue Components between a TWB and a BC

Beam in mixed uniform and non-uniform torsion, TWB	Beam in mixed transverse and axial loading, BC
Angle of twist or torque angle = $\theta$	Transverse displacement = $y$
$\theta \neq 0$ , some torsion rotation $\theta' \neq 0$ , no warping restraint $\theta'' \neq 0$ , some bimoment	$y \neq 0$ , some transverse displacement $y' \neq 0$ , slope, no fixed end $y'' \neq 0$ , some bending

(continue)

$C_w = I_{ww} = I$ . Warping constant Principal 2 <sup>nd</sup> order moment of area using sectorial area coordinates = $\int w^2 \cdot dA$ , in <sup>6</sup>	$I =$ moment of inertia  $\int y^2 \cdot dA$ , in <sup>4</sup>
$E \cdot C_w =$ Warping stiffness, kip- in <sup>4</sup> $G \cdot J =$ St. Venant torsion stiffness, kip- in <sup>2</sup>	$E \cdot I =$ flexural stiffness, kip- in <sup>2</sup> $N =$ tensile axial load in original coordinates, kip
$\theta =$ Torsion twist at a generic section $\theta'' =$ 2 <sup>nd</sup> derivative $\rightarrow$ Bimoment $B = -E \cdot C_w \cdot \theta''$	$y =$ Transverse displacement at a section $y'' =$ 2 <sup>nd</sup> derivative $\rightarrow$ bending moment $M = -E \cdot I \cdot y''$ , or $y'' = -M / E \cdot I$
$\theta = 0$ , no torsional rotation (clamped end or fork) $\theta' = 0$ , no slope, warping restraint (clamped end) $\theta'' = 0$ , no torsional curvature, $B = 0$ , (no end moment, pinned or free)	$y = 0$ , no transverse displacement (pinned end or roller, fixed end) $y' = 0$ , no slope, flexural restraint (fixed end, symmetry) $y'' = 0$ , no flexural curvature, $M = 0$ (no end moment, pinned end, roller)
Warping (lateral bending) axial stresses: $\sigma_w = -E \cdot w \cdot \theta''$	Vertical bending axial stresses: $\sigma_f = -E \cdot y \cdot y''$

Table 4. Torsional Conditions in TWB from AISC-DG 9

Physical	End	Mathematical
No rotation	Fixed (clamped) or pinned (fork)	$\theta = 0$
Cross-section cannot warp	Fixed (clamped) end	$\theta' = 0$
Cross section warps freely	Pinned or free	$\theta'' = 0$

## 2.8 The High Degree 9-DOF-Bar Element

The 9-DOF-BE is a beam-column element composed by a 3DOF rod element plus a 6-DOF beam-column element. The 3DOFs of the rod element operate in the field of axial displacements uncoupled with the transverse displacement and its derivatives without taking care of the 2<sup>nd</sup> order effect of the axial load. The 3DOF rod element has a quadratic expansion with one axial DOF at each end node plus a dimensionless DOF, the

average of axial displacements. The 6DOF beam-column element operates in the uncoupled field of transverse displacements and will be described with more detail.

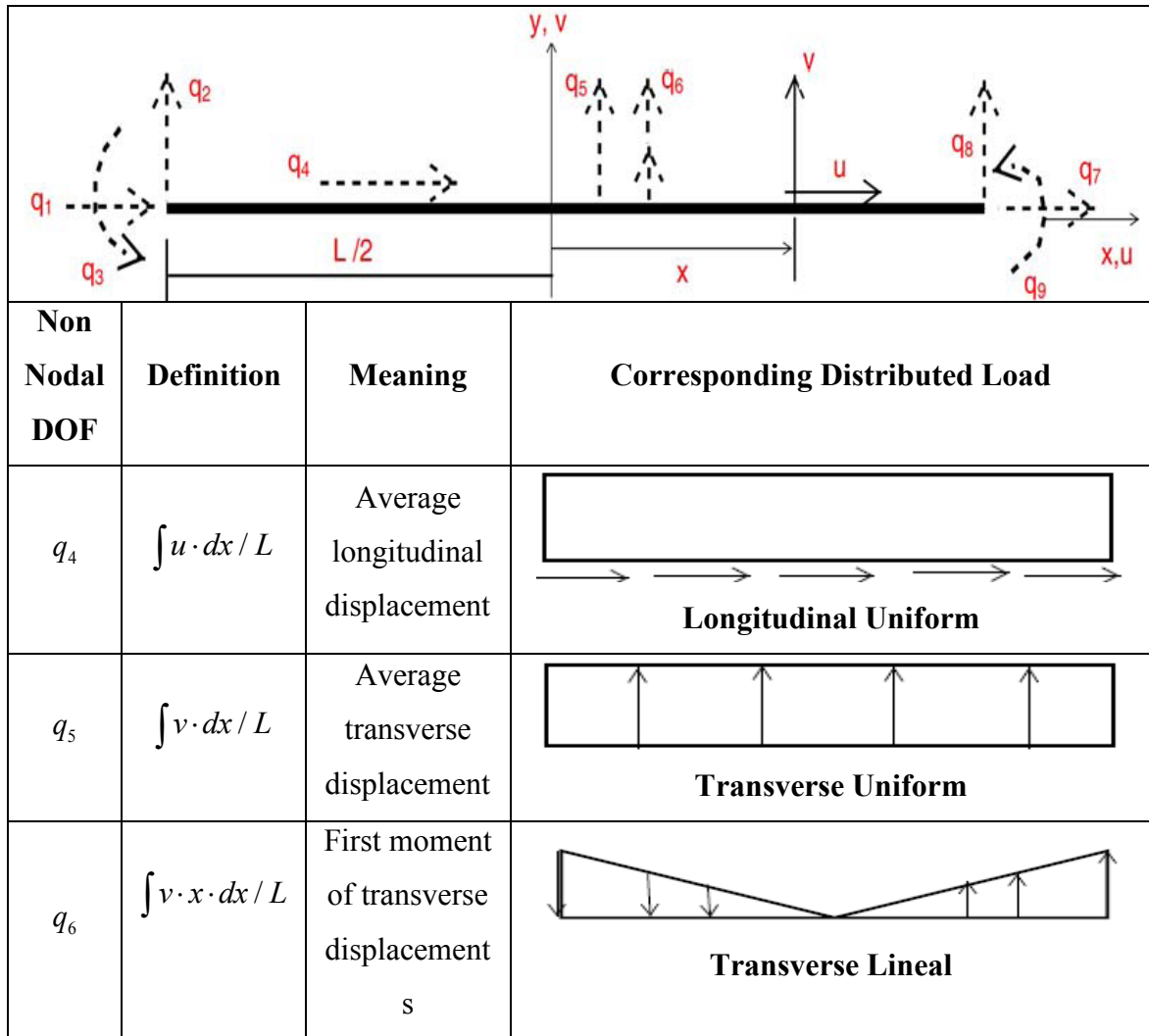


Figure 25. The 9DOF Frame Element with its Non-Nodal DOF

In order to avoid the presence of the element length  $L$  in the expressions of the entire matrix entries, normalized-dimensionless degrees of freedom divided by powers of the length were used as it could be seen in Table 5 and Figure 25.

Local axes are located at member midspan,  $A$  and  $I$  are the cross section area and inertia respectively, and  $Ce$  is a common dimensionless factor stated in Equation (20).

To avoid matrix singularity, a value different from zero, i.e. 1, must be assigned to the

cross section area to make possible the matrix inversion when running BMTORSWP. Table 6 shows one of the 9DOF components, the 6DOF stiffness matrix related to transverse displacement and derivatives.

Table 5. All the DOFs' Definitions in the 9DOF-BE

DOFs	Definitions
1 and 7	Axial displacements at each end divided by length, L
4	Average, non nodal, axial displacements divided by length, L
2 and 8	Transverse displacements at each end divided by L
3 and 7	Rotational displacement at each end
5	Average, non nodal, transverse displacement, divided by L
6	Average, non nodal, first moment of transverse displacement, divided by $L^2$

Table 6. The 9-DOF-BE Elastic Stiffness Matrix

$$K_{e9s} := \frac{4 \cdot E \cdot I}{L^3} \cdot \begin{pmatrix} 2 \cdot C_e & 0 & 0 & -3 \cdot C_e & 0 & 0 & C_e & 0 & 0 \\ 0 & 300 & 30 & 0 & -90 & 2520 & 0 & -210 & 15 \\ 0 & 30 & 4 & 0 & -15 & 210 & 0 & -15 & 1 \\ -3 \cdot C_e & 0 & 0 & 6 \cdot C_e & 0 & 0 & -3 \cdot C_e & 0 & 0 \\ 0 & -90 & -15 & 0 & 180 & 0 & 0 & -90 & 15 \\ 0 & 2520 & 210 & 0 & 0 & 25200 & 0 & -2520 & 210 \\ C_e & 0 & 0 & -3 \cdot C_e & 0 & 0 & 2 \cdot C_e & 0 & 0 \\ 0 & -210 & -15 & 0 & -90 & -2520 & 0 & 300 & -30 \\ 0 & 15 & 1 & 0 & 15 & 210 & 0 & -30 & 4 \end{pmatrix}$$

$$C_e = \frac{A}{2I} L^2 \therefore \text{if } A=1 \Rightarrow C_e = \frac{1}{2I} L^2 \quad (20)$$

### 2.8.1 Expediency and Importance of the Non Nodal DOF

Nodal degrees of freedom are precise points or directions at, along or around which only applied concentrated generalized loads could be accommodated. Conversely,

non nodal DOFs could accommodate actions operating on continuously distributed domains, i.e. complex geometric parameters other than localized directions.

Since a distributed load acts over a disperse domain in the element, it can be related to a dispersed displacement measured over the same domain. Non nodal degrees of freedom are "distributed directions" that could be defined by integrations or averages of geometric parameters (consistently and carefully chosen to avoid matrix singularity) over the element domain. Non nodal degrees of freedom cover the action of distributed loads, just as "concentrated directions or coordinates" are required to ensure continuity of the deformed configuration at the nodes, providing accommodation for complex loading cases without entailing to find equivalent nodal loads.

Deschapelles (2002) formulated a detailed description of a refined beam element with five degrees of freedom, one of them was the first non-nodal degree of freedom explicitly described in a finite element publication. Nevertheless, this type of degree of freedom had been used by him for quite a while. Deschapelles (1987) had shown before an explicit stiffness matrix containing non nodal DOF and making a reference to the development of his 12-DOF frame-shear element in Deschapelles (1978, 1984).

In the case of the 6DOF element used by BMTORSWP, this ground-breaking and handy non nodal DOF was able to capture entirely the uniformly distributed load in its corresponding consistent load vector. Thus, no loads components were projected to the other DOF. Moreover, the stiffness, and geometric matrixes were successfully developed, explicitly formulated and applied to a successful 2<sup>nd</sup> order analysis, and a soil contribution matrix was also provided for the solution of a wide range of EBEF problems. In fact, after a research spanning almost 2 years without the knowledge,

involvement or input of Deschappelles, a categorical statement can be made that no other reference to non nodal DOFs have been advanced in the literature of finite elements.

### **2.8.2 Adequacy of the 6-DOFs for the Transverse Displacement**

The 6-DOF-BE (which is a component of the 9-DOF-BE) corresponds to a 5<sup>th</sup> order polynomial acting in the field of transverse displacement  $v$ . The geometric and soil matrix are developed entirely in the field of transverse displacements, as well as the flexural elastic matrix. This element has a 5<sup>th</sup> order expansion with two nodal degrees of freedom at each end node plus two dimensionless degrees of freedom that symbolically are situated at midspan of each finite element.

The transverse displacement  $v$  of the beam-column is analogous to the angle of twist of a TWB with open cross section. The real open TWB under a linearly varying torque (Figure 26) undergoes a deformed rotational configuration governed by a 5<sup>th</sup> degree polynomial. Therefore, the beam-column should have a field of transverse displacements governed by an equal or higher-than a 5<sup>th</sup> degree polynomial.

The beam-column field of transverse displacements is ruled by a 5<sup>th</sup> degree polynomial. Thus, the 6DOF beam-column elastic line, which is a component of the 9DOF-BE can perfectly represent both a uniform and a linearly varying transverse load whose bending moments will be described at least by a 2<sup>nd</sup> and a 3<sup>rd</sup> degree polynomials respectively. On the other hand, the elastic lines will be described at least by a 4<sup>th</sup> and a 5<sup>th</sup> degree polynomials, respectively.

In fact, fields of transverse displacement containing powers lower than five cannot form the basis of a BC that provides a solution for general torsion problems. In addition, three derivatives are required to be produced in the output for design purposes.

Because, restrained warping shearing stresses depend on the angle of twist third derivative with respect to the axial coordinate, and warping normal stresses depend on the second derivative. The first derivative defines the free warping shearing stresses.

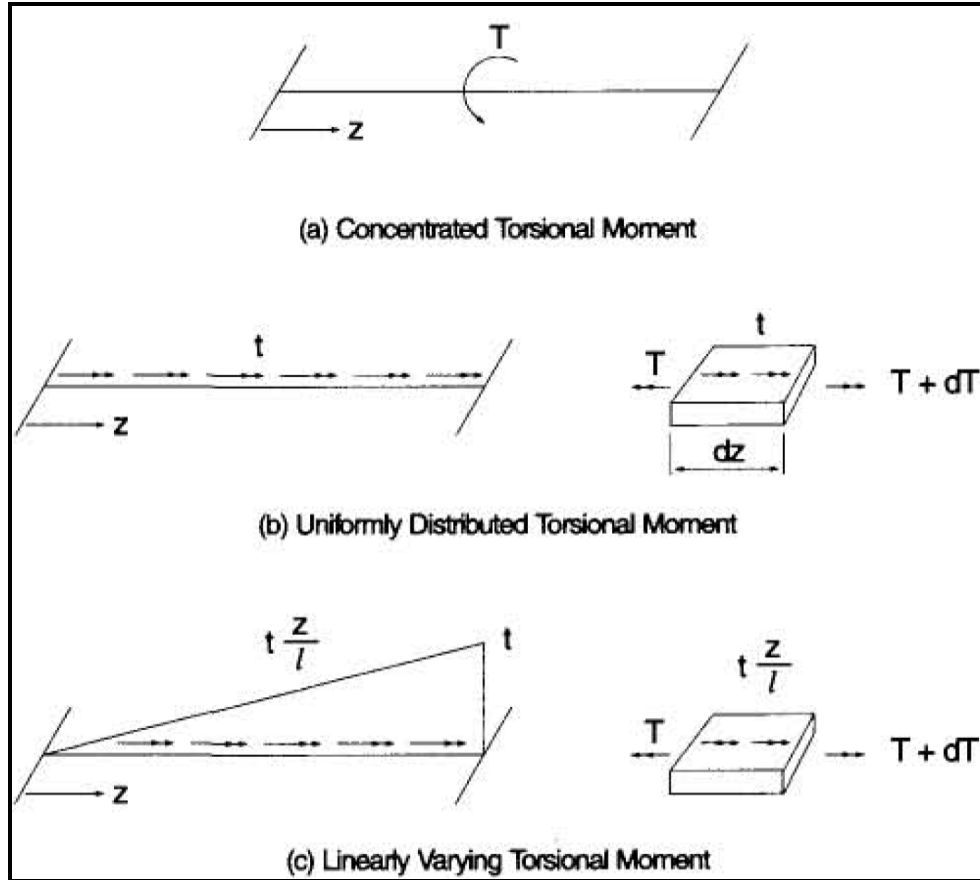


Figure 26. Torsional Loading Systems from AISC-DG9

It is pertinent to remark that this high order finite element can also accommodate in the corresponding non nodal load consistent vector shown in Equation (27) and Equation (28) all the torsional loading cases requested by the code just by using one algorithm. This additional feature makes the software application even more trustful.

### 2.8.3 Field of Transverse Displacements of the 6-DOF BE

The generic transversal field of displacements  $\{g\}$  at an arbitrary point along the 6-DOF BE is expressed with a 5<sup>th</sup> order polynomial in Equation (21). Where its variable of position  $\xi$  has been normalized, and it's  $\alpha$  coefficients are also called generalized

displacements. The generic displacement could also be expressed as a function of the Shape Function [N] shown in Equation (22) in a one column matrix version, as explained below. Identifying  $x$  instead of  $z$  as the longitudinal axis, the corresponding DOF are: First, the  $v(x(\xi))/L$  value for  $\xi = -1$ . Second, the  $v(x(\xi)),x$  value for  $\xi = -1$ . Third, the non-nodal integral  $(\int v(x(\xi))dx)/L^2$ . Fourth, the non-nodal integral  $(\int v(x(\xi)) \cdot x \cdot dx)/L^3$ . Fifth, the  $v(x(\xi))/L$  value for  $\xi = +1$ . Sixth, the  $v(x(\xi)),x$  value for  $\xi = +1$ .

$$v(\xi) \rightarrow \alpha_5 \cdot \xi^5 + \alpha_4 \cdot \xi^4 + \alpha_3 \cdot \xi^3 + \alpha_2 \cdot \xi^2 + \alpha_1 \cdot \xi + \alpha_0 \quad (21)$$

#### 2.8.4 Shape Function Matrix [N] of the 6-DOF BE

The generic displacement at an arbitrary point in the element  $\{g\}$  is numerically defined by the shape function matrix [N] pre-multiplied by the vector of specific (reference) displacements  $\{q\}$  along the DOFs. That is,  $\{g\} = \{q\} [N]$ . The shape function matrix [N] contains each of the interpolation polynomials of the position coordinate. [N] is shown in Equation (22) as a column vector for convenience. Where  $x = d\xi$ ,  $dx = ad\xi$ ,  $a = L/2$ , and  $L$  is the member length, and  $x$  is the longitudinal coordinate.

#### 2.8.5 Geometric Interpretation of the Shape Function [N]

The  $i^{\text{th}}$  shape function is the equation that describes the deformed configuration of the finite element elastic line when a unit displacement in the domain of the  $i^{\text{th}}$  degree of freedom is imposed while any other displacement along or around the rest of DOFs is prevented.

The  $i^{\text{th}}$  shape function is the equation describing the deformed configuration of the element elastic line when a unit displacement in the domain of the  $i^{\text{th}}$  degree of freedom is imposed when any other displacement along the rest of DOFs is prevented.



$$N := L \cdot \begin{pmatrix} \frac{21\xi^5}{8} - \frac{15\xi^4}{16} - 5\xi^3 + \frac{15\xi^2}{8} + \frac{15\xi}{8} - \frac{7}{16} \\ \frac{7\xi^5}{32} - \frac{5\xi^4}{32} - \frac{5\xi^3}{16} + \frac{3\xi^2}{16} + \frac{3\xi}{32} - \frac{1}{32} \\ \frac{15\xi^4}{8} - \frac{15\xi^2}{4} + \frac{15}{8} \\ \frac{105\xi^5}{4} - \frac{105\xi^3}{2} + \frac{105\xi}{4} \\ 5\xi^3 - \frac{15\xi^4}{16} - \frac{21\xi^5}{8} + \frac{15\xi^2}{8} - \frac{15\xi}{8} - \frac{7}{16} \\ \frac{7\xi^5}{32} + \frac{5\xi^4}{32} - \frac{5\xi^3}{16} - \frac{3\xi^2}{16} + \frac{3\xi}{32} + \frac{1}{32} \end{pmatrix} \quad (22)$$

### 2.8.6 The 6-DOF BE Elastic Matrix [B]

The elastic or strain matrix [B] is the second derivative of the shape function matrix expressed here with one subscript for each derivative [N<sub>xx</sub>]. Therefore, it is related to the generic curvature of the member elastic line. The elastic matrix is obtained by integrating the projections of all the infinitesimal stiffness contributions  $E \cdot I \cdot a d\xi$  along the member towards the geometric parameters chosen as the member DOFs. Where  $x = a\xi$ ,  $dx = a d\xi$ ,  $a = L/2$ ,  $L$  is the member length, and  $x$  is the longitudinal coordinate. Equation (23) states the analytical formulation of the elastic matrix, while its explicit numeric results appear in Equation (24) whose coordinate system has origin at midspan.

$$\int_{-1}^1 N_{xx} \cdot E \cdot I \cdot N_{xx}^T \cdot a d\xi \quad (23)$$

$$\left( \frac{4 \cdot E \cdot I}{L} \right) \cdot \begin{pmatrix} 300 & 30 & -90 & 2520 & -210 & 15 \\ 30 & 4 & -15 & 210 & -15 & 1 \\ -90 & -15 & 180 & 0 & -90 & 15 \\ 2520 & 210 & 0 & 25200 & -2520 & 210 \\ -210 & -15 & -90 & -2520 & 300 & -30 \\ 15 & 1 & 15 & 210 & -30 & 4 \end{pmatrix} \quad (24)$$

### 2.8.7 The 6-DOF BE Soil Normal Stiffness Contribution Matrix

The normal modulus of the foundation or soil modulus at interface is  $k_0$ . The 6-DOF-BE soil stiffness contribution matrix is expressed numerically in Equation (26) and was obtained by integrating Equation (25). Due to the Boussinesq effect, it is not an uncommon engineering practice to assign a unit value to the effective convex interface width  $b$  according to Terzaghi (1973); thus,  $k_0$  and  $k$  could be numerically equal except for their units:

$$\int_{-1}^1 N \cdot k_0 \cdot b \cdot N^T \cdot a \, d\xi \quad (25)$$

$$\frac{k_0 \cdot b \cdot L^3}{13860} \begin{pmatrix} 3000 & 140 & -2970 & 16800 & -30 & -25 \\ 140 & 8 & -165 & 630 & 25 & -3 \\ -2970 & -165 & 19800 & 0 & -2970 & 165 \\ 16800 & 630 & 0 & 352800 & -16800 & 630 \\ -30 & 25 & -2970 & -16800 & 3000 & -140 \\ -25 & -3 & 165 & 630 & -140 & 8 \end{pmatrix} \quad (26)$$

### 2.8.8 Load Consistent Vectors for the 6-DOF BE

The effect of infinitesimal loads applied along or around the element domain can be projected over the DOFs directions by means of the interpolation polynomials contained in the shape functions. The former operation provides a load consistent vector for any kind of distributed load. Particular distributed loads can be exclusively projected over distributed degrees of freedom. Different distributed normalized loads from a constant uniform load to one with a 5<sup>th</sup> order polynomial variation, are shown in Equation (27), where  $w$  is a distributed constant load expressed in lb/in (force/length), and  $W$  is a vector with powers of varying distributed loads.

In the case of a 6-DOF-BE, two load consistent vectors are shown in the first and second columns of the Equation (28) in matrix form one for a constant and other for a linearly varying distributed loads. It could be noticed that these 2 loadings do not project to DOFs other than the 3<sup>rd</sup> and 4<sup>th</sup> non nodal DOFs previously discussed.

This outstanding feature could very well be exploited in a number of finite element formulations. As it was already said, this fact was addressed in Deschappelles (2002). Unfortunately, in earlier papers, Deschappelles (1984, 1985, and 1987) restrained himself to just introduce finite element formulations of his own containing these suitable non nodal geometric parameters without a major discussion of their remarkable properties.

A great advantage of these non-nodal DOFs is that, in this particular problem, they could abridge in only one algorithm solutions for the 3<sup>rd</sup> and 4<sup>th</sup> order differential equations of members subjected to two different distributed loads. Concentrated loads in different locations could be handled by assigning each one node. Thus, it is not longer necessary to develop one algorithm for each version of Equation (17) for the BEF behavior or their corresponding two analog Equations (19) for the TWB behavior.

As a result of what has been previously stated, the  $i^{\text{th}}$  column of any stiffness matrix expresses the system of forces in the coordinates of all the element degrees of freedom compatible with a unit displacement along the  $i^{\text{th}}$  degree of freedom without any displacement in the coordinates of all other DOFs.

$$W^T \rightarrow \begin{pmatrix} w \\ \xi \cdot w \\ \xi^2 \cdot w \\ \xi^3 \cdot w \\ \xi^4 \cdot w \\ \xi^5 \cdot w \end{pmatrix} \quad (27)$$

$$\frac{1}{(L^2 \cdot w)} \cdot \left( \int_{-1}^1 N \cdot W \cdot a \, d\xi \right) \rightarrow \begin{pmatrix} 0 & 0 & \frac{2}{21} & -\frac{1}{21} & \frac{8}{105} & -\frac{34}{693} \\ 0 & 0 & \frac{1}{210} & -\frac{1}{630} & \frac{1}{315} & -\frac{1}{693} \\ 1 & 0 & \frac{1}{7} & 0 & \frac{1}{21} & 0 \\ 0 & 2 & 0 & \frac{2}{3} & 0 & \frac{10}{33} \\ 0 & 0 & \frac{2}{21} & \frac{1}{21} & \frac{8}{105} & \frac{34}{693} \\ 0 & 0 & -\frac{1}{210} & -\frac{1}{630} & -\frac{1}{315} & -\frac{1}{693} \end{pmatrix} \quad (28)$$

## 2.9 Software Application

BMCOLD or BMCOLDGP, under the name of BMTORSW or BMTORSWSP is the principal software application that will be used to analyze multi-span beams under mixed torsion. BMCOLDGP will be eventually used to analyze the contributions of bending shear and normal stresses.

BMTORSWSP provides the opportunity to bypass the complexities in the restrained warping analysis. And it predicts the restrained warping behavior better than that of the beam-column on elastic foundation (due to the absence of the soil hard-to-predict-behavior) and does not demand much of a computer memory or time.

BMCOLD (by Deschappelles) is a double precision adaptation of the program ZAPEL (by Deschappelles). It is a rather small (92KB) software application for Windows (32 bits) able to undertake the analysis of a bar in mixed bending and axial loading. It

can also be used to solve problems of the general torsion theory due to the BC analogy previously discussed. BMCOLD is based upon a nine degree-of freedom-beam-column element (9-DOF-BE) including 3 non nodal smeared DOFs.

As it has been seen before, the solution of restrained warping problems under distributed non uniform torsional loads requires a field of displacements expressed in a polynomial of 5<sup>th</sup> or higher order. That is to accommodate uniform and linearly varying torsional loads prescribed in manuals including Seaburg (1997) or Heins (1963.)

BMCOLD considers a continuous soil modulus that may vary according to 3 parameters between the element nodes, and a geometric parameter from an axial force considered positive in compression. The program works with or without the elastic foundation. However, the elastic soil input data is not necessary in open TWB torsion analysis but it is necessary in box girders distortion analysis.

Input forms show  $x$  as the longitudinal axis. BMCOLD processes applied concentrated forces parallel to the longitudinal axis  $x$ , parallel to transverse axis  $y$  and applied concentrated moments around axis  $z$ . End nodes of each element are numbered NI and NJ, where number NI is smaller than number NJ; and the local  $x$  axis is defined by the direction going from node NI to node NJ (Deschappelles, 2008-2011).

Information on nodes with any type of displacement restraint must be provided for any node N, three integers are given to cover conditions along the 3 global axes. Integer +1 denotes full restraint, zero indicates no restraint at all, and -1 implies the existence of a spring; thus, the number of springs increases each time the integer -1 is entered.

The data of spring constants must be given if an input on some negative integers have been written to indicate the existence of elastic restraints; otherwise the information of prescribed non zero displacements is omitted.

For each node  $n$ , an integer is given to define the direction along which the restraint operates or the displacement is specified. Integers 1, 2 and 3 correspond to directions  $x$  (longitudinal),  $y$  (vertical transverse) and  $z$  (lateral transverse) respectively. Values of spring constants or specified displacements are also specified in the input (Deschappelles, 2008-2011).

BMCOLDG is the version of BMCOLD able to export graphic data for applications like excel. BMTORSW is a 32-bit-software application adapted from the 32-bit-software application BMCOLDG, and named after beam-torsion-warping. While, BMTORSWP is a 64-bit-software application adapted from the 64-bit-software application BMCOLDGP, and named after beam-torsion-warping Portland.

## CHAPTER 3

### OBJECTIVES AND METHOD

To recommend a finite element size and to undertake single-span and multi-span problems including a single-span performance experiment by using the recommended size. Some user friendly software features and conclusions and recommendations will be also provided. The process will be undertaken in an orderly, logical, systematic fashion in accordance with the AISC-DG-9. Obrébski (2005) states that the application of the finite element method to single span bar analysis and torsion analysis can produce errors up to 394% and 270% respectively.

#### 3.1 Scope

This study is concerned with the analysis of thin-walled beams with open cross-sections subjected to first order mixed torsion. A software application based on the EBEF analogy and able to handle 3 parameter soils with axial loadings will be used.

This work is not concerned with the analysis of box girders, which are thin walled beams with closed sections whose governing differential equation is analogous to that of an EBEF as pointed by Vlasov (1961), Wright and Abdel-Samad (1968) and Hsu (1995) and confirmed by tests according to Heins (1981). However, it should be also pointed that the proposed software application is able to handle box girder distortional problems.

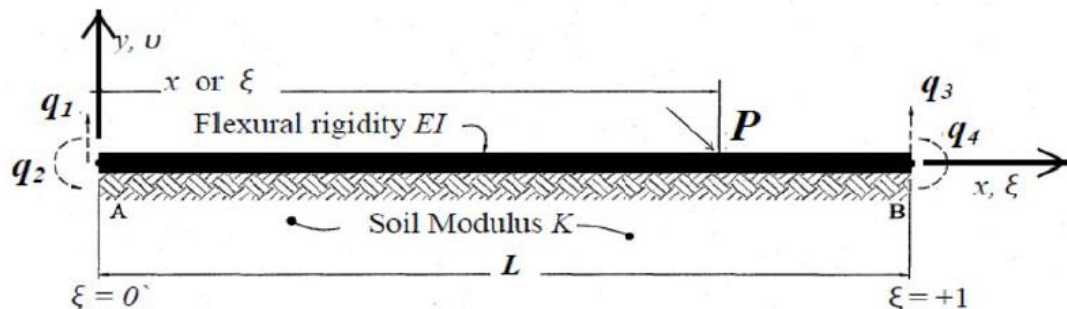


Figure 27. Four-DOF-BEF with Origin at Left Node

## 3.2 Methodology

The process will be undertaken in an orderly, logical, systematic manner. First, a finite element size calibration will be performed by means of a convergence study regarding the tensile geometric parameter of the EBC—analogue to  $G \cdot J$  in the TWB. Then, maximum and minimum dimensions will be recommended for the finite element.

A challenging performance experiment with an open TWB single supported at midspan under a torsional load at far end will be performed and analyzed with the software.

In this experiment, two conditions prone to errors pointed by Obrébski (2005) will be simultaneously fulfilled: This will be challenging, due to the fact that the application of the finite element method to single-span bars and torsion analysis can produce errors up to 394% and 270% respectively according to Obrébski (2005).

Examples 5.1, 5.4, 5.5 from the AISC DG 9, Boothby (1984), and Medwadowski (1985) will be reanalyzed and compared. Computations, graphics error comparisons, adjustments, printing, drawings, tables, and standards will be provided. No corrections will be imposed to the problems original data.

Comparisons will be made and conclusions and recommendations will also be provided. Finally, input forms, the Software Application Manual, and conclusions and recommendations will be provided

### 3.2.1 Convergence Study

The finite element used by the software is composed by two uncouple fields represented in a 3DOF rod element and a 6 DOF beam element. The 6 DOF-BC



described before is the tool that provides the algorithm to solve the 1<sup>st</sup> order general torsion problem analogous to the 2<sup>nd</sup> order beam column problem.

On the other hand, to study the convergence, one modus operandi could be to develop or select the exact stiffness matrix for the first order general torsion problem and compare its entries with those of the finite 6DOF-BE. For that purpose, the local axis and the DOFs must be similar in both matrixes.

Unfortunately to develop an exact 6x6-matrix with the DOFs prescribed in the 6DOF-BE will be a very difficult task due to the more complex nature of the non-nodal degrees of freedom.

Instead, the elastic and geometric finite element 6x6-matrixes will be transformed into local axis with origin at the left node of the member and then condensed to a 4x4-matrix. The transformation to local axis will be made independently for both the elastic and geometric matrixes.

Therefore, Equations (24) and (26) whose local axis coordinate systems have origin at midspan have been transformed into Equations (29) and (31) whose coordinate systems have origin at left node according to Figure 27. In Equation (30) appears a common factor.

$$K_{e6} := \frac{E \cdot I}{L} \cdot \begin{pmatrix} 1200 & 120 & -5400 & 10080 & -840 & 60 \\ 120 & 16 & -480 & 840 & -60 & 4 \\ -5400 & -480 & 25920 & -50400 & 4680 & -360 \\ 10080 & 840 & -50400 & 100800 & -10080 & 840 \\ -840 & -60 & 4680 & -10080 & 1200 & -120 \\ 60 & 4 & -360 & 840 & -120 & 16 \end{pmatrix} \quad (29)$$

$$\lambda := L \cdot \sqrt{\frac{N}{E \cdot I}} \quad (30)$$

$$K_{g6} := \lambda^2 \cdot \begin{pmatrix} \frac{100}{7} & \frac{23}{42} & -\frac{410}{7} & 100 & -\frac{40}{7} & \frac{5}{42} \\ \frac{23}{42} & \frac{4}{63} & -\frac{44}{21} & \frac{10}{3} & -\frac{5}{42} & -\frac{1}{126} \\ -\frac{410}{7} & -\frac{44}{21} & \frac{2080}{7} & -560 & \frac{290}{7} & -\frac{26}{21} \\ 100 & \frac{10}{3} & -560 & 1120 & -100 & \frac{10}{3} \\ -\frac{40}{7} & -\frac{5}{42} & \frac{290}{7} & -100 & \frac{100}{7} & -\frac{23}{42} \\ \frac{5}{42} & -\frac{1}{126} & -\frac{26}{21} & \frac{10}{3} & -\frac{23}{42} & \frac{4}{63} \end{pmatrix} \quad (31)$$

$$K_{red} \rightarrow \begin{pmatrix} 12 & 6 & -12 & 6 \\ 6 & 4 & -6 & 2 \\ -12 & -6 & 12 & -6 \\ 6 & 2 & -6 & 4 \end{pmatrix} \quad (32)$$

$$K_{gred} \rightarrow \begin{pmatrix} \frac{15 \cdot \lambda^2}{14} & \frac{\lambda^2}{28} & -\frac{15 \cdot \lambda^2}{14} & \frac{\lambda^2}{28} \\ \frac{\lambda^2}{28} & \frac{3 \cdot \lambda^2}{70} & -\frac{\lambda^2}{28} & -\frac{\lambda^2}{140} \\ -\frac{15 \cdot \lambda^2}{14} & -\frac{\lambda^2}{28} & \frac{15 \cdot \lambda^2}{14} & -\frac{\lambda^2}{28} \\ \frac{\lambda^2}{28} & -\frac{\lambda^2}{140} & -\frac{\lambda^2}{28} & \frac{3 \cdot \lambda^2}{70} \end{pmatrix} \quad (33)$$

Equations (32) and (33) present the finite element 4x4 matrixes with origin at left node were condensed independently from each other. It is important to make clear that the condensation must not be made independently for each matrix. This procedure is prone to error. In this work, instead, both the elastic and geometric 6x6 matrixes will be combined and subsequently condensed. See Equations (39) and (40).

Lighthfoot (1961) and Gere (1963) developed generalized equations for 2<sup>nd</sup> order beam-columns under mixed bending and axial loading. Petterson (1955), Chaudbary

(1982), Boothby (1984), Dvorkin (1988), and others developed generalized stiffness matrix for thin walled bars or just elastic line formulas like those in Roark (1982).

Nevertheless, the explicit version of the exact stiffness matrix corresponding to the solution of the governing differential equation of a prismatic beam under restraining warping developed by Medwadowski (1985) has been chosen in this work and it is reproduced in Equation (34). The complete entries of Medwadowski matrix are defined in Equations (35), (36), (37), and (38). Medwadowski matrix entries have been rearranged and normalized into dimensionless expressions according to the coordinate and DOF systems used for the development of the 6DOF finite element built into the software application BMTORSWP.

$$K_{DE} := \begin{pmatrix} -\delta(\lambda) & \gamma(\lambda) & \delta(\lambda) & \gamma(\lambda) \\ \gamma(\lambda) & \alpha(\lambda) & -\gamma(\lambda) & \beta(\lambda) \\ \delta(\lambda) & -\gamma(\lambda) & -\delta(\lambda) & -\gamma(\lambda) \\ \gamma(\lambda) & \beta(\lambda) & -\gamma(\lambda) & \alpha(\lambda) \end{pmatrix} \quad (34)$$

$$\alpha(\lambda) := \frac{\lambda \cdot \sinh(\lambda) - \lambda^2 \cdot \cosh(\lambda)}{2(\cosh(\lambda) - 1) - \lambda \cdot \sinh(\lambda)} \quad (35)$$

$$\beta(\lambda) := \frac{\lambda^2 - \lambda \cdot \sinh(\lambda)}{2(\cosh(\lambda) - 1) - \lambda \cdot \sinh(\lambda)} \quad (36)$$

$$\gamma(\lambda) := \frac{\lambda^2 - \lambda^2 \cdot \cosh(\lambda)}{2(\cosh(\lambda) - 1) - \lambda \cdot \sinh(\lambda)} \quad (37)$$

$$\delta(\lambda) := \frac{\lambda^3 \cdot \sinh(\lambda)}{2(\cosh(\lambda) - 1) - \lambda \cdot \sinh(\lambda)} \quad (38)$$

Equation (39) shows the first two columns and Equation (40) shows the last two columns of the condensed finite element stiffness matrix. Each entry combines both the elastic and geometric effect in a 4x4 matrix with origin at left node condensed from a 6x6

matrix with non nodal DOFs. Ten figures, each containing graphics of both the exact stiffness matrix for the thin-walled beam under mixed torsion and the finite element stiffness matrix for the analogue beam-column under mixed bending and axial load will be provided. Each chart corresponds to one of the 10 symmetric entries presented as functions of  $\lambda$  as defined in Equation (30). The corresponding form of  $\lambda$  for the analogue thin-walled beam is  $\lambda = Le/a = Le\sqrt{(G J/ E C_w)}$  as per the AISC definition. See Figure 28 to Figure 37 .

$$\left( \begin{array}{cc} \frac{15\lambda^4 + 1680\lambda^2 + 15120}{14\lambda^2 + 1260} & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \cdot & \cdot \\ \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \frac{3\lambda^6 + 1050\lambda^4 + 72240\lambda^2 + 1058400}{70\lambda^4 + 9240\lambda^2 + 264600} & \cdot & \cdot \\ \frac{15\lambda^4 + 1680\lambda^2 + 15120}{14\lambda^2 + 1260} & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \cdot & \cdot \\ \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \frac{210\lambda^4 - \lambda^6 + 19320\lambda^2 + 1058400}{140\lambda^4 + 18480\lambda^2 + 529200} & \cdot & \cdot \end{array} \right) \quad (39)$$

$$\left( \begin{array}{cc} \cdot & \cdot & \frac{15\lambda^4 + 1680\lambda^2 + 15120}{14\lambda^2 + 1260} & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} \\ \cdot & \cdot & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \frac{210\lambda^4 - \lambda^6 + 19320\lambda^2 + 1058400}{140\lambda^4 + 18480\lambda^2 + 529200} \\ \cdot & \cdot & \frac{15\lambda^4 + 1680\lambda^2 + 15120}{14\lambda^2 + 1260} & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} \\ \cdot & \cdot & \frac{\lambda^4 + 420\lambda^2 + 15120}{28\lambda^2 + 2520} & \frac{3\lambda^6 + 1050\lambda^4 + 72240\lambda^2 + 1058400}{70\lambda^4 + 9240\lambda^2 + 264600} \end{array} \right) \quad (40)$$

A more accurate convergence study will be made with the condensed matrix shown in Equations (39) and (40), which show a matrix very different from that of Equation (33). Current recommendations on the element size are for analyses of EBEF

with BMCOLDG, are based on a convergence study related to the soil parameter made with the 4x4 matrix developed exclusively from nodal DOFs as an upper limit for accuracy purposes.

Figure 28, Figure 29, Figure 30, Figure 31, Figure 32, Figure 32, Figure 33, Figure 34, Figure 35, Figure, 36, and figure 37 each contain the curves showing one entry of the finite element stiffness matrix (continuous curve) and its corresponding entry of the exact stiffness matrix in the 4DOF-TWB (dotted curve). More entries are not needed due to the symmetry of the matrixes.

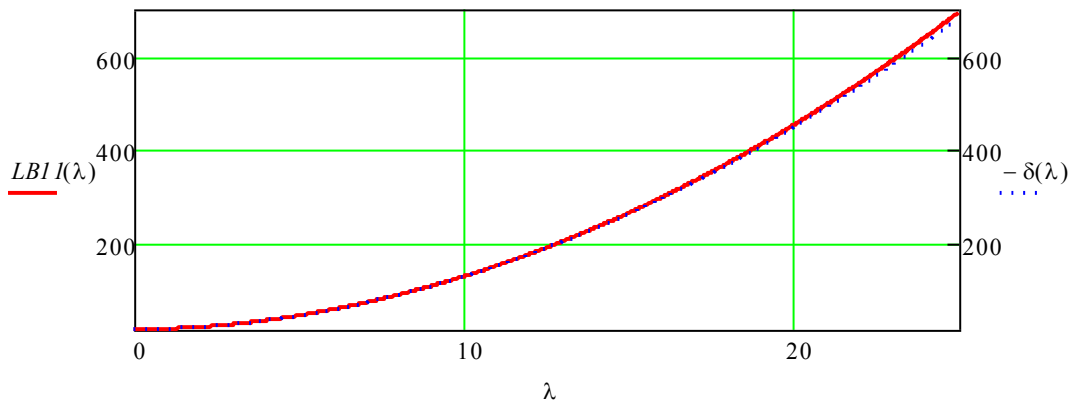


Figure 28. Entry 1-1 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

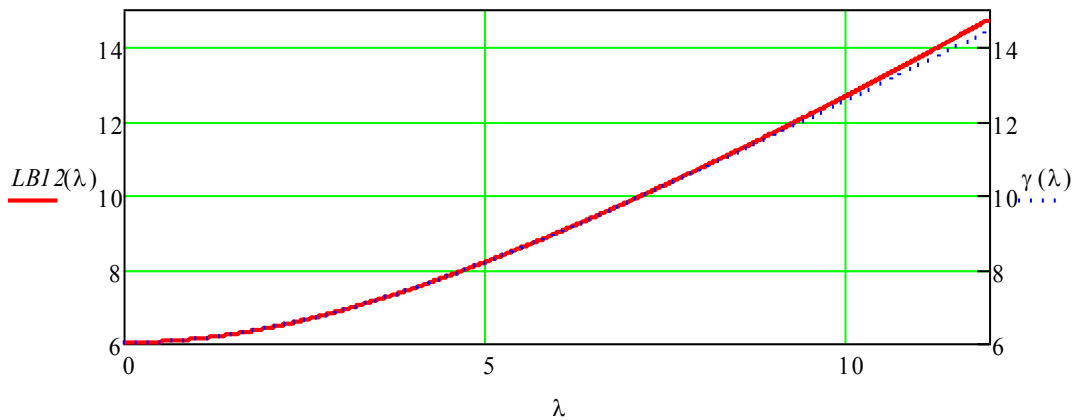


Figure 29. Entry 1-2 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

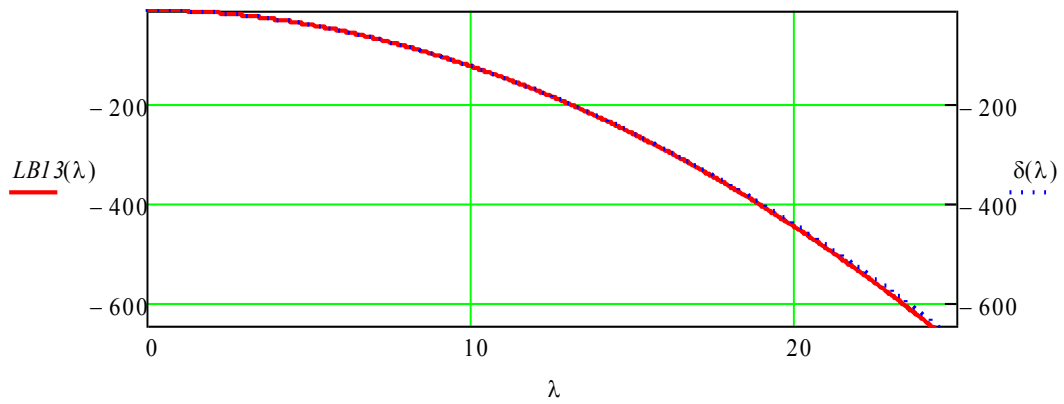


Figure 30. Entry 1-3 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

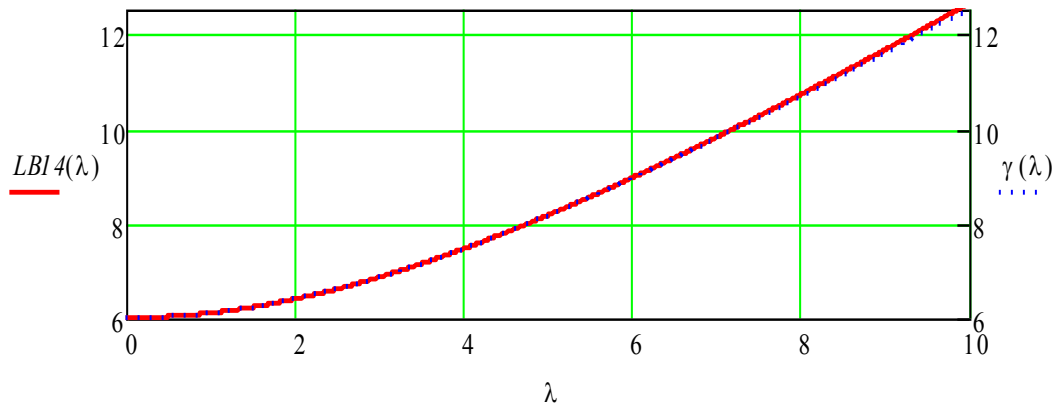


Figure 31. Entry 1-4 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

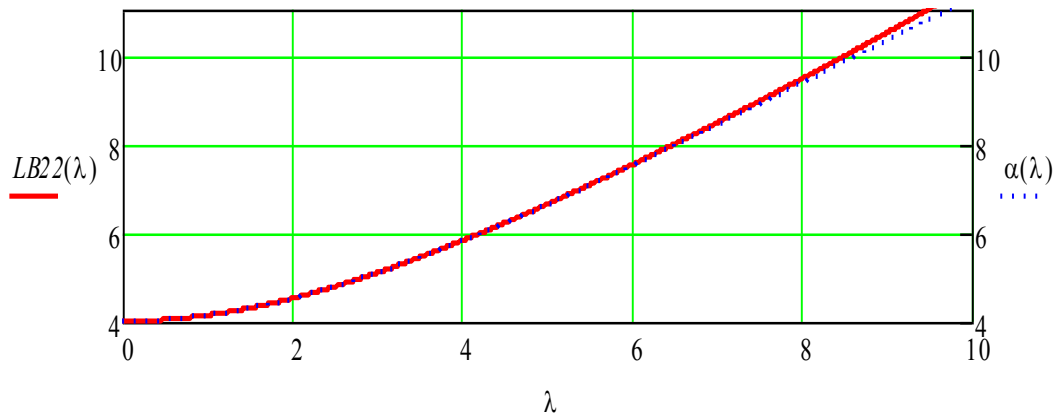


Figure 32. Entry 2-2 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

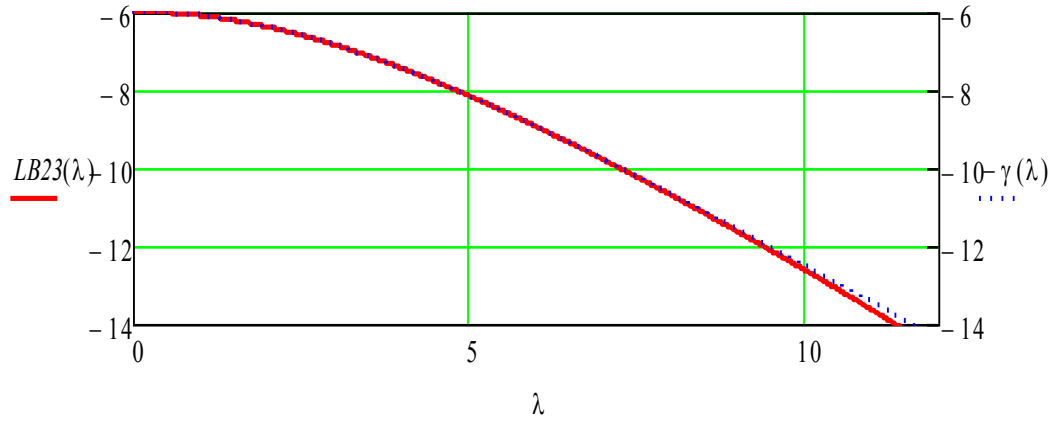


Figure 33. Entry 2-3 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

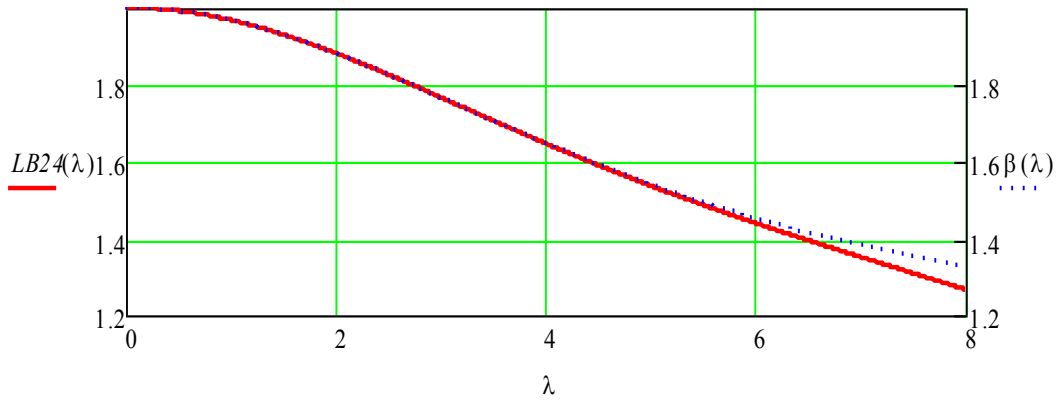


Figure 34. Entry 2-4 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

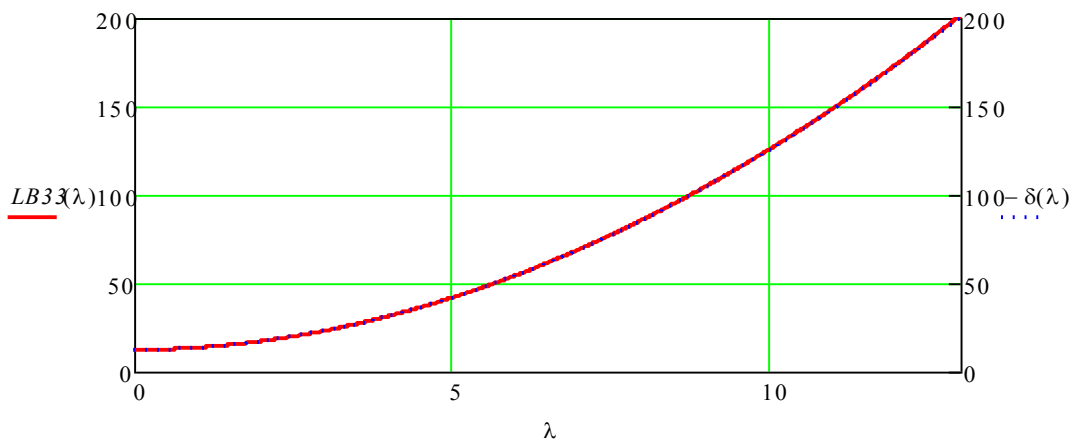


Figure 35. Entry 3-3 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

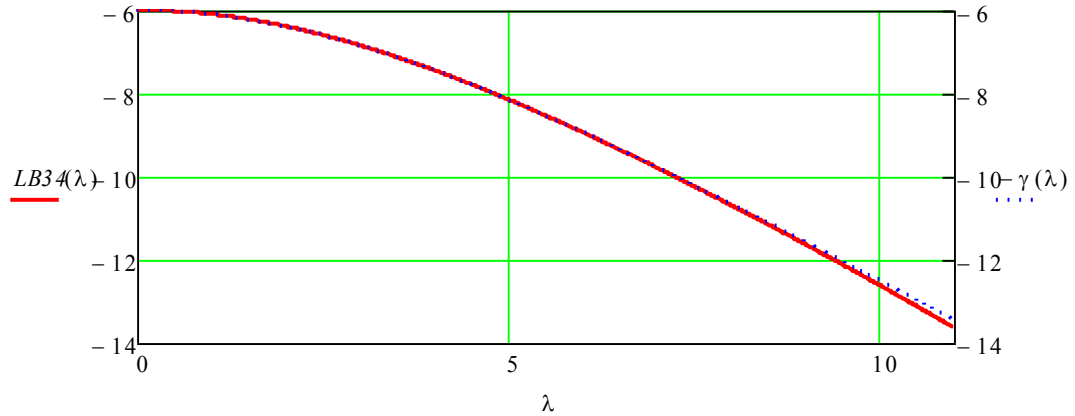


Figure 36. Entry 3-4 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

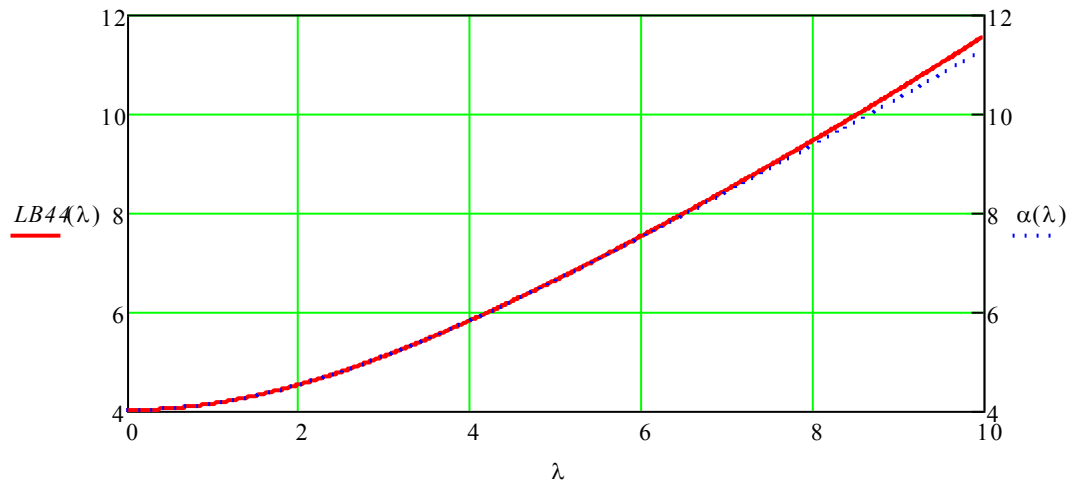


Figure 37. Entry 4-4 in the FE and Exact Stiffness Matrixes in the 4DOF-TWB

Entry Relative Errors of FE Condensed Stiffness Matrix at specific  $\lambda$  values will be discussed. For  $\lambda = 10$ , an unacceptable maximum error of 13% emerges in the stiffness entries  $LB24$  related to the DOFs 2 and 4 (slopes or minor angles in the BC elastic line). For a  $\lambda = 6$ , accuracy improves 15 times with a maximum error of 0.9%. For a  $\lambda = 5$ , accuracy improves 45 times with a maximum error of 0.29%. For a  $\lambda = 4$ , accuracy improves 200 times with a maximum error of 0.065%. For a  $\lambda = 2$ , the maximum error is 0.0004% and accuracy is not expected to keep improving considering other factors influencing errors. Even though, convergence can improve in theory, errors also depend on factor including but not limited to the final size of the global matrix



assembled (Wilkinson, 1963), as well as the precision of the algorithm built in the software. On the other hand, it is not a best practice to multiply the number of FEs ad infinitum. A discussion on this topic is not the theme of this work.

A preliminary advice would be to use  $\lambda$  ( $Le/a$ ) values between 2 and 4. Thus,  $2 \leq \lambda = Le/a \leq 4$  for the TWB. Thus,  $2a \leq Le \leq 4a$ , as it is shown in Equation (41). The minimum number of elements in a given span should be an integer  $Ne > (L/Le)$ .

Table 7. Relative % Entry Error in Condensed Matrix for Different  $\lambda$  Values

$1/a = \lambda$	Matrix of % Entry Errors
10	$\begin{pmatrix} 0.1894 & 0.9482 & 0.1894 & 0.9482 \\ 0.9482 & 2.5332 & 0.9482 & -13.3456 \\ 0.1894 & 0.9482 & 0.1894 & 0.9482 \\ 0.9482 & -13.3456 & 0.9482 & 2.5332 \end{pmatrix}$
6	$\begin{pmatrix} 0.0200 & 0.0602 & 0.0200 & 0.0602 \\ 0.0602 & 0.2481 & 0.0602 & -0.9083 \\ 0.0200 & 0.0602 & 0.0200 & 0.0602 \\ 0.0602 & -0.9083 & 0.0602 & 0.2481 \end{pmatrix}$
5	$\begin{pmatrix} 0.0075 & 0.0191 & 0.0075 & 0.0191 \\ 0.0191 & 0.0915 & 0.0191 & -0.2914 \\ 0.0075 & 0.0191 & 0.0075 & 0.0191 \\ 0.0191 & -0.2914 & 0.0191 & 0.0915 \end{pmatrix}$
4	$\begin{pmatrix} 0.0021 & 0.0043 & 0.0021 & 0.0043 \\ 0.0043 & 0.0240 & 0.0043 & -0.0653 \\ 0.0021 & 0.0043 & 0.0021 & 0.0043 \\ 0.0043 & -0.0653 & 0.0043 & 0.0240 \end{pmatrix}$
3	$\begin{pmatrix} 0.00033 & 0.00055 & 0.00033 & 0.00055 \\ 0.00055 & 0.00363 & 0.00055 & -0.00833 \\ 0.00033 & 0.00055 & 0.00033 & 0.00055 \\ 0.00055 & -0.00833 & 0.00055 & 0.00363 \end{pmatrix}$
2	$\begin{pmatrix} 0.00002 & 0.00003 & 0.00002 & 0.00003 \\ 0.00003 & 0.0002 & 0.00003 & -0.00039 \\ 0.00002 & 0.00003 & 0.00002 & 0.00003 \\ 0.00003 & -0.00039 & 0.00003 & 0.0002 \end{pmatrix}$
1	$\begin{pmatrix} 0.000000 & 0.000000 & 0.000000 & 0.000000 \\ 0.000000 & 0.000001 & 0.000000 & -0.000002 \\ 0.000000 & 0.000000 & 0.000000 & 0.000000 \\ 0.000000 & -0.000002 & 0.000000 & 0.000001 \end{pmatrix}$

$$Le \leq 2\sqrt{(EC_w/GJ)} = 2a < L/3 \quad (41)$$

$$Ne > 0.5L\sqrt{(GJ/EC_w)} = 0.5L/a > 3 \quad (42)$$

A similar study of the stiffness coefficients for a 4DOF-BC was made, and the comparison of the respective stiffness matrixes error-wise is shown and commented in Figure 38 and Figure 39. From the point of view of computer memory, it could be said that the 6DOF-BE increases accuracy and saves computer memory spaces.

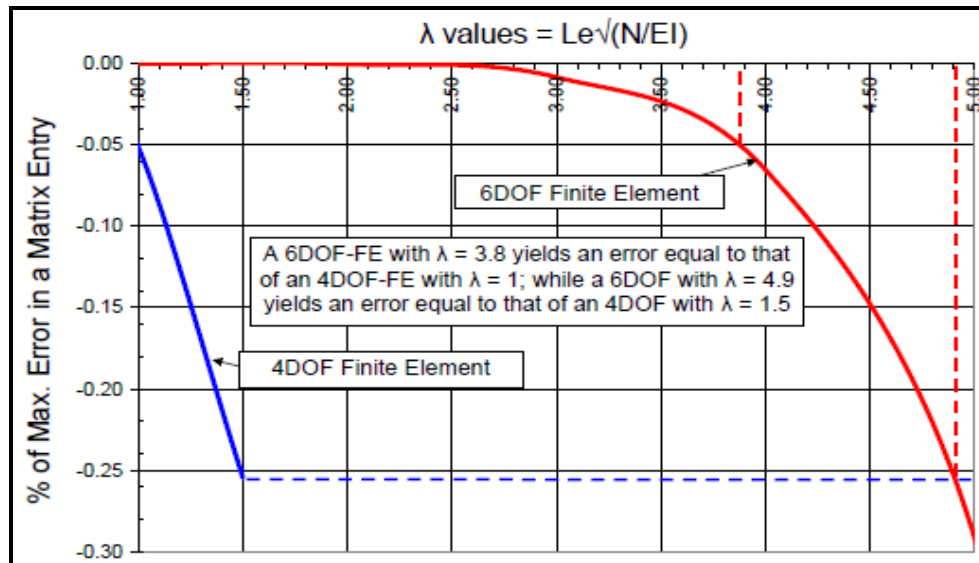


Figure 38. Four and 6DOF FE Stiffness Matrix Convergence for BC

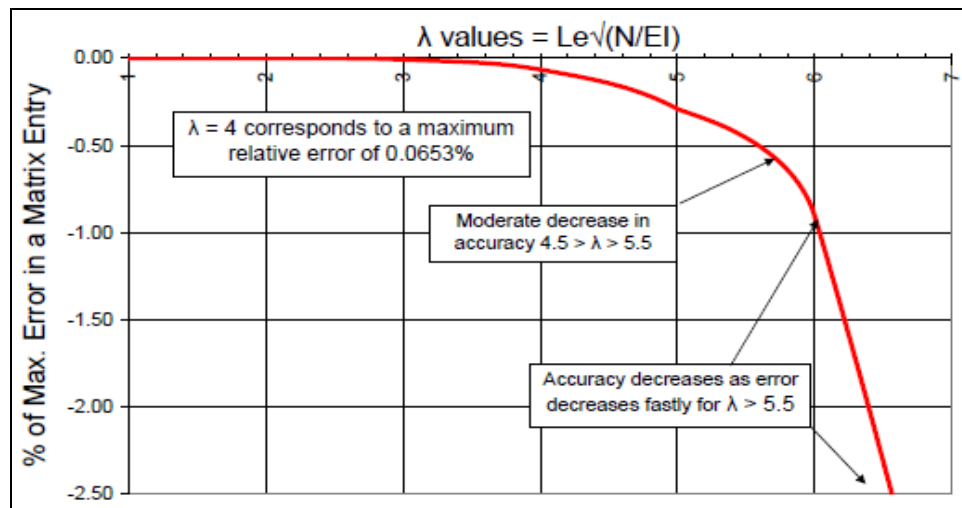


Figure 39. Six-DOF FE Stiffness Matrix Convergence for BC

As a sweeping statement, it could be said that the accuracy of the 6DOF beam column is 3.8 to 4 times that of the 4 DOF beam column. Only one 6DOF-BE is required to achieve the accuracy of four-4DOF-BE or five 6DOF-BE are required to achieve the accuracy of nineteen 4DOF-BE.

Therefore, one 6x6 matrix requires  $18 \times 1 + 3 = 21$  entries in a global matrix upper triangle and four 4x4 matrixes require  $7 \times 4 + 3 = 31$  entries in a global matrix upper triangle respectively. Similarly, five 6x6 matrixes require  $18 \times 5 + 3 = 93$  entries in a global matrix upper triangle and nineteen 4x4 matrixes require  $7 \times 19 + 3 = 135$  entries in a global matrix upper triangle respectively.

This difference will increase linearly as the number of required elements increases. In general the global matrix assembled with a higher degree finite element requires using less matrix entries in the memory to achieve a given accuracy; thus, the 6DOF is more convenient.

This advantage is in addition to the ones already discussed regarding the two non nodal DOFs properties, and the fact that for a solution of the mixed torsion problem a higher degree polynomial is mandatory.

### **3.2.2 List of Case Studies**

The examples to be solved will be the following: Two parametric experiments of a bar restrained under two different restraints at midspan and subjected to a torque at right end, and example analytically solved by Ugural (1987), examples 5.1, 5.4 and 5.5 from the AISC Design Guide 9, an example from Boothby (1984), an example from Medwadowski (1985). Trials with the finite elements size recommended in the convergence study will be undertaken and mesh adjustments will be made if any.

### 3.2.3 Case Study One

The first case study is a parametric experiment where a TWB is cantilevered at midspan and subjected to a 1k-in torque at far end. The data is as follows  $L = 15$  in at each side, of the support.  $E = 29000$  ksi,  $C_w = 0.01$  in<sup>6</sup>,  $GJ = 10$  kip-in<sup>2</sup>. The minimum element size is  $2\sqrt{EC_w/GJ} = 2\sqrt{290/10} \sim 11$  from Equation (41). The number of elements per span must be an integer larger than 15/11. As two elements did not work out, 3 elements were used from then on. Obrébski (2005) has commented on the difficulty presented by finite element formulations in the case of TWB and one span bars.

The cross section shown in Figure 40 is not that of the real bar in the parametric experiment. It just provides an order of magnitude for the bogus cross section torsional properties in order to make verisimilar the data handling.

Figure 40 shows the analog EBC subdivided in finite elements, Figure 41 shows an exaggerated EBC elastic line and Figure 42 shows the torque graph. More information can be found in Appendix A.

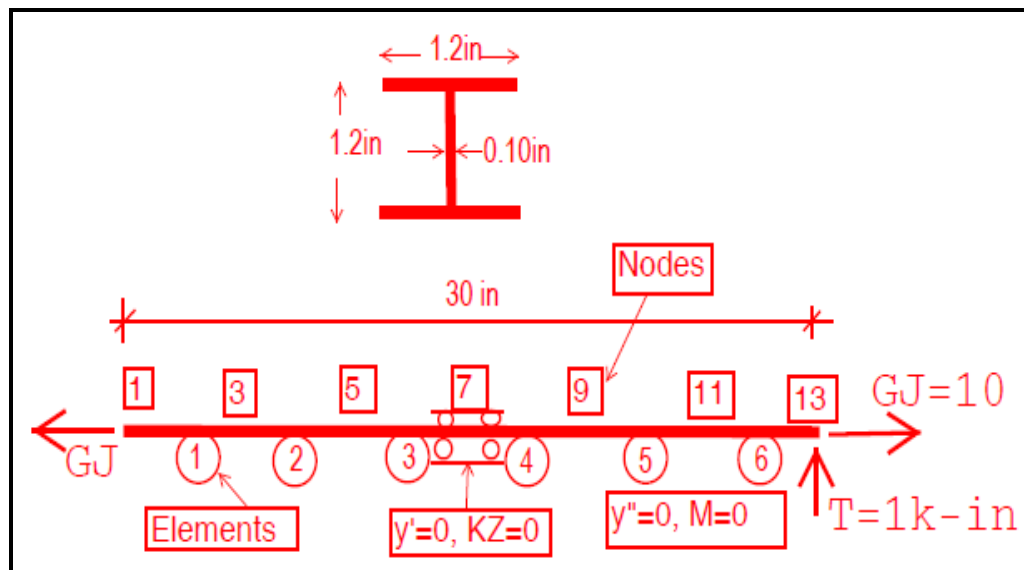


Figure 40. Model of Parametric Experiment to Test the Software

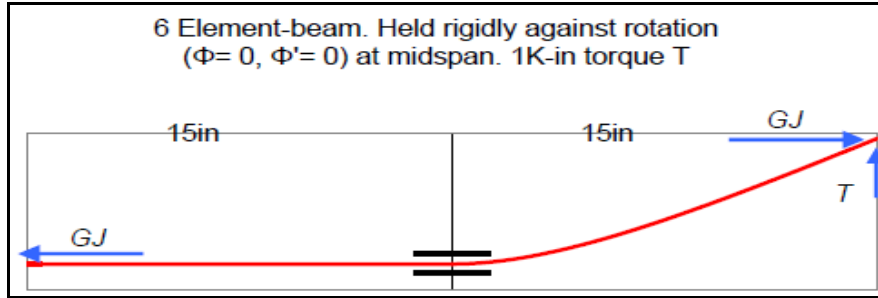


Figure 41. Elastic Line of the Beam Column Analogous to the TWB

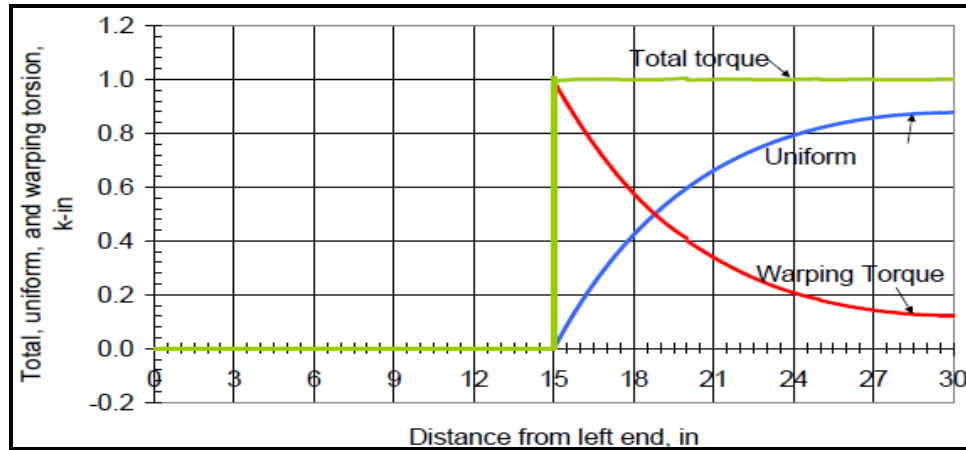


Figure 42. Asymptotic Behavior near Restraints from BMTORSWP

### 3.2.4 Case Study Two

This second case is a parametric experiment where the TWB is not cantilevered but just restrained against rotation at midspan. The data is the same as the first experiment. The results are shown in APPENDIX B including pictures; the input model, data, and forms; output data; and charts.

Three effects detected in the parametric experiment have been corroborated in the analysis: First, there is twist at the free unloaded near end; second, the twist at far end is the largest; and third, there could be a zero external torque with restrained warping along the full first span as seen in Figure 43 and Figure 44 and Figure 47. In addition to the contents in the respective APPENDIX B, more materials illustrating this problem could be found from Error! Reference source not found. to Error! Reference source not found.

These results are crucial to reformulate any previous advice on element size, particularly in the case of single span bars. Obrébski (2005) has commented on the inconvenience of both TWB and single span bars in the case of finite element formulations.

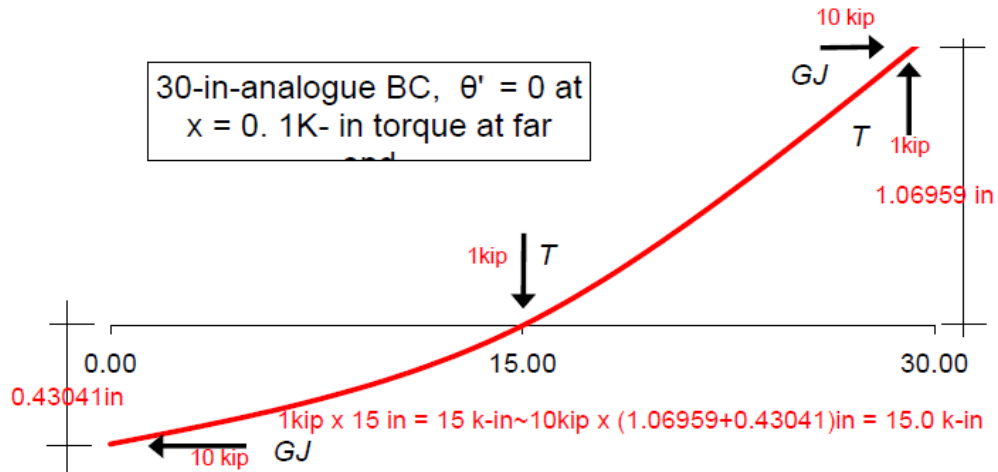


Figure 43. Analogue EBC in 2<sup>nd</sup> Order Equilibrium

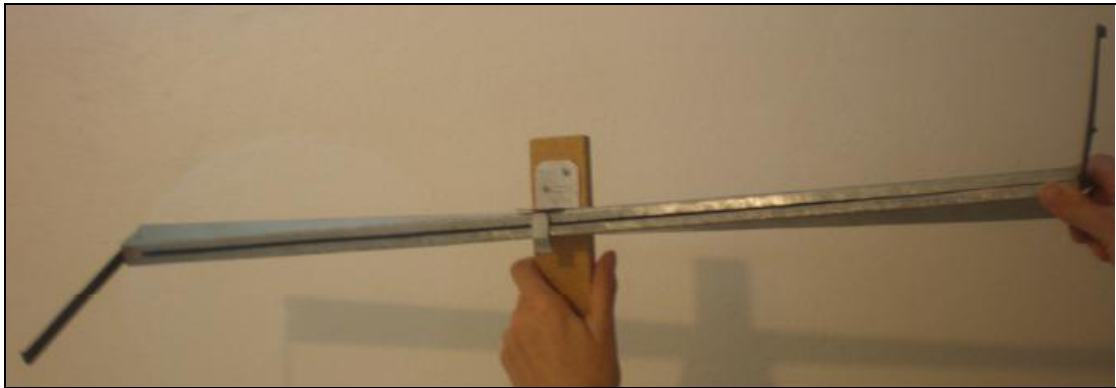


Figure 44. Zero External Torque with Restrained Warping at First Span

6EL									
6 Elements' beam. Torque angle restrained ad midspan. 1K-in torque at far end									
6	13	1	9	0	1	0	29000.		
1	1	2	3	0.03	1.	5.			
2	3	4	5	0.03	1.	5.			
3	5	6	7	0.03	1.	5.			
4	7	8	9	0.03	1.	5.			
5	9	10	11	0.03	1.	5.			
6	11	12	13	0.03	1.	5.			
1	7	0	1	0					
113		1.							

- GJ

-10.  
-10.  
-10.  
-10.  
-10.  
-10.

KX=KZ=0, KY=1 in Node 7

Total Torque

Figure 45. Input Data

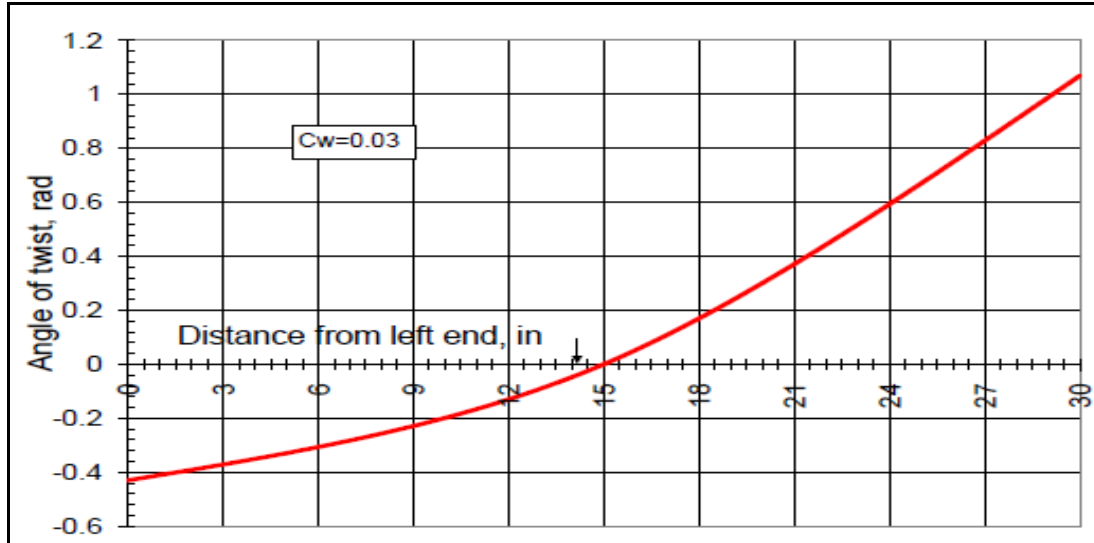


Figure 46. EBC with 6 Elements,  $\theta=0$  at Midspan, 1K-in Torque at Far End

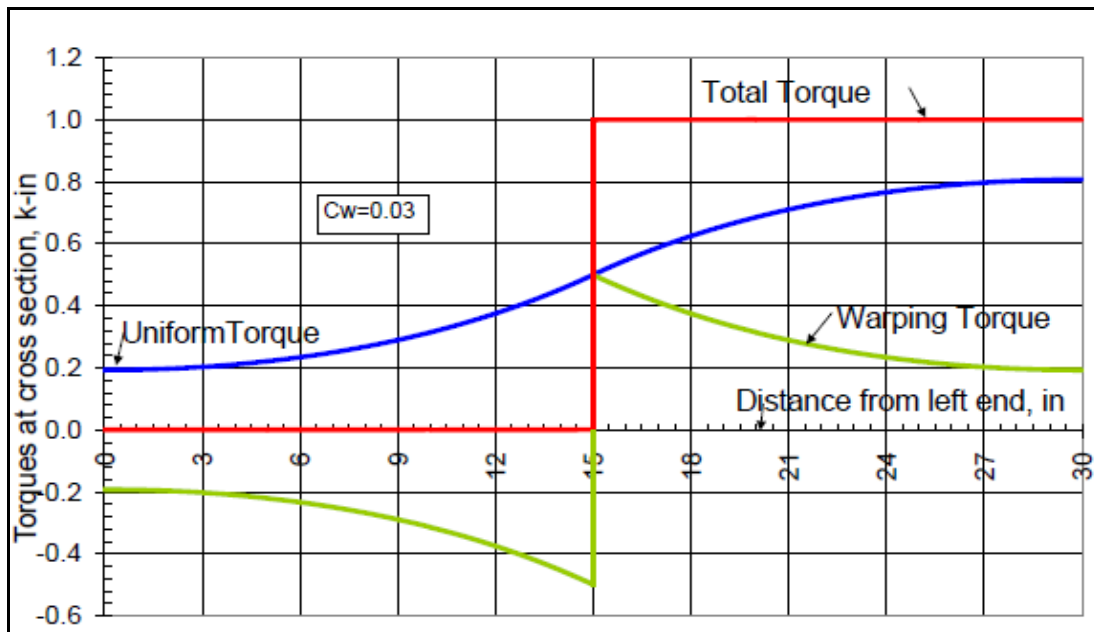


Figure 47. TWB with 6 Elements,  $\theta=0$  at Midspan, 1K-in Torque at Far End

### 3.2.5 Case Study Three

The case study three is a single span cantilevered TWB,  $L = 30$  in, under torsion held rigidly ( $\theta'(0) = 0$ ,  $\theta''(L) = 0$ ) in left end taken from pages 206-209 of Ugural (1987.) The TWB has been assigned the same cross section properties as in Case Study # 1 to compare BMTORSWP outputs with the analytic solution. The minimum element size is  $2\sqrt{EC_w/GJ} = 2\sqrt{290/10} \sim 11$  from Equation (41), and the number of elements should be

and integer larger than  $30/11 \sim 3$ , say 6 to edit the data from the former input form.

Figure 48 illustrates the analogous EBC, while Figure 49 and Figure 50 show the similarity of the finite element solution and the analytic solution of the torque diagram.

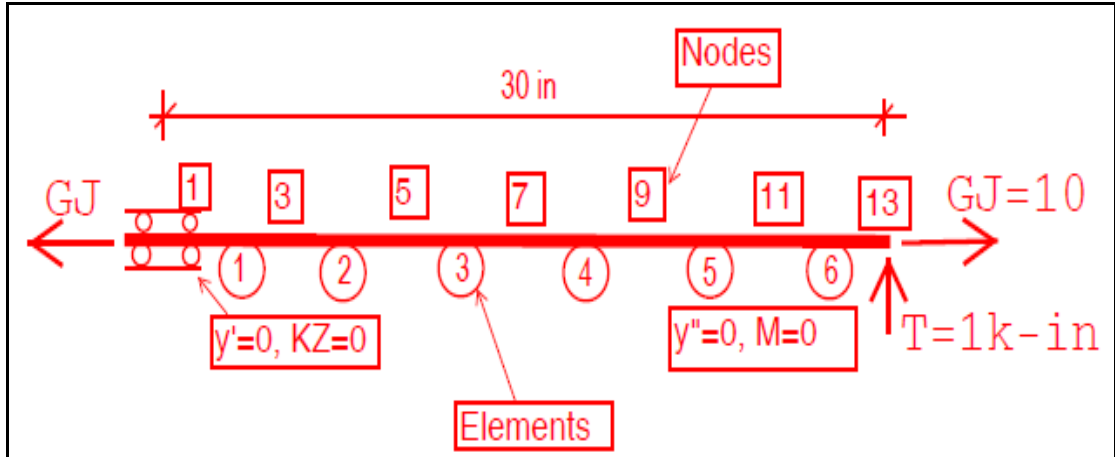


Figure 48. Model of Cantilevered Long Beam

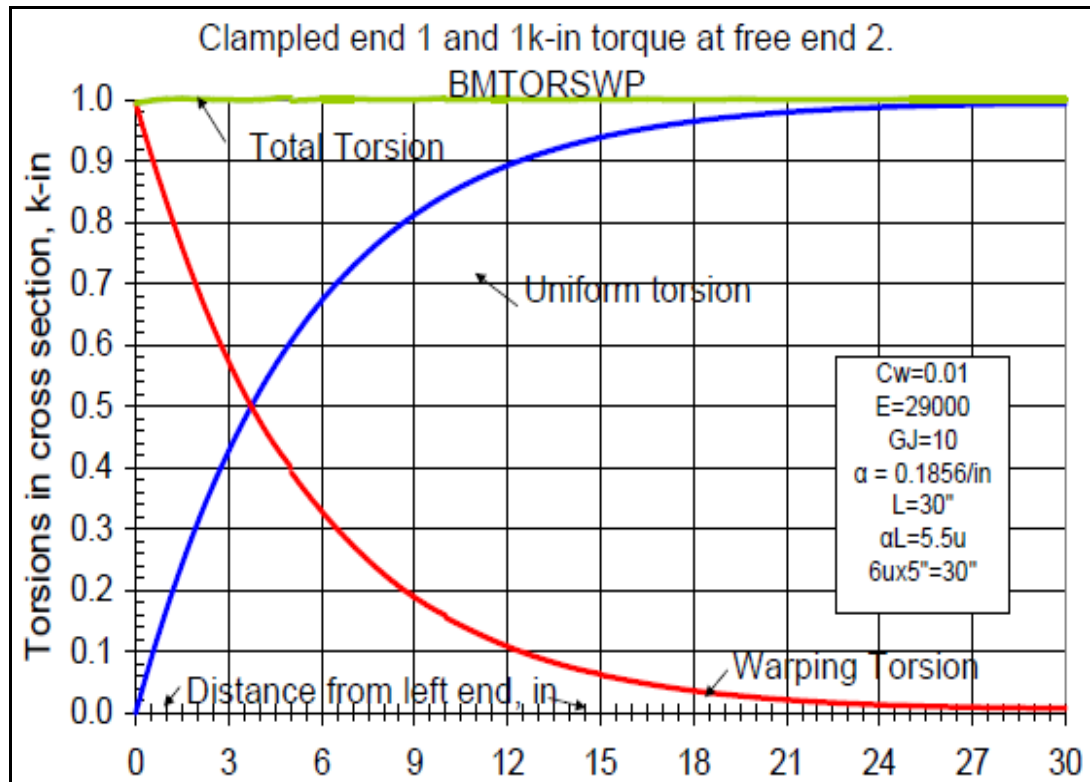


Figure 49.  $L=30''$ , TWB Clamped at Near End, Torque at Far End, BMTORSWP



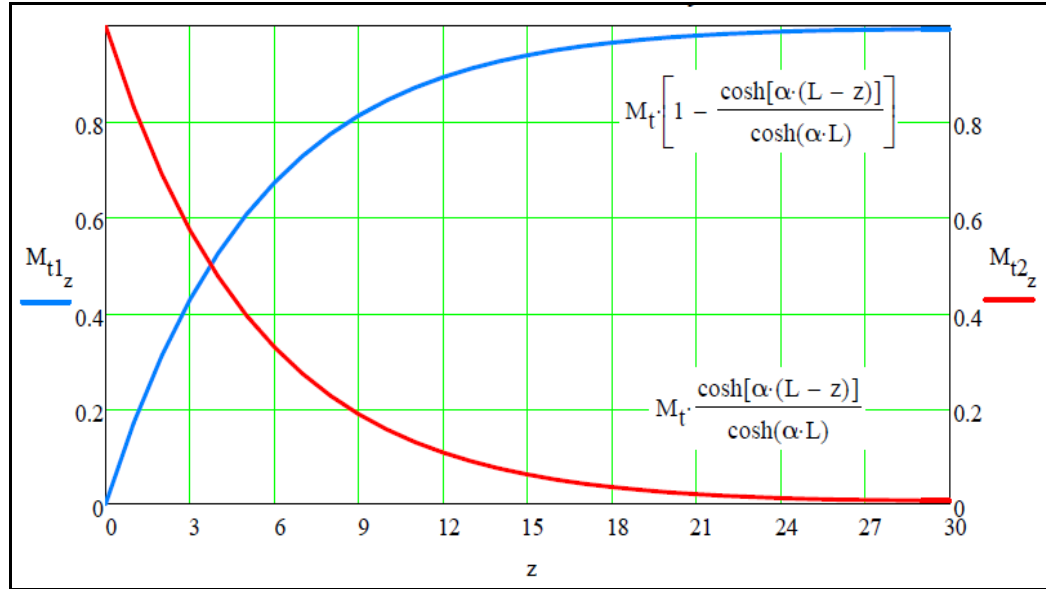


Figure 50.  $L=30''$ , Cantilevered Beam under Torsion, Exact Solution

More information can be found in APPENDIX C, whose charts show the asymptotic behavior of the twist angle derivatives increases as the order of derivative increases as seen in Figure 51. This is a very suitable feature of the particular finite element upon which the software is based.

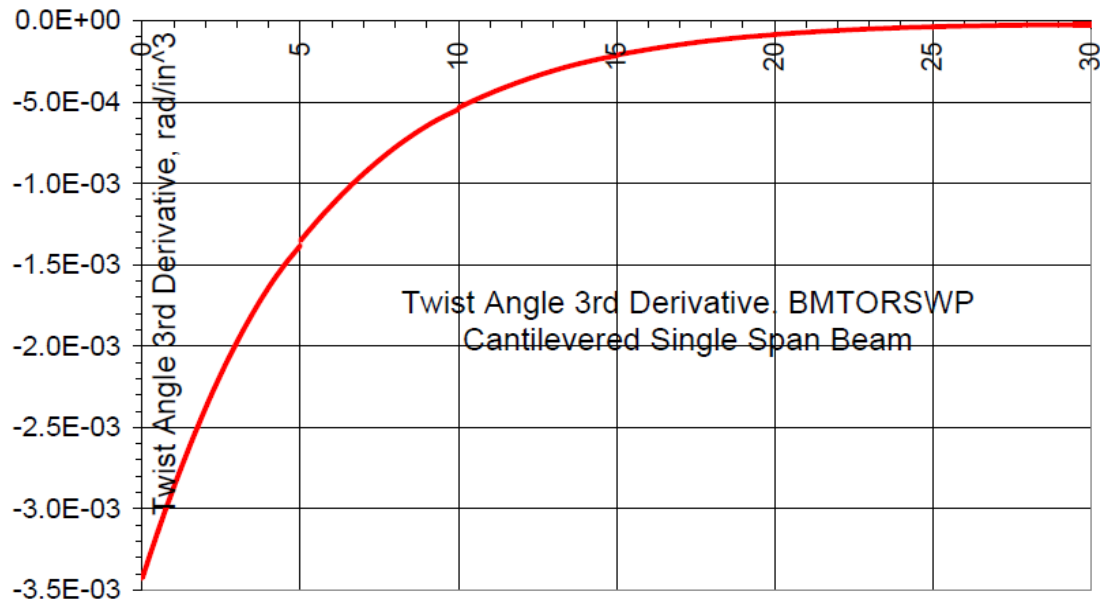


Figure 51. Twist Angle Third Derivative

### 3.2.6 Case Study Four

Examples 5.1 from the AISC DG-9 illustrated in Figure 52.a is solved, in which a W10x49 spans 15 ft (180 in) and supports a 15-kip factored load (10-kip service load) at midspan that acts at a 6 in eccentricity with respect to the shear center.

The stresses on the cross-section and the torsional rotation should be computed and compared with those provided by the AISC-DG9.

The ends are flexurally and torsionally pinned. The eccentric load can be resolved into a torsional moment and a load applied through the shear center as shown in Figure 52(b).

The resulting flexural and torsional loadings are illustrated in the same Figure 52. More detailed information can be found in APPENDIX D and a summary of results can be found from Figure 53 to Figure 58.

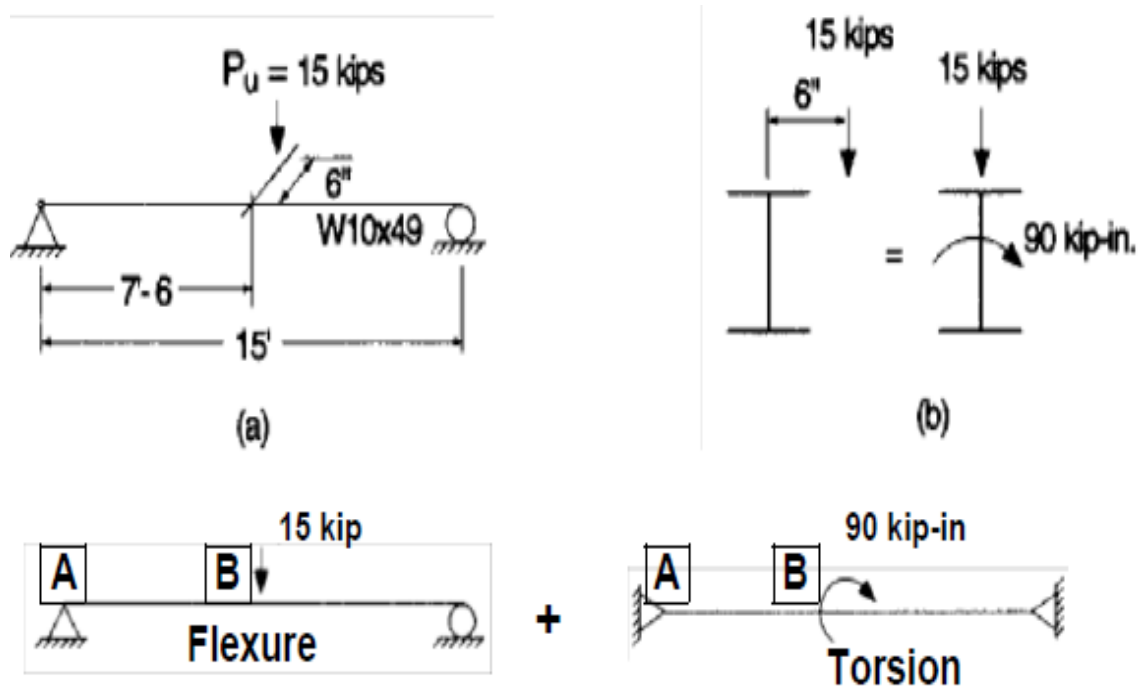


Figure 52. AISC-Design Guide 9, Example 5.1

Shear stress in web (w) at the TWB support:

$$\tau_{t\_wsBmtrs} := G \cdot t_w \cdot \left( \theta_{sBmtrs_1} \cdot \frac{1}{\text{in}} \right) = -6.11 \cdot \text{ksi} \quad \text{From BMTORSW}$$

$$\tau_{t\_wsAisc} := G \cdot t_w \cdot \left( \theta_{sAisc_1} \cdot \frac{1}{\text{in}} \right) = -6.16 \cdot \text{ksi} \quad \text{From AISC-DG9}$$

Shear stress in flange (f) at the support (s):

$$\tau_{t\_fsBmtrs} := G \cdot t_f \cdot \left( \theta_{sBmtrs_1} \cdot \frac{1}{\text{in}} \right) = -10.06 \cdot \text{ksi} \quad \text{From BMTORSW}$$

$$\tau_{t\_fsAisc} := G \cdot t_f \cdot \left( \theta_{sAisc_1} \cdot \frac{1}{\text{in}} \right) = -10.15 \cdot \text{ksi} \quad \text{From AISC-DG9}$$

Figure 53. Shear Stress in Web and Flange at Midspan

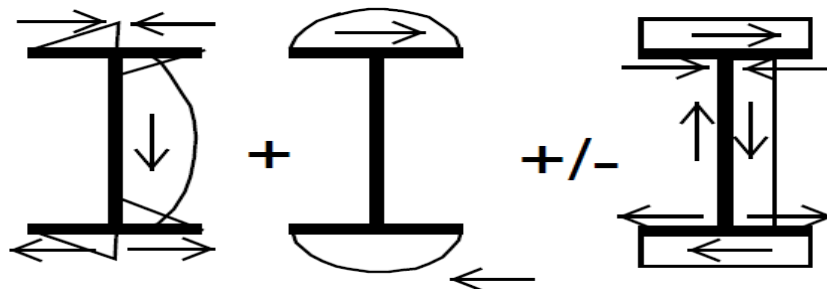


Figure 54. Two Solutions for Total Shear Stresses

Warping Shear Stress in flange at midspan (fm):

$$\tau_{w\_fmBmtrs} := -E \cdot S_{wl} \cdot \left( \theta_{mBmtrs_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -1.28 \cdot \text{ksi} \quad \text{BMTORSW}$$

$$\tau_{w\_fmAisc} := -E \cdot S_{wl} \cdot \left( \theta_{mAisc_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -1.28 \cdot \text{ksi} \quad \text{AISC-DG9}$$

Warping Shear Stress in flange at the support (fs):

$$\tau_{w\_fsBmtrs} := -E \cdot S_{wl} \cdot \left( \theta_{sBmtrs_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -0.57 \cdot \text{ksi} \quad \text{BMTORSW}$$

$$\tau_{w\_fsAisc} := -E \cdot S_{wl} \cdot \left( \theta_{sAisc_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -0.56 \cdot \text{ksi} \quad \text{AISC-DG9}$$

Figure 55. Shear Stress from Warping

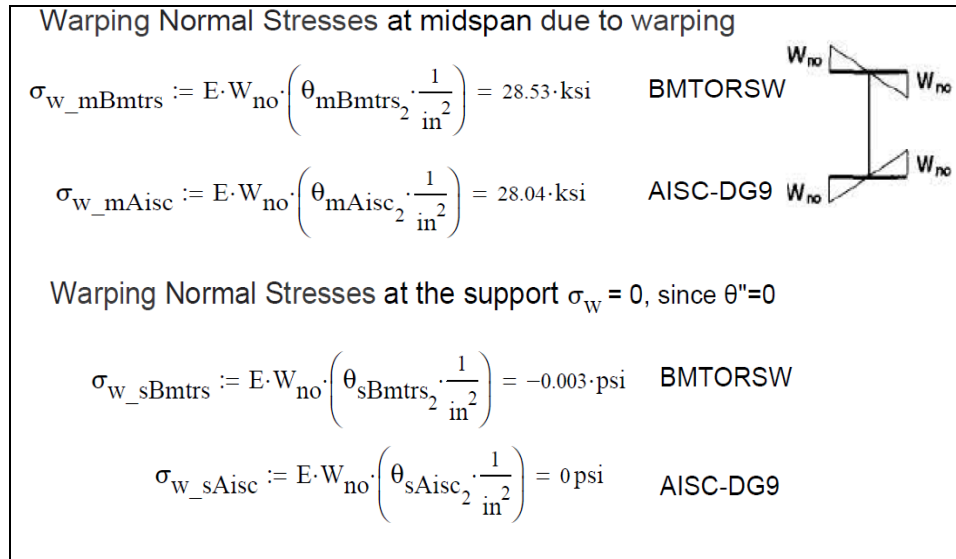


Figure 56. Normal Stresses due to Warping

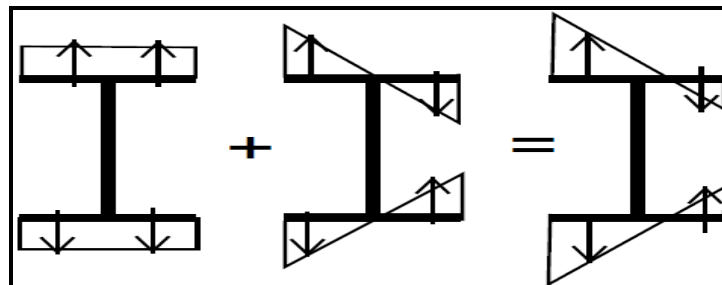


Figure 57. Superposition of Normal Stresses

Location	Normal Stresses			Shear Stresses			
	$\sigma_{iw}$	$\sigma_{ub}$	$f_{in}$	$\tau_{ut}$	$\tau_{iw}$	$\tau_{ub}$	$f_{iv}$
Midspan	<b>28.53</b>	<b>12.36</b>	<b>40.89</b>				
Flange	$\pm 28.1$	$\pm 12.4$	$\pm 40.4$	0	-1.28	$\pm 0.640$	-1.92
Web	----	----	----	0	----	$\pm 2.45$	-2.45
Support				<b>10.06</b>	<b>0.57</b>	<b>0.64</b>	<b>11.27</b>
Flange	0	0	0	-10.2	-0.564	$\pm 0.640$	-11.4
Web	----	----	----	-6.16	----	$\pm 2.45$	-8.61
Maximum		<b>40.89</b>	$\pm 40.4$			<b>11.27</b>	-11.4

BMTORSWSP figures without sign, next to AISC DG9 figures

Figure 58. Comparisons Total of Flexure and Torsion Stresses

### 3.2.7 Case Study Five

Examples 5.4 from the AISC DG-9 illustrated in The welded plate-girder shown in Figure 5.4a spans 25 ft (300 in.) and supports 310-kip and 420-kip factored loads (210-kip and 285-kip service loads). As illustrated in Figure 59, these concentrated loads are acting at a 3" eccentricity with respect to the shear center. Determine the stresses on the cross-section and the torsional rotation if the end conditions are assumed to be flexurally and torsionally pinned. The results and comparisons are found in Figure 60, Figure 61, and Figure 62. More detailed information can be found in APPENDIX E.

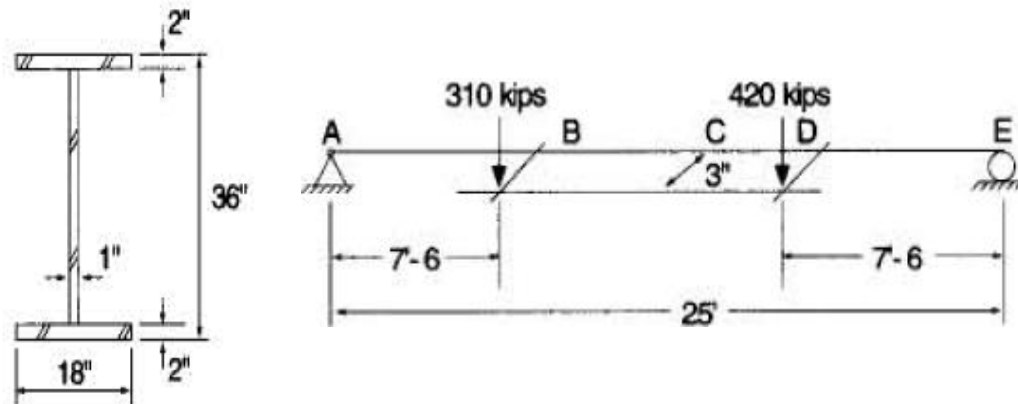


Figure 59. TWB Cross Section and Interest Points along Beam

Location		$\sigma_w$	$\sigma_b$	$f_{un}$
Point D	flange	-18.63		-45.63
	web	-18.4	$\pm 26.6$	-45.0
		—	—	—
Point E	flange	0	0	0
	web	—	—	—
Maximum				-45.63 -45.0

Figure 60. Discrepancy in Total Maximum Normal Stresses in Interest Points

Location		$\tau_f$	$\tau_w$	$\tau_b$	$f_{UV}$
Point D	flange	-4.73	-1.11		-8.21
	web	-4.47	-1.11	$\pm 2.37$	-7.95
		-2.24	—	$\pm 12.1$	-14.3
Point E	flange	-8.75	-0.84		-11.9
	web	-8.75	-0.87	$\pm 2.37$	-12.0
		-4.37	—	$\pm 12.1$	-16.5
<b>Maximum</b>					<b>-16.51</b>
					<b>-16.5</b>

Figure 61. No Discrepancy in Total Maximum Shear Stresses in Interest Points

From Appendix B, Case 3 with  $\bar{a} = 0.3$ , it is estimated that the maximum rotation will occur at approximately  $14\frac{1}{2}$  feet = 0.023 rad. ← 0.024 @ 12'-10"

Exact location for exact "a" 154" = 12'-10" →

Figure 62. Discrepancies on Maximum Twist Angle and Location

### 3.2.8 Case Study Six

The MC18x42.7 channel illustrated in Figure 5.5(a) spans 12ft (144 in.) and supports a uniformly distributed factored load of 16 kips/ft (2.4 kips/ft service load) acting through the centroid of the channel. Determine the stresses on the cross section and the torsional rotation.

The beam ends are flexurally and torsionally fixed. The eccentric load can be resolved into a torsional moment and a load applied through the shear center according to the AISC-DG9.

The resulting flexural and torsional loadings are taken from AISC-DG9 and illustrated in Figure 64; while Figure 63 shows the input model to be used in the software.

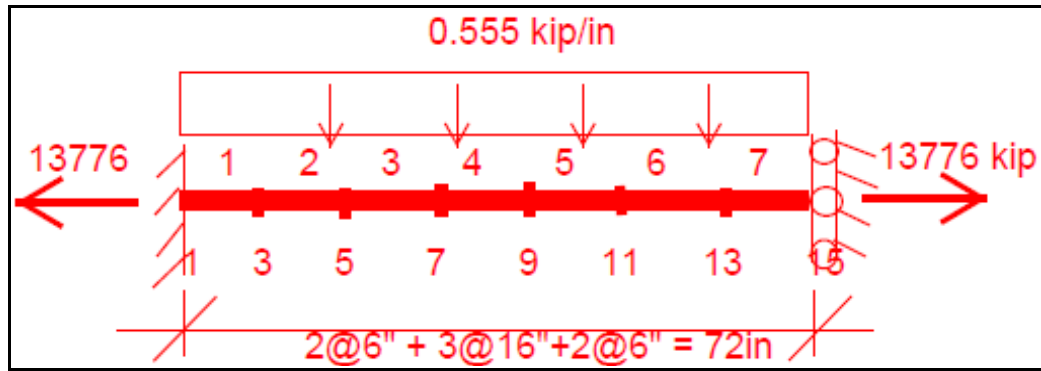


Figure 63. EBC Model for BMTORSWP

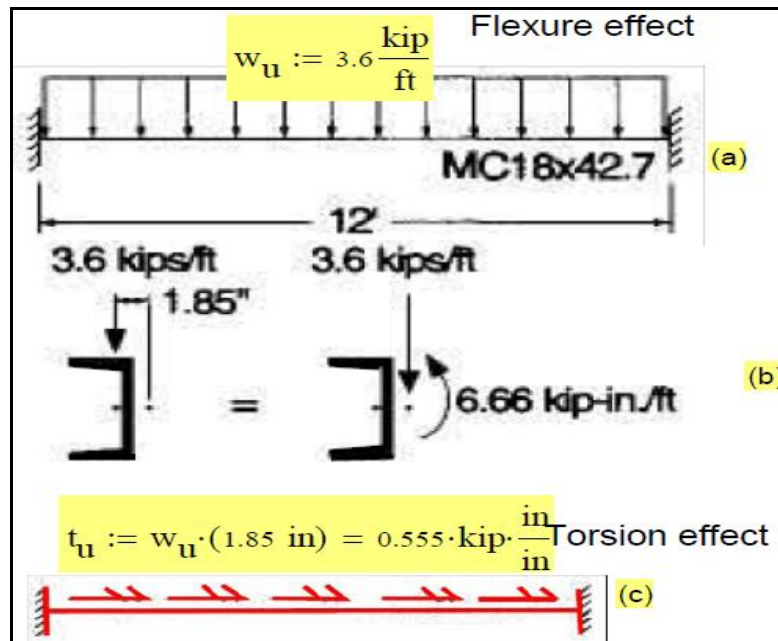


Figure 64. Flexure and Torsion Effect

$$\tau_{tBmt} = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 2.923 & 2.923 & 2.923 & 2.105 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at support  
0.2L  
0.5L  
L

$$\tau_t = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 2.843 & 2.843 & 2.843 & 2.047 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at support  
0.2L  
0.5L  
L

Figure 65. Shear Stress due to Pure Torsion along the Beam and Profile

$$\tau_{wBmt} = \begin{pmatrix} 0 & 1.306 & -1.013 & 0.703 \\ 0 & 0.595 & -0.462 & 0.321 \\ 0 & 0 & -0 & 0 \\ 0 & 1.306 & -1.013 & 0.703 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at support  
0.2L  
0.5L  
L

$$\tau_w = \begin{pmatrix} 0 & 1.303 & -1.011 & 0.702 \\ 0 & 0.599 & -0.465 & 0.323 \\ 0 & 0 & 0 & 0 \\ 0 & 1.303 & -1.011 & 0.702 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at support  
0.2L  
0.5L  
L

Figure 66. Shear Stress due to Warping Torsion along the Beam and Profile

$$\sigma_{wBmt} = \begin{pmatrix} 21.009 & 0 & -9.932 & 0 \\ 0.18 & 0 & -0.085 & 0 \\ -9.187 & 0 & 4.343 & 0 \\ 21.009 & 0 & -9.932 & 0 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at support  
0.2L  
0.5L  
L

$$\sigma_w = \begin{pmatrix} 20.078 & 0 & -9.491 & 0 \\ 0 & 0 & 0 & 0 \\ -8.729 & 0 & 4.127 & 0 \\ 20.078 & 0 & -9.491 & 0 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at support  
0.2L  
0.5L  
L

Figure 67. Warping Normal Stresses along the Beam and Profile

The real point where maximum angle of twist occur is not at 0.2L but at 0.204L. It can be seen in the curve made from BMTORSWP data. However, the results from the AISC DG9 are very close. The maximum service load rotation  $\theta$ , at midspan with the service-load torque is 0.0123 for the AISC and 0.0130 for BMTORSWP.

$$f_{unBmt} = \begin{pmatrix} 12.594 & -8.416 & -18.347 & 0 \\ -0.156 & -0.337 & -0.422 & 0 \\ -4.979 & 4.208 & 8.551 & 0 \\ 12.594 & -8.416 & -18.347 & 0 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at support  
0.2L  
0.5L  
L

$$f_{un} = \begin{pmatrix} 11.662 & -8.416 & -17.907 & 0 \\ -0.337 & -0.337 & -0.337 & 0 \\ -4.522 & 4.208 & 8.334 & 0 \\ 11.662 & -8.416 & -17.907 & 0 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at support  
0.2L  
0.5L  
L

Figure 68. Combined Normal Stresses along the Beam and Profile



Location	Point	$\sigma_w$	$\sigma_b$	$f_{un}$
Support	0	20.1(C)	8.41(T)	11.7(C) 12.594
	1	0	8.41(T)	8.41(T) -8.416
	2	9.49(T)	8.41(T)	17.9(T) -18.347
	3	0	0	0 0.
Midspan	0	8.73(T)	4.20(C)	4.53(T)
	1	0	4.20(C)	4.20(C)
	2	4.13(C)	4.20(C)	8.33(C)
	3	0	0	0
$z/l = 0.20$	0	0	—	—
	1	0	—	—
	2	0	—	—
	3	0	—	—
Maximum				17.9(T) -18.347

Figure 69. Total Normal Stresses along the Beam and Profile

The maximum normal stress (tension) occurs at the support at point 2 in the flange. A discrepancy of 2.5% is noticed: 17.907 ksi vs. 18.347 ksi from DG9 and the software respectively (Figure 68). The AISC DG9 data have been recalculated to the third decimal place to be compared with BMTORSWP output calculated to the third decimal place.

$$f_{uv} = \begin{pmatrix} 0 & 1.303 & 0.218 & 3.986 \\ 2.843 & 3.442 & 3.115 & 4.34 \\ 0 & 0 & 0 & 0 \\ 0 & 1.303 & -2.24 & -2.582 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at  
support  
0.2L  
0.5L  
L

$$f_{uvBmt} = \begin{pmatrix} 0 & 1.306 & 0.216 & 3.987 \\ 2.923 & 3.518 & 3.199 & 4.396 \\ 0 & 0 & -0 & 0 \\ 0 & 1.306 & -2.242 & -2.58 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at  
support  
0.2L  
0.5L  
L

Figure 70. Combined Normal Stresses along the Beam and Profile

Location	Point	$\tau_t$	$\tau_w$	$\tau_b$	$f_{UV}$
Support	0	0	0	0	0
	1	0	1.30←	—	1.30←
	2	0	1.01←	1.23→	0.22→
	3	0	0.702↓	3.28↓	3.98↓
Midspan	0	0	0	0	0
	1	0	0	0	0
	2	0	0	0	0
	3	0	0	0	0
$z/l = 0.20$	0	2.84↔	0	0	2.84↔ 2.923
	1	2.84↔	0.599←	—	3.44← 3.158
	2	2.84↔	0.465←	0.740→	3.12→ 3.199
	3	2.05↓	0.323↓	1.98↓	4.35↓ 4.396
<b>Maximum</b>					<b>4.35↓ 4.396</b>

Figure 71. Summary of Total Shear along the Beam and Profile

Figure 65, Figure 65, Figure 66, Figure 67, Figure 68, Figure 69, Figure 70, and Figure 71 contain the results of isolated and combined stresses and maximum angle of twist were presented. More comments, data, graphs and charts are provided in APPENDIX F.

### 3.2.9 Case Study Seven

Boothby (1984) proposed the use of the moment distribution method to analyze multi-span bars under restrained warping. A three-span continuous C12x30 beam with  $J = 0.864 \text{ in}^4$ ,  $C_w = 151 \text{ in}^6$ , maximum  $W_n = 11.7 \text{ in}^2$ ,  $G/E = 0.4$ ,  $E=29000\text{ksi}$  must be solved. The beam with its support conditions are shown in Figure 72. The beam was divided in half due to the physical and loading symmetries. See APPENDIX G.

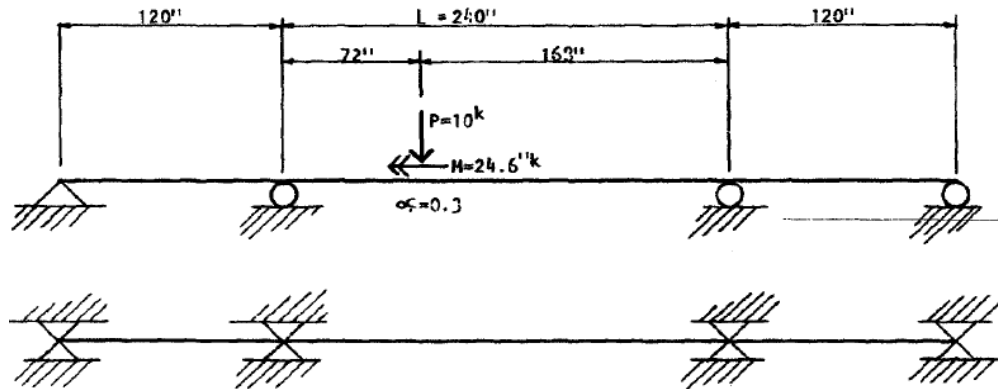


Figure 72. Boothby's Bending and Torsional Support Conditions

The properties of  $1/a = 0.04787/\text{in}$  or  $a = 20.9$  in; thus, each finite element could span  $2a$  to  $4a$ . Nevertheless, the elements will be chosen smaller as shown in Figure 73. The number each of the 23 elements is shown below the beam, and the node numbering is shown on top of the beam.

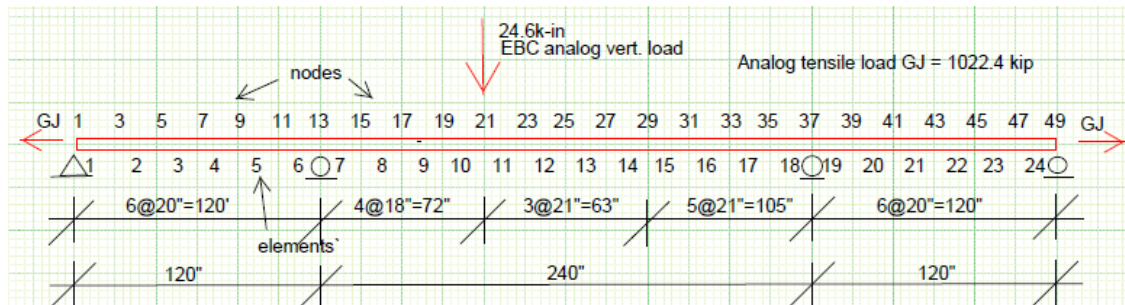


Figure 73. Model of EBC with Finite Elements

This is due to the fact that the program is based on a 9-DOF-beam-element. Whose three non nodal DOF are assigned to a fictitious middle span node. Therefore a structure with one element will carry 2 real nodes and one fictitious node; a structure with 2 elements will carry 3 real nodes and 2 fictitious nodes and so on and so forth. Therefore, a structure with 23 elements will carry 24 real nodes and 23 fictitious nodes.

The tensile analog force is  $-GJ = -0.4 * 0.864 \text{in}^4 * 29000 \text{ksi} = -1022 \text{ kip-in}^2$ .  $JBW$  is calculated by the formula  $3(NJ-NI+1)$ , being I and J the real nodes of any element with the maximum difference for  $J - I$ , such that  $I < J$ . That is, in our case  $JBW = 3(3-1+1) = 9$

(See APPENDIX F). The positive direction of forces and stresses along the cross section profile are those assigned to the beam under bending shear that appears in Figure 74.

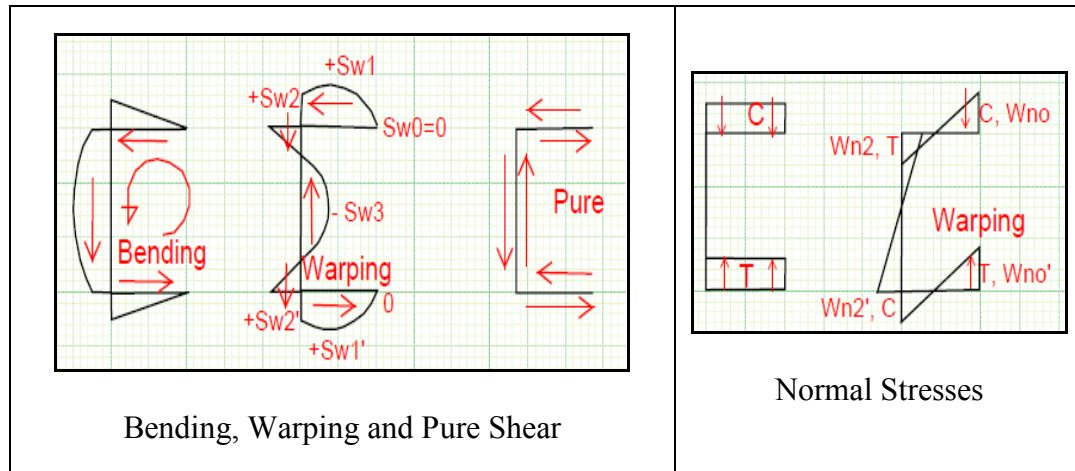


Figure 74. Stresses and Forces in Cross Section Profile

While there is a match between Boothby's and BMTORSWP charts (Figure 75 and Figure 76), it does not happen in the warping moment charts. Boothby's error on the warping moment is at the external sides of the first and last interior supports as it could be seen in Figure 77.

	Joint 2		C.O.=0.095	Joint 3	
	.523	.477		.477	.523
<i>FEB</i>		-363.1		+134.0	
Dist 1.	189.9	+173.2		-63.9	-70.1
C.O. 1		-6.1		+16.5	
Dist. 2	+2.9	+3.2		-7.9	-8.6
C.O. 2		-0.8		+0.3	
Dist. 3	+0.4	+0.4		-0.1	-0.2
	+193.2	-193.2		+78.9	-78.9
	<b>193.62</b>			<b>78.93</b>	

Figure 75. Bimoment Results. BMTORSWP Output Shown in Red

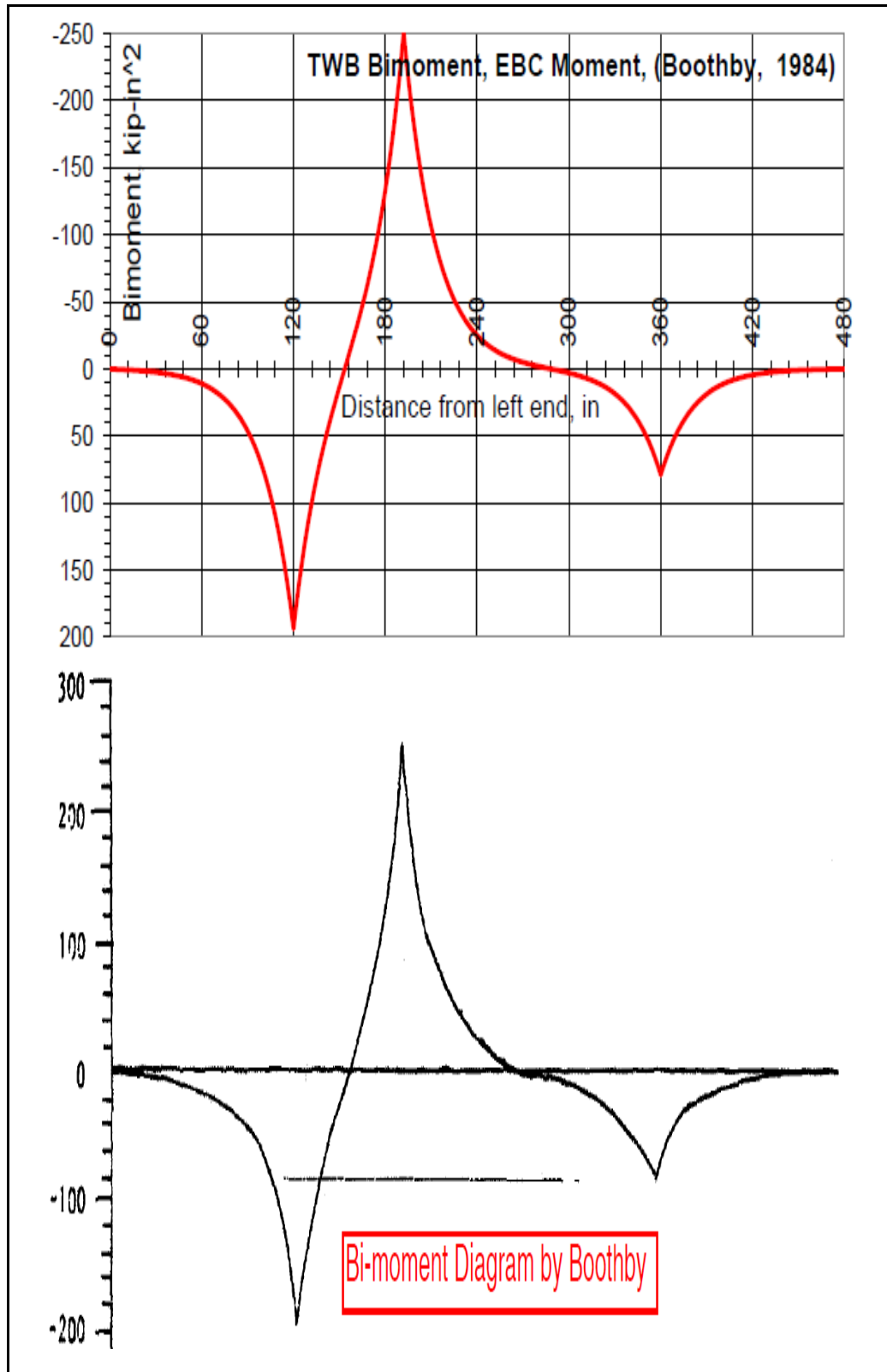


Figure 76. Bimoment Match between BMTORSWP and Boothby's

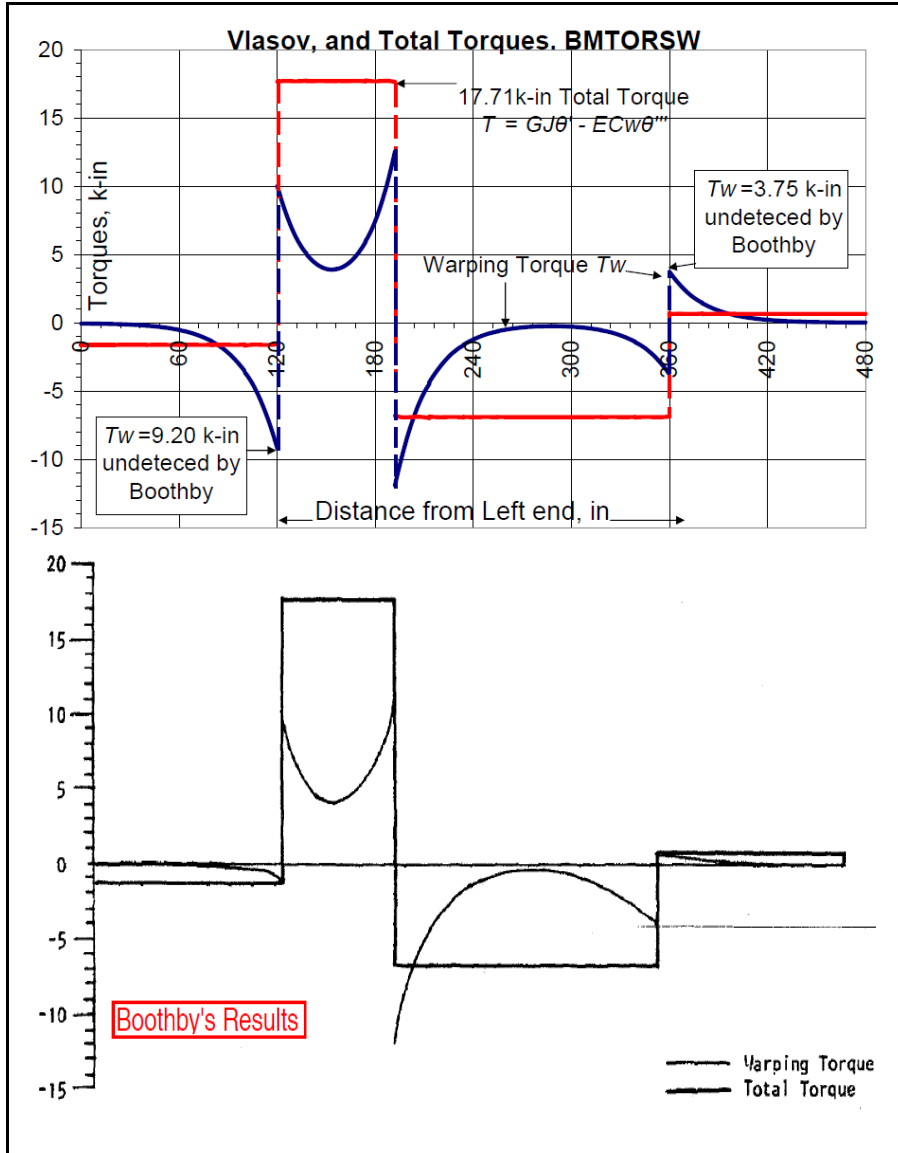


Figure 77. Total and Warping Torque from BMTORSWP and Boothby

Regarding stresses and strains Boothby made another mistake by selecting the wrong sectorial coordinate at the flange-web corner is illustrated in Figure 78 to evaluate the warping stresses. Nevertheless, it does not affect Boothby's Stresses at Right Side of 1st Interior Support (Figure 78). Regarding shear stresses, no discrepancies were found as shown in the same figure. More materials regarding this problem with a meticulous comprehensive study of stresses along the beam axial coordinate and along the cross sectional profile coordinate  $s$  may be found in APPENDIX G.

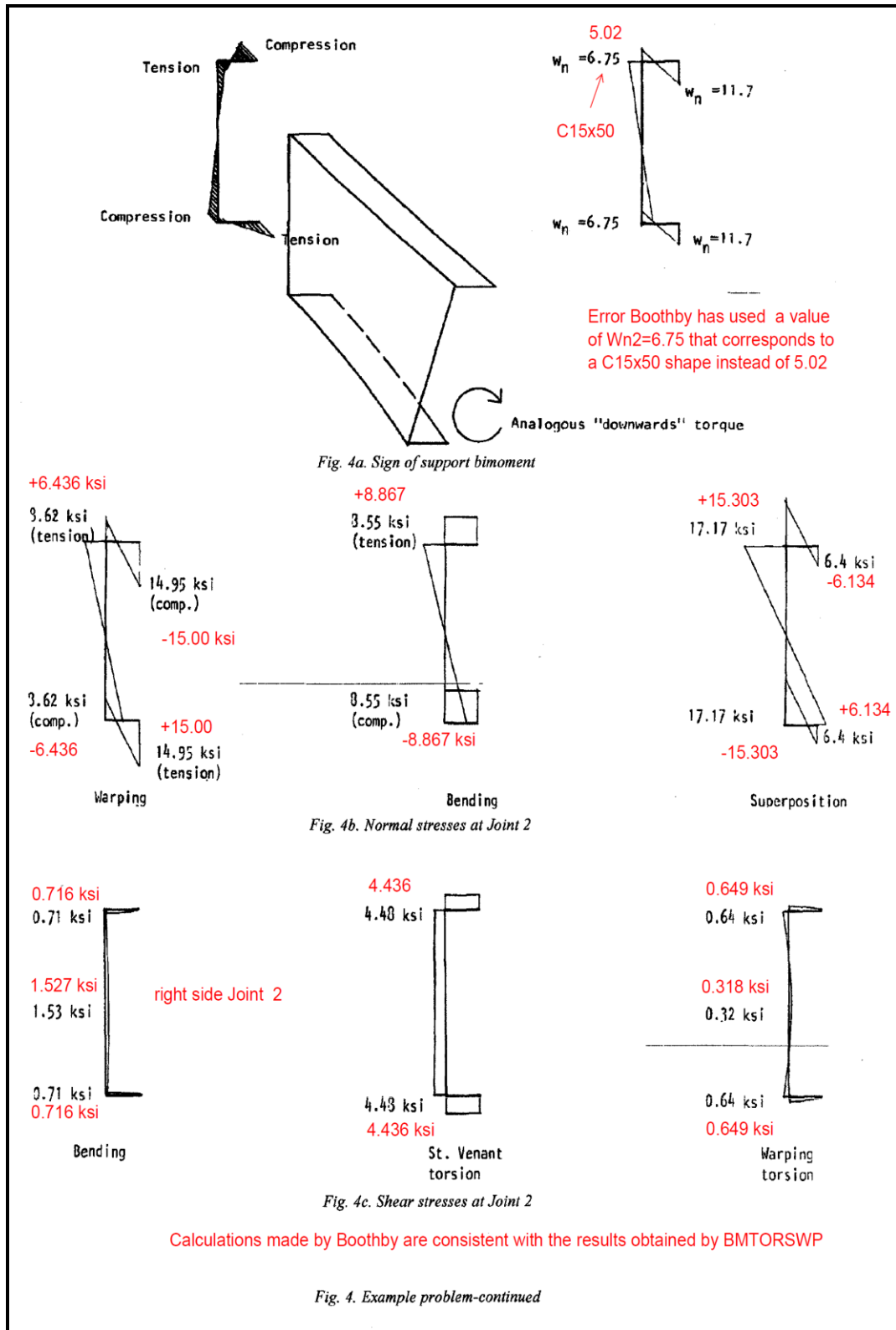


Figure 78. BMTORSWP and Boothby's Stresses at Right Side of 1<sup>st</sup> Interior Support

### 3.2.10 Case Study Eight

Consider the four-span continuous crane girder shown in Figure 79. The only torsional load in span AB and is due to the horizontal forces  $H = 6.7$  kip exerted by the wheels of the bridge at the top of the crane rail. The wheels' location causes the largest warping moments. The lateral load and the cross section of the crane girder are shown in Figure 81. The distance from the lateral force to the shear center profile is 8.365 in. The total torque at each wheel is  $T = 56.0$  kip-in. Figure 81 shows the values of the sectorial coordinate  $W_{ni}$  and the values of the first moment of the sectorial coordinate  $Sw$ . The St. Vt. torsion constant  $J$  is  $11.11 \text{ in}^4$ , and the sectorial moment of inertia  $C_w$  is  $81,989 \text{ in}^6$ .  $E = 29,000 \text{ kip/in}^2$ , and  $G = 11,154 \text{ kip/in}^2$  for steel. And finally, Figure 80 shows the input model to be used in BMTORSWP.

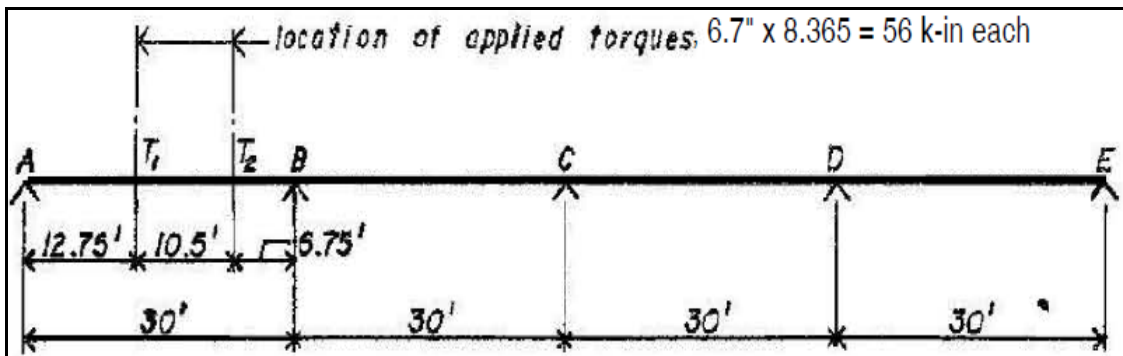


Figure 79. Continuous Crane Girder with Applied Torques

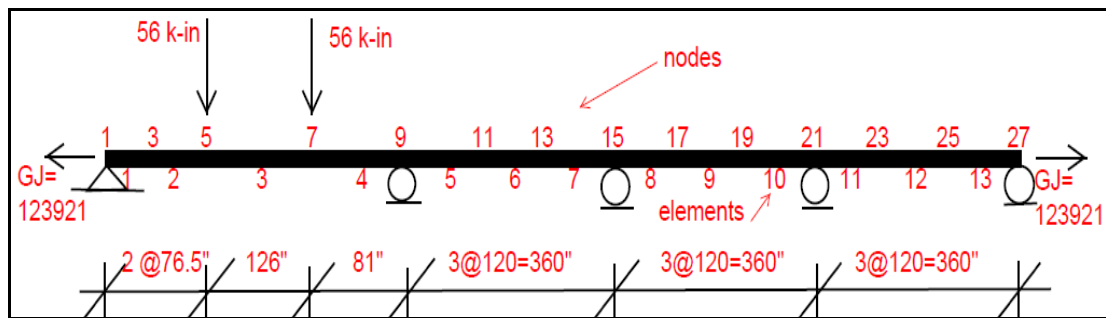


Figure 80. EBC Model for BMTORSWP



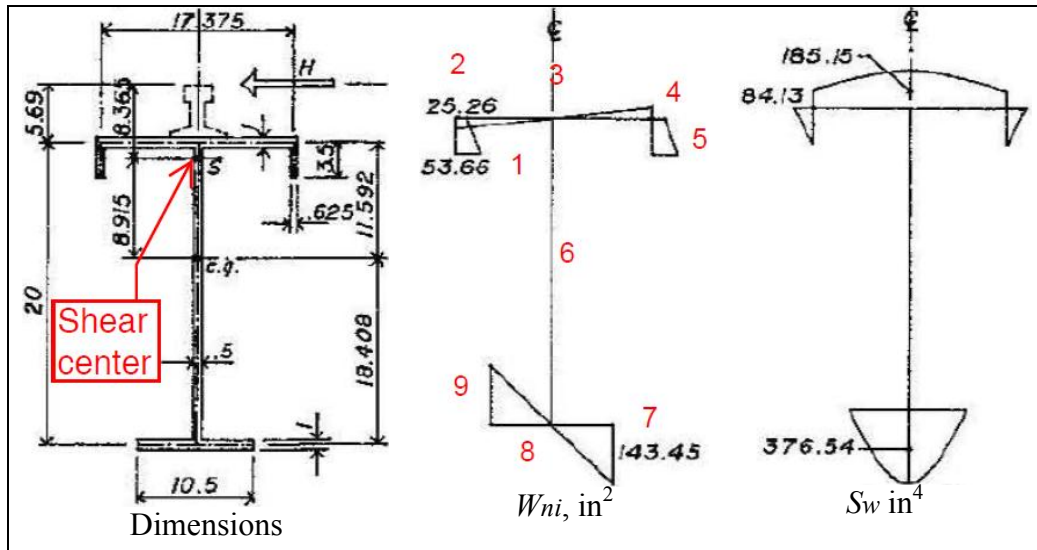


Figure 81. Section Properties and Profile Interest Points “s”

In APPENDIX H, material corresponding to the input model, input data, input forms, figures of output data with checks in notepad version, and excel processed output data and charts can be found. Charts are presented containing both partial and combined stresses along interest points of the beam and cross section profile.

Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP. Positive shear and axial stresses are assumed similar to those occurring at the cross section flanges and web when the beam undergoes bending.

There is a match between the form of BMTORSWP and Medwadowski charts (Figure 82). Nevertheless, Medwadowski committed a huge error in the scale of torques. He found torques 100 times larger than expected, which in terms of stresses is not acceptable at all as in can be seen in Figure 83.

The reasoning to support this statement is as follows: Given the hyperbolic nature of the functions involve in the angle of twist  $\theta(z)$ , each of its derivatives decreases an order of magnitude equal to the characteristic length “a”. On the other hand, the ratio

between bimoment  $B (= - ECw\theta''')$ ,  $\text{ksi-in}^2$ ) and warping moment  $T_w (= - ECw\theta''')$ ,  $\text{ksi-in}$ ) should be in the range of "a" ( $\sqrt{ECw/ GJ} = 138.5 \text{ in}$ ). Thus, using the Medwadowski bimoment data, the warping moment  $T_w$  is expected to be around  $(4000+3000) \text{ ksi-in}^2 / 138.50 \text{ in} \sim 50 \text{ k-in}$ ; as shown in the BMTORSWP chart.

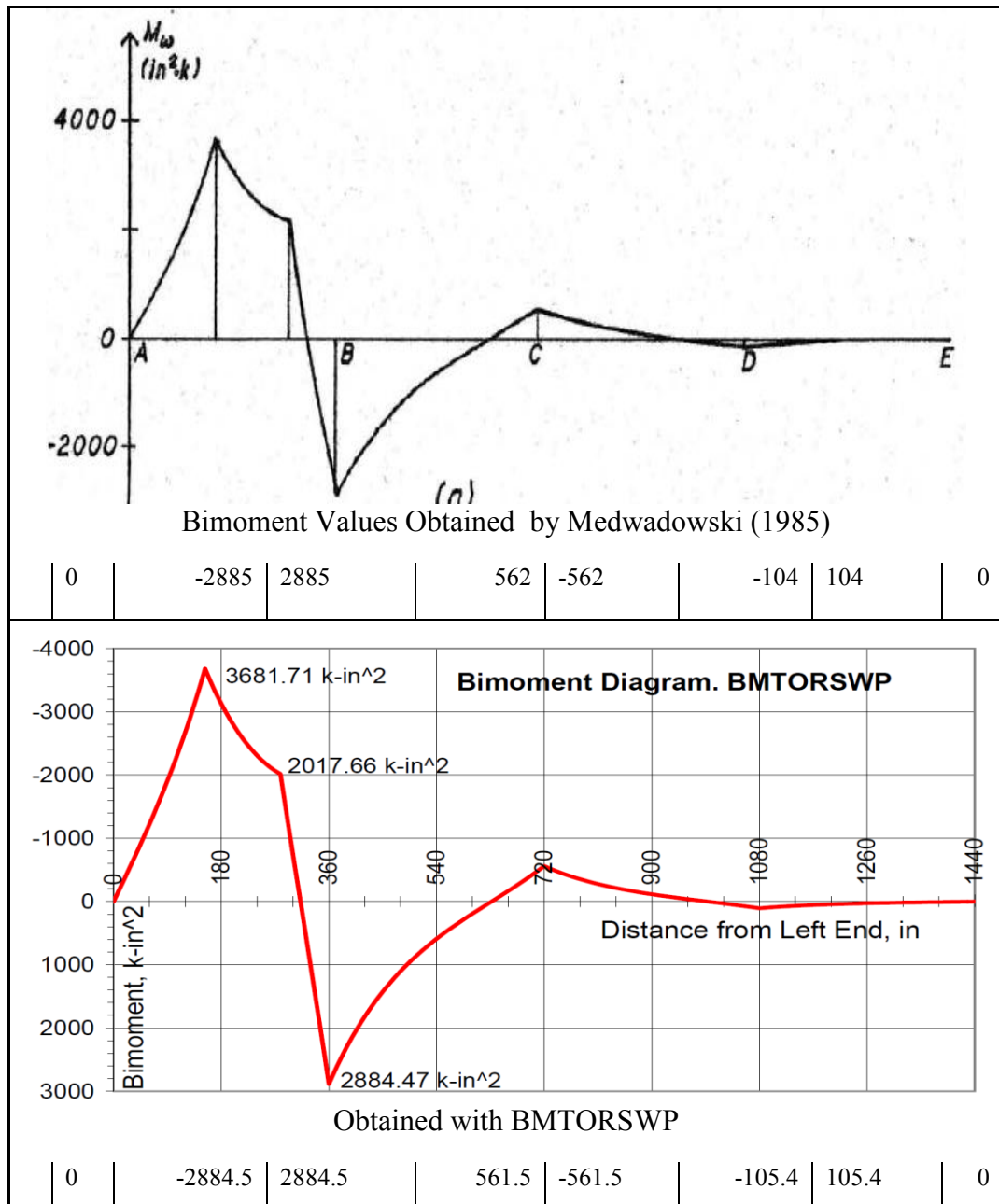
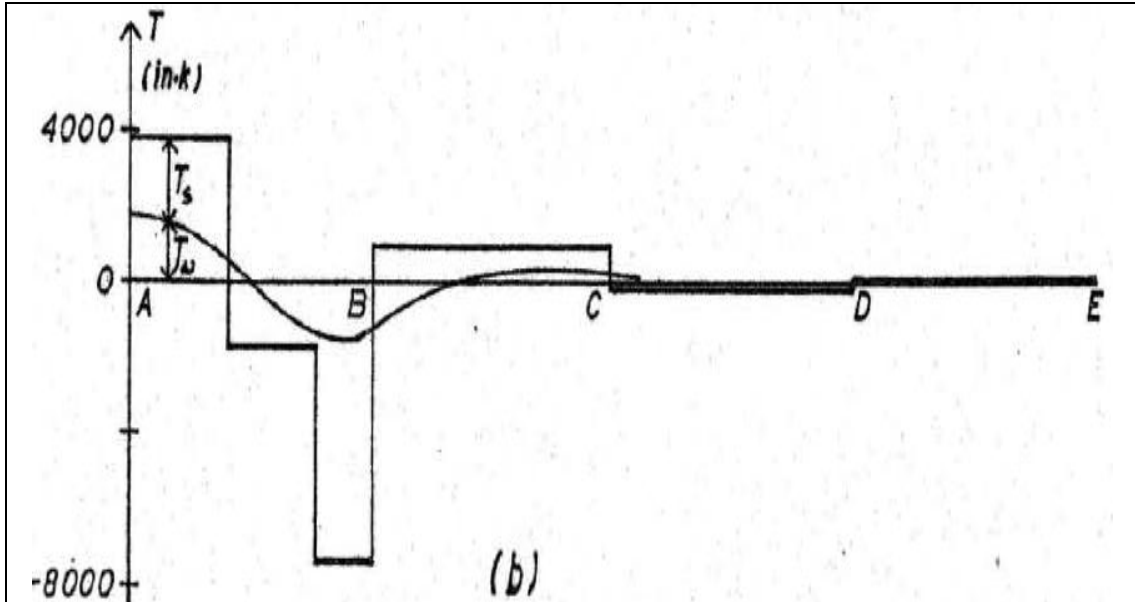


Figure 82. Comparison of Bimoment Charts by Medwadowski and with BMTORSWP



Medwadowski's with Scale Error. Given the hyperbolic nature of the functions involve in  $\theta(z)$ , each of its derivatives decreases an order of magnitude equal to the characteristic length "a". Thus  $B = -ECw\theta''$  should be around "a" ( $\sqrt{ECw/GJ} = 138.5$ ) times  $T_w = -ECw\theta''$ . From the bimoment chart,  $T_w$  could be reasonably expected to be around  $(4000+3000)/138.50 \sim 50$ ; that is, a two-figure-number as shown below

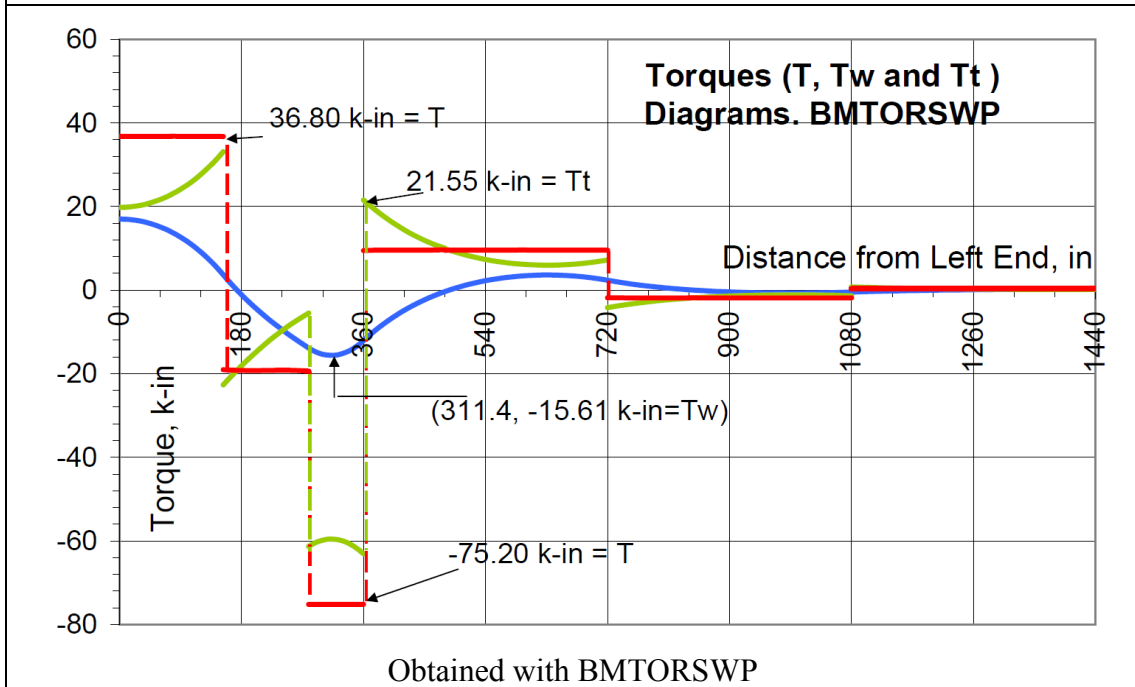


Figure 83. Comparison of Torque Charts by Medwadowski and with BMTORSWP

### 3.2.11 Results' Interpretation

According to Saadeé (2004-2005), polynomial interpolation functions are usually used for bending while the hyperbolic functions are often used for non uniform torsion. Nevertheless, the efficacy of BMTORSWP to represent hyperbolic functions has been proven despite the fact it is based upon a finite element developed by means of polynomial interpolation functions.

On the other hand, the solutions in the AISC-DG9 by Seaburg (1997) have been found correct. They were originated in the works by Heins (1963) and Seaburg, actualized by Seaburg (1997) and Carter. Nevertheless, in the case of C shape profiles, the AISC-DG9 omits the shear stress analysis in the profile interest point at  $s = 1$ . A computer solution allows the practitioner to analyze this shear using interpolation of the value for the sectorial coordinate.

Conversely, the errors found in the works by Boothby (1984) and Medwadowski (1985) provide evidence that the solution of restrained warping problem is prone to error. Therefore, it is suitable to bring in a tool as BMTORSWP able to simplify the problems involving first order general torsion theory. In addition to the complexity that structural analysis of TWB share with 2<sup>nd</sup> order analysis of beam-columns, the calculation of TWB stresses along the cross section profile is tiresome. And BMTORSWP provides the suitable option to export the data to a spreadsheet for a more systematic handling of stress computations.

As a comment on the trustiness of the finite element upon which the software is based, it is important to state that the accuracy of the 6DOF beam column is 3.8 to 4 times that of the 4 DOF beam column. Therefore, just one 6DOF-BE is required to

achieve the accuracy of four-4DOF-BE or five 6DOF-BE are required to achieve the accuracy of nineteen 4DOF-BE. On the other hand, one 6DOF-BE matrix occupies  $18 \times 1 + 3 = 21$  entries in a global matrix upper triangle, while four 4DOF-BE matrixes occupy  $7 \times 4 + 3 = 31$  entries in a global matrix upper triangle respectively. Similarly, five 6DOF-BE matrixes occupy  $18 \times 5 + 3 = 93$  entries in a global matrix upper triangle and nineteen 4DOF-BE matrixes occupy  $7 \times 19 + 3 = 135$  entries in a global matrix upper triangle respectively.

This difference will increase linearly as the number of required elements increases. In general the global matrix assembled with a higher degree finite element requires a lesser number of matrix entries in the program memory in order to achieve a given accuracy; thus, it is more convenient. This advantage of the 6DOF-BE is in addition to the ones related to its two non nodal DOFs, and adds to the fact that for a solution of the mixed torsion problem a higher degree polynomial is mandatory.

Regarding the aforementioned errors by Boothby and Medwadowski, it is pertinent to comment that perfect coincidence was found in the bimoment diagrams. However the chart of torques by Boothby contains 2 mistakes. At left side of first interior support and right side of last interior support, warping torques of 9.2 and 3.5 respectively were not detected by Boothby with corresponding errors of 700 and 300 per cent. See Chart Total and Warping Torque from BMTORSWP and Boothby.

Nevertheless, these errors would not be critical for design purposes. It was found that stresses calculations by Boothby correspond to the right side of the first interior support, and they are correct. Nevertheless, another mistake by Boothby was noticed

regarding the cross section sectorial coordinate at the flange-web corner:  $A W_n^2 = 6.75 \text{ in}^2$  corresponding to a C15x50 shape was used instead of  $5.02 \text{ in}^2$ .

Regarding the aforementioned errors by Medwadowski, a perfect coincidence was found in the bimoment diagrams. However the chart of torques by Medwadowski contains one error. He found torques 100 times larger than expected, which in terms of stresses is not acceptable at all.

The reasoning to sustain this statement is as follows: Given the hyperbolic nature of the functions involve in the angle of twist  $\theta(z)$ , each of its derivatives decreases an order of magnitude equal to the characteristic length "a". Therefore, the ratio between bimoment  $B (= - ECw\theta'')$ ,  $\text{ksi-in}^2$  and warping moment  $T_w (= - ECw\theta''')$ ,  $\text{ksi-in}$  should be in the range of "a" ( $\sqrt{ECw/ GJ} = 138.5 \text{ in}$ ). Thus, using the Medwadowski bimoment data, the warping moment  $T_w$  is expected to be around  $(4000+3000) \text{ ksi-in}^2 / 138.50 \text{ in} \sim 50 \text{ k-in}$ ; as shown in Figure 83.

## CHAPTER 4

### CONCLUSIONS AND RECOMMENDATIONS

It is appropriate for designers of thin-walled open cross sections frequently used in steel structures to count on straightforward and affordable tools enabling them to bypass the common difficulties related to the design of elements under restrained warping. BMTORSW or BMTORSWP fulfill very well all these requirements.

#### 4.1 Conclusions

Considering the asymptotic character of the governing hyperbolic functions, the use of smaller finite elements may be more appropriate near restraints. An additional consideration must be taken into account to check the equilibrium of forces when distributed loads are involved. Due to the fact that the location of the smeared DOF is unknown, a more refined mesh reduces the error when it comes to arbitrarily assigning a reasonable location for the smeared DOF action during the equilibrium checks.

The accuracy of the 6DOF beam column is 3.8 to 4 times that of the 4 DOF beam column as shown in Figure 38. Therefore, it could be said that “n” 6DOF-BE matrixes require  $18n+3$  entries in a global matrix upper triangle which, in terms of accuracy, are equivalent to  $3.8 \cdot n$  4DOF-BE matrixes that require about  $27n + 3$  ( $7(3.8n) + 3$ ) entries in a global matrix upper triangle respectively.

Two effects detected in the parametric experiments have been corroborated in the analysis: First, there is twist and restrained warping along the unloaded span; second, the twist at the loaded end is larger than that at the unloaded end.

## 4.2 Recommendations

Definitely BMTORSWP is the recommended software application to handle problems of multi-span beams subjected to restrained warping. It is straightforward, versatile and powerful enough. And it has survived the test of handling single span bar under restrained warping (Obréski, 2005).

The recommended size of the finite element to be used in BMTORSWP is between  $2a$  and  $4a$ ; where  $a = \sqrt{E \cdot C_w / G \cdot J}$  according to the AISC. Nevertheless, an additional advice is to use a minimum of 3 finite elements per span to reproduce the asymptotic behavior (which accentuates as  $C_w$  gets smaller Figure 50) of the angle of twist and its derivatives. Nodes are mandatory at restraints and points of application of concentrated loads. An additional recommendation is that smaller finite elements can be used neighboring the nodes.

Inexpensive experiments like that of the bar restrained against torsion at midspan could be used to experimentally measure cross section torsional properties, and to verify the range of linear behavior of  $GJ$ . They could also be used to study large deformation behavior, and to explain behavior of moving large slim structures like trucks and catamarans when struck or impacted near midspan.

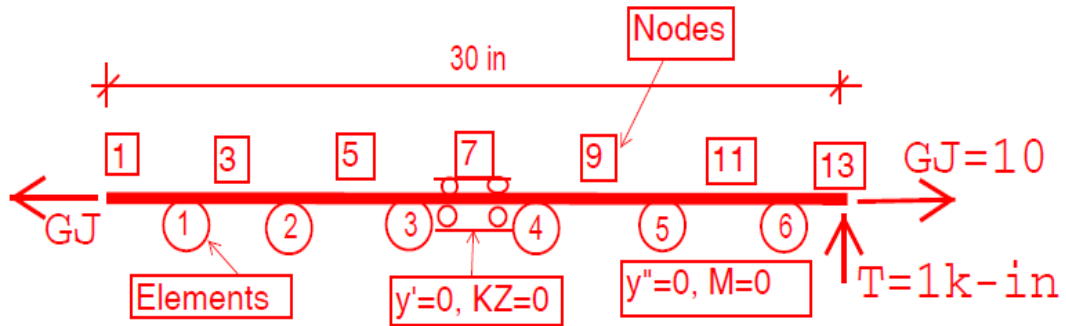
Finally, the software application is recommended to solve problems of distorted box girders. However, for a box girder solution, a study of convergence similar to the one presented in this work is recommended to be advanced. This more accurate convergence study must be made by combining the  $6 \times 6$  elastic and soil matrixes related to the field of transverse displacements.



## APPENDIX A

### BAR CANTILEVERED AT MIDSPAN

The equivalent beam-column model, input forms, input and output notepads as well as the charts can be found in the following pages of this part of the document.



Input model for Bar Cantilevered at Midspan

CMS

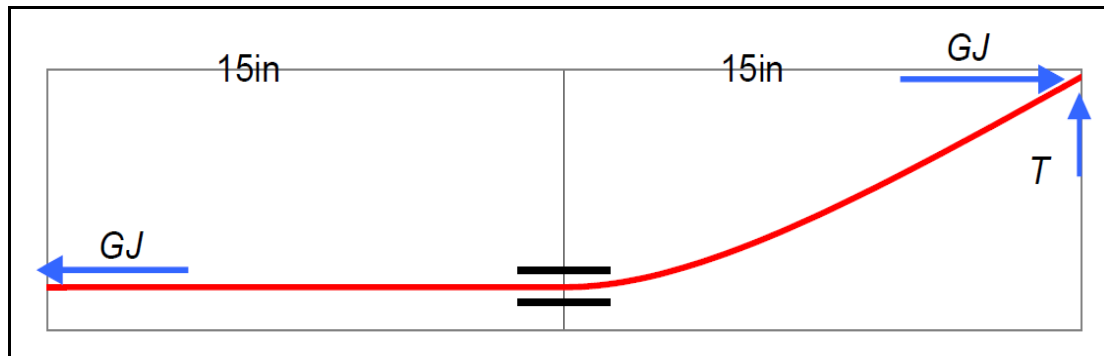
Held rigidly against rotation at midspan and subjected to a 1k-in

6	13	1	9	0	1	0	<span style="border: 1px solid red; padding: 2px;">29000.</span>	<span style="color: red;">E</span>	
1	1	2	3	<span style="border: 1px solid red; padding: 2px;">0.01</span>	1.	5.			-10.
2	3	4	5	<span style="border: 1px solid red; padding: 2px;">0.01</span>	1.	5.			-10.
3	5	6	7	<span style="border: 1px solid red; padding: 2px;">0.01</span>	1.	5.	→Cw	tensile load GJ	-10.
4	7	8	9	<span style="border: 1px solid red; padding: 2px;">0.01</span>	1.	5.			-10.
5	9	10	11	<span style="border: 1px solid red; padding: 2px;">0.01</span>	<span style="border: 1px solid red; padding: 2px;">1.</span>	<span style="border: 1px solid red; padding: 2px;">5.</span>			-10.
6	11	12	13	<span style="border: 1px solid red; padding: 2px;">0.01</span>	<span style="border: 1px solid red; padding: 2px;">1.</span>	<span style="border: 1px solid red; padding: 2px;">5.</span>			-10.
1	7	<span style="border: 1px solid red; padding: 2px;">0</span>	<span style="border: 1px solid red; padding: 2px;">1</span>	<span style="border: 1px solid red; padding: 2px;">1</span>					
113									

0 1 1 ← Cantilevered, KY=KZ=1

A figure to avoid matrix singularity

Input Notepad



Bar Cantilevered ( $\theta = 0, \theta'' = 0$ ) at Midspan with 1K-in Torque at far End

```

-----
                                cmsout
-----
YOU ARE USING COMPUTER PROGRAM BMTORSW,DEVELOPED BY DR. BERNARDO DESCHAPELLES
0      INPUT DATA FILE NAME IS = cms.txt
0      OUTPUT FILE NAME IS = cmsout.txt
0      STORAGE FILE FOR POST-PROCESSING WITH EXCEL = cmsgrf.grf
-----
Held rigidly against rotation at midspan and subjected to a 1k-in torque at far
0modulus of elasticity of the material= 29000. k/ft2
0ELEM nodes inertia length distrib. load AXIAL SOIL NORMAL MODULUS,Ksf angle
0      i   j   ft.4   ft   at i   at j   LOAD   1st END   2nd END   rad
1      1   3   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
2      3   5   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
3      5   7   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
4      7   9   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
5      9  11   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
6     11  13   0.01000  5.00  0.000  0.000  -10.00  0.0       0.0       0.000 00
0      INPUT DATA RELATED TO THE 1 SUPPORTS Tensile
1 7 0 1 1
0      INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2
113 1.00
0      FINAL SOLUTION FOUND AFTER 1 ITERATIONS
0      Output of nodal displacements in reference to global axes
0node displ. displ. displ. node displ. displ. displ.
      along x along y around z along x along y around z
      or nonn1 or nonn2 or nonn 3 or nonn1 or nonn2 or nonn 3
+ 1 0.0000E+00 0.0000E+00 0.0000E+00 2 0.0000E+00 0.0000E+00 0.0000E+00
+
+ 3 0.0000E+00 0.0000E+00 0.0000E+00 4 0.0000E+00 0.0000E+00 0.0000E+00
+
+ 5 0.0000E+00 0.0000E+00 0.0000E+00 6 0.0000E+00 0.0000E+00 0.0000E+00
+
+ 7 0.0000E+00 0.0000E+00 0.0000E+00 8 0.0000E+00 0.6168E-01 0.1475E-01
+
+ 9 0.0000E+00 0.1724E+00 0.5967E-01 10 0.0000E+00 0.3451E+00 0.3048E-01
+
+ 11 0.0000E+00 0.5363E+00 0.8202E-01 12 0.0000E+00 0.7486E+00 0.3586E-01
+
+ 13 0.0000E+00 0.9656E+00 0.8771E-01
-----
OUTPUT OF SOIL REACTIONS,STRESSES AND TRANSVERSE DISPLACEMENTS
-----
0      ELEMENT 1 DISPLACEMENTS IN INCIDENCES 1 2 3
      NODE 1 0.00000E+00 0.00000E+00 0.00000E+00
      NODE 2 0.00000E+00 0.00000E+00 0.00000E+00
      NODE 3 0.00000E+00 0.00000E+00 0.00000E+00
0      FORCES ACTING ALONG THE 9 DOF
      NODE 1 0.00000E+00 0.00000E+00 0.00000E+00
      NODE 2 0.00000E+00 0.00000E+00 0.00000E+00
      NODE 3 0.00000E+00 0.00000E+00 0.00000E+00
0ELEMNT 1, FROM NODE 1, TO NODE 3 - LENGTH = 5.00 ft
0 left half of span,at tenth points of length
      span span span span span span span
+ soil,k/ft 0.0 0.1 0.2 0.3 0.4 0.5
  shear,k 0.000 0.000 0.000 0.000 0.000 0.000
      0.00 0.00 0.00 0.00 0.00 0.00
      GJ=10; T = 1; L = 15; Page 1
      (- GJ*0.9656 + TL) = Bimomento = 5.344

```

Output Checks, Page 1

```

                                cmsout
bmom,kft      0.00      0.00      0.00      0.00      0.00      0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+ 0.5         0.6         0.7         0.8         0.9         1.0
soil,k/ft    0.000    0.000    0.000    0.000    0.000    0.000
shear,k      0.00     0.00     0.00     0.00     0.00     0.00
bmom,kft     0.00     0.00     0.00     0.00     0.00     0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 2 DISPLACEMENTS IN INCIDENCES 3 4 5
  NODE 3 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 4 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 5 0.00000E+00 0.00000E+00 0.00000E+00
0 FORCES ACTING ALONG THE 9 DOF
  NODE 3 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 4 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 5 0.00000E+00 0.00000E+00 0.00000E+00
OELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 5.00 ft
0 left half of span,at tenth points of length
span          span          span          span          span          span
+ 0.0         0.1         0.2         0.3         0.4         0.5
soil,k/ft    0.000    0.000    0.000    0.000    0.000    0.000
shear,k      0.00     0.00     0.00     0.00     0.00     0.00
bmom,kft     0.00     0.00     0.00     0.00     0.00     0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+ 0.5         0.6         0.7         0.8         0.9         1.0
soil,k/ft    0.000    0.000    0.000    0.000    0.000    0.000
shear,k      0.00     0.00     0.00     0.00     0.00     0.00
bmom,kft     0.00     0.00     0.00     0.00     0.00     0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 3 DISPLACEMENTS IN INCIDENCES 5 6 7
  NODE 5 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 6 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 7 0.00000E+00 0.00000E+00 0.00000E+00
0 FORCES ACTING ALONG THE 9 DOF
  NODE 5 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 6 0.00000E+00 0.00000E+00 0.00000E+00
  NODE 7 0.00000E+00 0.00000E+00 0.00000E+00
OELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 5.00 ft
0 left half of span,at tenth points of length
span          span          span          span          span          span
+ 0.0         0.1         0.2         0.3         0.4         0.5
soil,k/ft    0.000    0.000    0.000    0.000    0.000    0.000
shear,k      0.00     0.00     0.00     0.00     0.00     0.00
bmom,kft     0.00     0.00     0.00     0.00     0.00     0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+ 0.5         0.6         0.7         0.8         0.9         1.0
soil,k/ft    0.000    0.000    0.000    0.000    0.000    0.000
shear,k      0.00     0.00     0.00     0.00     0.00     0.00
bmom,kft     0.00     0.00     0.00     0.00     0.00     0.00
tdisp,ft     0.00000  0.00000  0.00000  0.00000  0.00000  0.00000
axial,k      0.00 AT 1st END and 0.00 AT 2nd END

```

Page 2

$$(-D_i + D_j) * GJ + V_j * L_{ij} = M_j - M_i, \text{ where } GJ=10, L_{ij}=5$$

$$-0.0 + 0.0 * 10 - 0 * 5 = 0$$

$$-0 + 0.0 = 0$$

Output Checks, Page 2

GJ=10; T = 1; L = 15;  
 (-GJ\*0.9656 + TL) = Bimomento = 5.344      cMsOut

---

0 ELEMENT 4 DISPLACEMENTS IN INCIDENCES 7 8 9

NODE 7	0.00000E+00	0.00000E+00	0.00000E+00
NODE 8	0.00000E+00	0.61681E-01	0.14750E-01
NODE 9	0.00000E+00	0.17239E+00	0.59675E-01

0 FORCES ACTING ALONG THE 9 DOF

NODE 7	0.00000E+00	-0.10000E+01	-0.53443E+01
NODE 8	0.00000E+00	-0.21316E-13	0.11369E-12
NODE 9	0.00000E+00	0.10000E+01	0.20682E+01

0 ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 5.00 ft  
 0 left half of span, at tenth points of length

span	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
bmom, kft	5.34	4.87	4.43	4.03	3.67	3.34
tdisp, ft	0.00000	0.00223	0.00866	0.01892	0.03265	0.04955
axial, k	0.00	AT 1st END and		0.00		AT 2nd END

0 right half of span, at tenth points of length

span	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
bmom, kft	3.34	3.04	2.76	2.51	2.28	2.07
tdisp, ft	0.04955	0.06934	0.09174	0.11652	0.14347	0.17239
axial, k	0.00	AT 1st END and		0.00		AT 2nd END

---

0 ELEMENT 5 DISPLACEMENTS IN INCIDENCES 9 10 11

NODE 9	0.00000E+00	0.17239E+00	0.59675E-01
NODE 10	0.00000E+00	0.34515E+00	0.30482E-01
NODE 11	0.00000E+00	0.53626E+00	0.82015E-01

0 FORCES ACTING ALONG THE 9 DOF

NODE 9	0.00000E+00	-0.10000E+01	-0.20682E+01
NODE 10	0.00000E+00	0.10658E-12	-0.11369E-11
NODE 11	0.00000E+00	0.10000E+01	0.70689E+00

0 ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 5.00 ft  
 0 left half of span, at tenth points of length

span	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
bmom, kft	2.07	1.88	1.70	1.54	1.39	1.25
tdisp, ft	0.17239	0.20309	0.23541	0.26919	0.30430	0.34061
axial, k	0.00	AT 1st END and		0.00		AT 2nd END

0 right half of span, at tenth points of length

span	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
bmom, kft	1.25	1.12	1.01	0.90	0.80	0.71
tdisp, ft	0.34061	0.37799	0.41634	0.45557	0.49557	0.53626
axial, k	0.00	AT 1st END and		0.00		AT 2nd END

---

0 ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13

NODE 11	0.00000E+00	0.53626E+00	0.82015E-01
NODE 12	0.00000E+00	0.74858E+00	0.35859E-01
NODE 13	0.00000E+00	0.96557E+00	0.87706E-01

0 FORCES ACTING ALONG THE 9 DOF

NODE 11	0.00000E+00	-0.10000E+01	-0.70689E+00
NODE 12	0.00000E+00	-0.63949E-13	0.56843E-12
NODE 13	0.00000E+00	0.10000E+01	-0.15632E-12

0 ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 5.00 ft  
 0 left half of span, at tenth points of length

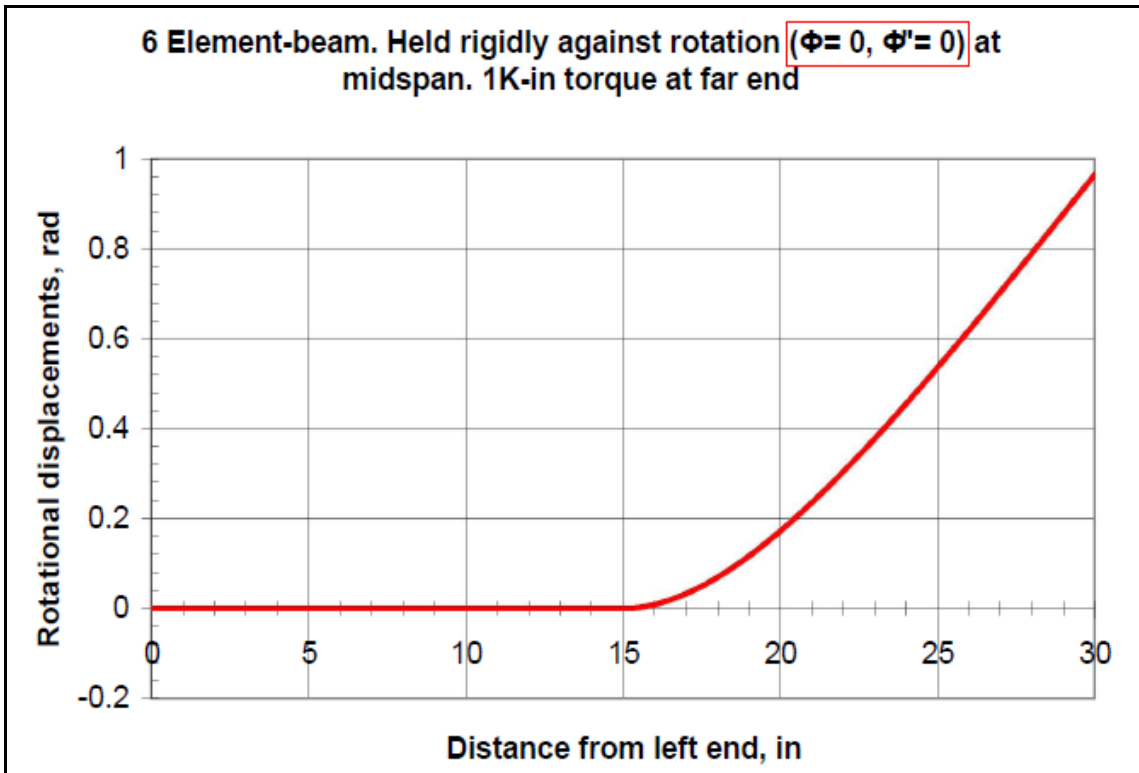
(-Di + Dj)\* GJ + Vj\*Lij = Mj-Mi, where GJ=10, Lij=5  
 -0.17239 + 0.53626)\*10 - 1\*5 = -1.3613  
 -2.07 + 0.71 = -1.36 OK

		cMsOut					
		span	span	span	span	span	span
		0.0	0.1	0.2	0.3	0.4	0.5
+	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
	bmom,kft	0.71	0.62	0.54	0.46	0.39	0.32
	tdisp,ft	0.53626	0.57756	0.61939	0.66169	0.70439	0.74742
	axial,k	0.00 AT 1st END and		0.00 AT 2nd END			
Oright half of span,at tenth points of length							
		span	span	span	span	span	span
		0.5	0.6	0.7	0.8	0.9	1.0
+	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
	bmom,kft	0.32	0.25	0.19	0.12	0.06	0.00
	tdisp,ft	0.74742	0.79073	0.83425	0.87793	0.92172	0.96557
	axial,k	0.00 AT 1st END and		0.00 AT 2nd END			

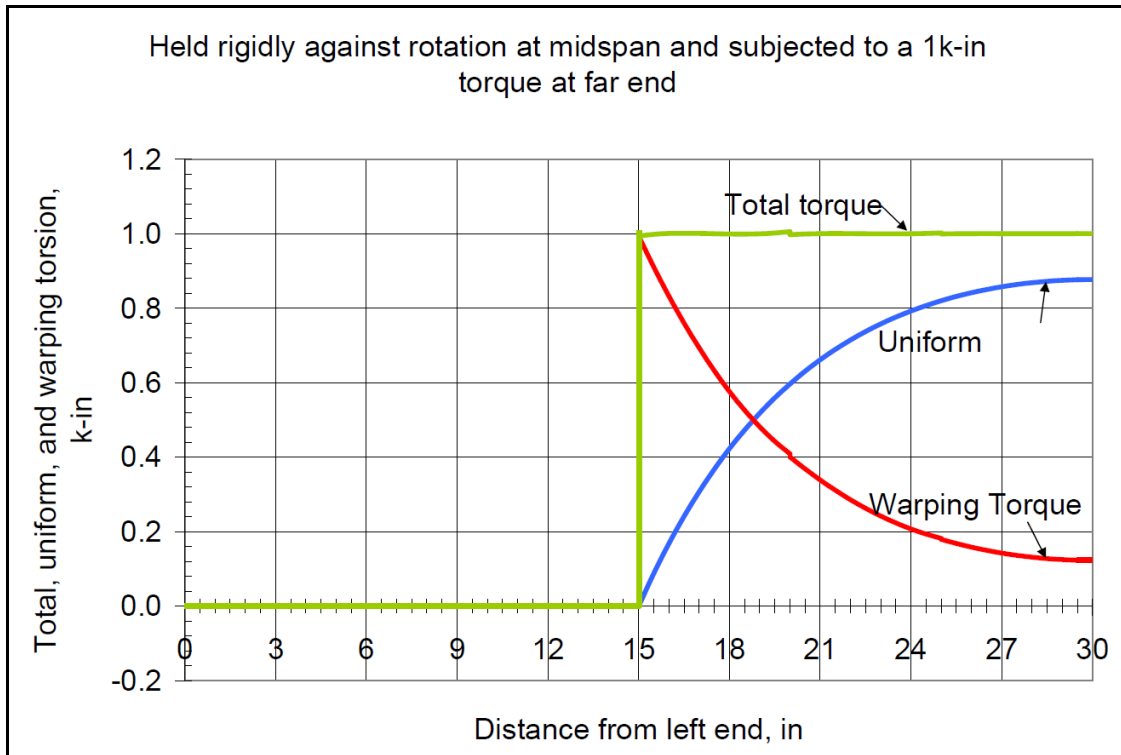
1

$(-D_i + D_j) * GJ + V_j * L_{ij} = M_j - M_i$ , where  $GJ=10$ ,  $L_{ij}=5$   
 $-0.53626 + 0.96557 * 10 - 1 * 5 = 0.7069$   
 $0 + 0.71 = 0.71$  OK

Output Checks, Last Page



Elastic Line of TWB Cantilevered at Midspan



Cross Section Torques Diagrams

Data Processed by Spreadsheet

elm	Z	$\Phi$	T	B(z)	$\Phi'$	$\Phi''$	$\Phi'''$	GJ $\Phi'$	-ECw $\Phi'''$	T
1	0	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	0.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	1	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	1.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	2	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	2.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	3	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	3.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	4	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	4.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
2	5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	5.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	6	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	6.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	7	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	7.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	8	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	8.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	9	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	9.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00

3	10	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	10.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	11	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	11.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	12	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	12.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	13	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	13.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	14	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
	14.5	0	0	0	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00E+00	0.00
4	15	0	-1	5.3443	0.00E+00	1.84E-02	-3.43E-03	0.00E+00	9.94E-01	0.99
	15.5	0.0022	-1	4.8666	8.80E-03	1.68E-02	-3.14E-03	8.80E-02	9.10E-01	1.00
	16	0.0087	-1	4.431	1.68E-02	1.53E-02	-2.87E-03	1.68E-01	8.33E-01	1.00
	16.5	0.0189	-1	4.0335	2.41E-02	1.39E-02	-2.62E-03	2.41E-01	7.60E-01	1.00
	17	0.0327	-1	3.6709	3.07E-02	1.27E-02	-2.39E-03	3.07E-01	6.93E-01	1.00
	17.5	0.0496	-1	3.3399	3.68E-02	1.15E-02	-2.18E-03	3.68E-01	6.32E-01	1.00
	18	0.0693	-1	3.0377	4.23E-02	1.05E-02	-1.99E-03	4.23E-01	5.77E-01	1.00
	18.5	0.0917	-1	2.7617	4.73E-02	9.52E-03	-1.82E-03	4.73E-01	5.26E-01	1.00
	19	0.1165	-1	2.5096	5.18E-02	8.66E-03	-1.66E-03	5.18E-01	4.82E-01	1.00
	19.5	0.1435	-1	2.2791	5.59E-02	7.86E-03	-1.53E-03	5.59E-01	4.43E-01	1.00
5	20	0.1724	-1	2.0682	5.97E-02	7.13E-03	-1.38E-03	5.97E-01	4.01E-01	1.00
	20.5	0.2031	-1	1.8752	6.31E-02	6.47E-03	-1.27E-03	6.31E-01	3.69E-01	1.00
	21	0.2354	-1	1.6984	6.62E-02	5.86E-03	-1.17E-03	6.62E-01	3.39E-01	1.00
	21.5	0.2692	-1	1.5362	6.89E-02	5.30E-03	-1.07E-03	6.89E-01	3.11E-01	1.00
	22	0.3043	-1	1.3873	7.15E-02	4.78E-03	-9.85E-04	7.15E-01	2.86E-01	1.00
	22.5	0.3406	-1	1.2504	7.37E-02	4.31E-03	-9.06E-04	7.37E-01	2.63E-01	1.00
	23	0.378	-1	1.1242	7.58E-02	3.88E-03	-8.34E-04	7.58E-01	2.42E-01	1.00
	23.5	0.4163	-1	1.0078	7.76E-02	3.48E-03	-7.71E-04	7.76E-01	2.24E-01	1.00
	24	0.4556	-1	0.9	7.93E-02	3.10E-03	-7.15E-04	7.93E-01	2.07E-01	1.00
	24.5	0.4956	-1	0.8	8.07E-02	2.76E-03	-6.67E-04	8.07E-01	1.93E-01	1.00
6	25	0.5363	-1	0.7069	8.20E-02	2.44E-03	-6.18E-04	8.20E-01	1.79E-01	1.00
	25.5	0.5776	-1	0.6199	8.32E-02	2.14E-03	-5.80E-04	8.32E-01	1.68E-01	1.00
	26	0.6194	-1	0.5382	8.42E-02	1.86E-03	-5.47E-04	8.42E-01	1.58E-01	1.00
	26.5	0.6617	-1	0.4612	8.50E-02	1.59E-03	-5.17E-04	8.50E-01	1.50E-01	1.00
	27	0.7044	-1	0.3882	8.58E-02	1.34E-03	-4.92E-04	8.58E-01	1.43E-01	1.00
	27.5	0.7474	-1	0.3185	8.64E-02	1.10E-03	-4.70E-04	8.64E-01	1.36E-01	1.00
	28	0.7907	-1	0.2516	8.69E-02	8.67E-04	-4.53E-04	8.69E-01	1.31E-01	1.00
	28.5	0.8342	-1	0.1868	8.72E-02	6.44E-04	-4.40E-04	8.72E-01	1.28E-01	1.00
	29	0.8779	-1	0.1236	8.75E-02	4.27E-04	-4.31E-04	8.75E-01	1.25E-01	1.00
	29.5	0.9217	-1	0.0616	8.77E-02	2.12E-04	-4.26E-04	8.77E-01	1.24E-01	1.00
	30	0.9656	-1	0	8.77E-02	-4.35E-07	-4.26E-04	8.77E-01	1.23E-01	1.00

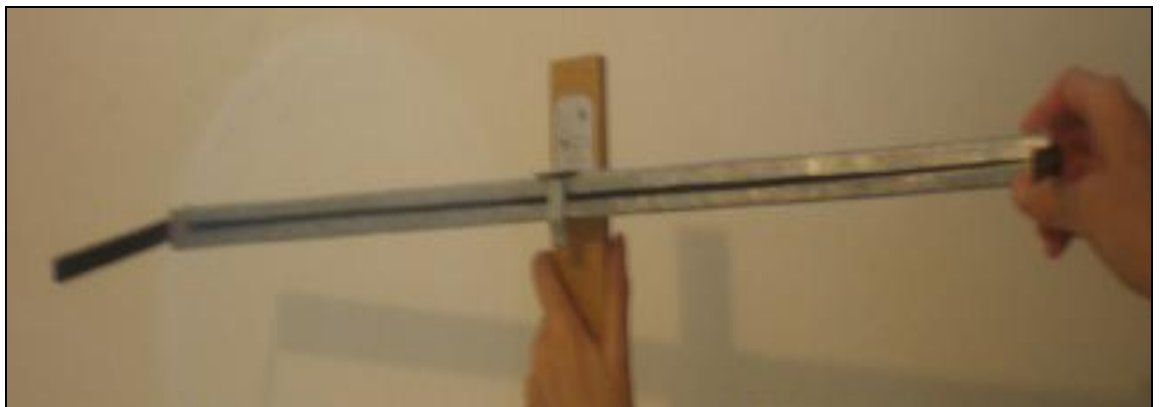
## APPENDIX B

### BAR RESTRAINED AGAINST TWIST AT MIDSPAN

The present material contents the input model, input data, input forms, figures of output data in notepad version, and excel processed output data and charts for the corresponding case. The asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP. The cross section shown in the input form is not that of the real bar in the parametric experiment. It just provides an order of magnitude for the bogus cross section torsional properties in order to make verisimilar the input data.

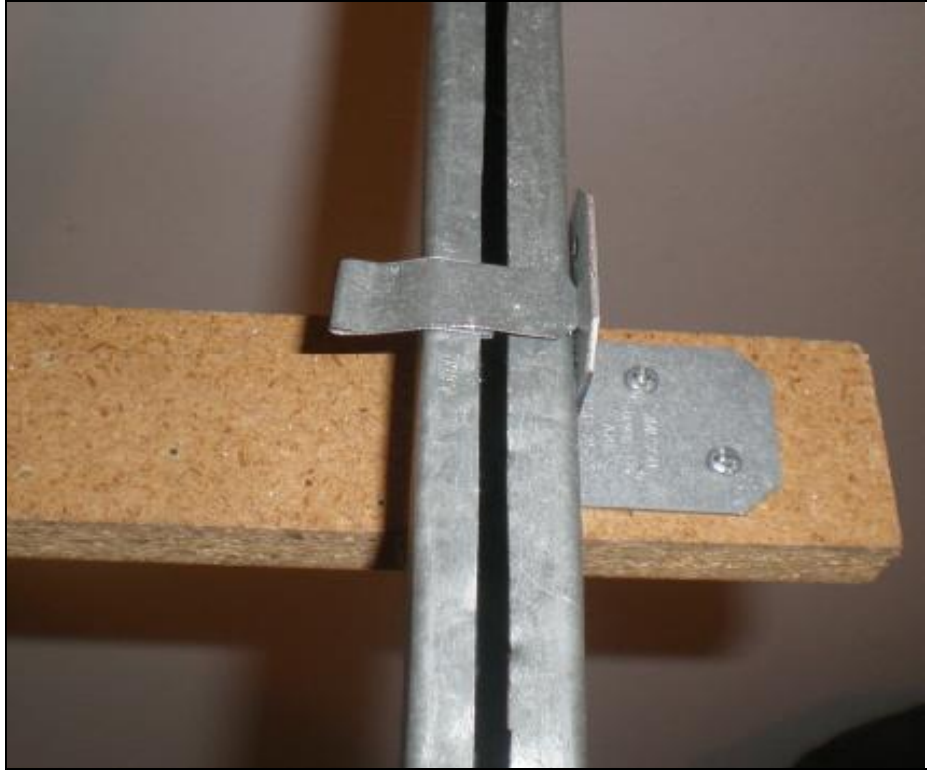


Straight TWB Restrained against Rotation at Midspan before Twisting Action

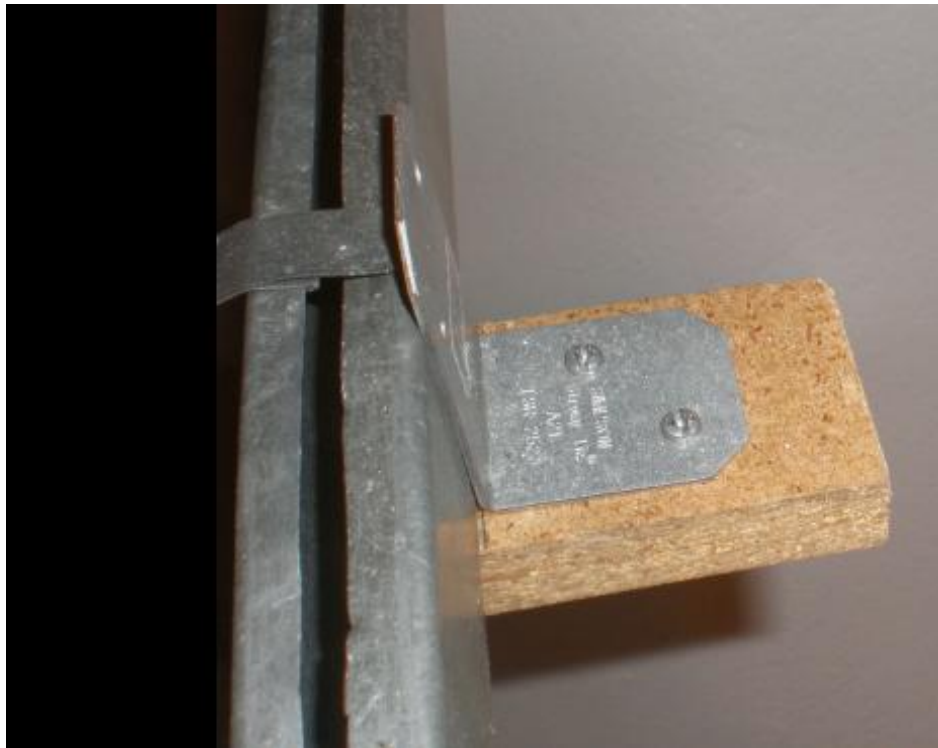


Straight TWB Restrained against Rotation at Midspan about to be Twisted





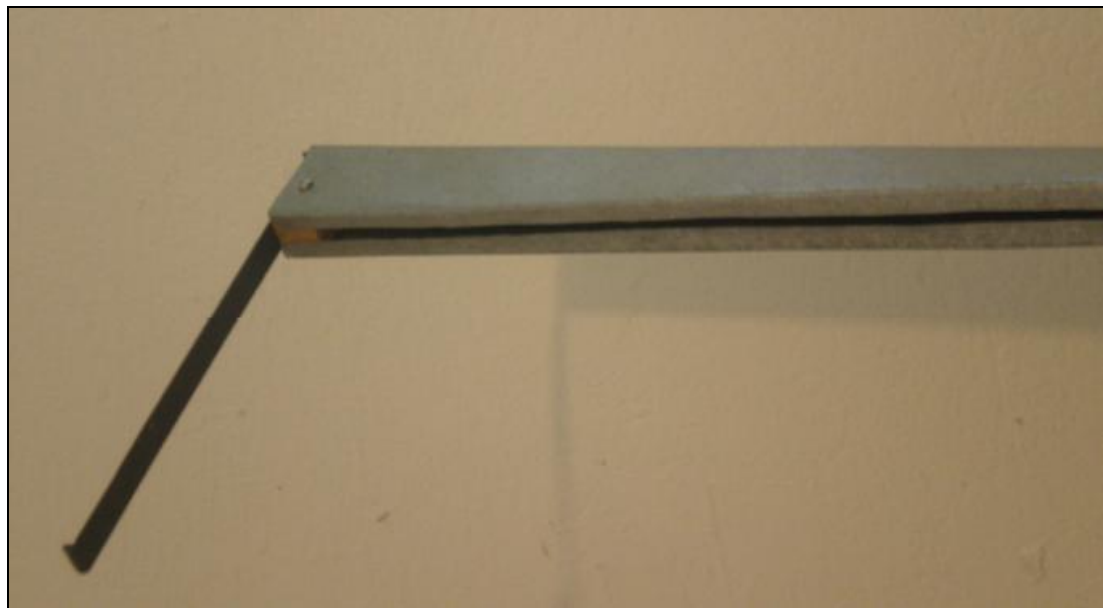
A Square Fastener at Right Side and a Separator at Top Prevent Rotation



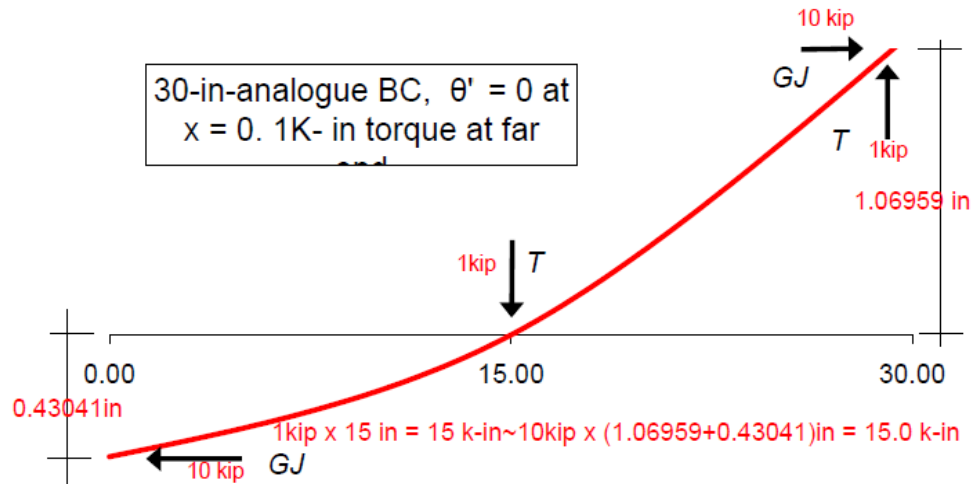
Detail of Right Fastener



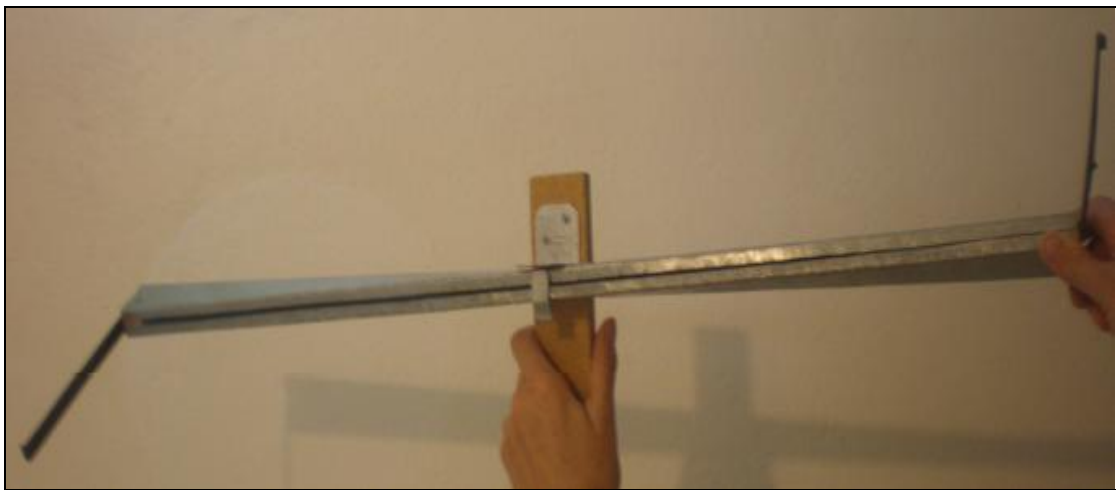
At Both Edges, Cross Section Distortion Prevented by a Piece of Wood Only Fastened at One Side with Centerline Guides for Measuring the Twist Angle



Elastic Line Detail at Free Near End without External Torque



Analogue EBC in 2<sup>nd</sup> Order Equilibrium with a 1 kip Reaction at Midspan Support



Zero External Torque with Restrained Warping at First Span  $T_w + T_t = T = 0$ .  
 $T_w$  &  $T_t \neq 0$

6EL

6 Elements' beam. Torque angle restrained ad midspan. 1K-in torque at far end

6	13	1	9	0	1	0	29000.
1	1	2	3	0.03	1.	5.	-10.
2	3	4	5	0.03	1.	5.	-10.
3	5	6	7	0.03	1.	5.	-10.
4	7	8	9	0.03	1.	5.	-10.
5	9	10	11	0.03	1.	5.	-10.
6	11	12	13	0.03	1.	5.	-10.
1	7	0	1	0			
113	1.						

$-GJ$

$KX=KZ=0, KY=1$  in Node 7

Total Torque

Input Data

PROGRAM BMTORSW INPUT FORM																																										
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43
ALPHAMERIC DESCRIPTION OF THE JOB																				FILE NAME:																						
Beam with torsion angle restrained at midspan subjected																																										
NEL	NOD	NSUP	NSPD	JBW	NFX	NFY	NFZ	ELASTICITY (E11.4)										GENE																								
6	1	3	1		9	0	1	0	2900 0.																																	
NE	NI	NC	NJ	CW			AREA	LENGTH	WTANG	SOIL NI			SOIL N																													
1	1	2	3	0, 0 3			1.	5.																																		
2	3	4	5	0, 0 3			1.	5.																																		
3	5	6	7	0, 0 3			1.	5.																																		
4	7	8	9	0, 0 3			1.	5.																																		
5	9	10	11	0, 0 3			1.	5.																																		
6	11	12	13	0, 0 3			1.	5.																																		
NF	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N																					
1	7	0	1	0																																						
NF	N	KD	SPRING			N	KD	SPRING			N	KD	SPRING																													
NF	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	N	LOAD																																
1	13	1.																																								
NF	ELEM	END	ROTATIONAL SPRING MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS																													
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	42	43

Input Form with Input Model, Left Side

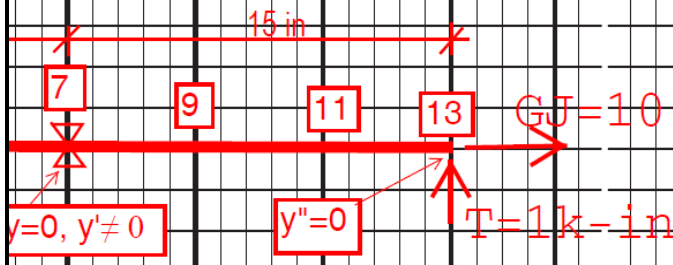
38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

NAME: subjected to torsion at one end

GENERAL INFORMATION

PARTIAL RESTRAINT AT END ? 1 FOR YES 0 FOR NO

NI	SOIL NJ	SOIL TI	SOIL TJ	ANGLE	- G*J	Wn @ NI	Wn @ NJ	1	2
					-10.				
					-10.				
					-10.				
					-10.				
					-10.				
					-10.				



KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	SUPPORTS

KD	SPRING	N	KD	SPRING	SPRINGS AT SUPPORTS

E	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	APPLIED NODAL FORCES

ROTATIONAL SPRING MODULUS	ELEM	END	ROTATIONAL SPRING MODULUS	ELEM	END	ROTATIONAL SPRING MODULUS	ELEM	END	SPRINGS AT ELEM. END, 1 OR 2, IF ANY

38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

Input Form with Input Model, Right Side

```

6eout
-----
YOU ARE USING COMPUTER PROGRAM BMTORSW, DEVELOPED BY DR. BERNARDO DESCHAPELLES
0 INPUT DATA FILE NAME IS = 6el.txt
0 OUTPUT FILE NAME IS = 6eout.txt
0 STORAGE FILE FOR POST-PROCESSING WITH EXCEL = 6egrp.grf
-----
6 Elements' beam. Torque angle restrained ad midspan. 1K-in torque at far end
0 modulus of elasticity of the material= 29000. k/ft2
0ELEM nodes inertia length distrib. load AXIAL SOIL NORMAL MODULUS, Ksf angle
0 i j ft.4 ft at i at j LOAD 1st END 2nd END rad
1 1 3 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
2 3 5 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
3 5 7 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
4 7 9 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
5 9 11 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
6 11 13 0.03000 5.00 0.000 0.000 -10.00 0.0 0.0 0.000 00
0 INPUT DATA RELATED TO THE 1 SUPPORTS
1 7 0 1 0
0 INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2
113 1.00
0 FINAL SOLUTION FOUND AFTER 1 ITERATIONS
0 Output of nodal displacements in reference to global axes
0node displ. displ. displ. node displ. displ. displ.
along x along y around z along x along y around z
or nonn1 or nonn2 or nonn 3 or nonn1 or nonn2 or nonn 3
+ 1 0.0000E+00 -0.4304E+00 0.1925E-01
+ 2 0.0000E+00 -0.3811E+00 0.8372E-02
+ 3 0.0000E+00 -0.3295E+00 0.2209E-01
+ 4 0.0000E+00 -0.2680E+00 0.1084E-01
+ 5 0.0000E+00 -0.1988E+00 0.3142E-01
+ 6 0.0000E+00 -0.1071E+00 0.1649E-01
+ 7 0.0000E+00 0.0000E+00 0.5000E-01
+ 8 0.0000E+00 0.1429E+00 0.2518E-01
+ 9 0.0000E+00 0.3012E+00 0.6858E-01
+ 10 0.0000E+00 0.4820E+00 0.3083E-01
+ 11 0.0000E+00 0.6705E+00 0.7791E-01
+ 12 0.0000E+00 0.8689E+00 0.3329E-01
+ 13 0.0000E+00 0.1070E+01 0.8075E-01
1kip x 15 in = 15 k-in ~ 10kip x
(1.06959+0.43041)in = 15.0
-----
OUTPUT OF SOIL REACTIONS, STRESSES AND TRANSVERSE DISPLACEMENTS
-----
0 ELEMENT 1 DISPLACEMENTS IN INCIDENCES 1 2 3
NODE 1 0.00000E+00 -0.43041E+00 0.19253E-01
NODE 2 0.00000E+00 -0.38111E+00 0.83720E-02
NODE 3 0.00000E+00 -0.32946E+00 0.22086E-01
0 FORCES ACTING ALONG THE 9 DOF
NODE 1 0.00000E+00 -0.46896E-12 -0.95923E-13
NODE 2 0.00000E+00 -0.16342E-12 -0.38654E-11
NODE 3 0.00000E+00 -0.85265E-12 0.10094E+01
0ELEMEN 1, FROM NODE 1, TO NODE 3 - LENGTH = 5.00 ft
0 left half of span, at tenth points of length
+ soil, k/ft span 0.0 0.1 0.2 0.3 0.4 0.5
shear, k 0.000 0.000 0.000 0.000 0.000 0.000
0.00 0.00 0.00 0.00 0.00 0.00
Page 1

```

Page 1 Output Data

```

6eout
bmom,kft      0.00      0.10      0.19      0.29      0.39      0.49
tdisp,ft     -0.43041   -0.42078   -0.41112   -0.40140   -0.39160   -0.38170
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+
soil,k/ft     0.000      0.000      0.000      0.000      0.000      0.000      0.000
shear,k       0.00      0.00      0.00      0.00      0.00      0.00      0.00
bmom,kft      0.49      0.59      0.69      0.79      0.90      1.01
tdisp,ft     -0.38170   -0.37165   -0.36143   -0.35101   -0.34037   -0.32946
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 2 DISPLACEMENTS IN INCIDENCES 3 4 5
  NODE 3 0.00000E+00 -0.32946E+00 0.22086E-01
  NODE 4 0.00000E+00 -0.26801E+00 0.10836E-01
  NODE 5 0.00000E+00 -0.19881E+00 0.31420E-01
0 FORCES ACTING ALONG THE 9 DOF
  NODE 3 0.00000E+00 -0.21316E-12 -0.10094E+01
  NODE 4 0.00000E+00 -0.56843E-13 0.34106E-11
  NODE 5 0.00000E+00 -0.28422E-13 0.23160E+01
0 ELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 5.00 ft
0 left half of span,at tenth points of length
span          span          span          span          span          span
+
soil,k/ft     0.00      0.10      0.20      0.30      0.40      0.50      0.000
shear,k       0.00      0.00      0.00      0.00      0.00      0.00      0.00
bmom,kft      1.01      1.12      1.24      1.36      1.48      1.60
tdisp,ft     -0.32946   -0.31827   -0.30675   -0.29488   -0.28262   -0.26994
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+
soil,k/ft     0.000      0.000      0.000      0.000      0.000      0.000      0.000
shear,k       0.00      0.00      0.00      0.00      0.00      0.00      0.00
bmom,kft      1.60      1.74      1.87      2.01      2.16      2.32
tdisp,ft     -0.26994   -0.25679   -0.24314   -0.22896   -0.21420   -0.19881
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 3 DISPLACEMENTS IN INCIDENCES 5 6 7
  NODE 5 0.00000E+00 -0.19881E+00 0.31420E-01
  NODE 6 0.00000E+00 -0.10711E+00 0.16489E-01
  NODE 7 0.00000E+00 0.00000E+00 0.50000E-01
0 FORCES ACTING ALONG THE 9 DOF
  NODE 5 0.00000E+00 -0.99476E-13 -0.23160E+01
  NODE 6 0.00000E+00 0.42633E-13 -0.22737E-12
  NODE 7 0.00000E+00 -0.28422E-13 0.43041E+01
0 ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 5.00 ft
0 left half of span,at tenth points of length
span          span          span          span          span          span
+
soil,k/ft     0.000      0.000      0.000      0.000      0.000      0.000      0.000
shear,k       0.00      0.00      0.00      0.00      0.00      0.00      0.00
bmom,kft      2.32      2.48      2.64      2.82      3.00      3.19
tdisp,ft     -0.19881   -0.18276   -0.16600   -0.14848   -0.13014   -0.11095
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span          span          span          span          span          span
+
soil,k/ft     0.000      0.000      0.000      0.000      0.000      0.000      0.000
shear,k       0.00      0.00      0.00      0.00      0.00      0.00      0.00
bmom,kft      3.19      3.40      3.61      3.83      4.06      4.30
tdisp,ft     -0.11095   -0.09084   -0.06975   -0.04762   -0.02439   0.00000
axial,k       0.00 AT 1st END and 0.00 AT 2nd END
Page 2
0 -(-0.19881)*10 +0*5= 1.988
4.30- 2.32 = 1.98 OK
(-Di + Dj)* GJ + Vj*L = Mj - Mi, where GJ=10

```

Page 2 Output Data

6eout										
-----										
0	ELEMENT	4	DISPLACEMENTS IN INCIDENCES			7	8	9		
	NODE	7	0.00000E+00	0.00000E+00	0.50000E-01					
	NODE	8	0.00000E+00	0.14289E+00	0.25178E-01					
	NODE	9	0.00000E+00	0.30119E+00	0.68580E-01					
0	FORCES ACTING ALONG THE 9 DOF									
	NODE	7	0.00000E+00	-0.10000E+01	-0.43041E+01					
	NODE	8	0.00000E+00	0.14211E-12	0.68212E-12					
	NODE	9	0.00000E+00	0.10000E+01	0.23160E+01					
ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 5.00 ft										
0 left half of span,at tenth points of length										
		span	span	span	span	span	span	span	span	
+		0.0	0.1	0.2	0.3	0.4	0.5			
	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000		
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00		
	bmom,kft	4.30	4.06	3.83	3.61	3.40	3.19	3.19		
	tdisp,ft	0.00000	0.02561	0.05238	0.08025	0.10916	0.13905	0.13905		
	axial,k	0.00 AT 1st END and			0.00 AT 2nd END					
Oright half of span,at tenth points of length										
		span	span	span	span	span	span	span	span	
+		0.5	0.6	0.7	0.8	0.9	1.0			
	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000		
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00		
	bmom,kft	3.19	3.00	2.82	2.64	2.48	2.32	2.32		
	tdisp,ft	0.13905	0.16986	0.20152	0.23400	0.26724	0.30119	0.30119		
	axial,k	0.00 AT 1st END and			0.00 AT 2nd END					
-----										
0	ELEMENT	5	DISPLACEMENTS IN INCIDENCES			9	10	11		
	NODE	9	0.00000E+00	0.30119E+00	0.68580E-01					
	NODE	10	0.00000E+00	0.48199E+00	0.30831E-01					
	NODE	11	0.00000E+00	0.67054E+00	0.77914E-01					
0	FORCES ACTING ALONG THE 9 DOF									
	NODE	9	0.00000E+00	-0.10000E+01	-0.23160E+01					
	NODE	10	0.00000E+00	-0.34106E-12	0.22737E-11					
	NODE	11	0.00000E+00	0.10000E+01	0.10094E+01					
ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 5.00 ft										
0 left half of span,at tenth points of length										
		span	span	span	span	span	span	span	span	
+		0.0	0.1	0.2	0.3	0.4	0.5			
	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000		
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00		
	bmom,kft	2.32	2.16	2.01	1.87	1.74	1.60	1.60		
	tdisp,ft	0.30119	0.33580	0.37104	0.40686	0.44321	0.48006	0.48006		
	axial,k	0.00 AT 1st END and			0.00 AT 2nd END					
Oright half of span,at tenth points of length										
		span	span	span	span	span	span	span	span	
+		0.5	0.6	0.7	0.8	0.9	1.0			
	soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000		
	shear,k	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00		
	bmom,kft	1.60	1.48	1.36	1.24	1.12	1.01	1.01		
	tdisp,ft	0.48006	0.51738	0.55512	0.59325	0.63173	0.67054	0.67054		
	axial,k	0.00 AT 1st END and			0.00 AT 2nd END					
-----										
0	ELEMENT	6	DISPLACEMENTS IN INCIDENCES			11	12	13		
	NODE	11	0.00000E+00	0.67054E+00	0.77914E-01					
	NODE	12	0.00000E+00	0.86889E+00	0.33295E-01					
	NODE	13	0.00000E+00	0.10696E+01	0.80747E-01					
0	FORCES ACTING ALONG THE 9 DOF									
	NODE	11	0.00000E+00	-0.10000E+01	-0.10094E+01					
	NODE	12	0.00000E+00	-0.51159E-12	-0.10914E-10					
	NODE	13	0.00000E+00	0.10000E+01	-0.68212E-12					
ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 5.00 ft										
0 left half of span,at tenth points of length										



		6eout						
		span	span	span	span	span	span	span
+	soil,k/ft	0.0	0.1	0.2	0.3	0.4	0.5	0.000
	shear,k	0.000	0.000	0.000	0.000	0.000	0.000	0.000
	bmom,kft	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
	tdisp,ft	1.01	0.90	0.79	0.69	0.59	0.49	0.49
	axial,k	0.67054	0.70963	0.74899	0.78857	0.82835	0.86830	0.86830
		0.00 AT 1st END and 0.00 AT 2nd END						
		Oright half of span,at tenth points of length						
		span	span	span	span	span	span	span
+	soil,k/ft	0.5	0.6	0.7	0.8	0.9	1.0	0.000
	shear,k	0.000	0.000	0.000	0.000	0.000	0.000	0.000
	bmom,kft	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00	-1.00
	tdisp,ft	0.49	0.39	0.29	0.19	0.10	0.00	0.00
	axial,k	0.86830	0.90840	0.94860	0.98888	1.02922	1.06959	1.06959
		0.00 AT 1st END and 0.00 AT 2nd END						

1

↑ T      ↑ L      ↑ GJ      ↑  $\Phi_{13}$       ↑  $-\Phi_1$

1kip x 15 in = 15 k-in ~ 10kip x (1.06959+0.43041)in = 15.0 k-in

Last Page Output Data

Table of Bimoment, Torque and Torque Angle and Derivatives

				GJ	10	ksi-in^2		
6 Elements-beam. Fork ad midspan. 1K-in torque at far end								
				E	29000	ksi/in^2		
				Cw	0.030	in^6		
elm	Z	$\Phi$	Torque	Bimoment	$\Phi'$	$\Phi''$	$\Phi'''$	
1	0.00	-0.43	0.00	0.000	1.925E-02	-2.452E-08	2.214E-04	
	0.50	-0.42	0	0.096	1.928E-02	1.107E-04	2.217E-04	
	1.00	-0.41	0	0.193	1.936E-02	2.217E-04	2.226E-04	
	1.50	-0.4	0	0.290	1.950E-02	3.334E-04	2.242E-04	
	2.00	-0.39	0	0.388	1.970E-02	4.460E-04	2.264E-04	
	2.50	-0.38	0	0.487	1.995E-02	5.599E-04	2.293E-04	
	3.00	-0.37	0	0.588	2.026E-02	6.754E-04	2.329E-04	
	3.50	-0.36	0	0.690	2.062E-02	7.929E-04	2.371E-04	
	4.00	-0.35	0	0.794	2.105E-02	9.126E-04	2.420E-04	
	4.50	-0.34	0	0.900	2.154E-02	1.035E-03	2.475E-04	
	5.00	-0.33	0	1.009	2.209E-02	1.160E-03	2.537E-04	
2								
	5.00	-0.33	0	1.009	2.209E-02	1.160E-03	2.542E-04	
	5.50	-0.32	0	1.121	2.270E-02	1.289E-03	2.610E-04	
	6.00	-0.31	0	1.237	2.338E-02	1.421E-03	2.687E-04	
	6.50	-0.29	0	1.355	2.412E-02	1.558E-03	2.772E-04	

elm	Z	$\Phi$	T	Bimom	$\Phi'$	$\Phi''$	$\Phi'''$
	7.00	-0.28	0	1.478	2.493E-02	1.699E-03	2.866E-04
	7.50	-0.27	0	1.605	2.582E-02	1.844E-03	2.968E-04
	8.00	-0.26	0	1.736	2.678E-02	1.996E-03	3.079E-04
	8.50	-0.24	0	1.873	2.782E-02	2.152E-03	3.198E-04
	9.00	-0.23	0	2.015	2.893E-02	2.316E-03	3.326E-04
	9.50	-0.21	0	2.162	3.013E-02	2.485E-03	3.462E-04
	10.00	-0.2	0	2.316	3.142E-02	2.662E-03	3.608E-04
3							
	10.00	-0.2	0	2.316	3.142E-02	2.662E-03	3.618E-04
	10.50	-0.18	0	2.477	3.280E-02	2.847E-03	3.772E-04
	11.00	-0.17	0	2.644	3.427E-02	3.039E-03	3.938E-04
	11.50	-0.15	0	2.819	3.584E-02	3.241E-03	4.118E-04
	12.00	-0.13	0	3.003	3.751E-02	3.451E-03	4.310E-04
	12.50	-0.11	0	3.195	3.929E-02	3.672E-03	4.516E-04
	13.00	-0.09	0	3.396	4.118E-02	3.903E-03	4.735E-04
	13.50	-0.07	0	3.607	4.320E-02	4.146E-03	4.966E-04
	14.00	-0.05	0	3.828	4.533E-02	4.400E-03	5.211E-04
	14.50	-0.02	0	4.060	4.760E-02	4.667E-03	5.469E-04
	15.00	0	0	4.304	5.000E-02	4.947E-03	5.739E-04
4							
	15.00	0	-1	4.304	5.000E-02	4.947E-03	-5.739E-04
	15.50	0.026	-1	4.060	5.240E-02	4.667E-03	-5.469E-04
	16.00	0.052	-1	3.828	5.467E-02	4.400E-03	-5.211E-04
	16.50	0.08	-1	3.607	5.680E-02	4.146E-03	-4.966E-04
	17.00	0.109	-1	3.396	5.882E-02	3.903E-03	-4.735E-04
	17.50	0.139	-1	3.195	6.071E-02	3.672E-03	-4.516E-04
	18.00	0.17	-1	3.003	6.249E-02	3.451E-03	-4.310E-04
	18.50	0.202	-1	2.819	6.416E-02	3.241E-03	-4.118E-04
	19.00	0.234	-1	2.644	6.573E-02	3.039E-03	-3.938E-04
	19.50	0.267	-1	2.477	6.720E-02	2.847E-03	-3.772E-04
	20.00	0.301	-1	2.316	6.858E-02	2.662E-03	-3.618E-04
5							
	20.00	0.301	-1	2.316	6.858E-02	2.662E-03	-3.608E-04
	20.50	0.336	-1	2.162	6.987E-02	2.485E-03	-3.462E-04
	21.00	0.371	-1	2.015	7.107E-02	2.316E-03	-3.326E-04
	21.50	0.407	-1	1.873	7.218E-02	2.152E-03	-3.198E-04
	22.00	0.443	-1	1.736	7.322E-02	1.996E-03	-3.079E-04

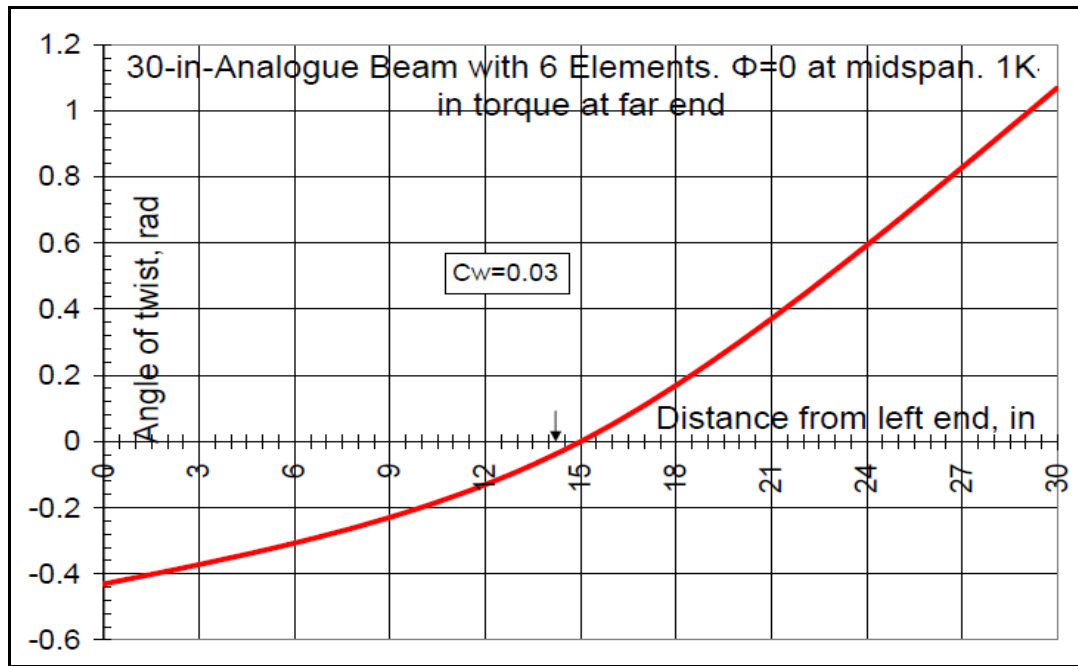
elm	Z	$\Phi$	T	Bimom	$\Phi'$	$\Phi''$	$\Phi'''$
	22.50	0.48	-1	1.605	7.418E-02	1.844E-03	-2.968E-04
	23.00	0.517	-1	1.478	7.507E-02	1.699E-03	-2.866E-04
	23.50	0.555	-1	1.355	7.588E-02	1.558E-03	-2.772E-04
	24.00	0.593	-1	1.237	7.662E-02	1.421E-03	-2.687E-04
	24.50	0.632	-1	1.121	7.730E-02	1.289E-03	-2.610E-04
	25.00	0.671	-1	1.009	7.791E-02	1.160E-03	-2.542E-04
6							
	25.00	0.671	-1	1.009	7.791E-02	1.160E-03	-2.537E-04
	25.50	0.71	-1	0.900	7.846E-02	1.035E-03	-2.475E-04
	26.00	0.749	-1	0.794	7.895E-02	9.126E-04	-2.420E-04
	26.50	0.789	-1	0.690	7.938E-02	7.929E-04	-2.371E-04
	27.00	0.828	-1	0.588	7.974E-02	6.754E-04	-2.329E-04
	27.50	0.868	-1	0.487	8.005E-02	5.599E-04	-2.293E-04
	28.00	0.908	-1	0.388	8.030E-02	4.460E-04	-2.264E-04
	28.50	0.949	-1	0.290	8.050E-02	3.334E-04	-2.242E-04
	29.00	0.989	-1	0.193	8.064E-02	2.217E-04	-2.226E-04
	29.50	1.029	-1	0.096	8.072E-02	1.107E-04	-2.217E-04
	30.00	1.07	-1	0.000	8.075E-02	-2.452E-08	-2.214E-04

Table of Excel Processed Data on Torques

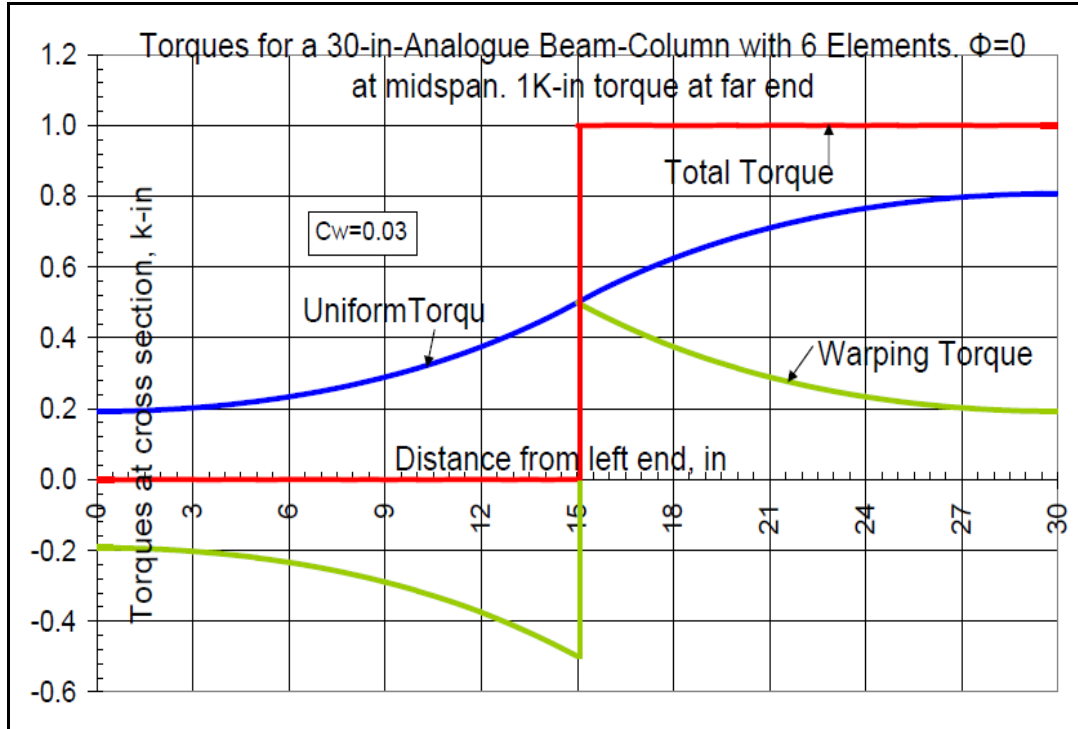
1st. order General Torsion Theory				
elm	Z	$GJ\Phi'$	$-EC_w\Phi'''$	T
1		St. Venant	Vlasov	Total
	0.00	1.925E-01	-1.926E-01	-1.180E-04
	0.50	1.928E-01	-1.929E-01	-7.900E-05
	1.00	1.936E-01	-1.937E-01	-6.200E-05
	1.50	1.950E-01	-1.951E-01	-5.400E-05
	2.00	1.970E-01	-1.970E-01	3.200E-05
	2.50	1.995E-01	-1.995E-01	9.000E-06
	3.00	2.026E-01	-2.026E-01	-2.300E-05
	3.50	2.062E-01	-2.063E-01	-7.700E-05
	4.00	2.105E-01	-2.105E-01	-4.000E-05
	4.50	2.154E-01	-2.153E-01	7.500E-05
	5.00	2.209E-01	-2.207E-01	1.810E-04
2				
	5.00	2.209E-01	-2.212E-01	-2.540E-04

elm	Z	GJΦ'	-ECwΦ'''	T
	5.50	2.270E-01	-2.271E-01	-7.000E-05
	6.00	2.338E-01	-2.338E-01	3.100E-05
	6.50	2.412E-01	-2.412E-01	3.600E-05
	7.00	2.493E-01	-2.493E-01	-4.200E-05
	7.50	2.582E-01	-2.582E-01	-1.600E-05
	8.00	2.678E-01	-2.679E-01	-7.300E-05
	8.50	2.782E-01	-2.782E-01	-2.600E-05
	9.00	2.893E-01	-2.894E-01	-6.200E-05
	9.50	3.013E-01	-3.012E-01	1.060E-04
	10.00	3.142E-01	-3.139E-01	3.040E-04
3				
	10.00	3.142E-01	-3.148E-01	-5.660E-04
	10.50	3.280E-01	-3.282E-01	-1.640E-04
	11.00	3.427E-01	-3.426E-01	9.400E-05
	11.50	3.584E-01	-3.583E-01	1.340E-04
	12.00	3.751E-01	-3.750E-01	1.300E-04
	12.50	3.929E-01	-3.929E-01	8.000E-06
	13.00	4.118E-01	-4.119E-01	-1.450E-04
	13.50	4.320E-01	-4.320E-01	-4.200E-05
	14.00	4.533E-01	-4.534E-01	-5.700E-05
	14.50	4.760E-01	-4.758E-01	1.970E-04
	15.00	5.000E-01	-4.993E-01	7.070E-04
4				
	15.00	5.000E-01	4.993E-01	9.993E-01
	15.50	5.240E-01	4.758E-01	9.998E-01
	16.00	5.467E-01	4.534E-01	1.000E+00
	16.50	5.680E-01	4.320E-01	1.000E+00
	17.00	5.882E-01	4.119E-01	1.000E+00
	17.50	6.071E-01	3.929E-01	1.000E+00
	18.00	6.249E-01	3.750E-01	9.999E-01
	18.50	6.416E-01	3.583E-01	9.999E-01
	19.00	6.573E-01	3.426E-01	9.999E-01
	19.50	6.720E-01	3.282E-01	1.000E+00
	20.00	6.858E-01	3.148E-01	1.001E+00
5	20.00	6.858E-01	3.139E-01	9.997E-01
	20.50	6.987E-01	3.012E-01	9.999E-01
	21.00	7.107E-01	2.894E-01	1.000E+00

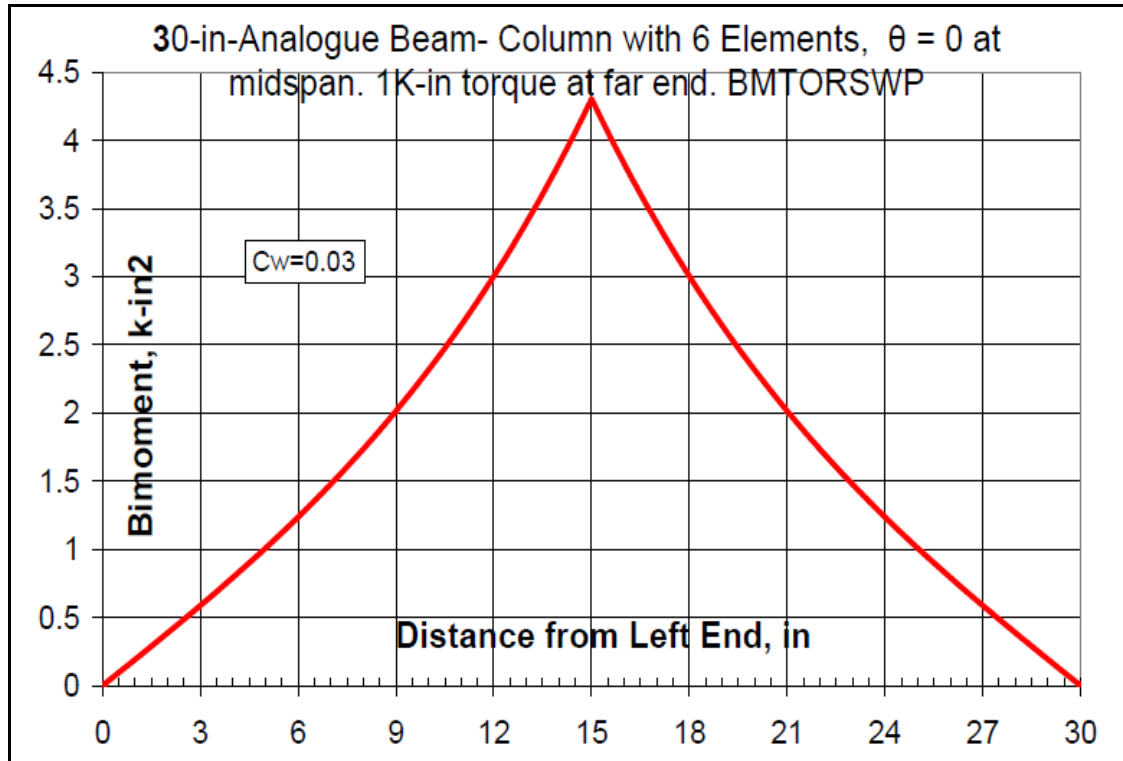
elm	Z	GJΦ'	-ECwΦ'''	T
	21.50	7.218E-01	2.782E-01	1.000E+00
	22.00	7.322E-01	2.679E-01	1.000E+00
	22.50	7.418E-01	2.582E-01	1.000E+00
	23.00	7.507E-01	2.493E-01	1.000E+00
	23.50	7.588E-01	2.412E-01	1.000E+00
	24.00	7.662E-01	2.338E-01	1.000E+00
	24.50	7.730E-01	2.271E-01	1.000E+00
	25.00	7.791E-01	2.212E-01	1.000E+00
6	25.00	7.791E-01	2.207E-01	9.998E-01
	25.50	7.846E-01	2.153E-01	9.999E-01
	26.00	7.895E-01	2.105E-01	1.000E+00
	26.50	7.938E-01	2.063E-01	1.000E+00
	27.00	7.974E-01	2.026E-01	1.000E+00
	27.50	8.005E-01	1.995E-01	1.000E+00
	28.00	8.030E-01	1.970E-01	1.000E+00
	28.50	8.050E-01	1.951E-01	1.000E+00
	29.00	8.064E-01	1.937E-01	1.000E+00
	29.50	8.072E-01	1.929E-01	1.000E+00
	30.00	8.075E-01	1.926E-01	1.000E+00



30-in-EBC with 6 Elements,  $\theta=0$  at Midspan, 1k-in Torque at Far End



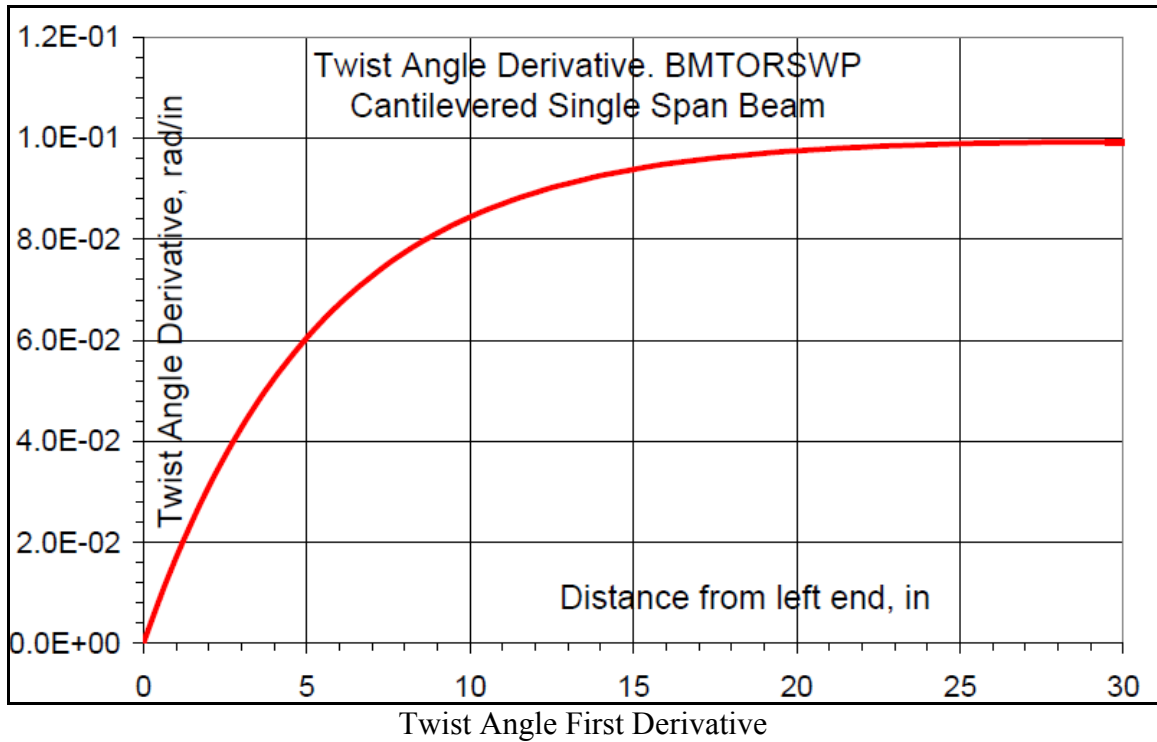
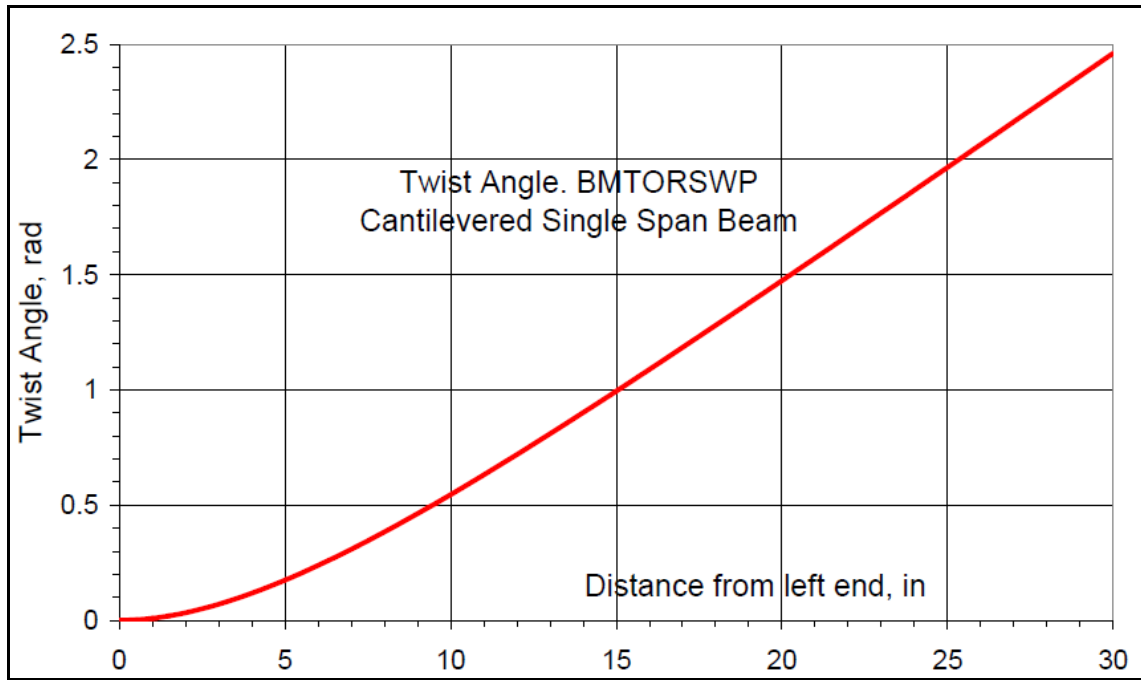
30-in-TWB with 6 Elements,  $\theta=0$  at Midspan, 1k-in Torque at Far End

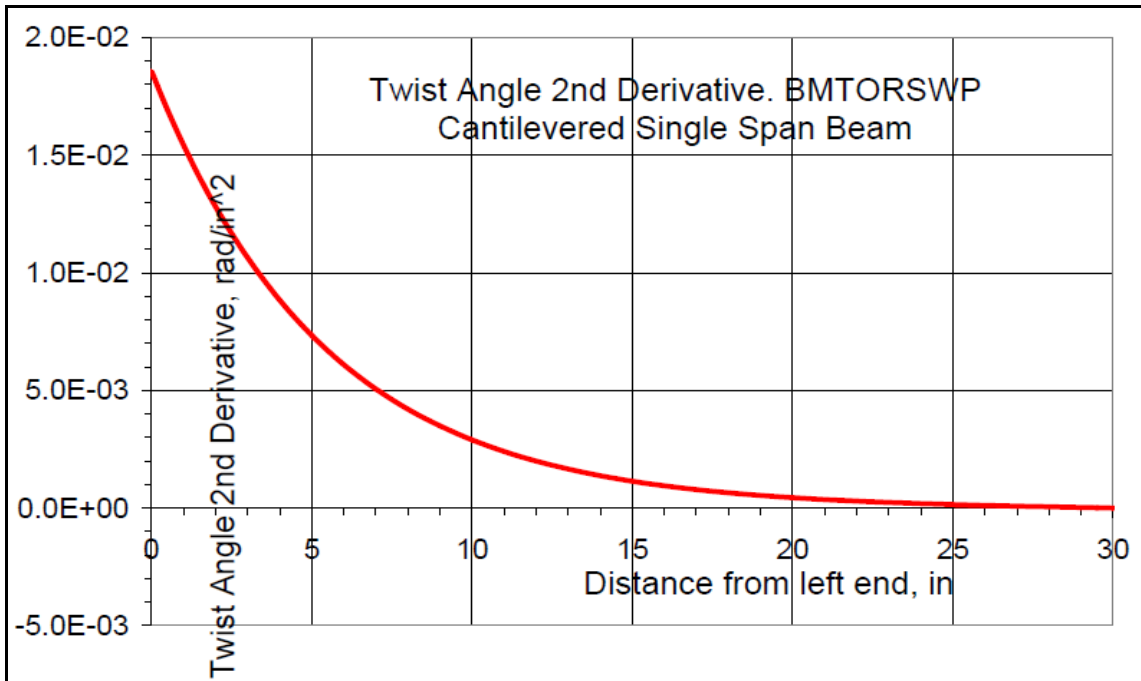


Bimoment 30-in-TWB with 6 Elements,  $\theta=0$  at Midspan, 1K-in at Far End

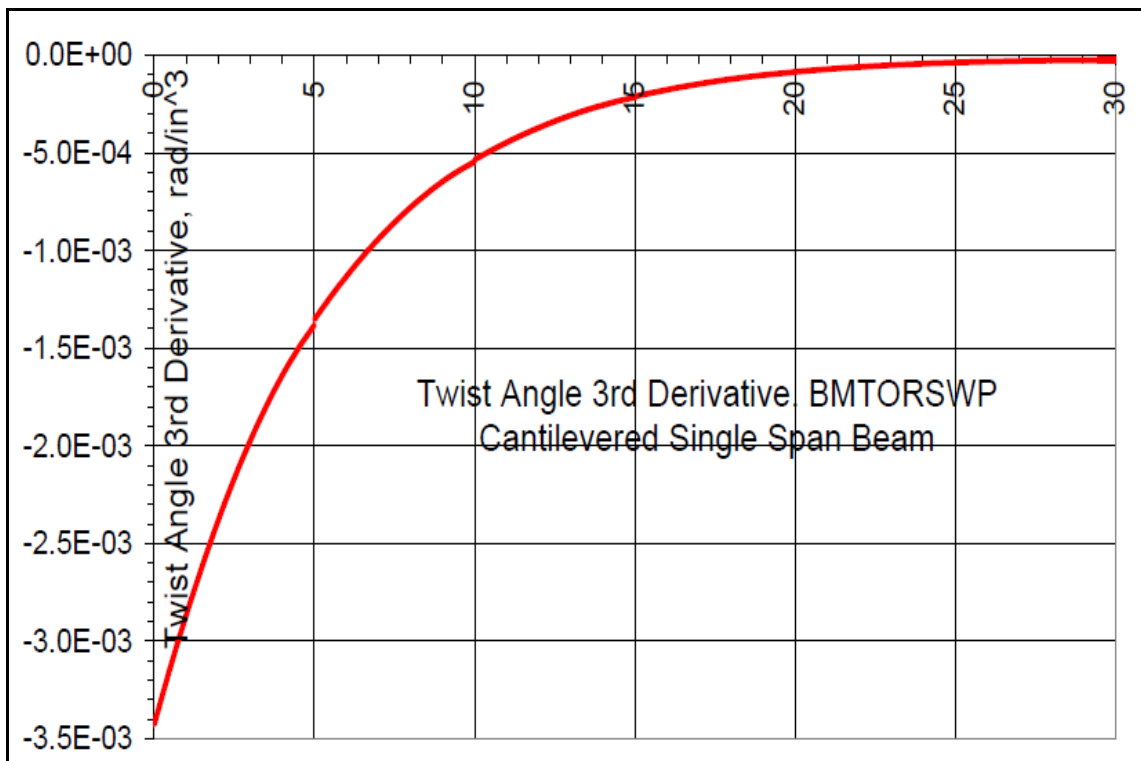
## APPENDIX C

### CANTILEVERED SINGLE SPAN BAR



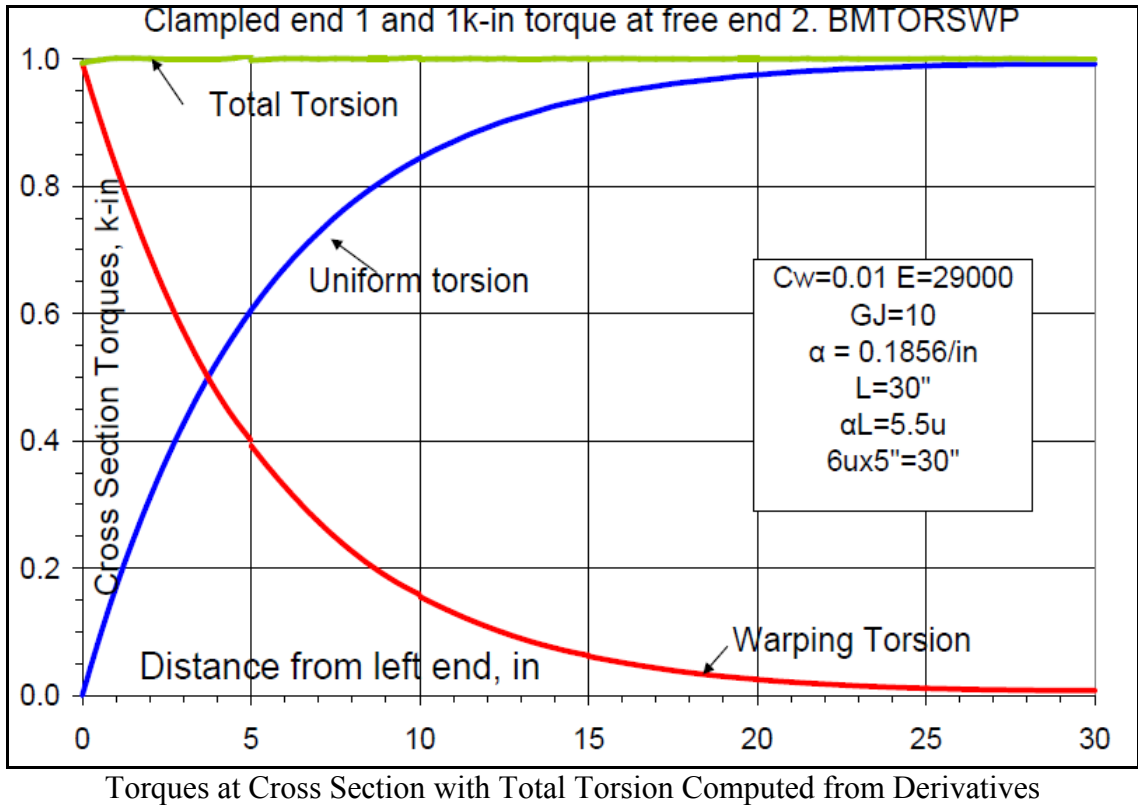
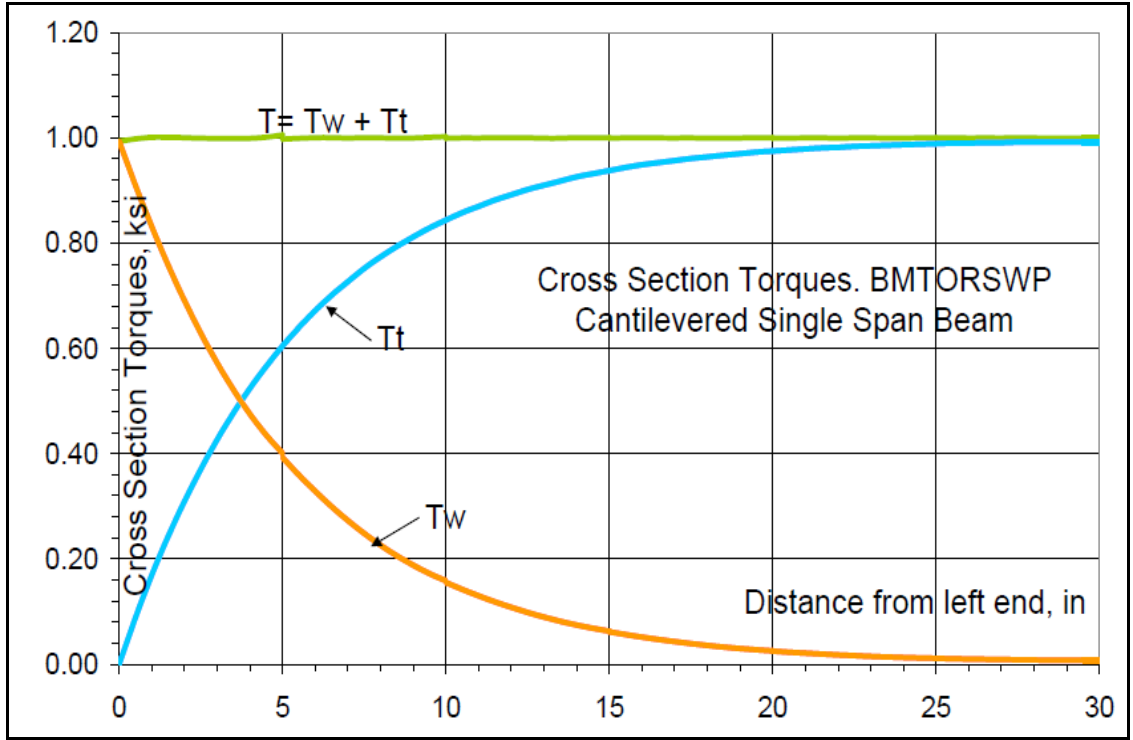


Twist Angle Second Derivative



Twist Angle Third Derivative





XS data

GJ=	10		L=	3.000E+01
E=	29000	T=	1k-in	$\lambda$ 1.857E-01
Cw=	0.01		$\lambda$ L=	5.571E+00

Single Span Cantilevered Bar under 1k-in-Torque at Far End

elm	Z	$\Phi$	$\Phi'$	$\Phi''$	$\Phi'''$	GJ $\Phi'$	ECw $\Phi'''$	T
1								
	0	0	0.00E+00	1.86E-02	-3.43E-03	0.000E+00	9.935E-01	0.994
	0.5	0.0023	8.87E-03	1.69E-02	-3.14E-03	8.866E-02	9.094E-01	0.998
	1	0.0087	1.70E-02	1.54E-02	-2.87E-03	1.695E-01	8.311E-01	1.001
	1.5	0.0191	2.43E-02	1.41E-02	-2.61E-03	2.431E-01	7.581E-01	1.001
	2	0.0329	3.10E-02	1.28E-02	-2.38E-03	3.102E-01	6.905E-01	1.001
	2.5	0.05	3.71E-02	1.17E-02	-2.17E-03	3.714E-01	6.284E-01	1.000
	3	0.07	4.27E-02	1.06E-02	-1.97E-03	4.271E-01	5.722E-01	0.999
	3.5	0.0926	4.78E-02	9.70E-03	-1.80E-03	4.779E-01	5.211E-01	0.999
	4	0.1177	5.24E-02	8.84E-03	-1.64E-03	5.242E-01	4.753E-01	1.000
	4.5	0.145	5.66E-02	8.05E-03	-1.50E-03	5.664E-01	4.353E-01	1.002
	5	0.1743	6.05E-02	7.33E-03	-1.38E-03	6.048E-01	4.008E-01	1.006
2								
	5	0.1743	6.05E-02	7.34E-03	-1.35E-03	6.048E-01	3.927E-01	0.997
	5.5	0.2054	6.40E-02	6.69E-03	-1.24E-03	6.398E-01	3.593E-01	0.999
	6	0.2382	6.72E-02	6.09E-03	-1.13E-03	6.718E-01	3.286E-01	1.000
	6.5	0.2725	7.01E-02	5.55E-03	-1.03E-03	7.009E-01	2.996E-01	1.000
	7	0.3083	7.27E-02	5.06E-03	-9.41E-04	7.274E-01	2.729E-01	1.000
	7.5	0.3452	7.52E-02	4.61E-03	-8.57E-04	7.515E-01	2.484E-01	1.000
	8	0.3834	7.74E-02	4.20E-03	-7.80E-04	7.736E-01	2.261E-01	1.000
	8.5	0.4226	7.94E-02	3.83E-03	-7.10E-04	7.936E-01	2.059E-01	1.000
	9	0.4627	8.12E-02	3.49E-03	-6.48E-04	8.119E-01	1.879E-01	1.000
	9.5	0.5037	8.29E-02	3.18E-03	-5.94E-04	8.286E-01	1.721E-01	1.001
	10	0.5455	8.44E-02	2.90E-03	-5.46E-04	8.438E-01	1.585E-01	1.002
3								
	10	0.5455	8.44E-02	2.90E-03	-5.35E-04	8.438E-01	1.552E-01	0.999
	10.5	0.5881	8.58E-02	2.64E-03	-4.90E-04	8.576E-01	1.421E-01	1.000
	11	0.6313	8.70E-02	2.41E-03	-4.48E-04	8.702E-01	1.299E-01	1.000
	11.5	0.6751	8.82E-02	2.19E-03	-4.09E-04	8.817E-01	1.185E-01	1.000
	12	0.7194	8.92E-02	2.00E-03	-3.72E-04	8.922E-01	1.080E-01	1.000
	12.5	0.7643	9.02E-02	1.82E-03	-3.39E-04	9.017E-01	9.828E-02	1.000
	13	0.8096	9.10E-02	1.66E-03	-3.09E-04	9.104E-01	8.949E-02	1.000
	13.5	0.8553	9.18E-02	1.51E-03	-2.81E-04	9.183E-01	8.152E-02	1.000
	14	0.9014	9.26E-02	1.38E-03	-2.57E-04	9.255E-01	7.444E-02	1.000
	14.5	0.9478	9.32E-02	1.25E-03	-2.35E-04	9.321E-01	6.818E-02	1.000
	15	0.9946	9.38E-02	1.14E-03	-2.17E-04	9.381E-01	6.281E-02	1.001
4								

	15	0.9946	9.38E-02	1.14E-03	-2.12E-04	9.381E-01	6.154E-02	1.000
	15.5	1.0416	9.44E-02	1.04E-03	-1.94E-04	9.435E-01	5.638E-02	1.000
	16	1.0889	9.49E-02	9.47E-04	-1.78E-04	9.485E-01	5.156E-02	1.000
elm	Z	$\Phi$	$\Phi'$	$\Phi''$	$\Phi'''$	GJ $\Phi'$	ECw $\Phi'''$	T
	16.5	1.1365	9.53E-02	8.62E-04	-1.62E-04	9.530E-01	4.710E-02	1.000
	17	1.1842	9.57E-02	7.84E-04	-1.48E-04	9.571E-01	4.295E-02	1.000
	17.5	1.2322	9.61E-02	7.13E-04	-1.35E-04	9.608E-01	3.915E-02	1.000
	18	1.2803	9.64E-02	6.49E-04	-1.23E-04	9.642E-01	3.570E-02	1.000
	18.5	1.3286	9.67E-02	5.90E-04	-1.12E-04	9.673E-01	3.260E-02	1.000
	19	1.377	9.70E-02	5.36E-04	-1.03E-04	9.701E-01	2.984E-02	1.000
	19.5	1.4256	9.73E-02	4.87E-04	-9.45E-05	9.727E-01	2.740E-02	1.000
	20	1.4743	9.75E-02	4.41E-04	-8.73E-05	9.750E-01	2.532E-02	1.000
5								
	20	1.4743	9.75E-02	4.42E-04	-8.56E-05	9.750E-01	2.482E-02	1.000
	20.5	1.5231	9.77E-02	4.01E-04	-7.87E-05	9.771E-01	2.283E-02	1.000
	21	1.572	9.79E-02	3.63E-04	-7.23E-05	9.790E-01	2.098E-02	1.000
	21.5	1.621	9.81E-02	3.28E-04	-6.64E-05	9.808E-01	1.927E-02	1.000
	22	1.6701	9.82E-02	2.96E-04	-6.10E-05	9.823E-01	1.770E-02	1.000
	22.5	1.7192	9.84E-02	2.67E-04	-5.61E-05	9.837E-01	1.627E-02	1.000
	23	1.7685	9.85E-02	2.40E-04	-5.17E-05	9.850E-01	1.499E-02	1.000
	23.5	1.8177	9.86E-02	2.15E-04	-4.77E-05	9.861E-01	1.384E-02	1.000
	24	1.8671	9.87E-02	1.92E-04	-4.43E-05	9.871E-01	1.284E-02	1.000
	24.5	1.9165	9.88E-02	1.71E-04	-4.13E-05	9.881E-01	1.198E-02	1.000
	25	1.9659	9.89E-02	1.51E-04	-3.89E-05	9.889E-01	1.127E-02	1.000
6								
	25	1.9659	9.89E-02	1.51E-04	-3.83E-05	9.889E-01	1.110E-02	1.000
	25.5	2.0153	9.90E-02	1.32E-04	-3.59E-05	9.896E-01	1.042E-02	1.000
	26	2.0648	9.90E-02	1.15E-04	-3.39E-05	9.902E-01	9.817E-03	1.000
	26.5	2.1144	9.91E-02	9.85E-05	-3.20E-05	9.907E-01	9.286E-03	1.000
	27	2.1639	9.91E-02	8.29E-05	-3.05E-05	9.912E-01	8.831E-03	1.000
	27.5	2.2135	9.92E-02	6.80E-05	-2.91E-05	9.916E-01	8.448E-03	1.000
	28	2.2631	9.92E-02	5.37E-05	-2.81E-05	9.919E-01	8.140E-03	1.000
	28.5	2.3127	9.92E-02	3.99E-05	-2.73E-05	9.921E-01	7.905E-03	1.000
	29	2.3623	9.92E-02	2.64E-05	-2.67E-05	9.923E-01	7.746E-03	1.000
	29.5	2.4119	9.92E-02	1.32E-05	-2.64E-05	9.924E-01	7.656E-03	1.000
	30	2.4615	9.92E-02	-2.69E-08	-2.64E-05	9.924E-01	7.644E-03	1.000

The column of total torque was calculated from the twist angle derivatives with a negligible error. The computer produces a more exact output for the total torque but with opposite sign convention. The charts show how the asymptotic behavior of the twist

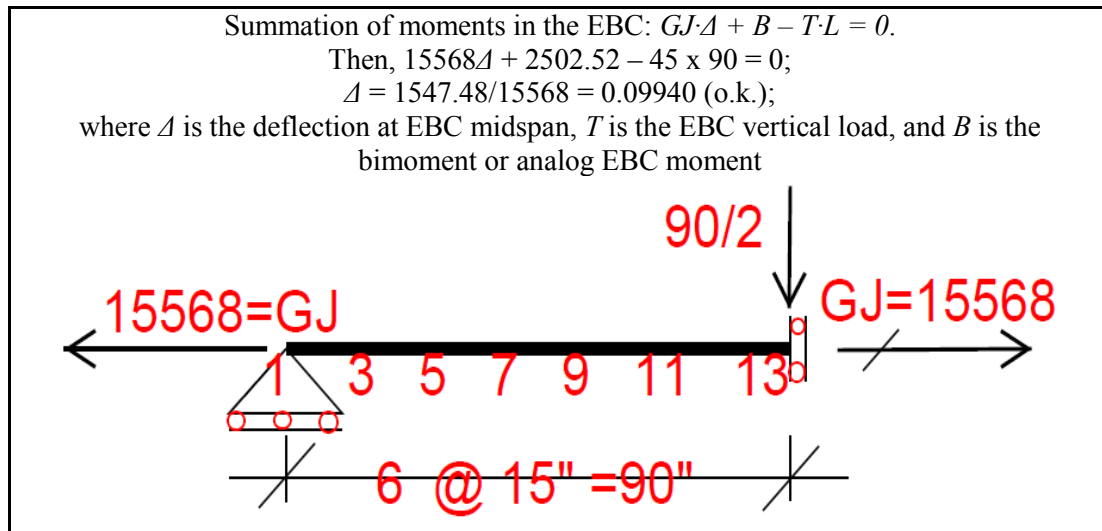
angle derivatives increases as the derivative order does. This is a very suitable feature of the finite element used by the software.

## APPENDIX D

### AISC-DG9-EXAMPLE 5.1

In this EBC of case study two, this material corresponds to the figure of the input model including its 2<sup>nd</sup> order equilibrium check, table of boundary conditions, stresses superposition strategy, input data, input forms, figures of output data in notepad version, and output charts and output data excel processed.

Normalized figures containing stresses computed by hand from the torque angle and derivatives, comparisons of combined flexure and torsional stresses, maximum twist angle for serviceability could be also found. Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP.



Node	TWB	EBC
A	No torque angle, $\theta = 0$	No transverse displacement $y = 0$
B	Maximum torsional rotation when $(d\theta/dx) = 0$	Maximum vertical transverse displacement when $(dy/dx) = 0$

Boundary Conditions for TWB and Analog EBC

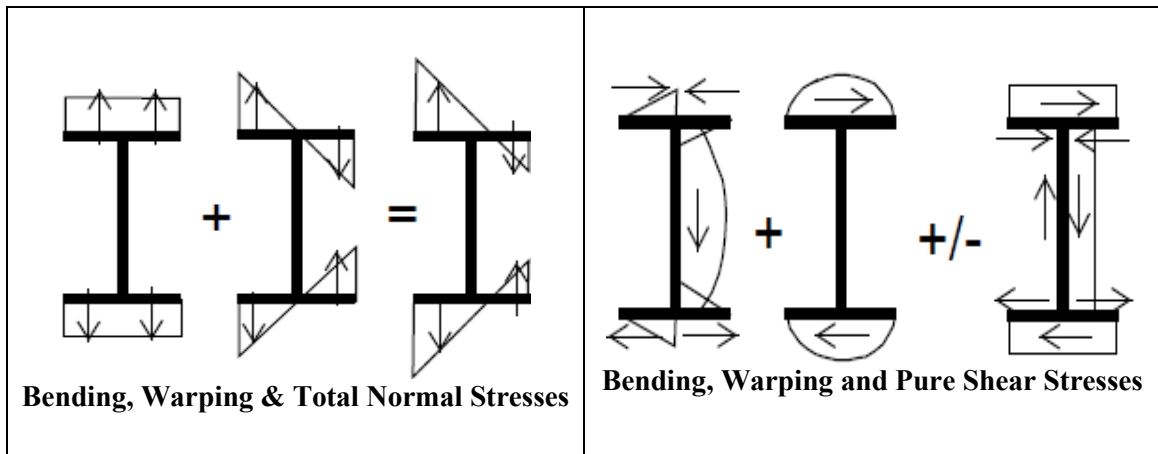
Restrained Warping Example 5.1. AISC-Design Guide 9. w10x49. T=90/2k-in

```

6 13 2 0 9 0 1 0 29000.
1 1 2 3 2070. 14.4 15. -15568
2 3 4 5 2070. 14.4 15. -15568
3 5 6 7 2070. 14.4 15. -15568
4 7 8 9 2070. 14.4 15. -15568
5 9 10 11 2070. 14.4 15. -15568
6 11 12 13 2070. 14.4 15. -15568
2 1 1 1 0 13 0 0 1 ← EBC rotational restraint
113 -45. ← Fixed end in the Equivalent Beam-Column

```

Notepad Input



Stresses Superposition Strategy

The figures containing the Stresses Superposition Strategy correspond to a sign convention based upon the different expected force directions for each of the interest points in the cross section. The AISC uses a positive convention for the stresses that correspond to a thin walled beam that undergoes a flexural deformation.

The asymptotic behavior of the thin-walled beam elastic line produces radical variations in zones near applied loads and supports. This behavior is successfully shown in the charts, due to the suitable high order finite element used by BMTORSWP.

PROGRAM																INPUT FORM															
BMTORSW																															
ALPHAMERIC DESCRIPTION OF THE JOB																FILE NAME:															
Design Guide Example 5.1 - w10x49 with 90 k-in torque at m																															
NEL	NOD	NSUP	NSPD	JBW	NFX	NFY	NFZ	ELASTICITY (E11.4)																							
6	13	2	0	9	0	1	0	29000.																							
NE	NI	NC	NJ	$C_w$			AREA	LENGTH	WTANG	SOIL NI				SOI																	
1	1	2	3	2070.			14.4	15.																							
2	3	4	5	2070.			14.4	15.																							
3	5	6	7	2070.			14.4	15.																							
4	7	8	9	2070.			14.4	15.																							
5	9	10	11	2070.			14.4	15.																							
6	11	12	13	2070.			14.4	15.																							
NF	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ											
2	1	1	1	1	13	0	0	1																							
NF	N	KD	SPRING				N	KD	SPRING				N	KD																	
NF	N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE													
1	13	-45.																													
NF	ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MOD																

Ex. 5.1 DG-9, Left Side Input Form

AUTHOR: B. DESCHAPELLES																																									
39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80
NAME:																																									
at mid span																																									
GENERAL INFORMATION																				PARTIAL RESTRAINT AT END ? 1 FOR YES 0 FOR NO																					
SOIL NJ		SOIL TI		SOIL TJ		ANGLE		- G*J				Wn @ NI		Wn @ NJ		1	2																								
								-15668.																																	
								S A M E																																	
								-15668.																																	
KY KZ N		KX KY KZ		N KX		KY KZ		N KX		KY KZ		SUPPORTS																													
SPRING				N KD		SPRING				SPRINGS AT SUPPORTS																															
N		LOAD VALUE		N		LOAD VALUE		N		LOAD VALUE		APPLIED NODAL FORCES																													
ROTATIONAL G MODULUS		ELEM		END		ROTATIONAL SPRING MODULUS		ELEM		END		ROTATIONAL SPRING MODULUS		SPRINGS AT ELEM. END, 1 OR 2, IF ANY																											
39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80

Ex. 5.1 DG-9, Right Side Input Form



510T

-----  
YOU ARE USING COMPUTER PROGRAM BMTORSW, DEVELOPED BY DR. BERNARDO DESCHAPELLES

INPUT DATA FILE NAME IS = 51.txt

OUTPUT FILE NAME IS = 51ot.txt

STORAGE FILE FOR POST-PROCESSING WITH EXCEL = 51grf.grf  
-----

Restrained Warping Example 5.1. AISC-Design Guide 9. W10x49. T=90/2k-in

modulus of elasticity of the material= 29000. k/ft<sup>2</sup>

ELEM	nodes	inertia	length	distrib.	load	AXIAL	SOIL	NORMAL	MODULUS,ksf	angle
	i	j	ft.4	ft	at i	at j	LOAD	1st END	2nd END	rad
1	1	3	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00
2	3	5	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00
3	5	7	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00
4	7	9	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00
5	9	11	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00
6	11	13	*****	15.00	0.000	0.000	*****	0.0	0.0	0.000 00

INPUT DATA RELATED TO THE 2 SUPPORTS

2 1 1 1 013 0 0 1

INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2

113 -45.00

FINAL SOLUTION FOUND AFTER 1 ITERATIONS

Output of nodal displacements in reference to global axes

node	displ. along x or nonn1	displ. along y or nonn2	displ. around z or nonn 3	node	displ. along x or nonn1	displ. along y or nonn2	displ. around z or nonn 3
1	0.0000E+00	0.0000E+00	-0.1604E-02	2	0.0000E+00	-0.1199E-01	-0.1991E-02
3	0.0000E+00	-0.2388E-01	-0.1567E-02	4	0.0000E+00	-0.3539E-01	-0.1896E-02
5	0.0000E+00	-0.4662E-01	-0.1451E-02	6	0.0000E+00	-0.5706E-01	-0.1701E-02
7	0.0000E+00	-0.6700E-01	-0.1252E-02	8	0.0000E+00	-0.7571E-01	-0.1393E-02
9	0.0000E+00	-0.8369E-01	-0.9559E-03	10	0.0000E+00	-0.8991E-01	-0.9546E-03
11	0.0000E+00	-0.9511E-01	-0.5467E-03	12	0.0000E+00	-0.9794E-01	-0.3606E-03
13	0.0000E+00	-0.9940E-01	0.0000E+00				

Noise disappears in the Excel File

Max. ↗

Max. ↗

-----  
OUTPUT OF SOIL REACTIONS, STRESSES AND TRANSVERSE DISPLACEMENTS  
-----

Page 1

Page 1, Output File

510T							
ELEMENT	1	DISPLACEMENTS IN INCIDENCES			1	2	3
NODE	1	0.00000E+00			0.00000E+00		-0.16045E-02
NODE	2	0.00000E+00			-0.11986E-01		-0.19915E-02
NODE	3	0.00000E+00			-0.23879E-01		-0.15667E-02
FORCES ACTING ALONG THE 9 DOF							
NODE	1	0.00000E+00			0.45000E+02		0.23919E-10
NODE	2	0.00000E+00			0.38613E-10		-0.89074E-10
NODE	3	0.00000E+00			-0.45000E+02		0.30326E+03
ELEMENT 1, FROM NODE 1, TO NODE 3 - LENGTH = 15.00 ft							
left half of span,at tenth points of length							
	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	45.00	45.00	45.00	45.00	45.00	45.00	45.00
bmom,kft	0.00	30.04	60.09	90.18	120.32	150.53	
tdisp,ft	0.00000	-0.00241	-0.00481	-0.00721	-0.00961	-0.01201	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		
right half of span,at tenth points of length							
	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	45.00	45.00	45.00	45.00	45.00	45.00	45.00
bmom,kft	150.53	180.83	211.23	241.76	272.43	303.26	
tdisp,ft	-0.01201	-0.01440	-0.01678	-0.01916	-0.02152	-0.02388	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		
-----							
ELEMENT	2	DISPLACEMENTS IN INCIDENCES			3	4	5
NODE	3	0.00000E+00			-0.23879E-01		-0.15667E-02
NODE	4	0.00000E+00			-0.35391E-01		-0.18964E-02
NODE	5	0.00000E+00			-0.46615E-01		-0.14514E-02
FORCES ACTING ALONG THE 9 DOF							
NODE	3	0.00000E+00			0.45000E+02		-0.30326E+03
NODE	4	0.00000E+00			-0.59101E-10		0.90580E-09
NODE	5	0.00000E+00			-0.45000E+02		0.62430E+03
ELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 15.00 ft							
left half of span,at tenth points of length							
	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	45.00	45.00	45.00	45.00	45.00	45.00	45.00
bmom,kft	303.26	334.26	365.46	396.87	428.52	460.41	
tdisp,ft	-0.02388	-0.02622	-0.02855	-0.03087	-0.03318	-0.03546	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		
right half of span,at tenth points of length							
	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	45.00	45.00	45.00	45.00	45.00	45.00	45.00
bmom,kft	460.41	492.58	525.03	557.79	590.87	624.30	
tdisp,ft	-0.03546	-0.03773	-0.03998	-0.04222	-0.04443	-0.04662	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		
-----							
ELEMENT	3	DISPLACEMENTS IN INCIDENCES			5	6	7
NODE	5	0.00000E+00			-0.46615E-01		-0.14514E-02
NODE	6	0.00000E+00			-0.57057E-01		-0.17006E-02

```

                    510T
NODE 7 0.00000E+00 -0.67000E-01 -0.12517E-02
FORCES ACTING ALONG THE 9 DOF
NODE 5 0.00000E+00 0.45000E+02 -0.62430E+03
NODE 6 0.00000E+00 -0.10010E-09 0.24890E-08
NODE 7 0.00000E+00 -0.45000E+02 0.98194E+03

ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 15.00 ft
left half of span,at tenth points of length
span span span span span span
0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 45.00 45.00 45.00 45.00 45.00 45.00
bmom,kft 624.30 658.09 692.26 726.84 761.85 797.30
tdisp,ft -0.04662 -0.04878 -0.05092 -0.05304 -0.05512 -0.05718
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 45.00 45.00 45.00 45.00 45.00 45.00
bmom,kft 797.30 833.21 869.61 906.52 943.95 981.94
tdisp,ft -0.05718 -0.05921 -0.06121 -0.06317 -0.06510 -0.06700
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 4 DISPLACEMENTS IN INCIDENCES 7 8 9
NODE 7 0.00000E+00 -0.67000E-01 -0.12517E-02
NODE 8 0.00000E+00 -0.75713E-01 -0.13927E-02
NODE 9 0.00000E+00 -0.83687E-01 -0.95591E-03
FORCES ACTING ALONG THE 9 DOF
NODE 7 0.00000E+00 0.45000E+02 -0.98194E+03
NODE 8 0.00000E+00 0.78843E-10 0.11085E-08
NODE 9 0.00000E+00 -0.45000E+02 0.13972E+04

ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 15.00 ft
left half of span,at tenth points of length
span span span span span span
0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 45.00 45.00 45.00 45.00 45.00 45.00
bmom,kft 981.94 1020.50 1059.66 1099.43 1139.85 1180.93
tdisp,ft -0.06700 -0.06886 -0.07068 -0.07246 -0.07420 -0.07590
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 45.00 45.00 45.00 45.00 45.00 45.00
bmom,kft 1180.93 1222.70 1265.18 1308.40 1352.39 1397.16
tdisp,ft -0.07590 -0.07755 -0.07916 -0.08072 -0.08223 -0.08369
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 5 DISPLACEMENTS IN INCIDENCES 9 10 11
NODE 9 0.00000E+00 -0.83687E-01 -0.95591E-03
NODE 10 0.00000E+00 -0.89910E-01 -0.95463E-03
NODE 11 0.00000E+00 -0.95112E-01 -0.54667E-03
FORCES ACTING ALONG THE 9 DOF
NODE 9 0.00000E+00 0.45000E+02 -0.13972E+04

```

```

                    510T
NODE 10          0.00000E+00      0.12910E-09      -0.23032E-08
NODE 11          0.00000E+00      -0.45000E+02      0.18943E+04

```

ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 15.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    45.00      45.00      45.00      45.00      45.00      45.00
bmom,kft  1397.16    1442.75    1489.18    1536.48    1584.68    1633.80
tdisp,ft  -0.08369    -0.08509    -0.08645    -0.08775    -0.08899    -0.09017
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    45.00      45.00      45.00      45.00      45.00      45.00
bmom,kft  1633.80    1683.88    1734.93    1787.00    1840.12    1894.30
tdisp,ft  -0.09017    -0.09128    -0.09234    -0.09333    -0.09426    -0.09511
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

```

```

-----
ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13
NODE 11          0.00000E+00      -0.95112E-01      -0.54667E-03
NODE 12          0.00000E+00      -0.97939E-01      -0.36065E-03
NODE 13          0.00000E+00      -0.99401E-01      0.00000E+00
FORCES ACTING ALONG THE 9 DOF
NODE 11          0.00000E+00      0.45000E+02      -0.18943E+04
NODE 12          0.00000E+00      -0.35084E-09      -0.22137E-08
NODE 13          0.00000E+00      -0.45000E+02      0.25025E+04

```

Max. Twist Angle



ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 15.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    45.00      45.00      45.00      45.00      45.00      45.00
bmom,kft  1894.30    1949.60    2006.03    2063.63    2122.43    2182.47
tdisp,ft  -0.09511    -0.09590    -0.09661    -0.09724    -0.09780    -0.09828
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    45.00      45.00      45.00      45.00      45.00      45.00
bmom,kft  2182.47    2243.79    2306.42    2370.39    2435.74    2502.52
tdisp,ft  -0.09828    -0.09868    -0.09899    -0.09922    -0.09935    -0.09940
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

```

3.- First entry row corresponds to the torque angle and the rest to each of its 3 dimensionless increasing derivatives

Notice that in the AISC calculations, the applied torque is negative with the sign convention used in AISC-DG 9:  $T_u = -90 \cdot \text{kip} \cdot \text{in}$

From AISC-DG 9, Appendix B, Case 3, with  $\alpha = 0.5$ , and  $L/a = 2.9$ :  
 **$\theta_m$  at midspan  $z/L=0.5$        $\theta_s$  at left support  $z/L=0$**

$$\theta_m := \frac{T_u}{G \cdot J} \cdot \begin{pmatrix} 0.09 \cdot L \\ 0 \cdot \text{in} \\ \frac{-0.44 \cdot \text{in}^2}{a} \\ \frac{-0.50 \cdot \text{in}^3}{a^2} \end{pmatrix} = \begin{pmatrix} -0.09 \\ 0 \\ 4.1 \times 10^{-5} \\ 7.5 \times 10^{-7} \end{pmatrix} \quad \theta_s := \frac{T_u}{G \cdot J} \cdot \begin{pmatrix} 0 \cdot L \\ 0.28 \text{in} \\ \frac{0 \cdot \text{in}^2}{a} \\ \frac{-0.22 \cdot \text{in}^3}{a^2} \end{pmatrix} = \begin{pmatrix} 0 \\ -1.62 \times 10^{-3} \\ 0 \\ 3.3 \times 10^{-7} \end{pmatrix}$$

Normalized Torque Angle and Derivatives (Increasing as per the Row Number) from AISC (Left) and BMTORSW

Shear stress in web (w) at the TWB support:

$$\tau_{t\_wsBmtrs} := G \cdot t_w \cdot \left( \theta_{sBmtrs_1} \cdot \frac{1}{\text{in}} \right) = -6.11 \cdot \text{ksi} \quad \text{From BMTORSW}$$

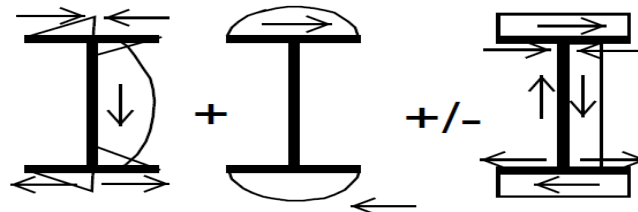
$$\tau_{t\_wsAisc} := G \cdot t_w \cdot \left( \theta_{sAisc_1} \cdot \frac{1}{\text{in}} \right) = -6.16 \cdot \text{ksi} \quad \text{From AISC-DG9}$$

Shear stress in flange (f) at the support (s):

$$\tau_{t\_fsBmtrs} := G \cdot t_f \cdot \left( \theta_{sBmtrs_1} \cdot \frac{1}{\text{in}} \right) = -10.06 \cdot \text{ksi} \quad \text{From BMTORSW}$$

$$\tau_{t\_fsAisc} := G \cdot t_f \cdot \left( \theta_{sAisc_1} \cdot \frac{1}{\text{in}} \right) = -10.15 \cdot \text{ksi} \quad \text{From AISC-DG9}$$

Shear Stress in Web and Flange at Midspan

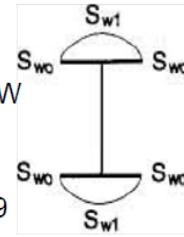


Two Solutions for Total Shear Stresses

Warping Shear Stress in flange at midspan (fm):

$$\tau_{w\_fmBmtrs} := -E \cdot S_{wl} \cdot \left( \theta_{mBmtrs_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -1.28 \cdot \text{ksi} \quad \text{BMTORSW}$$

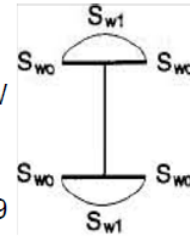
$$\tau_{w\_fmAisc} := -E \cdot S_{wl} \cdot \left( \theta_{mAisc_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -1.28 \cdot \text{ksi} \quad \text{AISC-DG9}$$



Warping Shear Stress in flange at the support (fs):

$$\tau_{w\_fsBmtrs} := -E \cdot S_{wl} \cdot \left( \theta_{sBmtrs_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -0.57 \cdot \text{ksi} \quad \text{BMTORSW}$$

$$\tau_{w\_fsAisc} := -E \cdot S_{wl} \cdot \left( \theta_{sAisc_3} \cdot \frac{1}{\text{in}^3} \right) \cdot \frac{1}{t_f} = -0.56 \cdot \text{ksi} \quad \text{AISC-DG9}$$

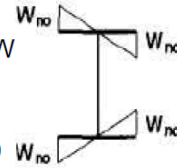


Shear Stress from Warping

Warping Normal Stresses at midspan due to warping

$$\sigma_{w\_mBmtrs} := E \cdot W_{no} \cdot \left( \theta_{mBmtrs_2} \cdot \frac{1}{\text{in}^2} \right) = 28.53 \cdot \text{ksi} \quad \text{BMTORSW}$$

$$\sigma_{w\_mAisc} := E \cdot W_{no} \cdot \left( \theta_{mAisc_2} \cdot \frac{1}{\text{in}^2} \right) = 28.04 \cdot \text{ksi} \quad \text{AISC-DG9}$$

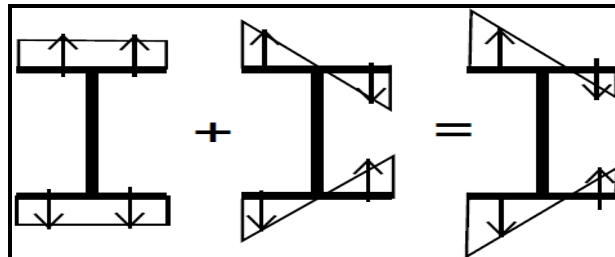


Warping Normal Stresses at the support  $\sigma_w = 0$ , since  $\theta''=0$

$$\sigma_{w\_sBmtrs} := E \cdot W_{no} \cdot \left( \theta_{sBmtrs_2} \cdot \frac{1}{\text{in}^2} \right) = -0.003 \cdot \text{psi} \quad \text{BMTORSW}$$

$$\sigma_{w\_sAisc} := E \cdot W_{no} \cdot \left( \theta_{sAisc_2} \cdot \frac{1}{\text{in}^2} \right) = 0 \cdot \text{psi} \quad \text{AISC-DG9}$$

Normal Stresses due to Warping

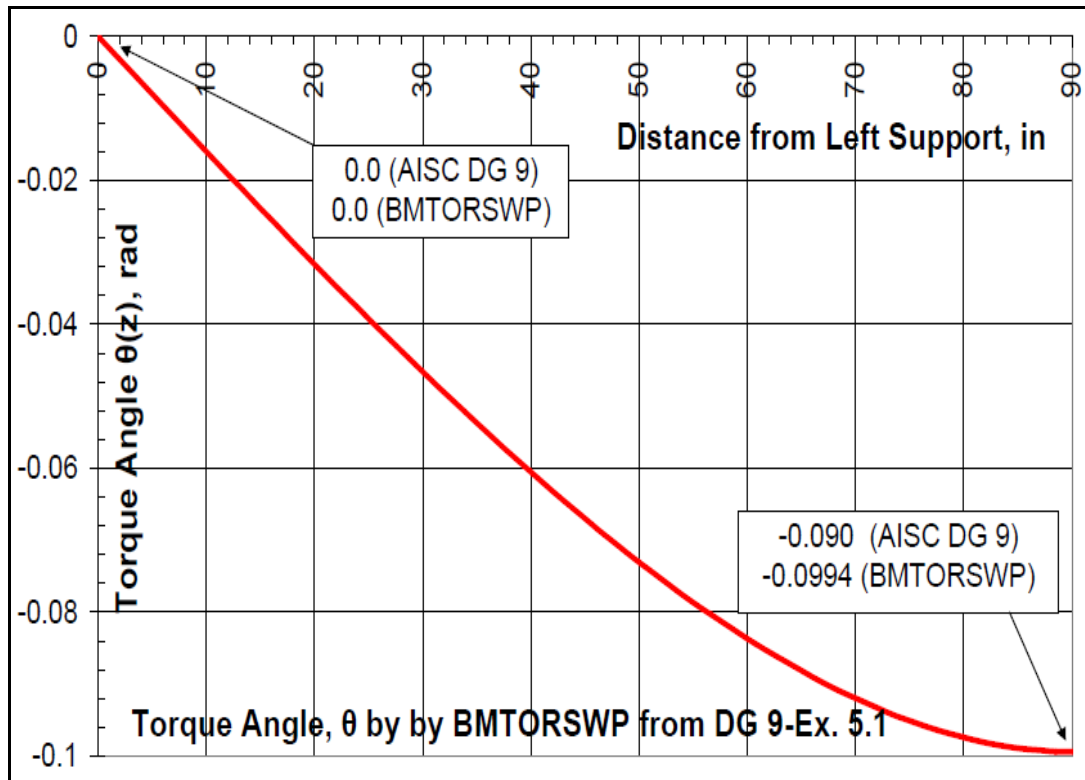


Superposition of Normal Stresses

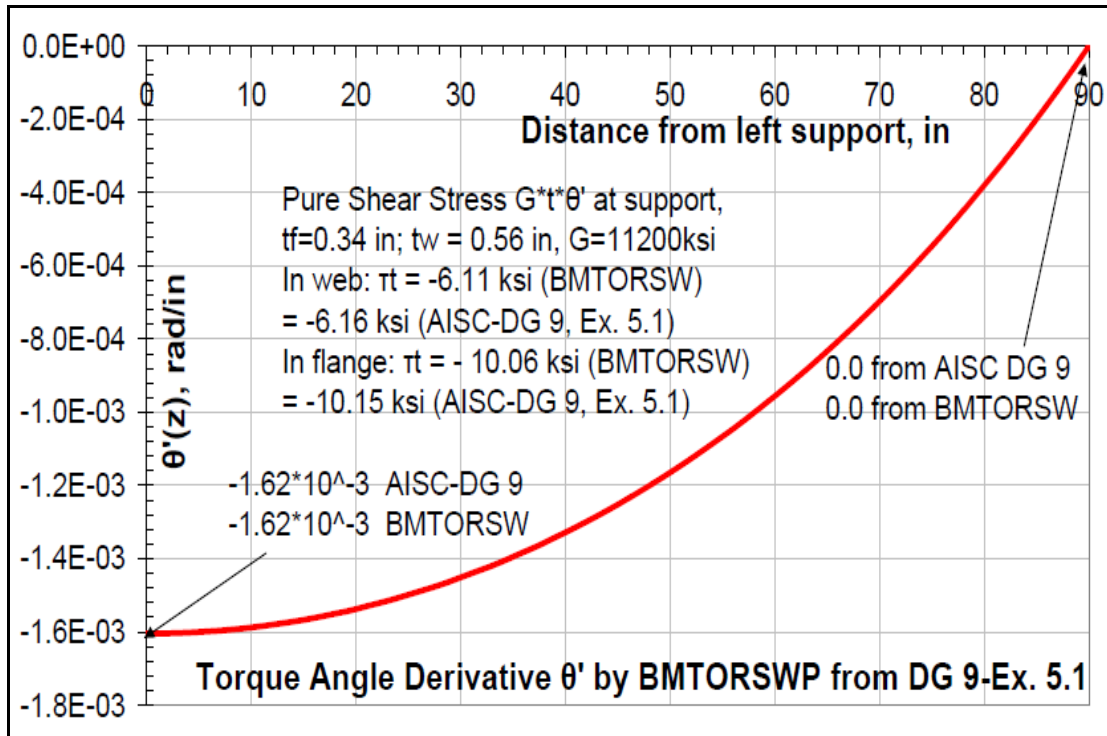
Location	Normal Stresses			Shear Stresses			
	$\sigma_{uw}$	$\sigma_{ub}$	$f_{un}$	$\tau_{ut}$	$\tau_{uw}$	$\tau_{ub}$	$f_{uv}$
Midspan	28.53	12.36	40.89				
Flange	±28.1	±12.4	±40.4	0	-1.28	±0.640	-1.92
Web	----	----	----	0	----	±2.45	-2.45
Support				10.06	0.57	0.64	11.27
Flange	0	0	0	-10.2	-0.564	±0.640	-11.4
Web	----	----	----	-6.16	----	±2.45	-8.61
Maximum		40.89	±40.4			11.27	-11.4

BMTORSWSP figures without sign, next to AISC DG9 figures

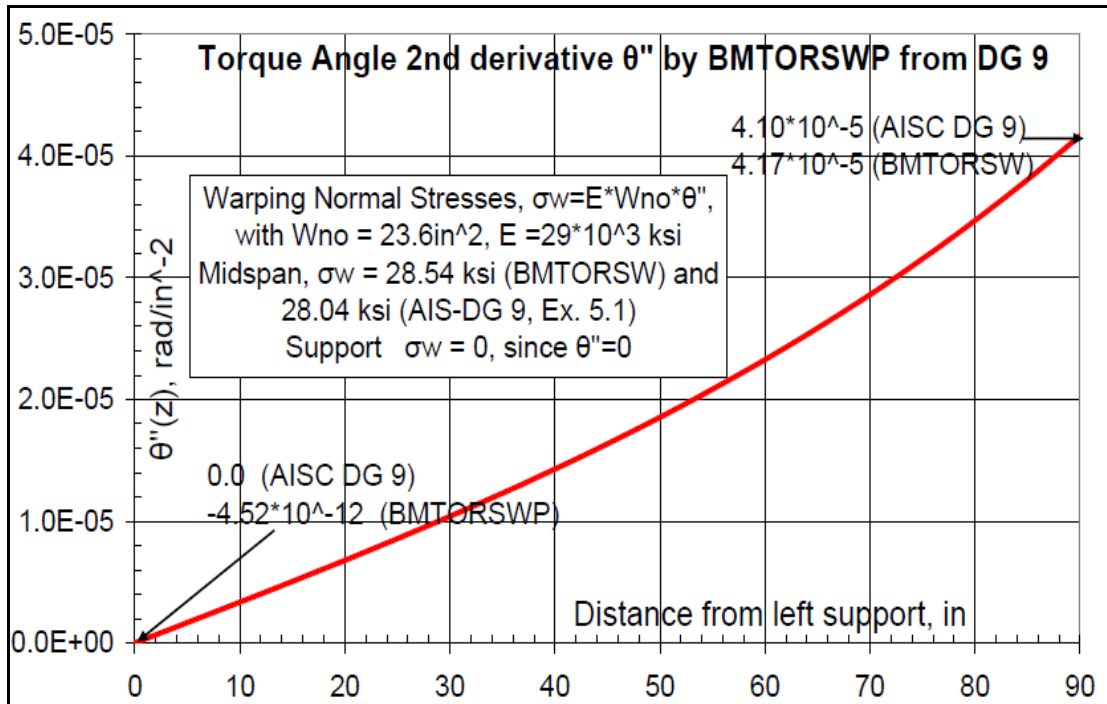
Comparisons Total of Flexure and Torsion Stresses



Torque Angle  $\theta$  of AISC-Design Guide 9, Ex. 5.1

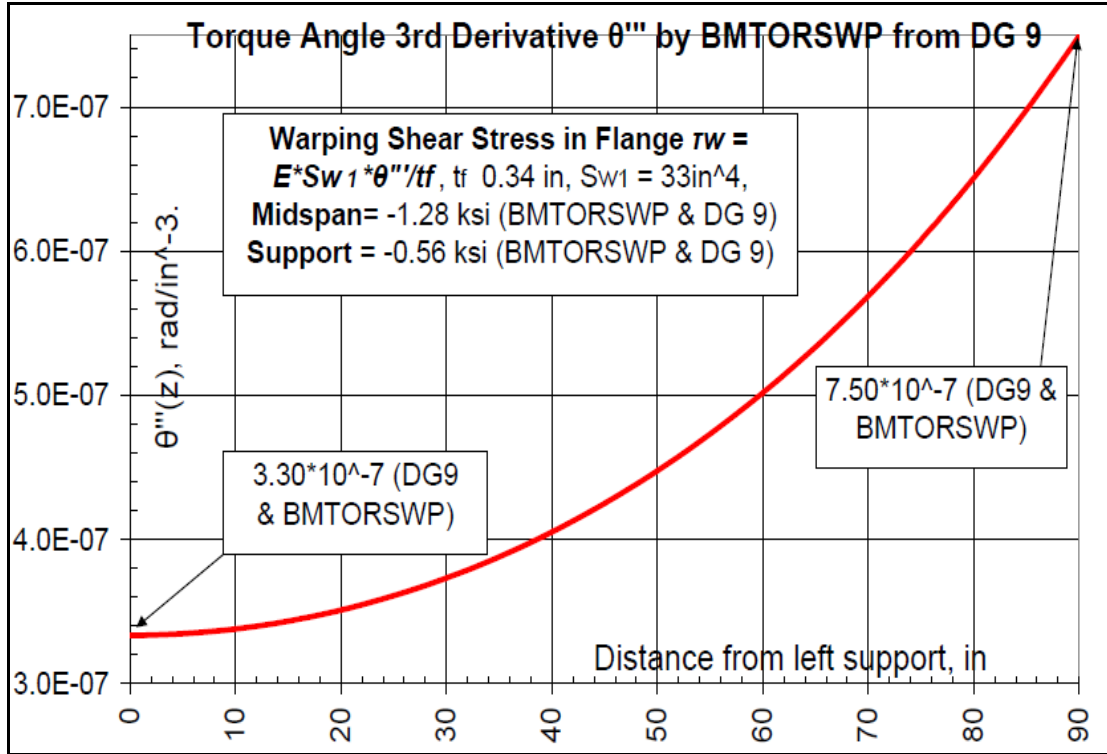


Torque Angle 1<sup>st</sup> Derivative  $\theta'$  of AISC-DG 9, Ex. 5.1

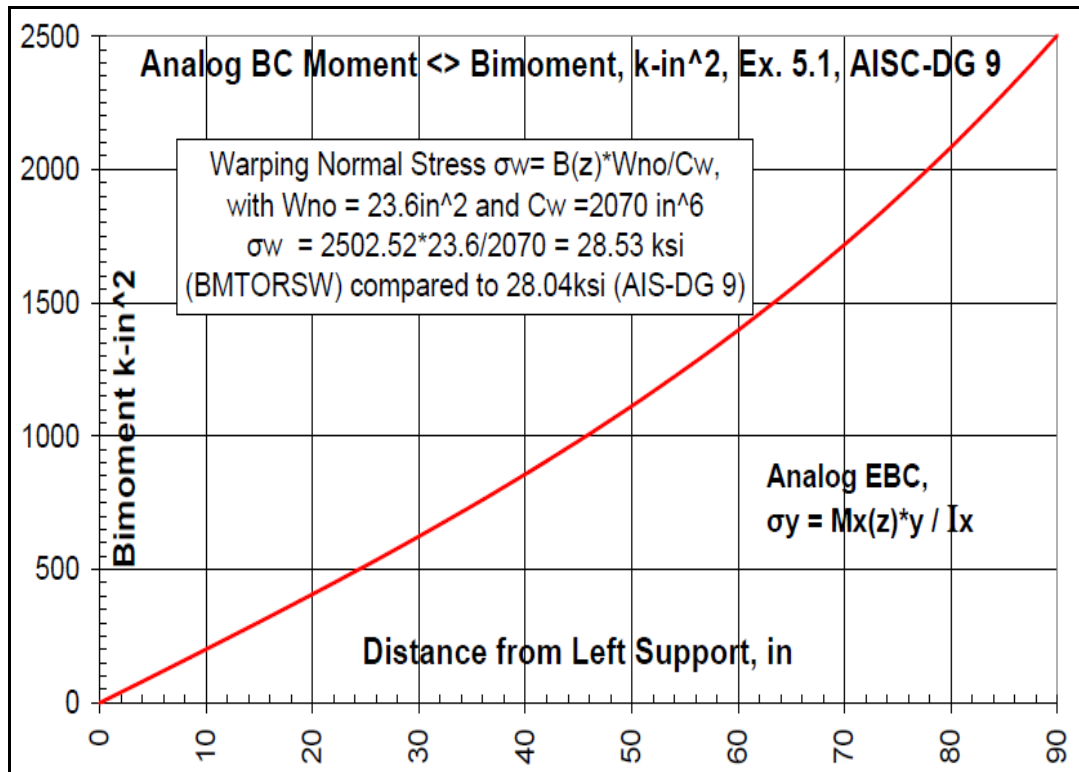


Torque Angle 2<sup>nd</sup> Derivative  $\theta''$  of AISC-DG 9, Ex. 5.1

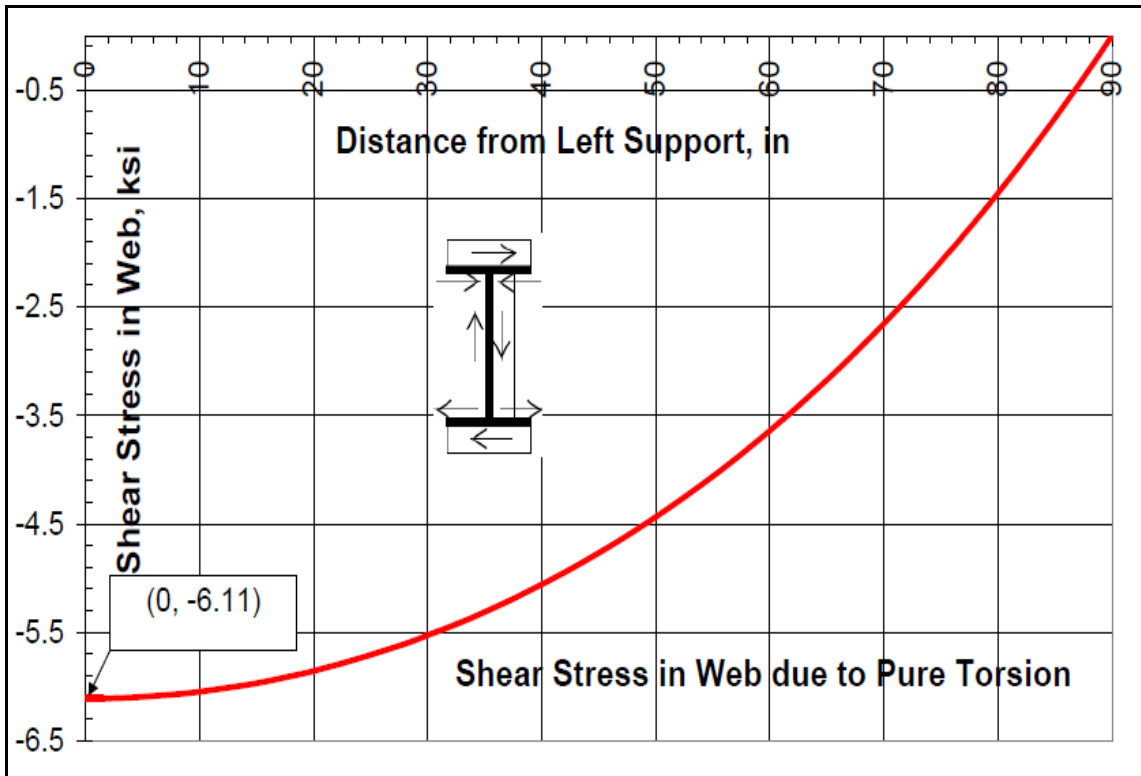




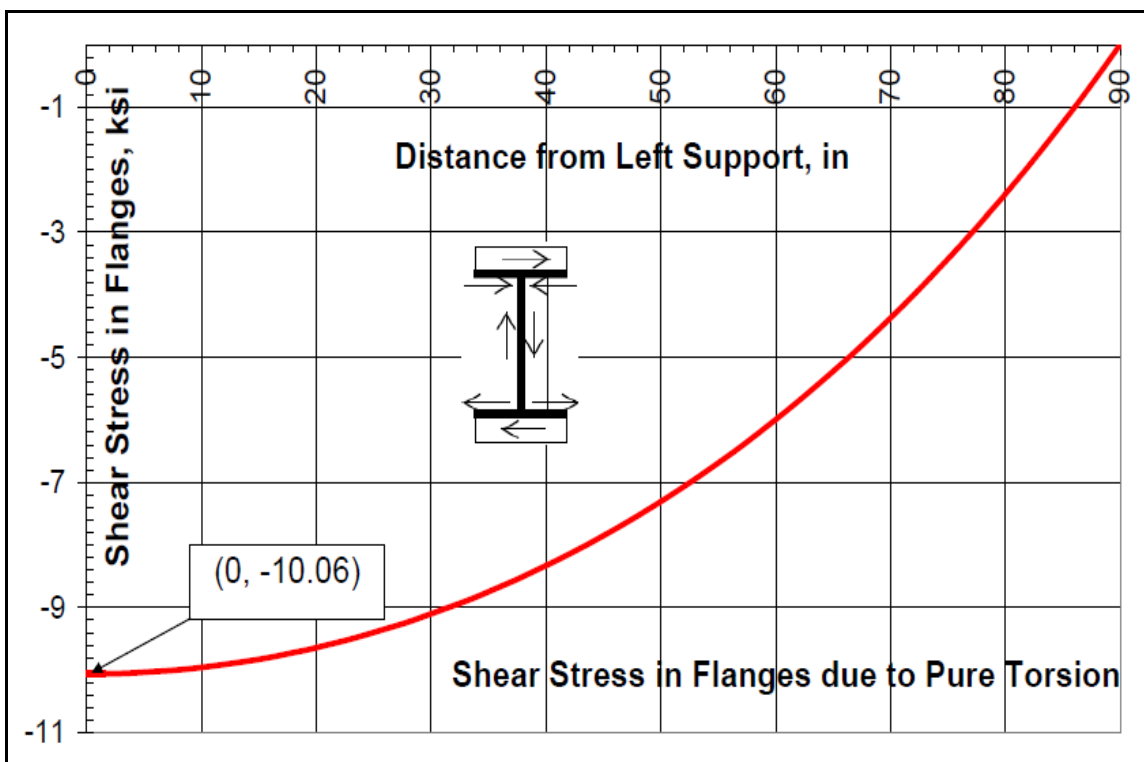
Torque Angle 3<sup>rd</sup> Derivative  $\theta'''$  of AISC-DG 9, Ex. 5.1



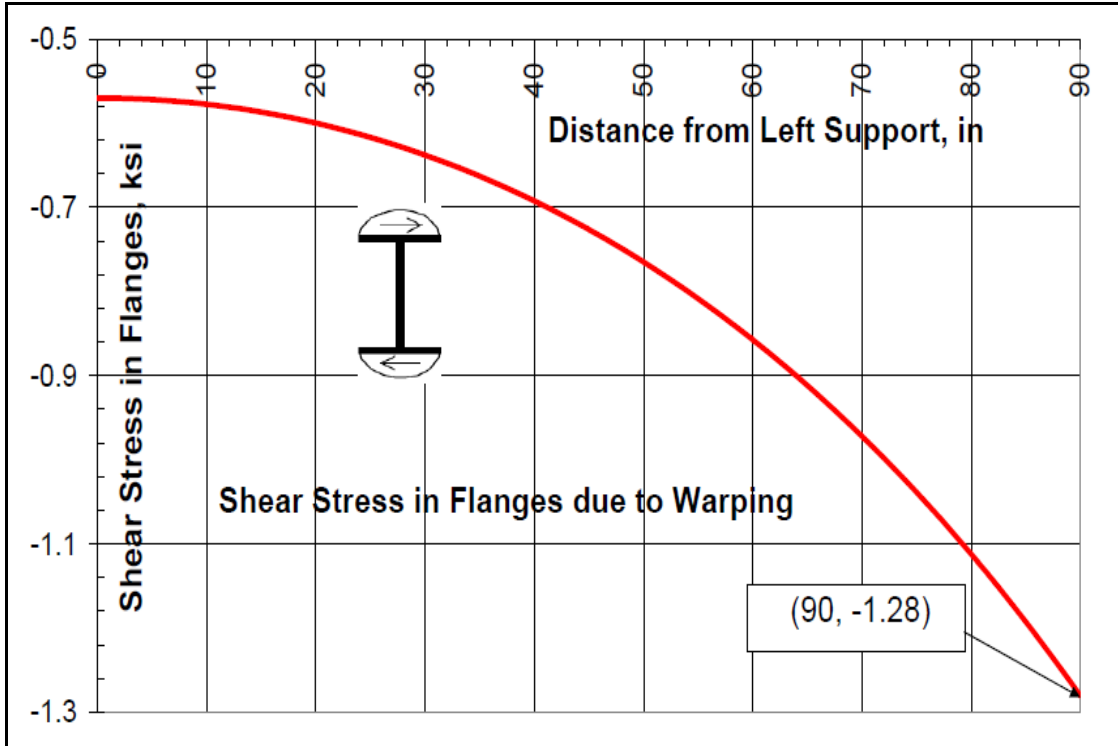
Bimoment from BMTORSWP of DG9-Ex. 5.1



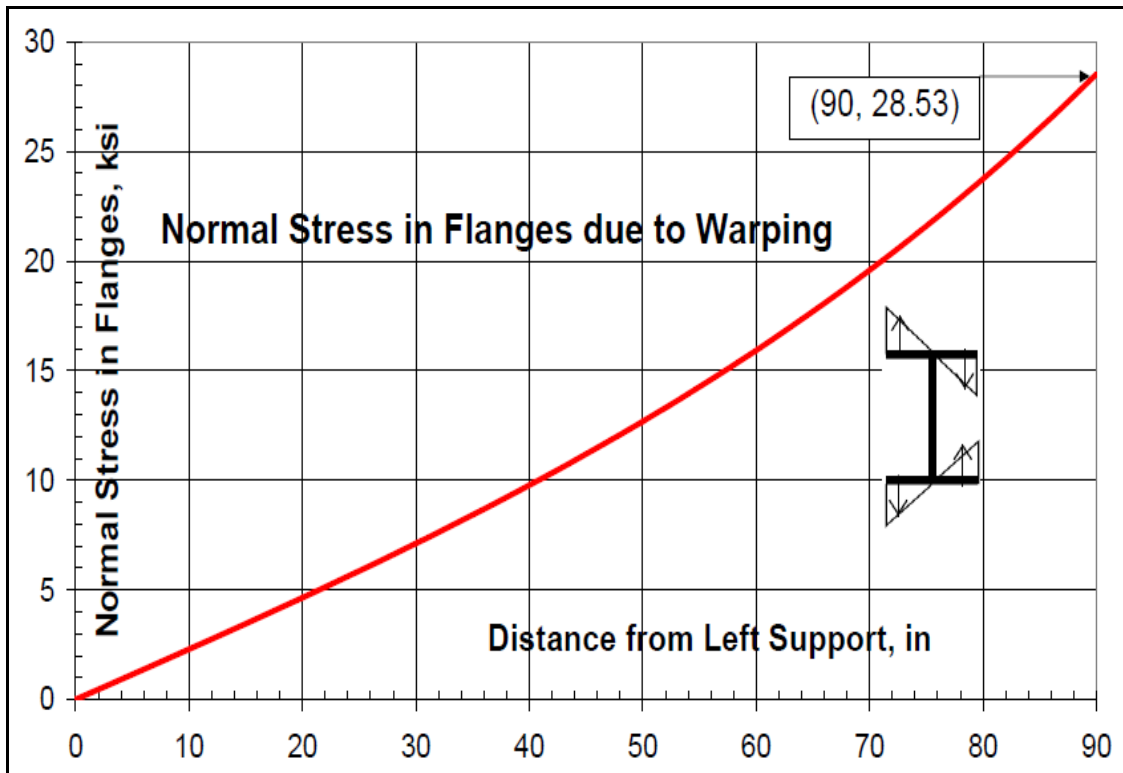
Shear Stress in Web Due to Pure Torsion



Shear Stress in Flanges Due to Pure Torsion



Shear Stress in Flanges Due to Warping



Normal Stress in Flanges Due to Warping

Table of Output for Graphics in Ex. 5.1

BMTORSW OUTPUT FOR GRAPHICS										
Elem	Z	$\theta$	Z	Moment	Z	$\theta'$	Z	$\theta''$	Z	$\theta'''$
1	0	0	0	0	0	-1.60E-03	0	-4.52E-12	0	3.34E-07
	1.5	-0.0024	1.5	30.0357	1.5	-1.60E-03	1.5	5.00E-07	1.5	3.34E-07
	3	-0.0048	3	60.0889	3	-1.60E-03	3	1.00E-06	3	3.34E-07
	4.5	-0.0072	4.5	90.1772	4.5	-1.60E-03	4.5	1.50E-06	4.5	3.34E-07
	6	-0.0096	6	120.3182	6	-1.60E-03	6	2.00E-06	6	3.35E-07
	7.5	-0.012	7.5	150.5293	7.5	-1.60E-03	7.5	2.51E-06	7.5	3.36E-07
	9	-0.0144	9	180.8283	9	-1.59E-03	9	3.01E-06	9	3.37E-07
	10.5	-0.0168	10.5	211.2328	10.5	-1.59E-03	10.5	3.52E-06	10.5	3.38E-07
	12	-0.0192	12	241.7605	12	-1.58E-03	12	4.03E-06	12	3.40E-07
	13.5	-0.0215	13.5	272.4294	13.5	-1.57E-03	13.5	4.54E-06	13.5	3.41E-07
	15	-0.0239	15	303.2572	15	-1.57E-03	15	5.05E-06	15	3.43E-07
2	15	-0.0239	15	303.2572	15	-1.57E-03	15	5.05E-06	15	3.43E-07
	16.5	-0.0262	16.5	334.2619	16.5	-1.56E-03	16.5	5.57E-06	16.5	3.45E-07
	18	-0.0286	18	365.4617	18	-1.55E-03	18	6.09E-06	18	3.48E-07
	19.5	-0.0309	19.5	396.8748	19.5	-1.54E-03	19.5	6.61E-06	19.5	3.50E-07
	21	-0.0332	21	428.5195	21	-1.53E-03	21	7.14E-06	21	3.53E-07
	22.5	-0.0355	22.5	460.4142	22.5	-1.52E-03	22.5	7.67E-06	22.5	3.56E-07
	24	-0.0377	24	492.5776	24	-1.51E-03	24	8.21E-06	24	3.59E-07
	25.5	-0.04	25.5	525.0284	25.5	-1.49E-03	25.5	8.75E-06	25.5	3.62E-07
	27	-0.0422	27	557.7856	27	-1.48E-03	27	9.29E-06	27	3.66E-07
	28.5	-0.0444	28.5	590.8683	28.5	-1.47E-03	28.5	9.84E-06	28.5	3.69E-07
	30	-0.0466	30	624.2958	30	-1.45E-03	30	1.04E-05	30	3.73E-07
3	30	-0.0466	30	624.2958	30	-1.45E-03	30	1.04E-05	30	3.73E-07
	31.5	-0.0488	31.5	658.0876	31.5	-1.44E-03	31.5	1.10E-05	31.5	3.77E-07
	33	-0.0509	33	692.2634	33	-1.42E-03	33	1.15E-05	33	3.82E-07
	34.5	-0.053	34.5	726.8432	34.5	-1.40E-03	34.5	1.21E-05	34.5	3.86E-07
	36	-0.0551	36	761.8471	36	-1.38E-03	36	1.27E-05	36	3.91E-07
	37.5	-0.0572	37.5	797.2956	37.5	-1.36E-03	37.5	1.33E-05	37.5	3.96E-07
	39	-0.0592	39	833.2093	39	-1.34E-03	39	1.39E-05	39	4.02E-07
	40.5	-0.0612	40.5	869.6092	40.5	-1.32E-03	40.5	1.45E-05	40.5	4.07E-07
	42	-0.0632	42	906.5166	42	-1.30E-03	42	1.51E-05	42	4.13E-07
	43.5	-0.0651	43.5	943.953	43.5	-1.28E-03	43.5	1.57E-05	43.5	4.19E-07
	45	-0.067	45	981.9402	45	-1.25E-03	45	1.64E-05	45	4.25E-07
4	45	-0.067	45	981.9402	45	-1.25E-03	45	1.64E-05	45	4.25E-07
	46.5	-0.0689	46.5	1020.5003	46.5	-1.23E-03	46.5	1.70E-05	46.5	4.32E-07
	48	-0.0707	48	1059.656	48	-1.20E-03	48	1.77E-05	48	4.38E-07
	49.5	-0.0725	49.5	1099.4301	49.5	-1.17E-03	49.5	1.83E-05	49.5	4.45E-07

(continued) Table of Output for Graphics in Ex. 5.1

	51	-0.0742	51	1139.8457	51	-1.15E-03	51	1.90E-05	51	4.53E-07
	52.5	-0.0759	52.5	1180.9264	52.5	-1.12E-03	52.5	1.97E-05	52.5	4.60E-07
	54	-0.0776	54	1222.6963	54	-1.09E-03	54	2.04E-05	54	4.68E-07
	55.5	-0.0792	55.5	1265.1796	55.5	-1.06E-03	55.5	2.11E-05	55.5	4.76E-07
	57	-0.0807	57	1308.4012	57	-1.02E-03	57	2.18E-05	57	4.84E-07
	58.5	-0.0822	58.5	1352.3864	58.5	-9.90E-04	58.5	2.25E-05	58.5	4.93E-07
	60	-0.0837	60	1397.1606	60	-9.56E-04	60	2.33E-05	60	5.02E-07
<b>5</b>										
	60	-0.0837	60	1397.1606	60	-9.56E-04	60	2.33E-05	60	5.02E-07
	61.5	-0.0851	61.5	1442.7502	61.5	-9.20E-04	61.5	2.40E-05	61.5	5.11E-07
	63	-0.0864	63	1489.1817	63	-8.84E-04	63	2.48E-05	63	5.20E-07
	64.5	-0.0877	64.5	1536.4822	64.5	-8.46E-04	64.5	2.56E-05	64.5	5.30E-07
	66	-0.089	66	1584.6793	66	-8.07E-04	66	2.64E-05	66	5.40E-07
	67.5	-0.0902	67.5	1633.801	67.5	-7.67E-04	67.5	2.72E-05	67.5	5.51E-07
	69	-0.0913	69	1683.8762	69	-7.25E-04	69	2.81E-05	69	5.62E-07
	70.5	-0.0923	70.5	1734.934	70.5	-6.83E-04	70.5	2.89E-05	70.5	5.73E-07
	72	-0.0933	72	1787.0041	72	-6.39E-04	72	2.98E-05	72	5.84E-07
	73.5	-0.0943	73.5	1840.1171	73.5	-5.93E-04	73.5	3.07E-05	73.5	5.96E-07
	75	-0.0951	75	1894.3038	75	-5.47E-04	75	3.16E-05	75	6.08E-07
<b>6</b>										
	75	-0.0951	75	1894.3038	75	-5.47E-04	75	3.16E-05	75	6.08E-07
	76.5	-0.0959	76.5	1949.5959	76.5	-4.99E-04	76.5	3.25E-05	76.5	6.20E-07
	78	-0.0966	78	2006.0257	78	-4.49E-04	78	3.34E-05	78	6.33E-07
	79.5	-0.0972	79.5	2063.6261	79.5	-3.98E-04	79.5	3.44E-05	79.5	6.46E-07
	81	-0.0978	81	2122.4307	81	-3.46E-04	81	3.54E-05	81	6.60E-07
	82.5	-0.0983	82.5	2182.4738	82.5	-2.92E-04	82.5	3.64E-05	82.5	6.74E-07
	84	-0.0987	84	2243.7904	84	-2.37E-04	84	3.74E-05	84	6.88E-07
	85.5	-0.099	85.5	2306.4164	85.5	-1.80E-04	85.5	3.84E-05	85.5	7.03E-07
	87	-0.0992	87	2370.3882	87	-1.22E-04	87	3.95E-05	87	7.18E-07
	88.5	-0.0994	88.5	2435.7433	88.5	-6.17E-05	88.5	4.06E-05	88.5	7.34E-07
	90	-0.0994	90	2502.5197	90	0.00E+00	90	4.17E-05	90	7.50E-07
<b>Elem</b>	<b>Z</b>	<b>θ</b>	<b>Z</b>	<b>Moment</b>	<b>Z</b>	<b>θ'</b>	<b>Z</b>	<b>θ''</b>	<b>Z</b>	<b>θ'''</b>

$\frac{\theta_{mAisc_0}}{LF} = -0.0624 \cdot \text{rad}$	Maximum rotation as per AISC
$\frac{\theta_{mBmtrs_0}}{LF} = -0.0663 \cdot \text{rad}$	Maximum rotation as per BMTORSW

Maximum Twist Angle

Table of Stresses. Ex. 5.5

PROPERTIES FOR SPREADSHEET										
			G = 11200 ksi							
			tw = 0.34 in							
			tf = 0.56 in							
			E = 29000 ksi							
			S.wl 33 in <sup>4</sup>							
			W.no 23.6 in <sup>2</sup>							
STRESSES COMPUTED FROM BMTORSW OUTPUT										
Elem	Z	θ	Pure torsion shear				Warping stresses			
			Z	web	Z	flange	Z	flange	Z	flange
1	0	0	0	-6.11	0	-10.06	0	0.000	0	-0.5699
	1.5	-0.0024	1.5	-6.11	1.5	-10.06	1.5	0.342	1.5	-0.5701
	3	-0.0048	3	-6.10	3	-10.05	3	0.685	3	-0.5706
	4.5	-0.0072	4.5	-6.10	4.5	-10.04	4.5	1.028	4.5	-0.5715
	6	-0.0096	6	-6.09	6	-10.02	6	1.372	6	-0.5727
	7.5	-0.012	7.5	-6.07	7.5	-10.00	7.5	1.716	7.5	-0.5742
	9	-0.0144	9	-6.06	9	-9.98	9	2.061	9	-0.5759
	10.5	-0.0168	10.5	-6.04	10.5	-9.95	10.5	2.408	10.5	-0.5781
	12	-0.0192	12	-6.02	12	-9.91	12	2.756	12	-0.5807
	13.5	-0.0215	13.5	-5.99	13.5	-9.87	13.5	3.106	13.5	-0.5834
	15	-0.0239	15	-5.97	15	-9.83	15	3.458	15	-0.5867
2				0.00		0.00		0.000		0.0000
	15	-0.0239	15	-5.97	15	-9.83	15	3.458	15	-0.5867
	16.5	-0.0262	16.5	-5.94	16.5	-9.78	16.5	3.811	16.5	-0.5903
	18	-0.0286	18	-5.90	18	-9.72	18	4.167	18	-0.5940
	19.5	-0.0309	19.5	-5.87	19.5	-9.67	19.5	4.525	19.5	-0.5983
	21	-0.0332	21	-5.83	21	-9.60	21	4.885	21	-0.6029
	22.5	-0.0355	22.5	-5.78	22.5	-9.53	22.5	5.249	22.5	-0.6079
	24	-0.0377	24	-5.74	24	-9.45	24	5.616	24	-0.6132
	25.5	-0.04	25.5	-5.69	25.5	-9.37	25.5	5.986	25.5	-0.6188
	27	-0.0422	27	-5.64	27	-9.29	27	6.359	27	-0.6248
	28.5	-0.0444	28.5	-5.59	28.5	-9.20	28.5	6.737	28.5	-0.6311
30	-0.0466	30	-5.53	30	-9.10	30	7.118	30	-0.6378	
3				0.00		0.00		0.000		0.0000
	30	-0.0466	30	-5.53	30	-9.10	30	7.118	30	-0.6379
	31.5	-0.0488	31.5	-5.46	31.5	-9.00	31.5	7.501	31.5	-0.6449
	33	-0.0509	33	-5.40	33	-8.90	33	7.891	33	-0.6523
	34.5	-0.053	34.5	-5.34	34.5	-8.79	34.5	8.288	34.5	-0.6602

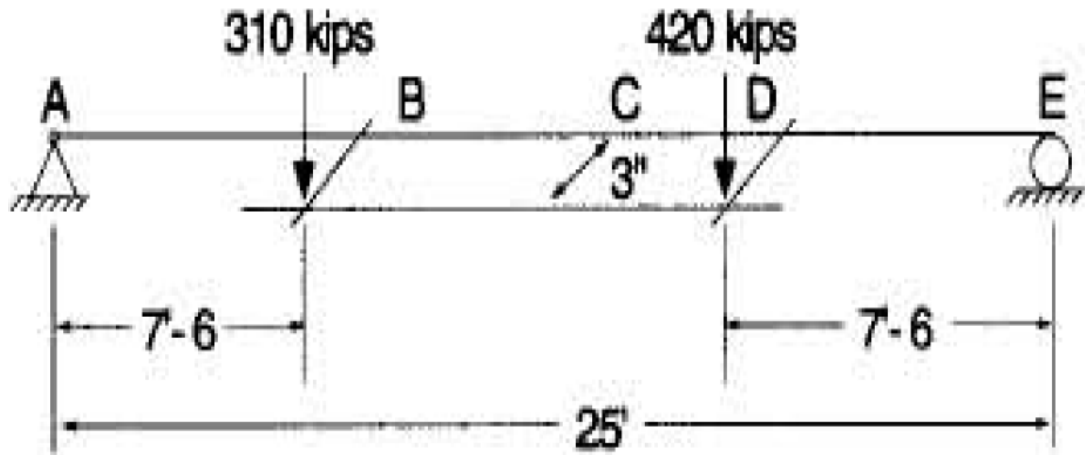
(continued) Table of Stresses. Ex. 5.5

4	36	-0.0551	36	-5.26	36	-8.67	36	8.685	36	-0.6685
	37.5	-0.0572	37.5	-5.19	37.5	-8.55	37.5	9.089	37.5	-0.6771
	39	-0.0592	39	-5.11	39	-8.42	39	9.499	39	-0.6861
	40.5	-0.0612	40.5	-5.03	40.5	-8.29	40.5	9.917	40.5	-0.6955
	42	-0.0632	42	-4.95	42	-8.15	42	10.334	42	-0.7054
	43.5	-0.0651	43.5	-4.86	43.5	-8.00	43.5	10.759	43.5	-0.7157
	45	-0.067	45	-4.77	45	-7.85	45	11.197	45	-0.7265
	46.5	-0.0689	46.5	-4.67	46.5	-7.70	46.5	11.635	46.5	-0.7374
	48	-0.0707	48	-4.57	48	-7.53	48	12.080	48	-0.7489
	49.5	-0.0725	49.5	-4.47	49.5	-7.36	49.5	12.531	49.5	-0.7608
	51	-0.0742	51	-4.36	51	-7.19	51	12.997	51	-0.7733
	52.5	-0.0759	52.5	-4.25	52.5	-7.01	52.5	13.462	52.5	-0.7861
	54	-0.0776	54	-4.14	54	-6.82	54	13.941	54	-0.7994
	55.5	-0.0792	55.5	-4.02	55.5	-6.62	55.5	14.427	55.5	-0.8133
	57	-0.0807	57	-3.90	57	-6.42	57	14.920	57	-0.8275
58.5	-0.0822	58.5	-3.77	58.5	-6.21	58.5	15.420	58.5	-0.8422	
60	-0.0837	60	-3.64	60	-6.00	60	15.926	60	-0.8574	
5				0.00		0.00		0.000		0.0000
	60	-0.0837	60	-3.64	60	-6.00	60	15.926	60	-0.8575
	61.5	-0.0851	61.5	-3.50	61.5	-5.77	61.5	16.446	61.5	-0.8731
	63	-0.0864	63	-3.37	63	-5.54	63	16.980	63	-0.8893
	64.5	-0.0877	64.5	-3.22	64.5	-5.31	64.5	17.521	64.5	-0.9061
	66	-0.089	66	-3.07	66	-5.06	66	18.068	66	-0.9233
	67.5	-0.0902	67.5	-2.92	67.5	-4.81	67.5	18.629	67.5	-0.9413
	69	-0.0913	69	-2.76	69	-4.55	69	19.197	69	-0.9596
	70.5	-0.0923	70.5	-2.60	70.5	-4.28	70.5	19.779	70.5	-0.9785
	72	-0.0933	72	-2.43	72	-4.01	72	20.375	72	-0.9980
73.5	-0.0943	73.5	-2.26	73.5	-3.72	73.5	20.977	73.5	-1.0180	
75	-0.0951	75	-2.08	75	-3.43	75	21.600	75	-1.0387	
6				0.00		0.00		0.000		0.0000
	75	-0.0951	75	-2.08	75	-3.43	75	21.600	75	-1.0389
	76.5	-0.0959	76.5	-1.90	76.5	-3.13	76.5	22.229	76.5	-1.0600
	78	-0.0966	78	-1.71	78	-2.82	78	22.873	78	-1.0819
	79.5	-0.0972	79.5	-1.52	79.5	-2.50	79.5	23.530	79.5	-1.1045
	81	-0.0978	81	-1.32	81	-2.17	81	24.200	81	-1.1277
	82.5	-0.0983	82.5	-1.11	82.5	-1.83	82.5	24.885	82.5	-1.1515
	84	-0.0987	84	-0.90	84	-1.49	84	25.583	84	-1.1761
	85.5	-0.099	85.5	-0.69	85.5	-1.13	85.5	26.295	85.5	-1.2012
	87	-0.0992	87	-0.46	87	-0.76	87	27.027	87	-1.2272
	88.5	-0.0994	88.5	-0.23	88.5	-0.39	88.5	27.773	88.5	-1.2537
90	-0.0994	90	0.00	90	0.00	90	28.533	90	-1.2808	

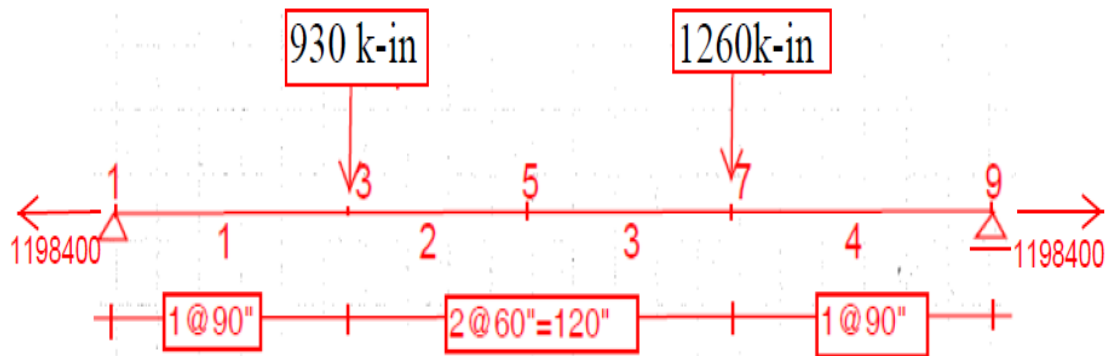
## APPENDIX E

### AISC-DG9-EXAMPLE 5.4

This material corresponds to the input model, input data, input forms, figures of output data with checks in notepad version, and excel processed output data and charts. An unsuccessful EBC model used before convergence study is presented. Charts are presented containing both partial and combined tresses along interest points of the beam and cross section profile. Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP.

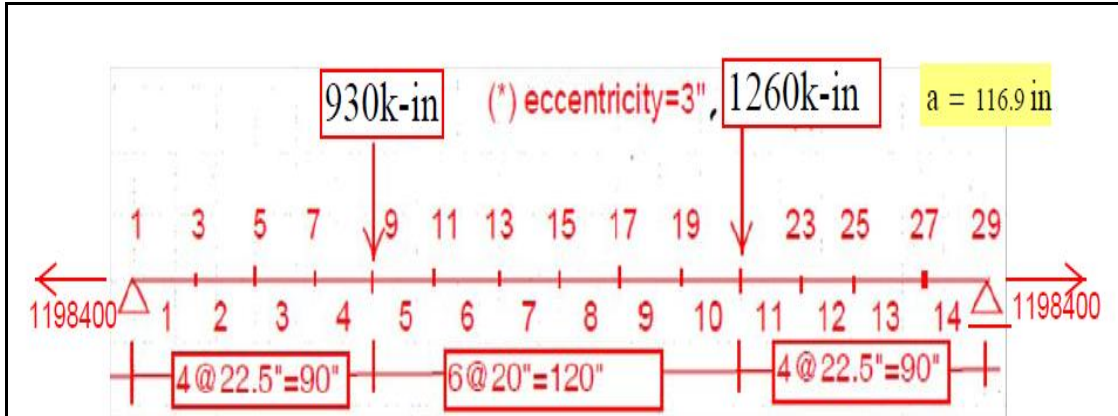


Original TWB from AISC DG9



Unsuccessful EBC Model Used Before Convergence Study





One Span between Consecutive Loads or Restraints and 3 FE in Spans

54b

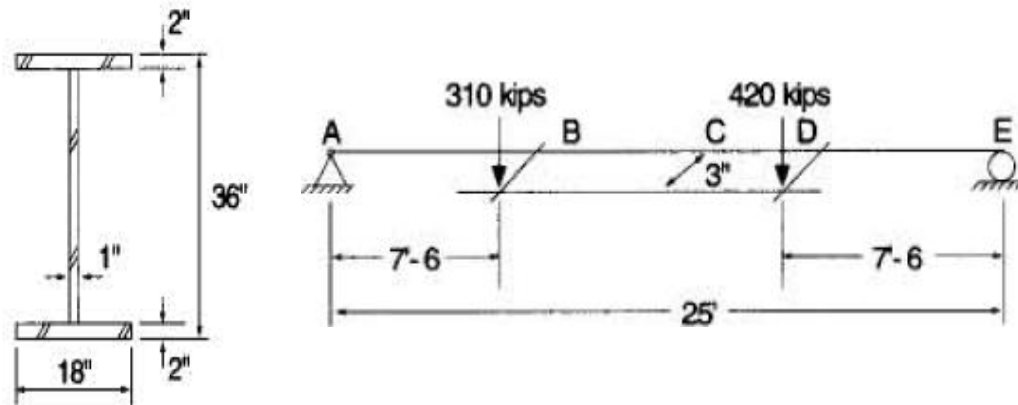
AISC Design Guide 9, Ex. 5.4

14	29	2	0	9	0	2	0	29000.		
1	1	2	3	5	6	4	0	0.	106.	22.5
2	3	4	5	5	6	4	0	0.	106.	22.5
3	5	6	7	5	6	4	0	0.	106.	22.5
4	7	8	9	5	6	4	0	0.	106.	22.5
5	9	10	11	5	6	4	0	0.	106.	20.0
6	11	12	13	5	6	4	0	0.	106.	20.0
7	13	14	15	5	6	4	0	0.	106.	20.0
8	15	16	17	5	6	4	0	0.	106.	20.0
9	17	18	19	5	6	4	0	0.	106.	20.0
10	19	20	21	5	6	4	0	0.	106.	20.0
11	21	22	23	5	6	4	0	0.	106.	22.5
12	23	24	25	5	6	4	0	0.	106.	22.5
13	25	26	27	5	6	4	0	0.	106.	22.5
14	27	28	29	5	6	4	0	0.	106.	22.5
2	1	1	1	0	29	0	1	0		
2	9	930.	21	1260.						

-GJ →

C<sub>w</sub>

Input Notepad, Ex. 5.4 DG9



TWB Cross Section and Interest Points along Beam

PROGRAM BMTORSW INPUT FORM																														
ALPHAMERIC DESCRIPTION OF THE JOB																													FILE NAME:	
AISC Design Guide 9, Ex. 5.4																														
NEL	NOD	NSUP	NSPD	JBW	NFX	NFY	NFZ	ELASTICITY (E11.4)																						
14	29	2	0	9	0	2	0	29 000.																						
NE	NI	NC	NJ	Cw				AREA	LENGTH	WTANG	SOIL NI			SOI																
1	1	2	3	5	6	4	0	0	10	6	2	2	5																	
2	3	4	5	SAME				SAME	2	2	5																			
3	5	6	7	SAME				SAME	2	2	5																			
4	7	8	9	SAME				SAME	2	2	5																			
5	9	10	11	5	6	4	0	0	10	6	2	0	0																	
6	11	12	13	5	6	4	0	0	10	6	2	0	0																	
7	13	14	15	5	6	4	0	0	10	6	2	0	0																	
8	15	16	17	5	6	4	0	0	10	6	2	0	0																	
9	17	18	19	5	6	4	0	0	10	6	2	0	0																	
10	19	20	21	5	6	4	0	0	10	6	2	0	0																	
11	21	22	23	5	6	4	0	0	10	6	2	2	5																	
12	23	24	25	5	6	4	0	0	10	6	2	2	5																	
13	25	26	27	5	6	4	0	0	10	6	2	2	5																	
14	27	28	29	5	6	4	0	0	10	6	2	2	5																	
NF	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ										
2	1	1	1	0	29	0	1	0																						
NF	N	KD	SPRING					N	KD	SPRING					N	KD														
NF	N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE												
2	9	93 0.			21	12 60.																								
NF	ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MOD															

Input Form, Left Side, Ex. 5.4 DG9

AUTHOR: B. DESCHAPELLES

40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

ME:

GENERAL INFORMATION

PARTIAL RESTRAINT  
AT END ? 1 FOR YES  
0 FOR NO

SOIL NJ	SOIL TI	SOIL TJ	ANGLE	- G*J				Wn @ NI	Wn @ NJ	1	2
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		
					-1	1	9	84	00		

Y	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	SUPPORTS							
---	----	---	----	----	----	---	----	----	----	---	----	----	----	----------	--	--	--	--	--	--	--

SPRING				N	KD	SPRING				SPRINGS AT SUPPORTS							
--------	--	--	--	---	----	--------	--	--	--	---------------------	--	--	--	--	--	--	--

N	LOAD VALUE				N	LOAD VALUE				N	LOAD VALUE				APPLIED NODAL FORCES							
---	------------	--	--	--	---	------------	--	--	--	---	------------	--	--	--	----------------------	--	--	--	--	--	--	--

ROTATIONAL MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS			SPRINGS AT ELEM. END, 1 OR 2, IF ANY			
--------------------	--	--	------	-----	---------------------------	--	--	------	-----	---------------------------	--	--	--------------------------------------	--	--	--

40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

Input Form, Right Side, Ex. 5.4 DG9

54OUT

-----  
YOU ARE USING COMPUTER PROGRAM BMTORSW, DEVELOPED BY DR. BERNARDO DESCHAPELLES

INPUT DATA FILE NAME IS = 54b.txt

OUTPUT FILE NAME IS = 54bout.txt

STORAGE FILE FOR POST-PROCESSING WITH EXCEL = 54bgrf.grf  
-----

AISC Design Guide 9, Ex. 5.4

modulus of elasticity of the material= 29000. k/ft<sup>2</sup>

ELEM	nodes	inertia	length	distrib.	load	AXIAL	SOIL NORMAL	MODULUS, Ksf	angle	
	i	j	ft.4	ft	at i	at j	LOAD	1st END	2nd END	rad
1	1	3	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
2	3	5	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
3	5	7	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
4	7	9	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
5	9	11	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
6	11	13	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
7	13	15	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
8	15	17	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
9	17	19	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
10	19	21	*****	20.00	0.000	0.000	*****	0.0	0.0	0.000 00
11	21	23	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
12	23	25	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
13	25	27	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00
14	27	29	*****	22.50	0.000	0.000	*****	0.0	0.0	0.000 00

INPUT DATA RELATED TO THE 2 SUPPORTS      KX=KY=KZ restrained at Node 1  
2 1 1 1 029 0 1 0      ←      KY restrained at Node 29

INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2  
2 9 930.00211260.00

FINAL SOLUTION FOUND AFTER 1 ITERATIONS

output of nodal displacements in reference to global axes

node	displ.	displ.	displ.	node	displ.	displ.	displ.
	along x	along y	around z		along x	along y	around z
	or nonn1	or nonn2	or nonn 3		or nonn1	or nonn2	or nonn 3
1	0.0000E+00	0.0000E+00	0.3696E-03	2	0.0000E+00	0.4140E-02	0.6878E-03
3	0.0000E+00	0.8247E-02	0.3605E-03	4	0.0000E+00	0.1222E-01	0.6535E-03
5	0.0000E+00	0.1608E-01	0.3328E-03	6	0.0000E+00	0.1967E-01	0.5836E-03
7	0.0000E+00	0.2308E-01	0.2856E-03	8	0.0000E+00	0.2605E-01	0.4755E-03
9	0.0000E+00	0.2877E-01	0.2171E-03	10	0.0000E+00	0.3071E-01	0.3035E-03
11	0.0000E+00	0.3242E-01	0.1476E-03				

Page 1

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```

54OUT
12 0.0000E+00 0.3367E-01 0.1896E-03
13 0.0000E+00 0.3469E-01 0.8011E-04
14 0.0000E+00 0.3527E-01 0.7735E-04
15 0.0000E+00 0.3562E-01 0.1250E-04
16 0.0000E+00 0.3552E-01 -0.3672E-04
17 0.0000E+00 0.3518E-01 -0.5717E-04
18 0.0000E+00 0.3437E-01 -0.1559E-03
19 0.0000E+00 0.3331E-01 -0.1310E-03
20 0.0000E+00 0.3174E-01 -0.2837E-03
21 0.0000E+00 0.2990E-01 -0.2110E-03
22 0.0000E+00 0.2720E-01 -0.4765E-03
23 0.0000E+00 0.2419E-01 -0.2919E-03
24 0.0000E+00 0.2068E-01 -0.6042E-03
25 0.0000E+00 0.1695E-01 -0.3477E-03
26 0.0000E+00 0.1290E-01 -0.6867E-03
27 0.0000E+00 0.8719E-02 -0.3803E-03
28 0.0000E+00 0.4380E-02 -0.7272E-03
29 0.0000E+00 0.0000E+00 -0.3911E-03

```

-----  
OUTPUT OF SOIL REACTIONS, STRESSES AND TRANSVERSE DISPLACEMENTS  
-----

```

ELEMENT 1 DISPLACEMENTS IN INCIDENCES 1 2 3
NODE 1 0.00000E+00 0.00000E+00 0.36955E-03
NODE 2 0.00000E+00 0.41404E-02 0.68780E-03
NODE 3 0.00000E+00 0.82468E-02 0.36045E-03
FORCES ACTING ALONG THE 9 DOF
NODE 1 0.00000E+00 -0.10290E+04 -0.14755E-08
NODE 2 0.00000E+00 0.36738E-09 0.11494E-07
NODE 3 0.00000E+00 0.10290E+04 -0.13270E+05

```

ELEMENT 1, FROM NODE 1, TO NODE 3 - LENGTH = 22.50 ft

left half of span, at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00
bmom, kft	0.00	-1318.87	-2638.23	-3958.57	-5280.37	-6604.14
tdisp, ft	0.00000	0.00083	0.00166	0.00249	0.00332	0.00415
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

right half of span, at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00
bmom, kft	-6604.14	-7930.35	-9259.51	-10592.10	-11928.62	-13269.57
tdisp, ft	0.00415	0.00497	0.00580	0.00662	0.00743	0.00825
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

-----  
ELEMENT 2 DISPLACEMENTS IN INCIDENCES 3 4 5  
(-Di + Dj)\* GJ + Vj\*L = Mj-Mi, where GJ=1198400 Page 2  
0.00825\*1198400+(-1029)\*22.5 = 9886.8 - 23152.25 = - 13265.7  
Mj-Mi = -13269.57-0 = -13269.6

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54OUT

NODE	3	0.00000E+00	0.82468E-02	0.36045E-03
NODE	4	0.00000E+00	0.12216E-01	0.65349E-03
NODE	5	0.00000E+00	0.16082E-01	0.33282E-03

FORCES ACTING ALONG THE 9 DOF

NODE	3	0.00000E+00	-0.10290E+04	0.13270E+05
NODE	4	0.00000E+00	-0.17447E-08	0.66930E-08
NODE	5	0.00000E+00	0.10290E+04	-0.27033E+05

ELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 22.50 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	
bmom,kft	-13269.57	-14615.43	-15966.72	-17323.93	-18687.57	-20058.14	
tdisp,ft	0.00825	0.00906	0.00986	0.01066	0.01145	0.01224	
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	
bmom,kft	-20058.14	-21436.15	-22822.11	-24216.54	-25619.95	-27032.86	
tdisp,ft	0.01224	0.01302	0.01380	0.01457	0.01533	0.01608	
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

-----

ELEMENT	3	DISPLACEMENTS IN INCIDENCES	5	6	7	
NODE	5	0.00000E+00	0.16082E-01			0.33282E-03
NODE	6	0.00000E+00	0.19668E-01			0.58359E-03
NODE	7	0.00000E+00	0.23077E-01			0.28562E-03

FORCES ACTING ALONG THE 9 DOF

NODE	5	0.00000E+00	-0.10290E+04	0.27033E+05
NODE	6	0.00000E+00	-0.72288E-09	-0.50648E-07
NODE	7	0.00000E+00	0.10290E+04	-0.41802E+05

ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 22.50 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	
bmom,kft	-27032.86	-28455.80	-29889.30	-31333.88	-32790.09	-34258.46	
tdisp,ft	0.01608	0.01683	0.01756	0.01829	0.01901	0.01971	
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	
bmom,kft	-34258.46	-35739.54	-37233.87	-38742.02	-40264.53	-41801.99	
tdisp,ft	0.01971	0.02041	0.02109	0.02177	0.02243	0.02308	
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

-----

ELEMENT	4	DISPLACEMENTS IN INCIDENCES	7	8	9	
NODE	7	0.00000E+00	0.23077E-01			0.28562E-03
NODE	8	0.00000E+00	0.26054E-01			0.47550E-03
NODE	9	0.00000E+00	0.28775E-01			0.21710E-03

(-Di + Dj)\* GJ + Vj\*L = Mj-Mi, where GJ=1198400  
 -0.01608+0.02308)1198400 -1029\*22.5= 8388.8-23152.5=14763.7  
 -41801.99+27032.86 =-14769.13 OK

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54OUT

FORCES ACTING ALONG THE 9 DOF

NODE	7	0.00000E+00	-0.10290E+04	0.41802E+05
NODE	8	0.00000E+00	-0.63679E-09	0.64662E-07
NODE	9	0.00000E+00	0.10290E+04	-0.58126E+05

ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 22.50 ft Warp, Torque Bimoment

left half of span, at tenth points of length

	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00
bmom, kft	-41801.99	-43354.95	-44923.99	-46509.69	-48112.65	-49733.45
tdisp, ft	0.02308	0.02371	0.02434	0.02494	0.02554	0.02612
axial, k	0.00	AT 1st END and			0.00	AT 2nd END

right half of span, at tenth points of length

	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00	-1029.00
bmom, kft	-49733.45	-51372.71	-53031.01	-54708.99	-56407.27	-58126.46
tdisp, ft	0.02612	0.02668	0.02723	0.02776	0.02828	0.02877
axial, k	0.00	AT 1st END and			0.00	AT 2nd END

---

ELEMENT 5 DISPLACEMENTS IN INCIDENCES 9 10 11

NODE	9	0.00000E+00	0.28775E-01	0.21710E-03
NODE	10	0.00000E+00	0.30712E-01	0.30347E-03
NODE	11	0.00000E+00	0.32417E-01	0.14765E-03

FORCES ACTING ALONG THE 9 DOF

NODE	9	0.00000E+00	-0.99000E+02	0.58126E+05
NODE	10	0.00000E+00	0.99470E-09	0.23257E-07
NODE	11	0.00000E+00	0.99000E+02	-0.55741E+05

ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 20.00 ft

left half of span, at tenth points of length

	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-99.00	-99.00	-99.00	-99.00	-99.00	-99.00
bmom, kft	-58126.46	-57812.63	-57515.74	-57235.70	-56972.45	-56725.89
tdisp, ft	0.02877	0.02920	0.02961	0.03001	0.03040	0.03077
axial, k	0.00	AT 1st END and			0.00	AT 2nd END

right half of span, at tenth points of length

	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-99.00	-99.00	-99.00	-99.00	-99.00	-99.00
bmom, kft	-56725.89	-56495.95	-56282.57	-56085.69	-55905.25	-55741.19
tdisp, ft	0.03077	0.03113	0.03147	0.03180	0.03212	0.03242
axial, k	0.00	AT 1st END and			0.00	AT 2nd END

---

ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13

NODE	11	0.00000E+00	0.32417E-01	0.14765E-03
NODE	12	0.00000E+00	0.33668E-01	0.18965E-03
NODE	13	0.00000E+00	0.34693E-01	0.80110E-04

FORCES ACTING ALONG THE 9 DOF

NODE	11	0.00000E+00	-0.99000E+02	0.55741E+05
NODE	12	0.00000E+00	0.72169E-08	-0.27496E-07

(-Di + Dj)\* GJ + Vj\*L = Mj-Mi, where GJ=1198400 Page 4

-0.02877+0.03242)\*1198400-99\*20=4374.16-1980=2394.16

-55741.19+58126.46=2385.27 OK

```

                    54OUT
NODE 13            0.00000E+00            0.99000E+02            -0.54994E+05
ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 20.00 ft
left half of span,at tenth points of length
span            span            span            span            span            span
0.0            0.1            0.2            0.3            0.4            0.5
soil,k/ft      0.000            0.000            0.000            0.000            0.000            0.000
shear,k        -99.00            -99.00            -99.00            -99.00            -99.00            -99.00
bmom,kft      -55741.19          -55593.47          -55462.04          -55346.87          -55247.92          -55165.16
tdisp,ft       0.03242            0.03271            0.03298            0.03324            0.03349            0.03372
axial,k        0.00 AT 1st END and            0.00 AT 2nd END

right half of span,at tenth points of length
span            span            span            span            span            span
0.5            0.6            0.7            0.8            0.9            1.0
soil,k/ft      0.000            0.000            0.000            0.000            0.000            0.000
shear,k        -99.00            -99.00            -99.00            -99.00            -99.00            -99.00
bmom,kft      -55165.16          -55098.57          -55048.13          -55013.82          -54995.63          -54993.57
tdisp,ft       0.03372            0.03394            0.03415            0.03435            0.03453            0.03469
axial,k        0.00 AT 1st END and            0.00 AT 2nd END
-----
ELEMENT 7 DISPLACEMENTS IN INCIDENCES 13 14 15 Max. Twist Angle
NODE 13            0.00000E+00            0.34693E-01            0.80110E-04
NODE 14            0.00000E+00            0.35270E-01            0.77350E-04
NODE 15            0.00000E+00            0.35621E-01            0.12498E-04
FORCES ACTING ALONG THE 9 DOF
NODE 13            0.00000E+00            -0.99000E+02            0.54994E+05
NODE 14            0.00000E+00            -0.81297E-08            0.37656E-07
NODE 15            0.00000E+00            0.99000E+02            -0.55862E+05
ELEMENT 7, FROM NODE 13, TO NODE 15 - LENGTH = 20.00 ft
left half of span,at tenth points of length
span            span            span            span            span            span
0.0            0.1            0.2            0.3            0.4            0.5
soil,k/ft      0.000            0.000            0.000            0.000            0.000            0.000
shear,k        -99.00            -99.00            -99.00            -99.00            -99.00            -99.00
bmom,kft      -54993.57          -55007.62          -55037.79          -55084.10          -55146.55          -55225.16
tdisp,ft       0.03469            0.03485            0.03499            0.03511            0.03523            0.03533
axial,k        0.00 AT 1st END and            0.00 AT 2nd END

right half of span,at tenth points of length
span            span            span            span            span            span
0.5            0.6            0.7            0.8            0.9            1.0
soil,k/ft      0.000            0.000            0.000            0.000            0.000            0.000
shear,k        -99.00            -99.00            -99.00            -99.00            -99.00            -99.00
bmom,kft      -55225.16          -55319.96          -55430.97          -55558.22          -55701.76          -55861.63
tdisp,ft       0.03533            0.03541            0.03548            0.03554            0.03559            0.03562
axial,k        0.00 AT 1st END and            0.00 AT 2nd END
-----
ELEMENT 8 DISPLACEMENTS IN INCIDENCES 15 16 17
NODE 15            0.00000E+00            0.35621E-01            0.12498E-04
NODE 16            0.00000E+00            0.35516E-01            -0.36718E-04
NODE 17            0.00000E+00            0.35180E-01            -0.57173E-04
FORCES ACTING ALONG THE 9 DOF
NODE 15            0.00000E+00            -0.99000E+02            0.55862E+05
NODE 16            0.00000E+00            -0.11674E-08            -0.73443E-07
NODE 17            0.00000E+00            0.99000E+02            -0.58371E+05
ELEMENT 8, FROM NODE 15, TO NODE 17 - LENGTH = 20.00 ft

```



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54OUT
left half of span,at tenth points of length
span span span span span span
soil,k/ft 0.0 0.1 0.2 0.3 0.4 0.5
shear,k -99.00 -99.00 -99.00 -99.00 -99.00 -99.00
bmom,kft -55861.63 -56037.87 -56230.53 -56439.67 -56665.36 -56907.65
tdisp,ft 0.03562 0.03564 0.03564 0.03563 0.03561 0.03557
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
soil,k/ft 0.5 0.6 0.7 0.8 0.9 1.0
shear,k -99.00 -99.00 -99.00 -99.00 -99.00 -99.00
bmom,kft -56907.65 -57166.62 -57442.34 -57734.91 -58044.39 -58370.88
tdisp,ft 0.03557 0.03552 0.03546 0.03538 0.03529 0.03518
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 9 DISPLACEMENTS IN INCIDENCES 17 18 19
NODE 17 0.00000E+00 0.35180E-01 -0.57173E-04
NODE 18 0.00000E+00 0.34366E-01 -0.15591E-03
NODE 19 0.00000E+00 0.33307E-01 -0.13095E-03
FORCES ACTING ALONG THE 9 DOF
NODE 17 0.00000E+00 -0.99000E+02 0.58371E+05
NODE 18 0.00000E+00 0.75810E-08 0.12084E-06
NODE 19 0.00000E+00 0.99000E+02 -0.62595E+05

ELEMENT 9, FROM NODE 17, TO NODE 19 - LENGTH = 20.00 ft

left half of span,at tenth points of length
span span span span span span
soil,k/ft 0.0 0.1 0.2 0.3 0.4 0.5
shear,k -99.00 -99.00 -99.00 -99.00 -99.00 -99.00
bmom,kft -58370.88 -58714.48 -59075.30 -59453.42 -59848.97 -60262.06
tdisp,ft 0.03518 0.03506 0.03492 0.03477 0.03461 0.03443
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
soil,k/ft 0.5 0.6 0.7 0.8 0.9 1.0
shear,k -99.00 -99.00 -99.00 -99.00 -99.00 -99.00
bmom,kft -60262.06 -60692.81 -61141.35 -61607.81 -62092.33 -62595.05
tdisp,ft 0.03443 0.03423 0.03402 0.03380 0.03356 0.03331
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 10 DISPLACEMENTS IN INCIDENCES 19 20 21
NODE 19 0.00000E+00 0.33307E-01 -0.13095E-03
NODE 20 0.00000E+00 0.31737E-01 -0.28373E-03
NODE 21 0.00000E+00 0.29900E-01 -0.21100E-03
FORCES ACTING ALONG THE 9 DOF
NODE 19 0.00000E+00 -0.99000E+02 0.62595E+05
NODE 20 0.00000E+00 -0.78221E-08 0.13106E-06
NODE 21 0.00000E+00 0.99000E+02 -0.68658E+05

ELEMENT 10, FROM NODE 19, TO NODE 21 - LENGTH = 20.00 ft
Warp. Torque Bimoment

left half of span,at tenth points of length
span span span span span span

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                    54OUT
soil,k/ft 0.0      0.1      0.2      0.3      0.4      0.5
shear,k   -99.00   -99.00   -99.00   -99.00   -99.00   -99.00
bmom,kft -62595.05 -63116.11 -63655.67 -64213.88 -64790.92 -65386.94
tdisp,ft  0.03331  0.03304  0.03275  0.03245  0.03214  0.03180
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span span
soil,k/ft 0.5      0.6      0.7      0.8      0.9      1.0
shear,k   -99.00   -99.00   -99.00   -99.00   -99.00   -99.00
bmom,kft -65386.94 -66002.13 -66636.67 -67290.73 -67964.52 -68658.22
tdisp,ft  0.03180  0.03146  0.03109  0.03071  0.03031  0.02990
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 11 DISPLACEMENTS IN INCIDENCES 21 22 23
NODE 21 0.00000E+00 0.29900E-01 -0.21100E-03
NODE 22 0.00000E+00 0.27197E-01 -0.47648E-03
NODE 23 0.00000E+00 0.24192E-01 -0.29194E-03
FORCES ACTING ALONG THE 9 DOF
NODE 21 0.00000E+00 0.11610E+04 0.68658E+05
NODE 22 0.00000E+00 0.55354E-09 0.35221E-07
NODE 23 0.00000E+00 -0.11610E+04 -0.49376E+05

ELEMENT 11, FROM NODE 21, TO NODE 23 - LENGTH = 22.50 ft

left half of span,at tenth points of length
span span span span span span span
soil,k/ft 0.0      0.1      0.2      0.3      0.4      0.5
shear,k   1161.00  1161.00  1161.00  1161.00  1161.00  1161.00
bmom,kft -68658.22 -66627.53 -64621.55 -62639.54 -60680.77 -58744.51
tdisp,ft  0.02990  0.02941  0.02891  0.02838  0.02784  0.02727
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span span
soil,k/ft 0.5      0.6      0.7      0.8      0.9      1.0
shear,k   1161.00  1161.00  1161.00  1161.00  1161.00  1161.00
bmom,kft -58744.51 -56830.03 -54936.64 -53063.63 -51210.30 -49375.96
tdisp,ft  0.02727  0.02669  0.02609  0.02547  0.02484  0.02419
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 12 DISPLACEMENTS IN INCIDENCES 23 24 25
NODE 23 0.00000E+00 0.24192E-01 -0.29194E-03
NODE 24 0.00000E+00 0.20676E-01 -0.60415E-03
NODE 25 0.00000E+00 0.16951E-01 -0.34769E-03
FORCES ACTING ALONG THE 9 DOF
NODE 23 0.00000E+00 0.11610E+04 0.49376E+05
NODE 24 0.00000E+00 0.16209E-08 -0.21178E-07
NODE 25 0.00000E+00 -0.11610E+04 -0.31931E+05

ELEMENT 12, FROM NODE 23, TO NODE 25 - LENGTH = 22.50 ft

left half of span,at tenth points of length
span span span span span span span
soil,k/ft 0.0      0.1      0.2      0.3      0.4      0.5
shear,k   1161.00  1161.00  1161.00  1161.00  1161.00  1161.00

```

54OUT  
 bmom,kft -49375.96 -47559.94 -45761.57 -43980.16 -42215.08 -40465.65  
 tdisp,ft 0.02419 0.02353 0.02285 0.02216 0.02145 0.02073  
 axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1161.00	1161.00	1161.00	1161.00	1161.00	1161.00
bmom,kft	-40465.65	-38731.23	-37011.18	-35304.85	-33611.63	-31930.87
tdisp,ft	0.02073	0.02000	0.01925	0.01850	0.01773	0.01695
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

-----  
 ELEMENT 13 DISPLACEMENTS IN INCIDENCES 25 26 27  
 NODE 25 0.00000E+00 0.16951E-01 -0.34769E-03  
 NODE 26 0.00000E+00 0.12896E-01 -0.68672E-03  
 NODE 27 0.00000E+00 0.87188E-02 -0.38033E-03  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 25 0.00000E+00 0.11610E+04 0.31931E+05  
 NODE 26 0.00000E+00 0.97435E-09 -0.15432E-07  
 NODE 27 0.00000E+00 -0.11610E+04 -0.15674E+05

ELEMENT 13, FROM NODE 25, TO NODE 27 - LENGTH = 22.50 ft

left half of span,at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1161.00	1161.00	1161.00	1161.00	1161.00	1161.00
bmom,kft	-31930.87	-30261.95	-28604.26	-26957.18	-25320.10	-23692.42
tdisp,ft	0.01695	0.01616	0.01537	0.01456	0.01375	0.01293
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

right half of span,at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1161.00	1161.00	1161.00	1161.00	1161.00	1161.00
bmom,kft	-23692.42	-22073.52	-20462.81	-18859.69	-17263.56	-15673.84
tdisp,ft	0.01293	0.01210	0.01126	0.01042	0.00957	0.00872
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

-----  
 ELEMENT 14 DISPLACEMENTS IN INCIDENCES 27 28 29  
 NODE 27 0.00000E+00 0.87188E-02 -0.38033E-03  
 NODE 28 0.00000E+00 0.43796E-02 -0.72724E-03  
 NODE 29 0.00000E+00 0.00000E+00 -0.39108E-03  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 27 0.00000E+00 0.11610E+04 0.15674E+05  
 NODE 28 0.00000E+00 -0.33970E-09 -0.77334E-08  
 NODE 29 0.00000E+00 -0.11610E+04 -0.17972E-08

ELEMENT 14, FROM NODE 27, TO NODE 29 - LENGTH = 22.50 ft

left half of span,at tenth points of length

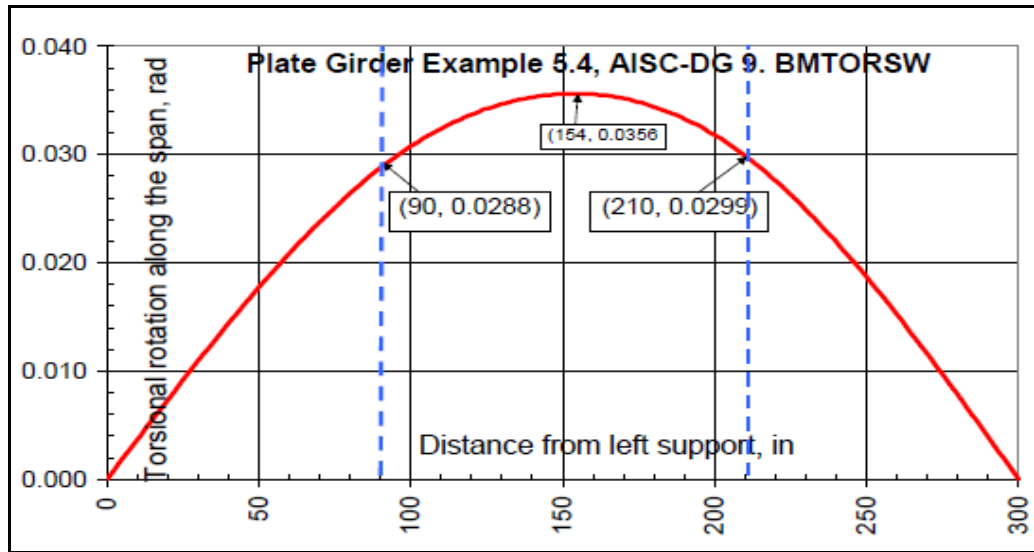
	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1161.00	1161.00	1161.00	1161.00	1161.00	1161.00
bmom,kft	-15673.84	-14089.93	-12511.25	-10937.21	-9367.23	-7800.72
tdisp,ft	0.00872	0.00786	0.00700	0.00613	0.00526	0.00439
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

Page 8

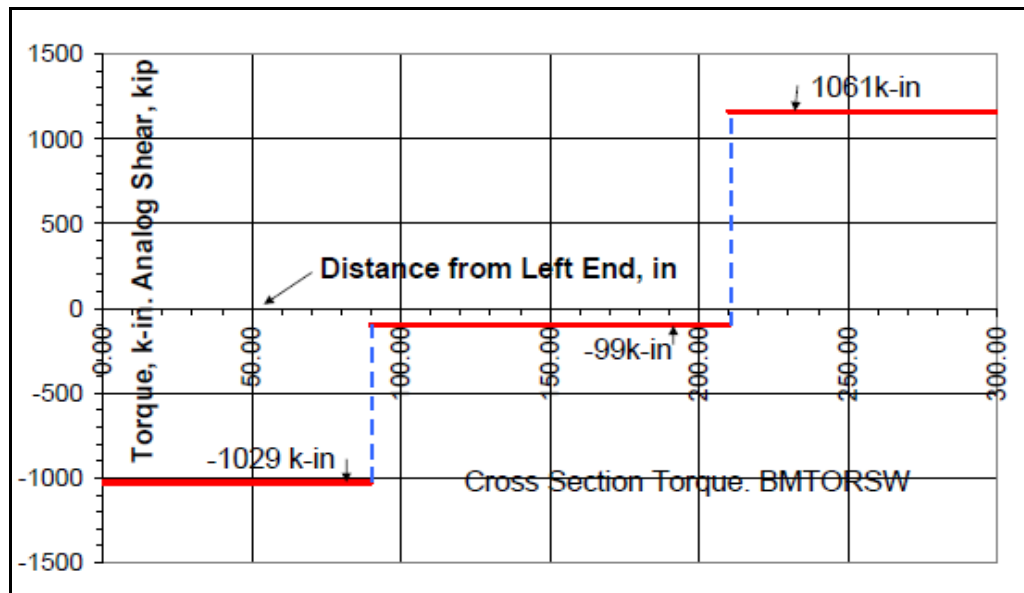
$(-D_i + D_j) * GJ + V_j * L = M_j - M_i$ , where  $GJ=1198400$   
 $(-0.00872 + 0) * 1198400 + 1161 * 22.5 = 0 - -15673.84$  54OUT  
 $-10450.05 + 26122.5 = 15675.45$  OK  
 right half of span, at tenth points of length  

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	1161.00	1161.00	1161.00	1161.00	1161.00	1161.00
bmom, kft	-7800.72	-6237.11	-4675.81	-3116.24	-1557.83	0.00
tdisp, ft	0.00439	0.00351	0.00264	0.00176	0.00088	0.00000
axial, k	0.00	AT 1st END and			0.00 AT 2nd END	

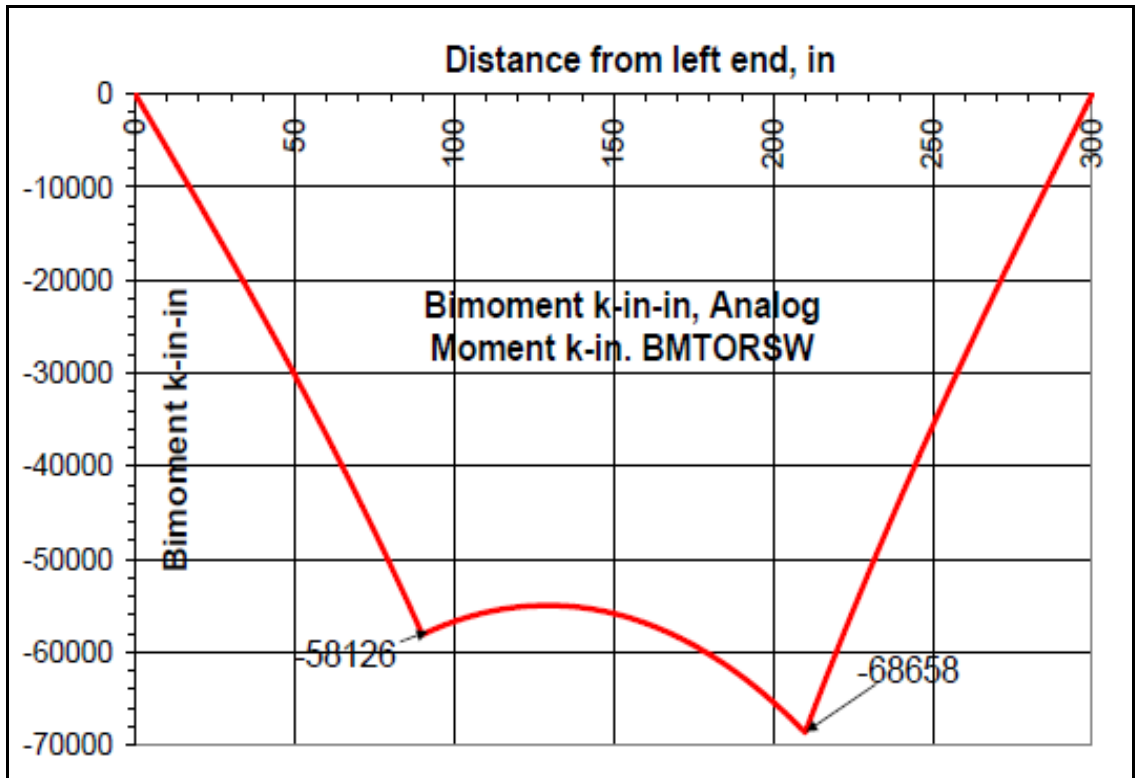
Last Page. Output Notepad. AISC Ex 5.4



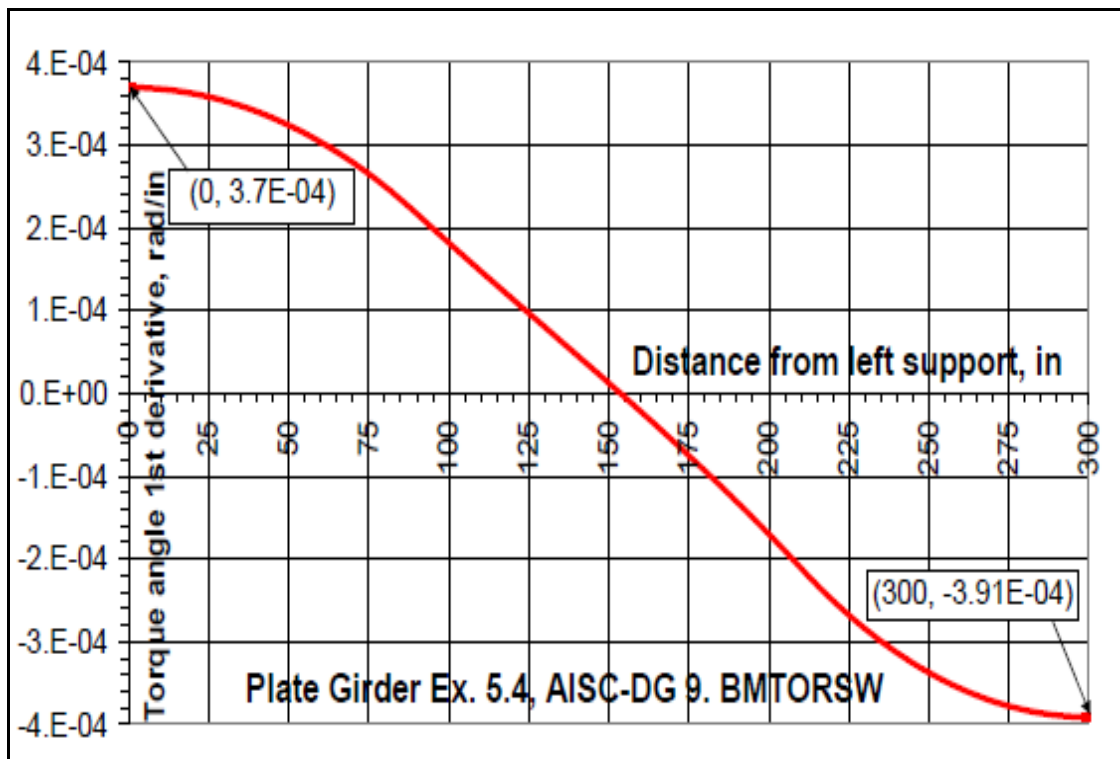
Twist Angle for Ultimate Loads in the TWB



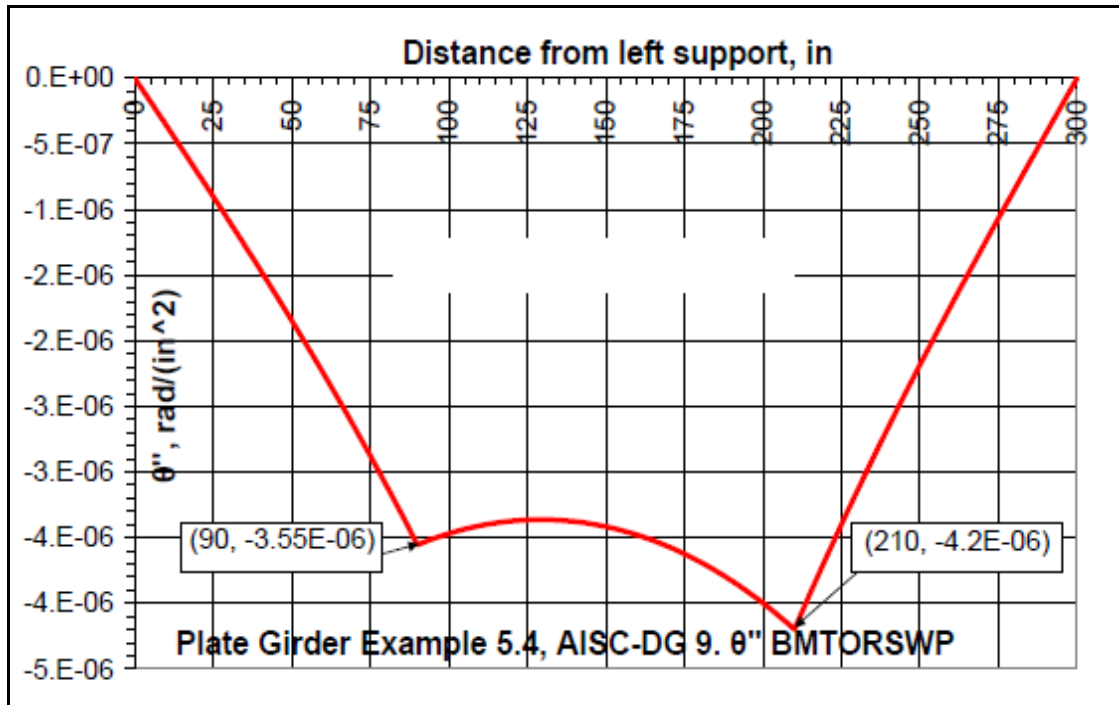
TWB Torque and BC Analogue Shear from BMTORSWP



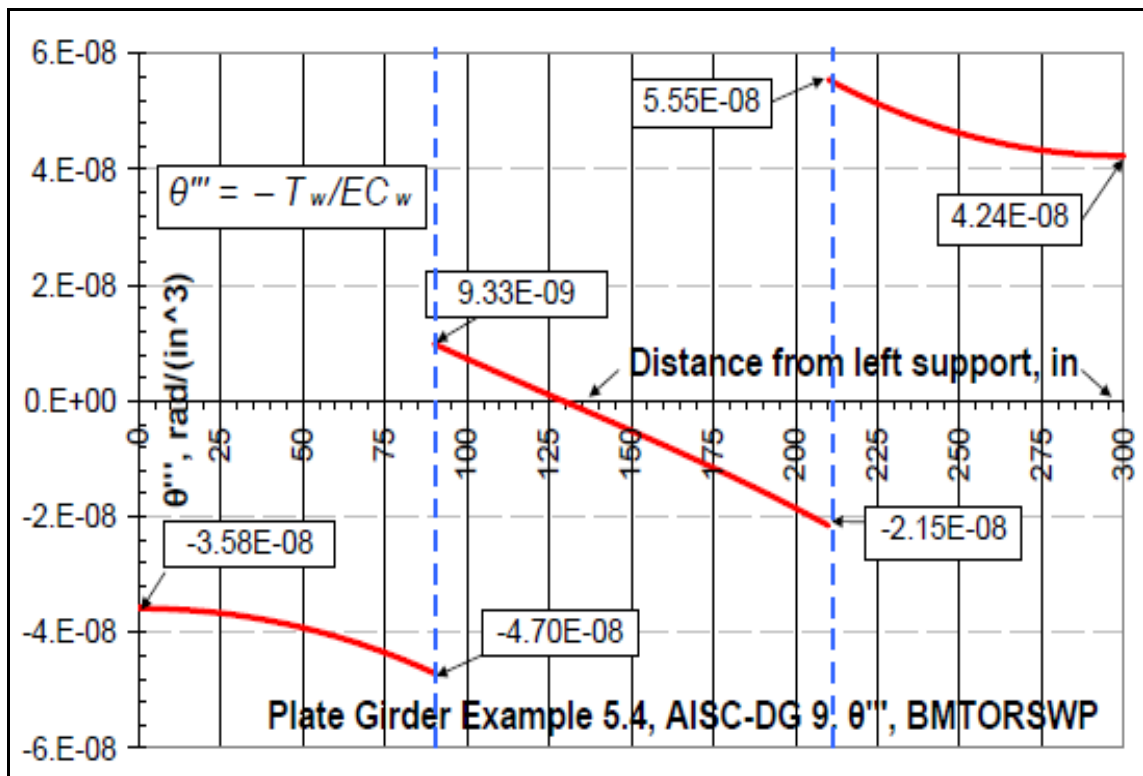
TWB Bimoment and BC Analogue Moment from BMTORSW



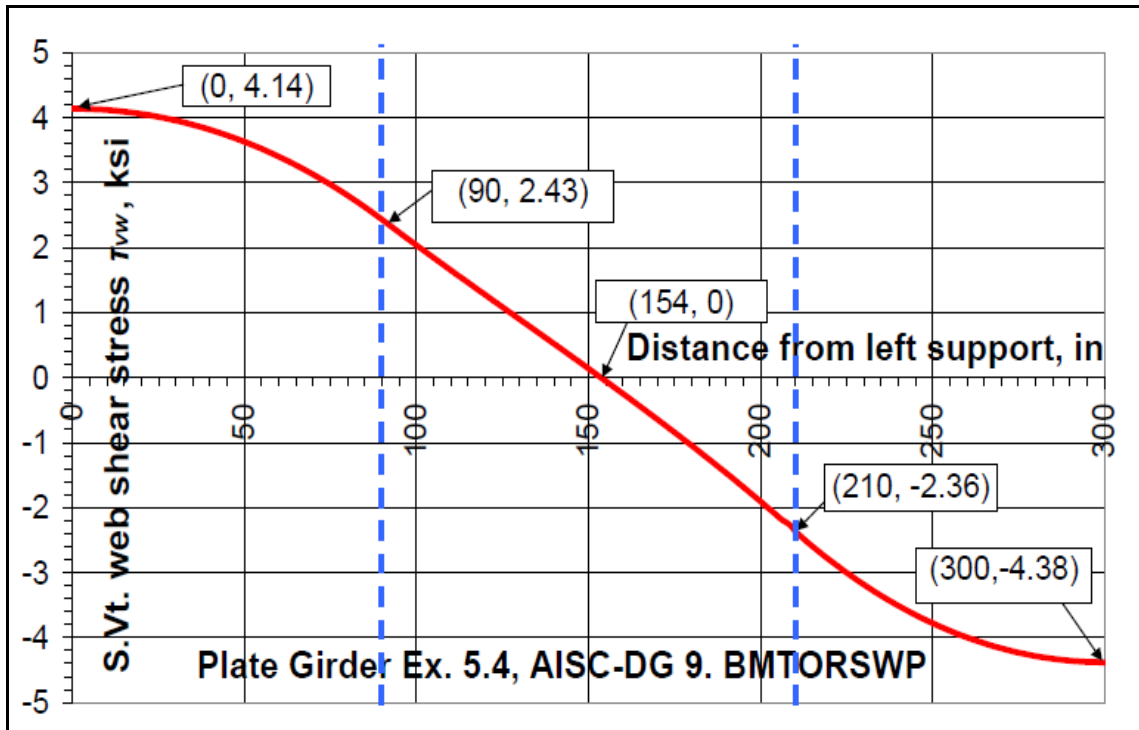
Twist Angle Derivative



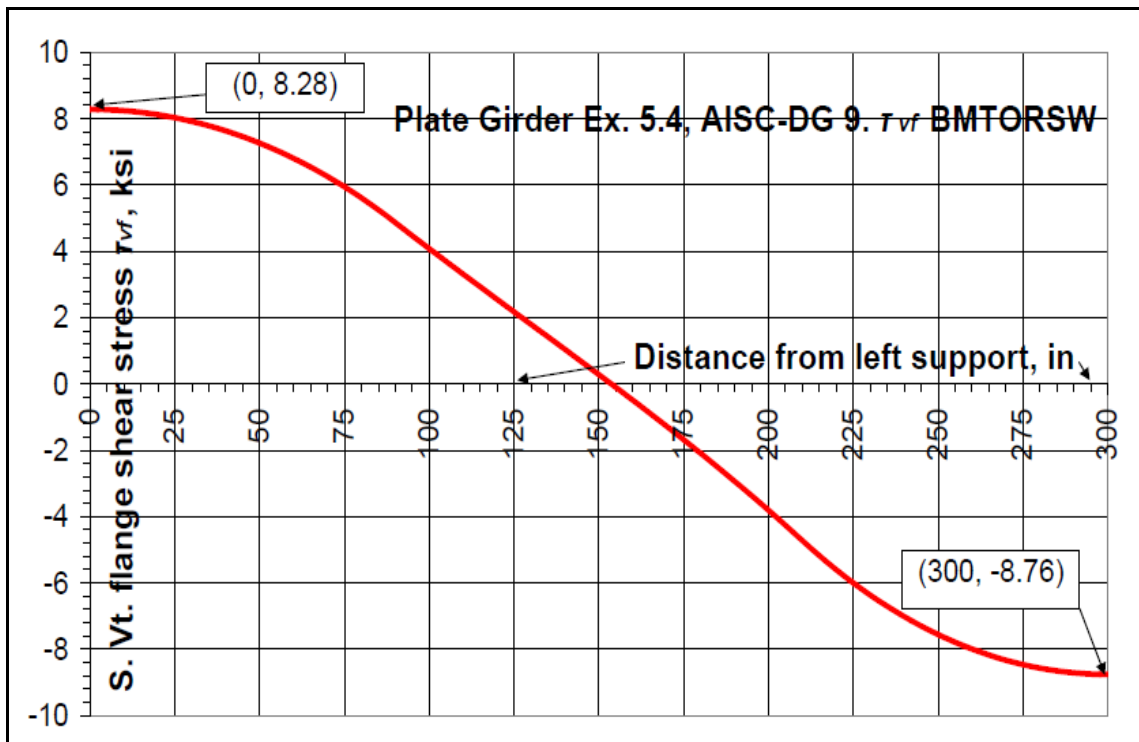
Twist Angle 2<sup>nd</sup> Derivative



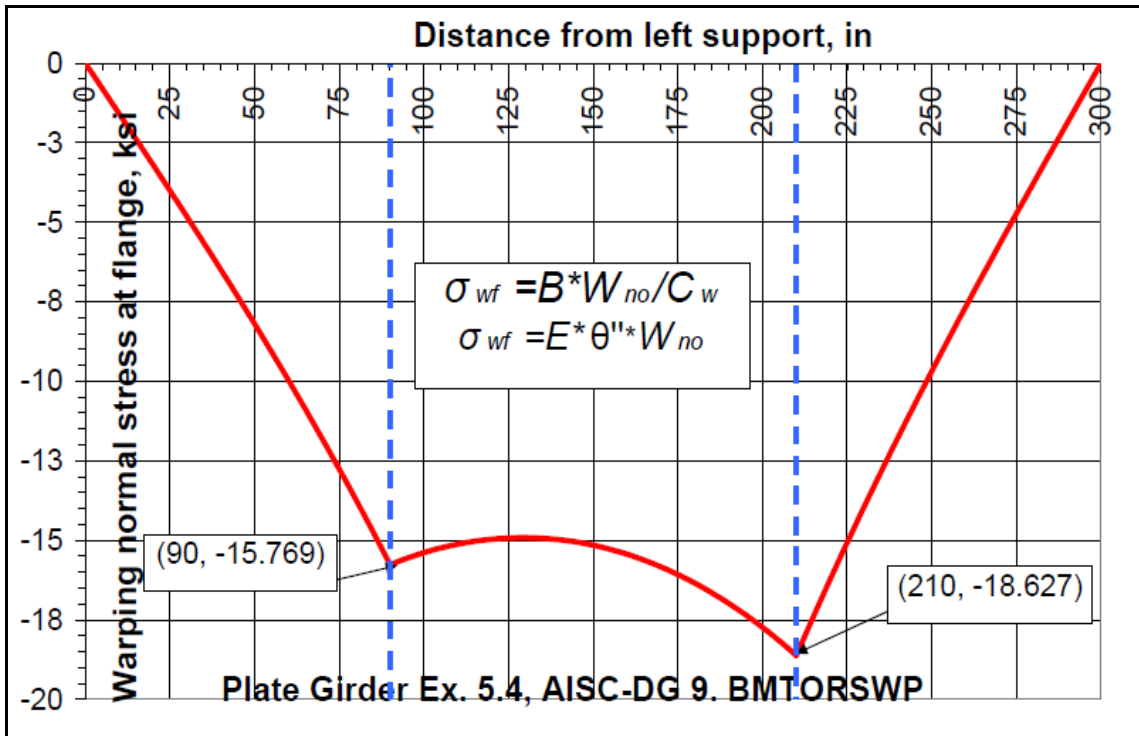
Twist Angle Third Derivative



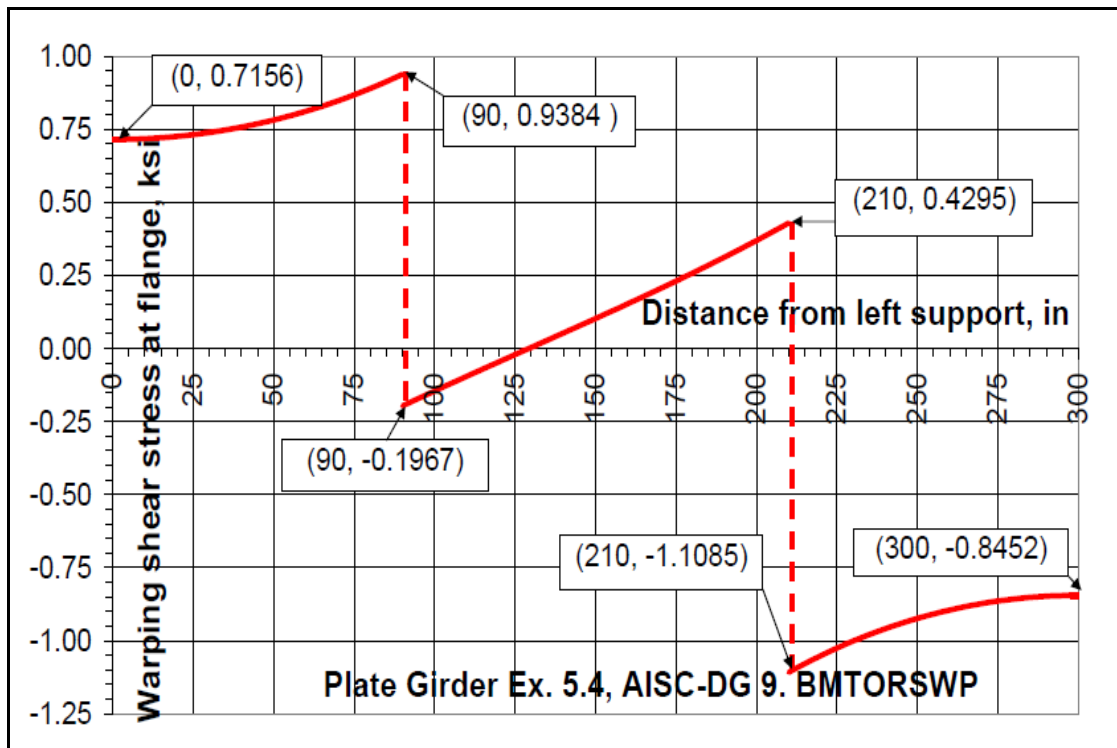
St. Vt. Shear Stress in TWB Web



St. Vt. Shear Stress in TWB Flange



Warping Normal Stresses at TWB Flange



Warping Shear Stresses at TWB Flange

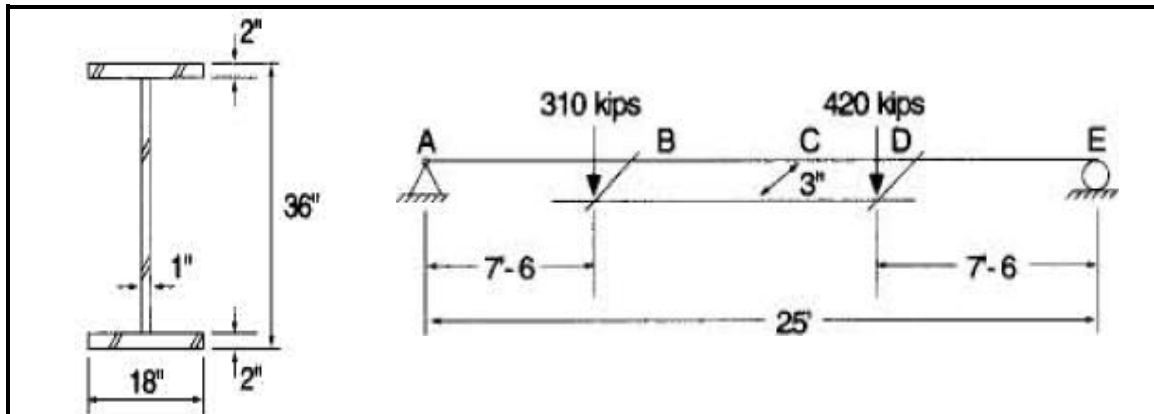


Location		$\sigma_w$	$\sigma_b$	$f_{un}$
Point D	flange web	- 18.63		-45.63
		-18.4	$\pm 26.6$	-45.0
		—	—	—
Point E	flange web	0	0	0
		—	—	—
				-45.63
Maximum				-45.0

Discrepancy in Total Maximum Normal Stresses in Interest Points

Location		$\tau_f$	$\tau_w$	$\tau_b$	$f_{uv}$
Point D	flange web	- 4.73	- 1.11		- 8.21
		-4.47	-1.11	$\pm 2.37$	- 7.95
		-2.24	—	$\pm 12.1$	-14.3
Point E	flange web	- 8.75	-0.84		-11.9
		-8.75	-0.87	$\pm 2.37$	-12.0
		-4.37	—	$\pm 12.1$	-16.5
Maximum					-16.51
Maximum					-16.5

No Discrepancy in Total Maximum Shear Stresses in Interest Points



TWB Cross Section and Interest Points along Beam

From Appendix B, Case 3 with  $\bar{a} = 0.3$ , it is estimated that the maximum rotation will occur at approximately  $14\frac{1}{2}$  feet  
 Exact location for exact "a"  $154" = 12'-10"$   $\rightarrow$  = 0.023 rad.  $\leftarrow$   $0.024 @ 12'-10"$

Discrepancies on Maximum Twist Angle and Location

Table BMTORSW Output for Graphics

BMTORSW P								
Elem	Z	θ	Z	Moment	Z	θ'	Z	θ''
1								
	0	0.000E+00	0	0	0	3.696E-04	0	2.940E-13
	2.25	8.000E-04	2.25	-1318.87	2.25	3.695E-04	2.25	-8.064E-08
	4.5	1.700E-03	4.5	-2638.23	4.5	3.692E-04	4.5	-1.613E-07
	6.75	2.500E-03	6.75	-3958.57	6.75	3.687E-04	6.75	-2.420E-07
	9	3.300E-03	9	-5280.37	9	3.681E-04	9	-3.228E-07
	11.25	4.100E-03	11.25	-6604.14	11.25	3.673E-04	11.25	-4.038E-07
	13.5	5.000E-03	13.5	-7930.35	13.5	3.663E-04	13.5	-4.849E-07
	15.75	5.800E-03	15.75	-9259.51	15.75	3.651E-04	15.75	-5.661E-07
	18	6.600E-03	18	-10592.1	18	3.637E-04	18	-6.476E-07
	20.25	7.400E-03	20.25	-11928.6	20.25	3.622E-04	20.25	-7.293E-07
	22.5	8.200E-03	22.5	-13269.6	22.5	3.605E-04	22.5	-8.113E-07
2								
	22.5	8.200E-03	22.5	-13269.6	22.5	3.605E-04	22.5	-8.113E-07
	24.75	9.100E-03	24.75	-14615.4	24.75	3.585E-04	24.75	-8.936E-07
	27	9.900E-03	27	-15966.7	27	3.564E-04	27	-9.762E-07
	29.25	1.070E-02	29.25	-17323.9	29.25	3.541E-04	29.25	-1.059E-06
	31.5	1.150E-02	31.5	-18687.6	31.5	3.517E-04	31.5	-1.143E-06
	33.75	1.220E-02	33.75	-20058.1	33.75	3.490E-04	33.75	-1.226E-06
	36	1.300E-02	36	-21436.2	36	3.461E-04	36	-1.311E-06
	38.25	1.380E-02	38.25	-22822.1	38.25	3.431E-04	38.25	-1.395E-06
	40.5	1.460E-02	40.5	-24216.5	40.5	3.399E-04	40.5	-1.481E-06
	42.75	1.530E-02	42.75	-25620	42.75	3.364E-04	42.75	-1.566E-06
	45	1.610E-02	45	-27032.9	45	3.328E-04	45	-1.653E-06
3								
	45	1.610E-02	45	-27032.9	45	3.328E-04	45	-1.653E-06
	47.25	1.680E-02	47.25	-28455.8	47.25	3.290E-04	47.25	-1.740E-06
	49.5	1.760E-02	49.5	-29889.3	49.5	3.250E-04	49.5	-1.827E-06
	51.75	1.830E-02	51.75	-31333.9	51.75	3.208E-04	51.75	-1.916E-06
	54	1.900E-02	54	-32790.1	54	3.164E-04	54	-2.005E-06
	56.25	1.970E-02	56.25	-34258.5	56.25	3.118E-04	56.25	-2.095E-06
	58.5	2.040E-02	58.5	-35739.5	58.5	3.069E-04	58.5	-2.185E-06

(continued) Table BMTORSW Output for Graphics

	60.75	2.110E-02	60.75	-37233.9	60.75	3.019E-04	60.75	-2.276E-06
	63	2.180E-02	63	-38742	63	2.967E-04	63	-2.369E-06
	65.25	2.240E-02	65.25	-40264.5	65.25	2.913E-04	65.25	-2.462E-06
	67.5	2.310E-02	67.5	-41802	67.5	2.856E-04	67.5	-2.556E-06
4								
	67.5	2.310E-02	67.5	-41802	67.5	2.856E-04	67.5	-2.556E-06
	69.75	2.370E-02	69.75	-43354.9	69.75	2.798E-04	69.75	-2.651E-06
	72	2.430E-02	72	-44924	72	2.737E-04	72	-2.747E-06
	74.25	2.490E-02	74.25	-46509.7	74.25	2.674E-04	74.25	-2.844E-06
	76.5	2.550E-02	76.5	-48112.6	76.5	2.609E-04	76.5	-2.942E-06
	78.75	2.610E-02	78.75	-49733.5	78.75	2.542E-04	78.75	-3.041E-06
	81	2.670E-02	81	-51372.7	81	2.472E-04	81	-3.141E-06
	83.25	2.720E-02	83.25	-53031	83.25	2.400E-04	83.25	-3.242E-06
	85.5	2.780E-02	85.5	-54709	85.5	2.326E-04	85.5	-3.345E-06
	87.75	2.830E-02	87.75	-56407.3	87.75	2.250E-04	87.75	-3.449E-06
	90	2.880E-02	90	-58126.5	90	2.171E-04	90	-3.554E-06
5								
	90	2.880E-02	90	-58126.5	90	2.171E-04	90	-3.554E-06
	92	2.920E-02	92	-57812.6	92	2.100E-04	92	-3.535E-06
	94	2.960E-02	94	-57515.7	94	2.030E-04	94	-3.516E-06
	96	3.000E-02	96	-57235.7	96	1.959E-04	96	-3.499E-06
	98	3.040E-02	98	-56972.4	98	1.890E-04	98	-3.483E-06
	100	3.080E-02	100	-56725.9	100	1.820E-04	100	-3.468E-06
	102	3.110E-02	102	-56496	102	1.751E-04	102	-3.454E-06
	104	3.150E-02	104	-56282.6	104	1.682E-04	104	-3.441E-06
	106	3.180E-02	106	-56085.7	106	1.613E-04	106	-3.429E-06
	108	3.210E-02	108	-55905.2	108	1.545E-04	108	-3.418E-06
	110	3.240E-02	110	-55741.2	110	1.476E-04	110	-3.408E-06
6								
	110	3.240E-02	110	-55741.2	110	1.476E-04	110	-3.408E-06
	112	3.270E-02	112	-55593.5	112	1.408E-04	112	-3.399E-06
	114	3.300E-02	114	-55462	114	1.341E-04	114	-3.391E-06
	116	3.320E-02	116	-55346.9	116	1.273E-04	116	-3.384E-06
	118	3.350E-02	118	-55247.9	118	1.205E-04	118	-3.378E-06

(continued) Table BMTORSW Output for Graphics

	120	3.370E-02	120	-55165.2	120	1.138E-04	120	-3.373E-06
	122	3.390E-02	122	-55098.6	122	1.070E-04	122	-3.369E-06
	124	3.420E-02	124	-55048.1	124	1.003E-04	124	-3.366E-06
	126	3.430E-02	126	-55013.8	126	9.356E-05	126	-3.364E-06
	128	3.450E-02	128	-54995.6	128	8.683E-05	128	-3.362E-06
	130	3.470E-02	130	-54993.6	130	8.011E-05	130	-3.362E-06
7								
	130	3.470E-02	130	-54993.6	130	8.011E-05	130	-3.362E-06
	132	3.480E-02	132	-55007.6	132	7.338E-05	132	-3.363E-06
	134	3.500E-02	134	-55037.8	134	6.666E-05	134	-3.365E-06
	136	3.510E-02	136	-55084.1	136	5.992E-05	136	-3.368E-06
	138	3.520E-02	138	-55146.5	138	5.318E-05	138	-3.372E-06
	140	3.530E-02	140	-55225.2	140	4.644E-05	140	-3.376E-06
	142	3.540E-02	142	-55320	142	3.968E-05	142	-3.382E-06
	144	3.550E-02	144	-55431	144	3.291E-05	144	-3.389E-06
	146	3.550E-02	146	-55558.2	146	2.612E-05	146	-3.397E-06
	148	3.560E-02	148	-55701.8	148	1.932E-05	148	-3.406E-06
	150	3.560E-02	150	-55861.6	150	1.250E-05	150	-3.415E-06
8								
	150	3.560E-02	150	-55861.6	150	1.250E-05	150	-3.415E-06
	152	3.560E-02	152	-56037.9	152	5.657E-06	152	-3.426E-06
	154	3.560E-02	154	-56230.5	154	-1.207E-06	154	-3.438E-06
	156	3.560E-02	156	-56439.7	156	-8.095E-06	156	-3.451E-06
	158	3.560E-02	158	-56665.4	158	-1.501E-05	158	-3.464E-06
	160	3.560E-02	160	-56907.6	160	-2.195E-05	160	-3.479E-06
	162	3.550E-02	162	-57166.6	162	-2.893E-05	162	-3.495E-06
	164	3.550E-02	164	-57442.3	164	-3.594E-05	164	-3.512E-06
	166	3.540E-02	166	-57734.9	166	-4.298E-05	166	-3.530E-06
	168	3.530E-02	168	-58044.4	168	-5.006E-05	168	-3.549E-06
	170	3.520E-02	170	-58370.9	170	-5.717E-05	170	-3.569E-06
9								
	170	3.520E-02	170	-58370.9	170	-5.717E-05	170	-3.569E-06
	172	3.510E-02	172	-58714.5	172	-6.433E-05	172	-3.590E-06
	174	3.490E-02	174	-59075.3	174	-7.153E-05	174	-3.612E-06

(continued) Table BMTORSW Output for Graphics

	176	3.480E-02	176	-59453.4	176	-7.878E-05	176	-3.635E-06
	178	3.460E-02	178	-59849	178	-8.607E-05	178	-3.659E-06
	180	3.440E-02	180	-60262.1	180	-9.342E-05	180	-3.684E-06
	182	3.420E-02	182	-60692.8	182	-1.008E-04	182	-3.711E-06
	184	3.400E-02	184	-61141.4	184	-1.083E-04	184	-3.738E-06
	186	3.380E-02	186	-61607.8	186	-1.158E-04	186	-3.767E-06
	188	3.360E-02	188	-62092.3	188	-1.233E-04	188	-3.796E-06
	190	3.330E-02	190	-62595	190	-1.310E-04	190	-3.827E-06
10								
	190	3.330E-02	190	-62595	190	-1.310E-04	190	-3.827E-06
	192	3.300E-02	192	-63116.1	192	-1.386E-04	192	-3.859E-06
	194	3.280E-02	194	-63655.7	194	-1.464E-04	194	-3.892E-06
	196	3.250E-02	196	-64213.9	196	-1.542E-04	196	-3.926E-06
	198	3.210E-02	198	-64790.9	198	-1.621E-04	198	-3.961E-06
	200	3.180E-02	200	-65386.9	200	-1.701E-04	200	-3.998E-06
	202	3.150E-02	202	-66002.1	202	-1.781E-04	202	-4.035E-06
	204	3.110E-02	204	-66636.7	204	-1.862E-04	204	-4.074E-06
	206	3.070E-02	206	-67290.7	206	-1.944E-04	206	-4.114E-06
	208	3.030E-02	208	-67964.5	208	-2.027E-04	208	-4.155E-06
	210	2.990E-02	210	-68658.2	210	-2.110E-04	210	-4.198E-06
11								
	210	2.990E-02	210	-68658.2	210	-2.110E-04	210	-4.198E-06
	212.25	2.940E-02	212.25	-66627.5	212.25	-2.203E-04	212.25	-4.074E-06
	214.5	2.890E-02	214.5	-64621.6	214.5	-2.293E-04	214.5	-3.951E-06
	216.75	2.840E-02	216.75	-62639.5	216.75	-2.381E-04	216.75	-3.830E-06
	219	2.780E-02	219	-60680.8	219	-2.466E-04	219	-3.710E-06
	221.25	2.730E-02	221.25	-58744.5	221.25	-2.548E-04	221.25	-3.592E-06
	223.5	2.670E-02	223.5	-56830	223.5	-2.627E-04	223.5	-3.475E-06
	225.75	2.610E-02	225.75	-54936.6	225.75	-2.704E-04	225.75	-3.359E-06
	228	2.550E-02	228	-53063.6	228	-2.778E-04	228	-3.244E-06
	230.25	2.480E-02	230.25	-51210.3	230.25	-2.850E-04	230.25	-3.131E-06
	232.5	2.420E-02	232.5	-49376	232.5	-2.919E-04	232.5	-3.019E-06
12								
	232.5	2.420E-02	232.5	-49376	232.5	-2.919E-04	232.5	-3.019E-06

(continued) Table BMTORSW Output for Graphics

	234.75	2.350E-02	234.75	-47559.9	234.75	-2.986E-04	234.75	-2.908E-06
	237	2.280E-02	237	-45761.6	237	-3.050E-04	237	-2.798E-06
	239.25	2.220E-02	239.25	-43980.2	239.25	-3.112E-04	239.25	-2.689E-06
	241.5	2.140E-02	241.5	-42215.1	241.5	-3.171E-04	241.5	-2.581E-06
	243.75	2.070E-02	243.75	-40465.6	243.75	-3.228E-04	243.75	-2.474E-06
	246	2.000E-02	246	-38731.2	246	-3.283E-04	246	-2.368E-06
	248.25	1.930E-02	248.25	-37011.2	248.25	-3.335E-04	248.25	-2.263E-06
	250.5	1.850E-02	250.5	-35304.9	250.5	-3.384E-04	250.5	-2.159E-06
	252.75	1.770E-02	252.75	-33611.6	252.75	-3.432E-04	252.75	-2.055E-06
	255	1.700E-02	255	-31930.9	255	-3.477E-04	255	-1.952E-06
13								
	255	1.700E-02	255	-31930.9	255	-3.477E-04	255	-1.952E-06
	257.25	1.620E-02	257.25	-30262	257.25	-3.520E-04	257.25	-1.850E-06
	259.5	1.540E-02	259.5	-28604.3	259.5	-3.560E-04	259.5	-1.749E-06
	261.75	1.460E-02	261.75	-26957.2	261.75	-3.598E-04	261.75	-1.648E-06
	264	1.370E-02	264	-25320.1	264	-3.634E-04	264	-1.548E-06
	266.25	1.290E-02	266.25	-23692.4	266.25	-3.668E-04	266.25	-1.449E-06
	268.5	1.210E-02	268.5	-22073.5	268.5	-3.700E-04	268.5	-1.350E-06
	270.75	1.130E-02	270.75	-20462.8	270.75	-3.729E-04	270.75	-1.251E-06
	273	1.040E-02	273	-18859.7	273	-3.756E-04	273	-1.153E-06
	275.25	9.600E-03	275.25	-17263.6	275.25	-3.781E-04	275.25	-1.055E-06
	277.5	8.700E-03	277.5	-15673.8	277.5	-3.803E-04	277.5	-9.583E-07
14								
	277.5	8.700E-03	277.5	-15673.8	277.5	-3.803E-04	277.5	-9.583E-07
	279.75	7.900E-03	279.75	-14089.9	279.75	-3.824E-04	279.75	-8.615E-07
	282	7.000E-03	282	-12511.3	282	-3.842E-04	282	-7.649E-07
	284.25	6.100E-03	284.25	-10937.2	284.25	-3.858E-04	284.25	-6.687E-07
	286.5	5.300E-03	286.5	-9367.23	286.5	-3.872E-04	286.5	-5.727E-07
	288.75	4.400E-03	288.75	-7800.72	288.75	-3.884E-04	288.75	-4.769E-07
	291	3.500E-03	291	-6237.11	291	-3.894E-04	291	-3.813E-07
	293.25	2.600E-03	293.25	-4675.81	293.25	-3.901E-04	293.25	-2.859E-07
	295.5	1.800E-03	295.5	-3116.24	295.5	-3.907E-04	295.5	-1.905E-07
	297.75	9.000E-04	297.75	-1557.83	297.75	-3.910E-04	297.75	-9.525E-08
	300	0.000E+00	300	0	300	-3.911E-04	300	3.473E-13

Table 3<sup>rd</sup> Derivative and Stresses Processed

Elem	Pure torsion shear						Warping stresses			
	Z	θ'''	Z	stress		Z	axial		shear	
				web	flange		flange	Z	flange	
1										
		0-3.584E-08	0	4.14	0	8.28	0	0.00	0	0.72
	2.25	-3.584E-08	2.25	4.14	2.25	8.28	2.25	-0.36	2.25	0.72
	4.5	-3.586E-08	4.5	4.14	4.5	8.27	4.5	-0.72	4.5	0.72
	6.75	-3.590E-08	6.75	4.13	6.75	8.26	6.75	-1.07	6.75	0.72
	9	-3.594E-08	9	4.12	9	8.25	9	-1.43	9	0.72
	11.25	-3.600E-08	11.25	4.11	11.25	8.23	11.25	-1.79	11.25	0.72
	13.5	-3.608E-08	13.5	4.10	13.5	8.21	13.5	-2.15	13.5	0.72
	15.75	-3.616E-08	15.75	4.09	15.75	8.18	15.75	-2.51	15.75	0.72
	18	-3.626E-08	18	4.07	18	8.15	18	-2.87	18	0.72
	20.25	-3.638E-08	20.25	4.06	20.25	8.11	20.25	-3.24	20.25	0.73
22.5	-3.650E-08	22.5	4.04	22.5	8.08	22.5	-3.60	22.5	0.73	
2				0.00		0.00		0.00		0.00
	22.5	-3.650E-08	22.5	4.04	22.5	8.08	22.5	-3.60	22.5	0.73
	24.75	-3.664E-08	24.75	4.02	24.75	8.03	24.75	-3.96	24.75	0.73
	27	-3.680E-08	27	3.99	27	7.98	27	-4.33	27	0.73
	29.25	-3.696E-08	29.25	3.97	29.25	7.93	29.25	-4.70	29.25	0.74
	31.5	-3.715E-08	31.5	3.94	31.5	7.88	31.5	-5.07	31.5	0.74
	33.75	-3.734E-08	33.75	3.91	33.75	7.82	33.75	-5.44	33.75	0.75
	36	-3.755E-08	36	3.88	36	7.75	36	-5.82	36	0.75
	38.25	-3.777E-08	38.25	3.84	38.25	7.69	38.25	-6.19	38.25	0.75
	40.5	-3.801E-08	40.5	3.81	40.5	7.61	40.5	-6.57	40.5	0.76
	42.75	-3.826E-08	42.75	3.77	42.75	7.54	42.75	-6.95	42.75	0.76
45	-3.853E-08	45	3.73	45	7.45	45	-7.33	45	0.77	
3				0.00		0.00		0.00		0.00
	45	-3.853E-08	45	3.73	45	7.45	45	-7.33	45	0.77
	47.25	-3.881E-08	47.25	3.68	47.25	7.37	47.25	-7.72	47.25	0.77
	49.5	-3.910E-08	49.5	3.64	49.5	7.28	49.5	-8.11	49.5	0.78
	51.75	-3.941E-08	51.75	3.59	51.75	7.19	51.75	-8.50	51.75	0.79
	54	-3.973E-08	54	3.54	54	7.09	54	-8.90	54	0.79
	56.25	-4.007E-08	56.25	3.49	56.25	6.98	56.25	-9.30	56.25	0.80

(continued) Table 3<sup>rd</sup> Derivative and Stresses Processed

	58.5	-4.042E-08	58.5	3.44	58.5	6.87	58.5	-9.69	58.5	0.81
	60.75	-4.079E-08	60.75	3.38	60.75	6.76	60.75	-10.10	60.75	0.81
	63	-4.117E-08	63	3.32	63	6.65	63	-10.51	63	0.82
	65.25	-4.157E-08	65.25	3.26	65.25	6.53	65.25	-10.92	65.25	0.83
	67.5	-4.198E-08	67.5	3.20	67.5	6.40	67.5	-11.34	67.5	0.84
4				0.00		0.00		0.00		0.00
	67.5	-4.199E-08	67.5	3.20	67.5	6.40	67.5	-11.34	67.5	0.84
	69.75	-4.242E-08	69.75	3.13	69.75	6.27	69.75	-11.76	69.75	0.85
	72	-4.286E-08	72	3.07	72	6.13	72	-12.19	72	0.86
	74.25	-4.332E-08	74.25	2.99	74.25	5.99	74.25	-12.62	74.25	0.86
	76.5	-4.380E-08	76.5	2.92	76.5	5.84	76.5	-13.05	76.5	0.87
	78.75	-4.429E-08	78.75	2.85	78.75	5.69	78.75	-13.49	78.75	0.88
	81	-4.480E-08	81	2.77	81	5.54	81	-13.94	81	0.89
	83.25	-4.533E-08	83.25	2.69	83.25	5.38	83.25	-14.38	83.25	0.91
	85.5	-4.587E-08	85.5	2.61	85.5	5.21	85.5	-14.84	85.5	0.92
	87.75	-4.643E-08	87.75	2.52	87.75	5.04	87.75	-15.30	87.75	0.93
	90	-4.700E-08	90	2.43	90	4.86	90	-15.77	90	0.94
5				0.00		0.00		0.00		0.00
	90	9.852E-09	90	2.43	90	4.86	90	-15.77	90	-0.20
	92	9.334E-09	92	2.35	92	4.70	92	-15.68	92	-0.19
	94	8.818E-09	94	2.27	94	4.55	94	-15.60	94	-0.18
	96	8.304E-09	96	2.19	96	4.39	96	-15.53	96	-0.17
	98	7.792E-09	98	2.12	98	4.23	98	-15.45	98	-0.16
	100	7.283E-09	100	2.04	100	4.08	100	-15.39	100	-0.15
	102	6.775E-09	102	1.96	102	3.92	102	-15.33	102	-0.14
	104	6.270E-09	104	1.88	104	3.77	104	-15.27	104	-0.13
	106	5.767E-09	106	1.81	106	3.61	106	-15.21	106	-0.12
	108	5.266E-09	108	1.73	108	3.46	108	-15.17	108	-0.11
	110	4.767E-09	110	1.65	110	3.31	110	-15.12	110	-0.10
6				0.00		0.00		0.00		0.00
	110	4.764E-09	110	1.65	110	3.31	110	-15.12	110	-0.10
	112	4.266E-09	112	1.58	112	3.15	112	-15.08	112	-0.09
	114	3.769E-09	114	1.50	114	3.00	114	-15.05	114	-0.08
	116	3.273E-09	116	1.43	116	2.85	116	-15.01	116	-0.07



(continued) Table 3<sup>rd</sup> Derivative and Stresses Processed

	118	2.778E-09	118	1.35	118	2.70	118	-14.99	118	-0.06
	120	2.283E-09	120	1.27	120	2.55	120	-14.97	120	-0.05
	122	1.789E-09	122	1.20	122	2.40	122	-14.95	122	-0.04
	124	1.295E-09	124	1.12	124	2.25	124	-14.93	124	-0.03
	126	8.022E-10	126	1.05	126	2.10	126	-14.93	126	-0.02
	128	3.100E-10	128	0.97	128	1.94	128	-14.92	128	-0.01
	130	-1.815E-10	130	0.90	130	1.79	130	-14.92	130	0.00
7				0.00		0.00		0.00		0.00
	130	-1.849E-10	130	0.90	130	1.79	130	-14.92	130	0.00
	132	-6.765E-10	132	0.82	132	1.64	132	-14.92	132	0.01
	134	-1.169E-09	134	0.75	134	1.49	134	-14.93	134	0.02
	136	-1.662E-09	136	0.67	136	1.34	136	-14.94	136	0.03
	138	-2.156E-09	138	0.60	138	1.19	138	-14.96	138	0.04
	140	-2.650E-09	140	0.52	140	1.04	140	-14.98	140	0.05
	142	-3.146E-09	142	0.44	142	0.89	142	-15.01	142	0.06
	144	-3.642E-09	144	0.37	144	0.74	144	-15.04	144	0.07
	146	-4.139E-09	146	0.29	146	0.59	146	-15.07	146	0.08
	148	-4.637E-09	148	0.22	148	0.43	148	-15.11	148	0.09
	150	-5.135E-09	150	0.14	150	0.28	150	-15.15	150	0.10
8				0.00		0.00		0.00		0.00
	150	-5.139E-09	150	0.14	150	0.28	150	-15.15	150	0.10
	152	-5.639E-09	152	0.06	152	0.13	152	-15.20	152	0.11
	154	-6.141E-09	154	-0.01	154	-0.03	154	-15.25	154	0.12
	156	-6.646E-09	156	-0.09	156	-0.18	156	-15.31	156	0.13
	158	-7.152E-09	158	-0.17	158	-0.34	158	-15.37	158	0.14
	160	-7.661E-09	160	-0.25	160	-0.49	160	-15.44	160	0.15
	162	-8.173E-09	162	-0.32	162	-0.65	162	-15.51	162	0.16
	164	-8.686E-09	164	-0.40	164	-0.81	164	-15.58	164	0.17
	166	-9.202E-09	166	-0.48	166	-0.96	166	-15.66	166	0.18
	168	-9.720E-09	168	-0.56	168	-1.12	168	-15.75	168	0.19
	170	-1.024E-08	170	-0.64	170	-1.28	170	-15.84	170	0.20
9				0.00		0.00		0.00		0.00
	170	-1.024E-08	170	-0.64	170	-1.28	170	-15.84	170	0.20
	172	-1.077E-08	172	-0.72	172	-1.44	172	-15.93	172	0.22

(continued) Table 3<sup>rd</sup> Derivative and Stresses Processed

	174	-1.129E-08	174	-0.80	174	-1.60	174	-16.03	174	0.23
	176	-1.182E-08	176	-0.88	176	-1.76	176	-16.13	176	0.24
	178	-1.236E-08	178	-0.96	178	-1.93	178	-16.23	178	0.25
	180	-1.290E-08	180	-1.05	180	-2.09	180	-16.35	180	0.26
	182	-1.344E-08	182	-1.13	182	-2.26	182	-16.47	182	0.27
	184	-1.399E-08	184	-1.21	184	-2.43	184	-16.59	184	0.28
	186	-1.454E-08	186	-1.30	186	-2.59	186	-16.71	186	0.29
	188	-1.509E-08	188	-1.38	188	-2.76	188	-16.84	188	0.30
	190	-1.565E-08	190	-1.47	190	-2.93	190	-16.98	190	0.31
10				0.00		0.00		0.00		0.00
	190	-1.565E-08	190	-1.47	190	-2.93	190	-16.98	190	0.31
	192	-1.621E-08	192	-1.55	192	-3.10	192	-17.12	192	0.32
	194	-1.678E-08	194	-1.64	194	-3.28	194	-17.27	194	0.34
	196	-1.735E-08	196	-1.73	196	-3.45	196	-17.42	196	0.35
	198	-1.793E-08	198	-1.82	198	-3.63	198	-17.57	198	0.36
	200	-1.851E-08	200	-1.91	200	-3.81	200	-17.74	200	0.37
	202	-1.910E-08	202	-1.99	202	-3.99	202	-17.90	202	0.38
	204	-1.970E-08	204	-2.09	204	-4.17	204	-18.08	204	0.39
	206	-2.030E-08	206	-2.18	206	-4.35	206	-18.25	206	0.41
	208	-2.090E-08	208.6	-2.27	208	-4.54	208	-18.44	208	0.42
	210	-2.151E-08	210	-2.36	210	-4.73	210	-18.63	210	0.43
11				0.00		0.00		0.00		0.00
	210	5.552E-08	210	-2.36	210	-4.73	210	-18.63	210	-1.11
	212.25	5.484E-08	212.25	-2.47	212.25	-4.93	212.3	-18.08	212.25	-1.09
	214.5	5.418E-08	214.5	-2.57	214.5	-5.14	214.5	-17.53	214.5	-1.08
	216.75	5.354E-08	216.75	-2.67	216.75	-5.33	216.8	-16.99	216.75	-1.07
	219	5.292E-08	219	-2.76	219	-5.52	219	-16.46	219	-1.06
	221.25	5.232E-08	221.25	-2.85	221.25	-5.71	221.3	-15.94	221.25	-1.04
	223.5	5.173E-08	223.5	-2.94	223.5	-5.88	223.5	-15.42	223.5	-1.03
	225.75	5.117E-08	225.75	-3.03	225.75	-6.06	225.8	-14.90	225.75	-1.02
	228	5.062E-08	228	-3.11	228	-6.22	228	-14.39	228	-1.01
	230.25	5.010E-08	230.25	-3.19	230.25	-6.38	230.3	-13.89	230.25	-1.00
	232.5	4.960E-08	232.5	-3.27	232.5	-6.54	232.5	-13.40	232.5	-0.99
12				0.00		0.00		0.00		0.00

(continued) Table 3<sup>rd</sup> Derivative and Stresses Processed

	232.5	4.959E-08	232.5	-3.27	232.5	-6.54	232.5	-13.40	232.5	-0.99
	234.75	4.910E-08	234.75	-3.34	234.75	-6.69	234.8	-12.90	234.75	-0.98
	237	4.863E-08	237	-3.42	237	-6.83	237	-12.41	237	-0.97
	239.25	4.818E-08	239.25	-3.49	239.25	-6.97	239.3	-11.93	239.25	-0.96
	241.5	4.775E-08	241.5	-3.55	241.5	-7.10	241.5	-11.45	241.5	-0.95
	243.75	4.733E-08	243.75	-3.62	243.75	-7.23	243.8	-10.98	243.75	-0.95
	246	4.693E-08	246	-3.68	246	-7.35	246	-10.51	246	-0.94
	248.25	4.655E-08	248.25	-3.74	248.25	-7.47	248.3	-10.04	248.25	-0.93
	250.5	4.619E-08	250.5	-3.79	250.5	-7.58	250.5	-9.58	250.5	-0.92
	252.75	4.584E-08	252.75	-3.84	252.75	-7.69	252.8	-9.12	252.75	-0.92
13	255	4.551E-08	255	-3.89	255	-7.79	255	-8.66	255	-0.91
	257.25	4.519E-08	257.25	-3.94	257.25	-7.88	257.3	-8.21	257.25	-0.90
	259.5	4.490E-08	259.5	-3.99	259.5	-7.97	259.5	-7.76	259.5	-0.90
	261.75	4.462E-08	261.75	-4.03	261.75	-8.06	261.8	-7.31	261.75	-0.89
	264	4.435E-08	264	-4.07	264	-8.14	264	-6.87	264	-0.89
	266.25	4.411E-08	266.25	-4.11	266.25	-8.22	266.3	-6.43	266.25	-0.88
	268.5	4.388E-08	268.5	-4.14	268.5	-8.29	268.5	-5.99	268.5	-0.88
	270.75	4.366E-08	270.75	-4.18	270.75	-8.35	270.8	-5.55	270.75	-0.87
	273	4.346E-08	273	-4.21	273	-8.41	273	-5.12	273	-0.87
	275.25	4.328E-08	275.25	-4.23	275.25	-8.47	275.3	-4.68	275.25	-0.86
14	277.5	4.312E-08	277.5	-4.26	277.5	-8.52	277.5	-4.25	277.5	-0.86
	279.75	4.297E-08	279.75	-4.28	279.75	-8.57	279.8	-3.82	279.75	-0.86
	282	4.283E-08	282	-4.30	282	-8.61	282	-3.39	282	-0.86
	284.25	4.271E-08	284.25	-4.32	284.25	-8.64	284.3	-2.97	284.25	-0.85
	286.5	4.261E-08	286.5	-4.34	286.5	-8.67	286.5	-2.54	286.5	-0.85
	288.75	4.253E-08	288.75	-4.35	288.75	-8.70	288.8	-2.12	288.75	-0.85
	291	4.245E-08	291	-4.36	291	-8.72	291	-1.69	291	-0.85
	293.25	4.240E-08	293.25	-4.37	293.25	-8.74	293.3	-1.27	293.25	-0.85
	295.5	4.236E-08	295.5	-4.38	295.5	-8.75	295.5	-0.85	295.5	-0.85
	297.75	4.234E-08	297.75	-4.38	297.75	-8.76	297.8	-0.42	297.75	-0.85
	300	4.233E-08	300	-4.38	300	-8.76	300	0.00	300	-0.85

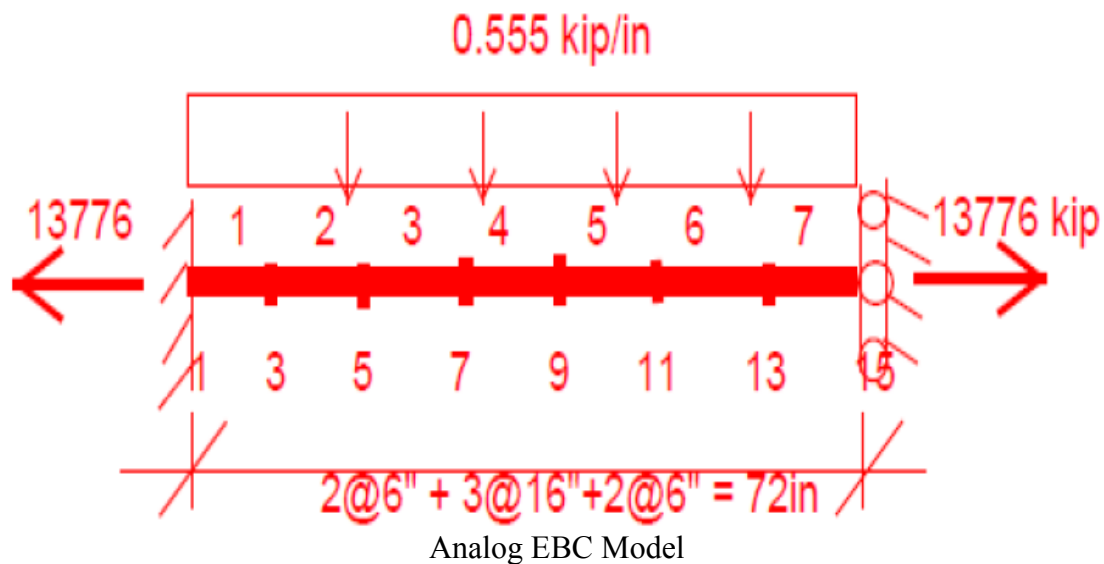
## APPENDIX F

### AISC-DG9-EXAMPLE 5.5

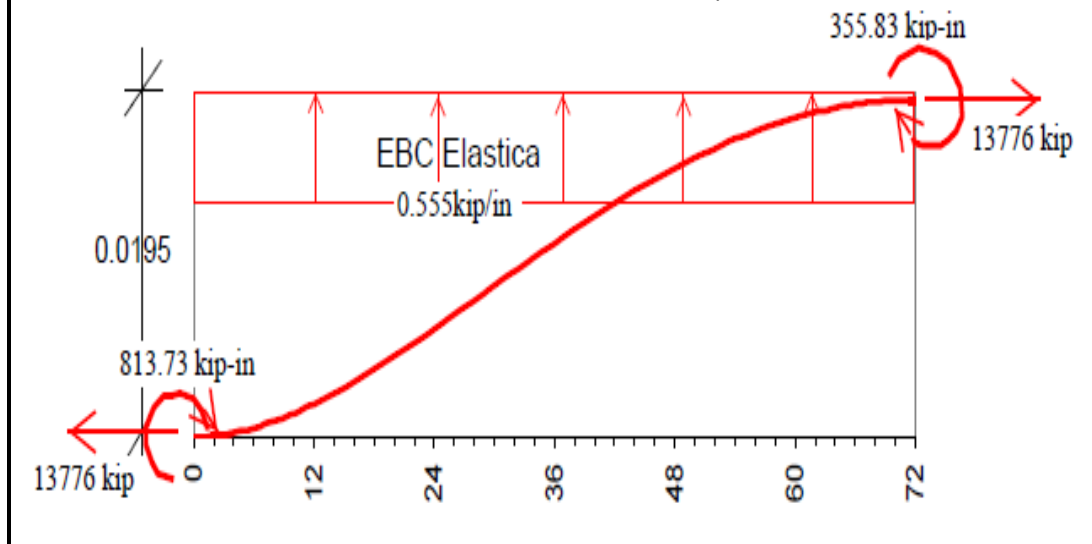
In the EBC of case study four, this material corresponds to the input model, input data, input forms, figures of output data with checks in notepad version, and excel processed output data and charts.

Charts are presented containing both partial and combined tresses along interest points of the beam and cross section profile. Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP.

It is important to notice that the exact location of the smeared DOF representing the distributed load is unknown. Nevertheless, it is assumed to be at midspan of the element for the purpose of checking summation of moments. This criterion has provided good results as it could be seen in pages 2 and four of the output notepads.



From summation of moments in the EBC  $\Delta = (B1 + B2 - wL^2/2) / GJ = (813.73 + 353.83 - 0.555 \times 72^2/2) / 13776 = 0.01967 \sim 0.0195$  (Channel MC18x47. BMTORSWP).



Elastic Line and BMTORSWP Model for DG9-Ex. 5.5

5-5

AISC Design Guide 9, Example 5.5, MC18x42.7 channel

```

7 15 2 0 9 0 0 0 29000.
1 1 2 3 852. 12. 6. -13776. 0.555 0.555
2 3 4 5 852. 12. 6. -13776. 0.555 0.555
3 5 6 7 852. 12. 16. -13776. 0.555 0.555
4 7 8 9 852. 12. 16. -13776. 0.555 0.555
5 9 10 11 852. 12. 16. -13776. 0.555 0.555
6 11 12 13 852. 12. 6. -13776. 0.555 0.555
7 13 14 15 852. 12. 6. -13776. 0.555 0.555
2 1 1 1 15 0 0 1 ← KX=KY=0 & KZ=1

```

KX=KY=KZ=1: Fixed end

Notepad Input Data for BMTORSWP.

PROGRAM BMTORSW INPUT FORM																																									
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	
ALPHAMERIC DESCRIPTION OF THE JOB																																	FILE NAME:								
AISC DESIGN GUIDE 9 EXAMPLE 5.5 ON WARPING																																									
NEL	NOD	NSUP	NSPD	JBW	NFX	NFY	NFZ	ELASTICITY (E11.4)																																	
7	15	2	0	9	0	0	0	29000.																																	
NE	NI	NC	NJ	Cw			AREA	LENGTH	WTANG	SOIL NI		SOI																													
1	1	2	3	852.			12.	6.																																	
2	3	4	5	852.			12.	6.																																	
3	5	6	7	852.			12.	16.																																	
4	7	8	9	852.			12.	16.																																	
5	9	10	11	852.			12.	16.																																	
6	11	12	13	852.			12.	6.																																	
7	13	14	15	852.			12.	6.																																	
NF	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ																					
2	1	1	1	1	15	0	0	1																																	
NF	N	KD	SPRING						N	KD	SPRING						N	KD																							
NF	N	LOAD VALUE				N	LOAD VALUE				N	LOAD VALUE				N	LOAD VALUE				N	LOAD VALUE																			
NF	ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MOD																										
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40	41	

Input Form Left Side

AUTHOR: B. DESCHAPELLES

39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

NAME:

GENERAL INFORMATION

PARTIAL RESTRAINT  
AT END ? 1 FOR YES  
0 FOR NO

SOIL NJ	SOIL TI	SOIL TJ	ANGLE	- G*J			Wn @ NI	Wn @ NJ	1	2
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				
				- 1 3 7 7 6 .	0. 5 5 5	0. 5 5 5				

KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	SUPPORTS		

SPRING			N	KD	SPRING			SPRINGS AT SUPPORTS		

N	LOAD VALUE			N	LOAD VALUE			N	LOAD VALUE			APPLIED NODAL FORCES		

ROTATIONAL G MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS			ELEM	END	ROTATIONAL SPRING MODULUS			SPRINGS AT ELEM. END, 1 OR 2, IF ANY		

39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80

Input Form Right Side

```

e1eme-6ft
5-50
-----
YOU ARE USING COMPUTER PROGRAM BMTORSW,DEVELOPED BY DR. BERNARDO DESCHAPELLES

INPUT DATA FILE NAME IS = 5-5.txt

OUTPUT FILE NAME IS = 5-5o.txt

STORAGE FILE FOR POST-PROCESSING WITH EXCEL = 5-5grf.grf
-----
AISC Design Guide 9, Example 5.5, MC18x42.7 channel
modulus of elasticity of the material= 29000. k/ft2

ELEM nodes inertia length distrib. load AXIAL SOIL NORMAL MODULUS,Ksf angle
i j ft.4 ft at i at j LOAD 1st END 2nd END rad
1 1 3***** 6.00 0.555 0.555***** 0.0 0.0 0.000 00
2 3 5***** 6.00 0.555 0.555***** 0.0 0.0 0.000 00
3 5 7***** 16.00 0.555 0.555***** 0.0 0.0 0.000 00
4 7 9***** 16.00 0.555 0.555***** 0.0 0.0 0.000 00
5 9 11***** 16.00 0.555 0.555***** 0.0 0.0 0.000 00
6 11 13***** 6.00 0.555 0.555***** 0.0 0.0 0.000 00
7 13 15***** 6.00 0.555 0.555***** 0.0 0.0 0.000 00

INPUT DATA RELATED TO THE 2 SUPPORTS
2 1 1 1 115 0 0 1

FINAL SOLUTION FOUND AFTER 1 ITERATIONS

Output of nodal displacements in reference to global axes

node displ. displ. displ. node displ. displ. displ.
along x along y around z along x along y around z
or nonn1 or nonn2 or nonn 3 or nonn1 or nonn2 or nonn 3

1 0.0000E+00 0.0000E+00 0.0000E+00 2 0.0000E+00 0.1835E-03 0.4518E-04
3 0.0000E+00 0.5367E-03 0.1699E-03 4 0.0000E+00 0.1178E-02 0.1172E-03
5 0.0000E+00 0.1939E-02 0.2898E-03 6 0.0000E+00 0.4763E-02 0.5041E-03
7 0.0000E+00 0.7923E-02 0.4174E-03 8 0.0000E+00 0.1118E-01 0.5311E-03
9 0.0000E+00 0.1426E-01 0.3493E-03 10 0.0000E+00 0.1662E-01 0.3555E-03
11 0.0000E+00 0.1850E-01 0.1686E-03 12 0.0000E+00 0.1893E-01 0.6387E-04
13 0.0000E+00 0.1927E-01 0.8589E-04 14 0.0000E+00 0.1944E-01 0.2155E-04
15 0.0000E+00 0.1953E-01 0.0000E+00

-----
OUTPUT OF SOIL REACTIONS,STRESSES AND TRANSVERSE DISPLACEMENTS

```



ELEMENT	1	DISPLACEMENTS IN INCIDENCES		
NODE	1	0.00000E+00	0.00000E+00	0.00000E+00
NODE	2	0.00000E+00	0.18348E-03	0.45178E-04
NODE	3	0.00000E+00	0.53673E-03	0.16991E-03
FORCES ACTING ALONG THE 9 DOF				
NODE	1	0.00000E+00	-0.39960E+02	-0.81373E+03
NODE	2	0.00000E+00	0.33300E+01	-0.31143E-10
NODE	3	0.00000E+00	0.36630E+02	0.59135E+03

ELEMENT 1, FROM NODE 1, TO NODE 3 - LENGTH = 6.00 ft

left half of span, at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-39.96	-39.63	-39.29	-38.96	-38.63	-38.30
bmom, kft	813.73	789.93	766.50	743.41	720.68	698.29
tdisp, ft	0.00000	0.00001	0.00002	0.00005	0.00009	0.00014
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

right half of span, at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-38.30	-37.96	-37.63	-37.30	-36.96	-36.63
bmom, kft	698.29	676.24	654.53	633.14	612.09	591.35
tdisp, ft	0.00014	0.00020	0.00027	0.00035	0.00044	0.00054
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

72in x 0.555kip/in = 39.96kip

ELEMENT	2	DISPLACEMENTS IN INCIDENCES		
NODE	3	0.00000E+00	0.53673E-03	0.16991E-03
NODE	4	0.00000E+00	0.11780E-02	0.11724E-03
NODE	5	0.00000E+00	0.19390E-02	0.28979E-03
FORCES ACTING ALONG THE 9 DOF				
NODE	3	0.00000E+00	-0.36630E+02	-0.59135E+03
NODE	4	0.00000E+00	0.33300E+01	-0.15864E-09
NODE	5	0.00000E+00	0.33300E+02	0.40088E+03

ELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 6.00 ft

left half of span, at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-36.63	-36.30	-35.96	-35.63	-35.30	-34.97
bmom, kft	591.35	570.94	550.84	531.05	511.56	492.38
tdisp, ft	0.00054	0.00064	0.00076	0.00088	0.00101	0.00115
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

right half of span, at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-34.97	-34.63	-34.30	-33.97	-33.63	-33.30
bmom, kft	492.38	473.50	454.91	436.62	418.61	400.88
tdisp, ft	0.00115	0.00129	0.00144	0.00160	0.00177	0.00194
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

ELEMENT	3	DISPLACEMENTS IN INCIDENCES		
NODE	5	0.00000E+00	0.19390E-02	0.28979E-03

Page 2  
 $(-D_i + D_j) * GJ + V_j * L - 3.33xL/2 = M_j - M_i$ , where  $GJ=13776$   
 $-0.00054 + 0.00194)13776 - 33.3 * 6 - 9.99 = 19.29 - 199.8 - 9.99 = -189.98$   
 $400.88 - 591.35 = -190.47$  OK

5-50				
NODE	6	0.00000E+00	0.47633E-02	0.50413E-03
NODE	7	0.00000E+00	0.79232E-02	0.41740E-03
FORCES ACTING ALONG THE 9 DOF				
NODE	5	0.00000E+00	-0.33300E+02	-0.40088E+03
NODE	6	0.00000E+00	0.88800E+01	0.41688E-10
NODE	7	0.00000E+00	0.24420E+02	0.21560E+02
ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 16.00 ft				
left half of span,at tenth points of length				
	span	span	span	span
	0.0	0.1	0.2	0.3
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-33.30	-32.41	-31.52	-30.64
bmom,kft	400.88	354.97	310.99	268.88
tdisp,ft	0.00194	0.00242	0.00294	0.00350
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
	span	span	span	span
	0.4	0.5		
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-29.75	-28.86		
bmom,kft	228.57	190.01		
tdisp,ft	0.00408	0.00468		
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
right half of span,at tenth points of length				
	span	span	span	span
	0.5	0.6	0.7	0.8
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-28.86	-27.97	-27.08	-26.20
bmom,kft	190.01	153.14	117.91	84.26
tdisp,ft	0.00468	0.00530	0.00594	0.00660
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
	span	span	span	span
	0.9	1.0		
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-25.31	-24.42		
bmom,kft	52.16	21.56		
tdisp,ft	0.00726	0.00792		
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
-----				
ELEMENT	4	DISPLACEMENTS IN INCIDENCES		
NODE	7	0.00000E+00	0.79232E-02	0.41740E-03
NODE	8	0.00000E+00	0.11183E-01	0.53113E-03
NODE	9	0.00000E+00	0.14257E-01	0.34926E-03
FORCES ACTING ALONG THE 9 DOF				
NODE	7	0.00000E+00	-0.24420E+02	-0.21560E+02
NODE	8	0.00000E+00	0.88800E+01	0.11555E-09
NODE	9	0.00000E+00	0.15540E+02	-0.21087E+03
ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 16.00 ft				
left half of span,at tenth points of length				
	span	span	span	span
	0.0	0.1	0.2	0.3
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-24.42	-23.53	-22.64	-21.76
bmom,kft	21.56	-7.59	-35.34	-61.71
tdisp,ft	0.00792	0.00859	0.00926	0.00992
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
	span	span	span	span
	0.4	0.5		
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-20.87	-19.98		
bmom,kft	-86.75	-110.49		
tdisp,ft	0.01058	0.01123		
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
right half of span,at tenth points of length				
	span	span	span	span
	0.5	0.6	0.7	0.8
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-19.98	-19.09	-18.20	-17.32
bmom,kft	-110.49	-132.97	-154.22	-174.27
tdisp,ft	0.01123	0.01187	0.01249	0.01310
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
	span	span	span	span
	0.9	1.0		
soil,k/ft	0.000	0.000	0.000	0.000
shear,k	-16.43	-15.54		
bmom,kft	-193.14	-210.87		
tdisp,ft	0.01369	0.01426		
axial,k	0.00	AT 1st END	and	0.00 AT 2nd END
-----				
ELEMENT	5	DISPLACEMENTS IN INCIDENCES		
NODE	9	0.00000E+00	0.14257E-01	0.34926E-03
NODE	10	0.00000E+00	0.16622E-01	0.35553E-03
NODE	11	0.00000E+00	0.18503E-01	0.16864E-03
FORCES ACTING ALONG THE 9 DOF				

5-50  
 NODE 9 0.00000E+00 -0.15540E+02 0.21087E+03  
 NODE 10 0.00000E+00 0.88800E+01 0.79967E-10  
 NODE 11 0.00000E+00 0.66600E+01 -0.32998E+03

ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 16.00 ft

left half of span,at tenth points of length  
 span span span span span span span  
 0.0 0.1 0.2 0.3 0.4 0.5  
 soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000  
 shear,k -15.54 -14.65 -13.76 -12.88 -11.99 -11.10  
 bmom,kft -210.87 -227.48 -242.99 -257.43 -270.82 -283.17  
 tdisp,ft 0.01426 0.01480 0.01533 0.01583 0.01630 0.01674  
 axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length  
 span span span span span span span  
 0.5 0.6 0.7 0.8 0.9 1.0  
 soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000  
 shear,k -11.10 -10.21 -9.32 -8.44 -7.55 -6.66  
 bmom,kft -283.17 -294.50 -304.84 -314.19 -322.56 -329.98  
 tdisp,ft 0.01674 0.01716 0.01754 0.01790 0.01822 0.01850  
 axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----  
 ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13  
 NODE 11 0.00000E+00 0.18503E-01 0.16864E-03  
 NODE 12 0.00000E+00 0.18927E-01 0.63868E-04  
 NODE 13 0.00000E+00 0.19268E-01 0.85888E-04  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 11 0.00000E+00 -0.66600E+01 0.32998E+03  
 NODE 12 0.00000E+00 0.33300E+01 0.21120E-08  
 NODE 13 0.00000E+00 0.33300E+01 -0.34940E+03

ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 6.00 ft

left half of span,at tenth points of length  
 span span span span span span span  
 0.0 0.1 0.2 0.3 0.4 0.5  
 soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000  
 shear,k -6.66 -6.33 -5.99 -5.66 -5.33 -4.99  
 bmom,kft -329.98 -332.51 -334.92 -337.19 -339.32 -341.33  
 tdisp,ft 0.01850 0.01860 0.01870 0.01878 0.01887 0.01895  
 axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length  
 span span span span span span span  
 0.5 0.6 0.7 0.8 0.9 1.0  
 soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000  
 shear,k -4.99 -4.66 -4.33 -4.00 -3.66 -3.33  
 bmom,kft -341.33 -343.20 -344.95 -346.56 -348.04 -349.40  
 tdisp,ft 0.01895 0.01902 0.01909 0.01916 0.01921 0.01927  
 axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----  
 ELEMENT 7 DISPLACEMENTS IN INCIDENCES 13 14 15  
 NODE 13 0.00000E+00 0.19268E-01 0.85888E-04  
 NODE 14 0.00000E+00 0.19441E-01 0.21550E-04  
 NODE 15 0.00000E+00 0.19527E-01 0.00000E+00  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 13 0.00000E+00 -0.33300E+01 0.34940E+03  
 NODE 14 0.00000E+00 0.33300E+01 -0.54570E-11  
 NODE 15 0.00000E+00 -0.86629E-10 -0.35583E+03

Page 4  
 (-Di + Dj)\* GJ + Vj\*L - 3.33L/2 = Mj-Mi, where GJ=13776  
 -0.01850+0.01927)\*13776-3.33\*6-9.99=10.6-19.98- 9.99= -19.37  
 -349.40+329.98=-19.42 OK

5-50

ELEMENT 7, FROM NODE 13, TO NODE 15 - LENGTH = 6.00 ft

left half of span, at tenth points of length

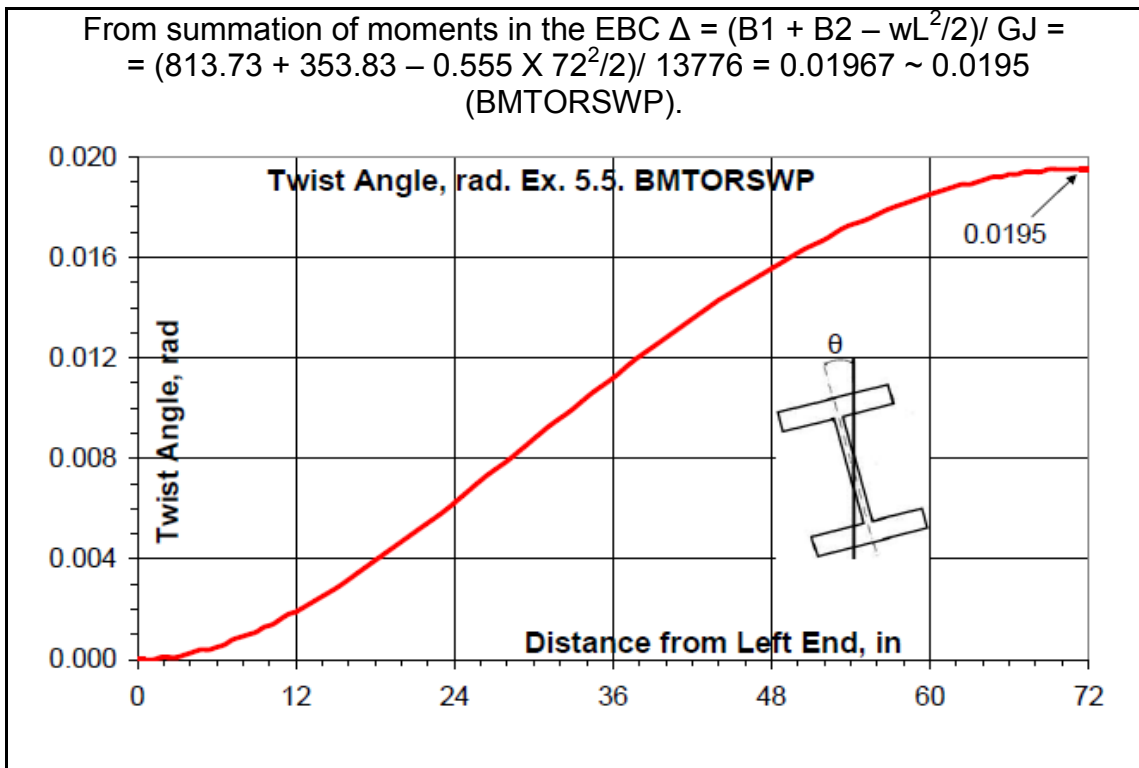
	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-3.33	-3.00	-2.66	-2.33	-2.00	-1.67
bmom, kft	-349.40	-350.62	-351.71	-352.68	-353.51	-354.22
tdisp, ft	0.01927	0.01932	0.01936	0.01940	0.01943	0.01946
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

right half of span, at tenth points of length

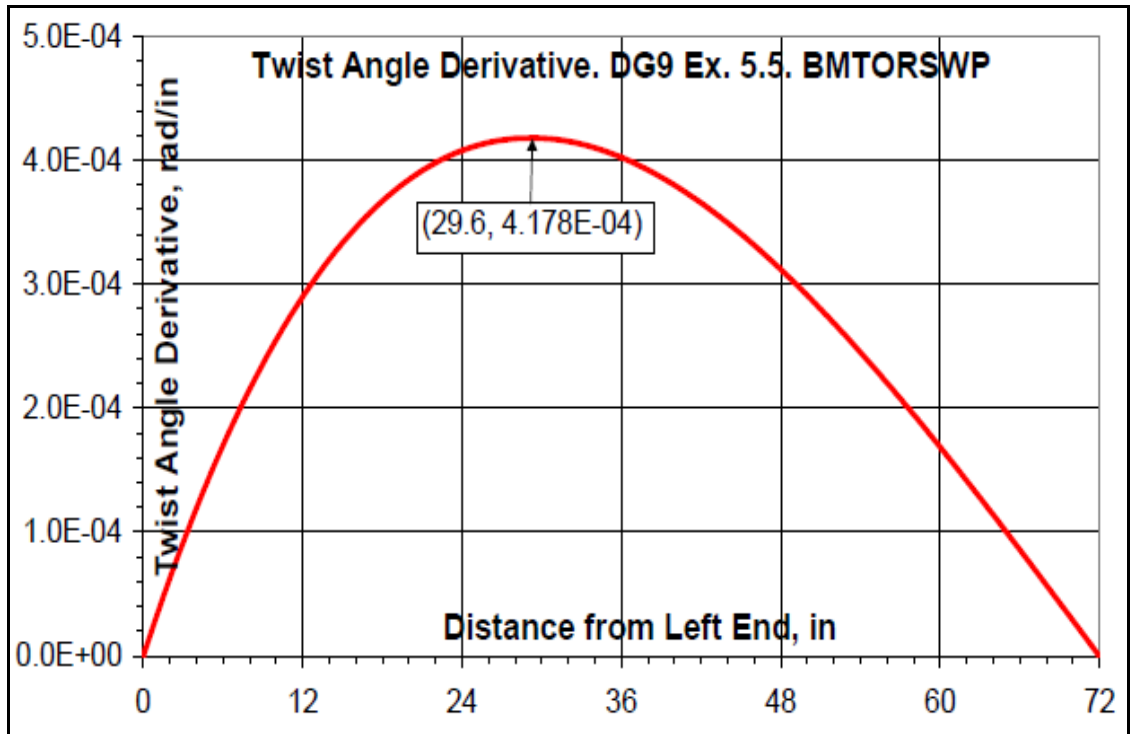
	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-1.67	-1.33	-1.00	-0.67	-0.33	0.00
bmom, kft	-354.22	-354.80	-355.25	-355.57	-355.76	-355.83
tdisp, ft	0.01946	0.01949	0.01950	0.01952	0.01952	0.01953
axial, k	0.00 AT 1st END and			0.00 AT 2nd END		

Local extreme twist angle and bi-moment in the TWB. Local extreme analog transverse displacement and moment in the EBC

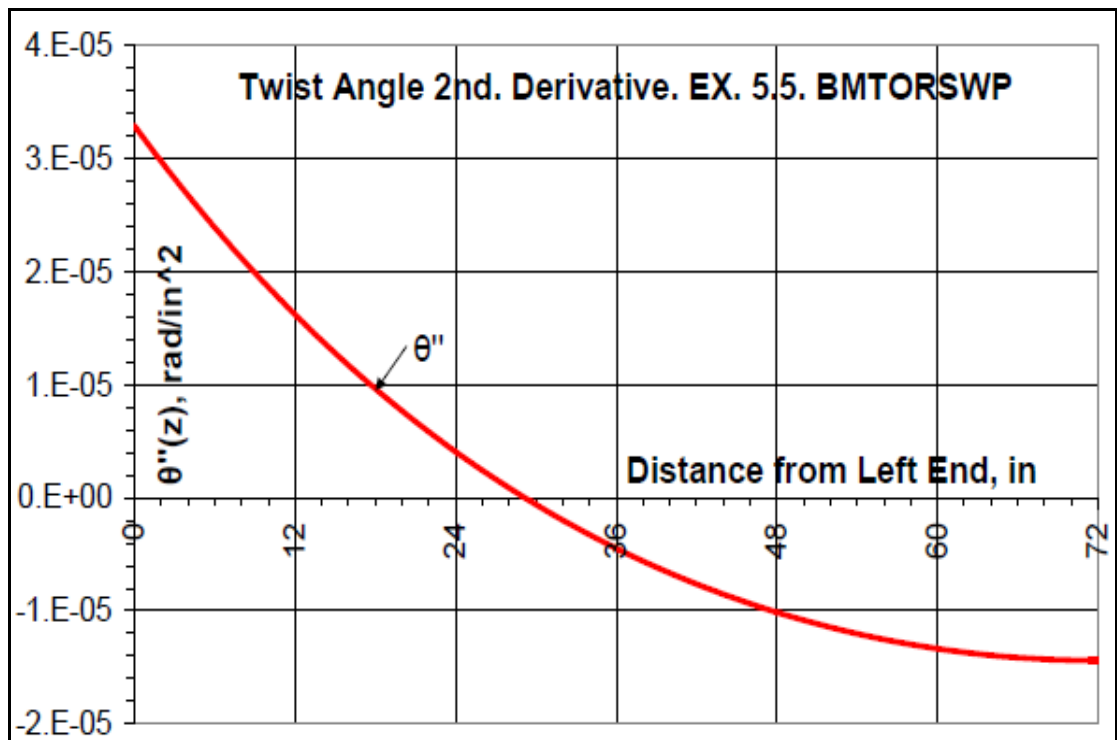
Output Page 5



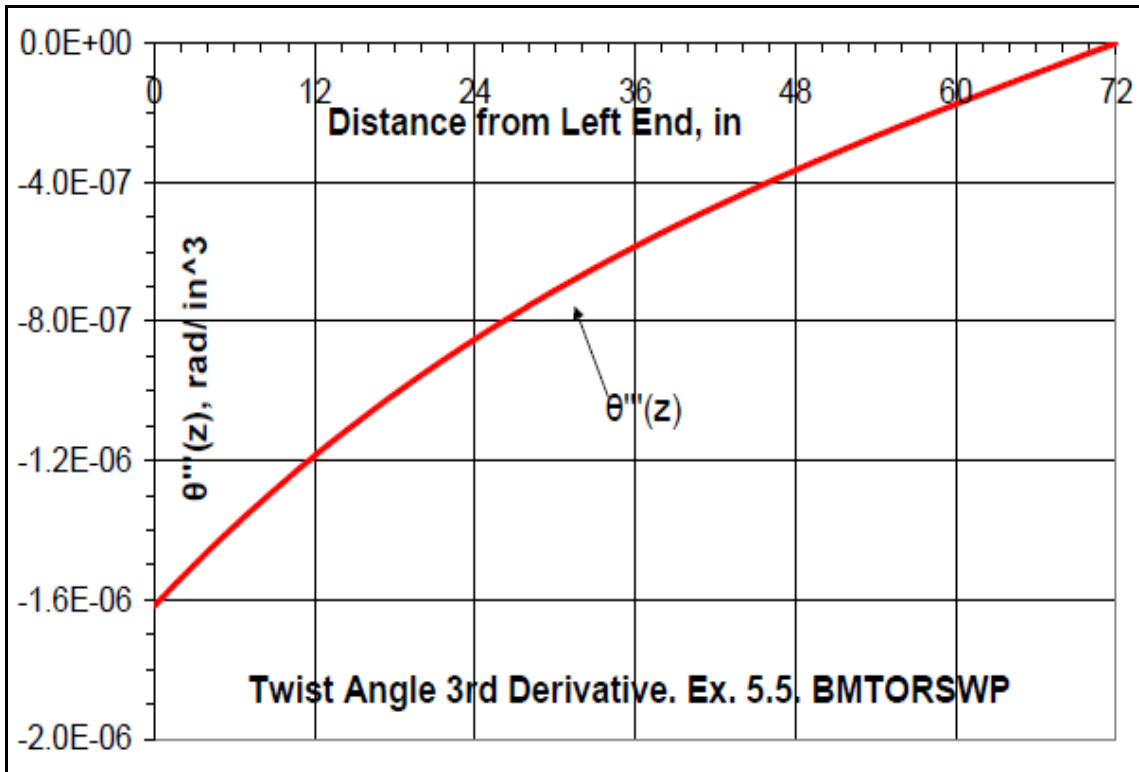
Twist Angle, Ex. 5.5 from BMTORSWP



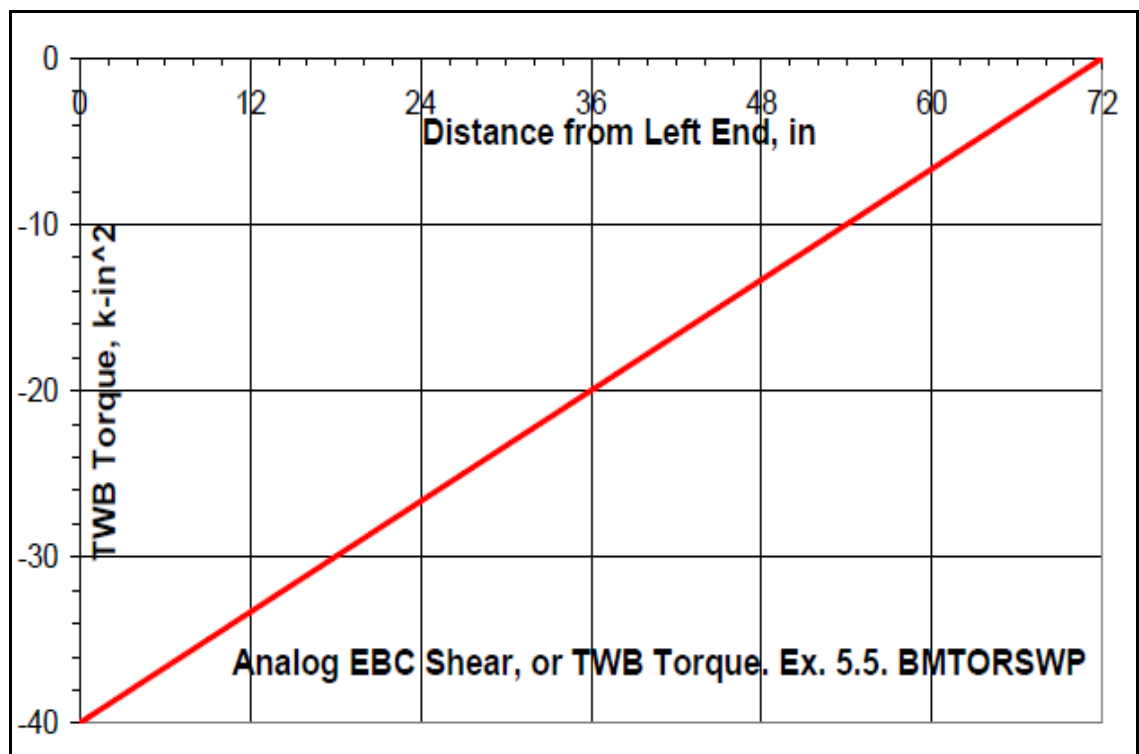
Twist Angle Derivative of DG9 Ex. 5.5 from BMTORSWP



Twist Angle 2<sup>nd</sup> Derivative of DG9 Ex. 5.5 from BMTORSWP

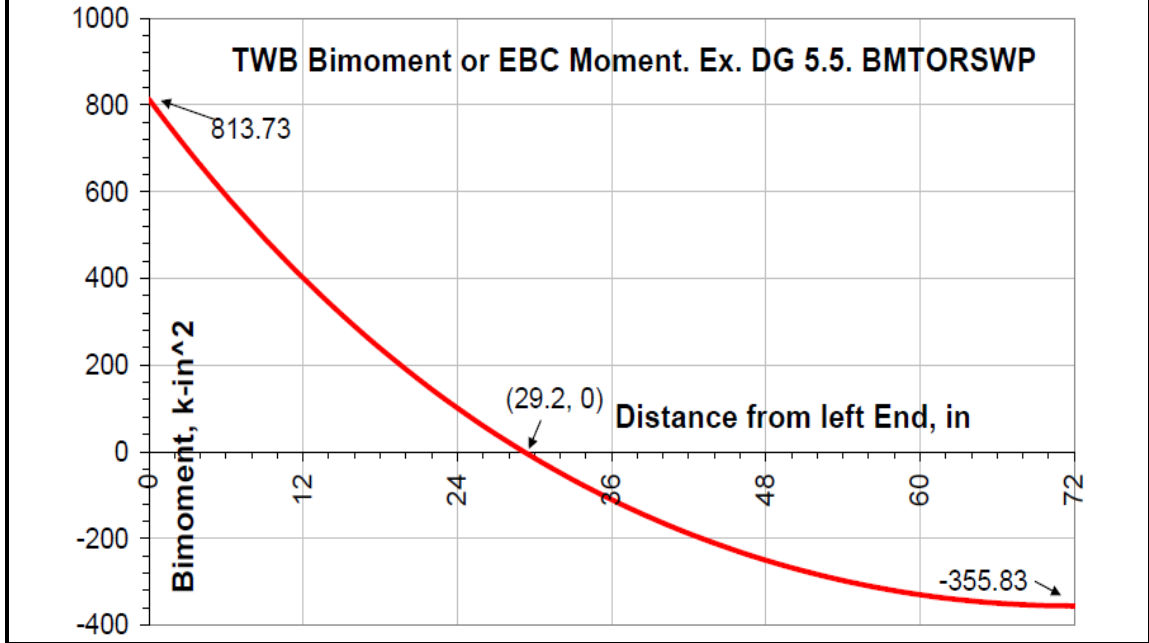


Twist Angle 3<sup>rd</sup> Derivative of DG9 Ex. 5.5 from BMTORSWP

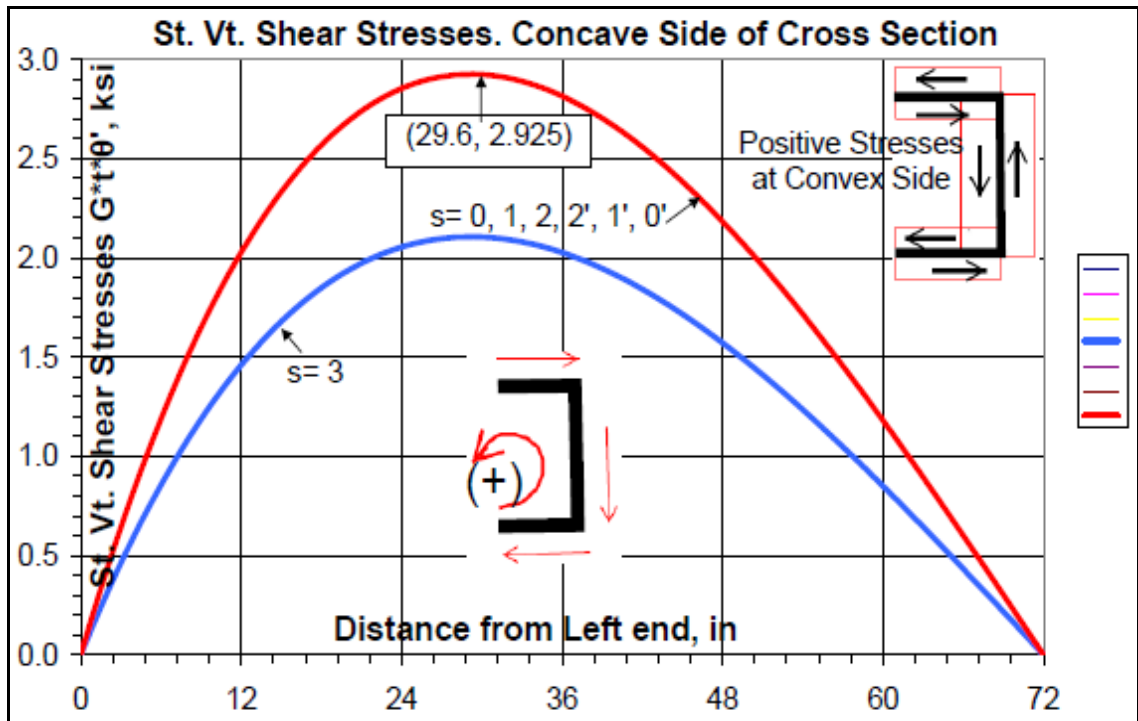


BC Shear and TWB Torque of DG9 Ex. 5.5 from BMTORSWP

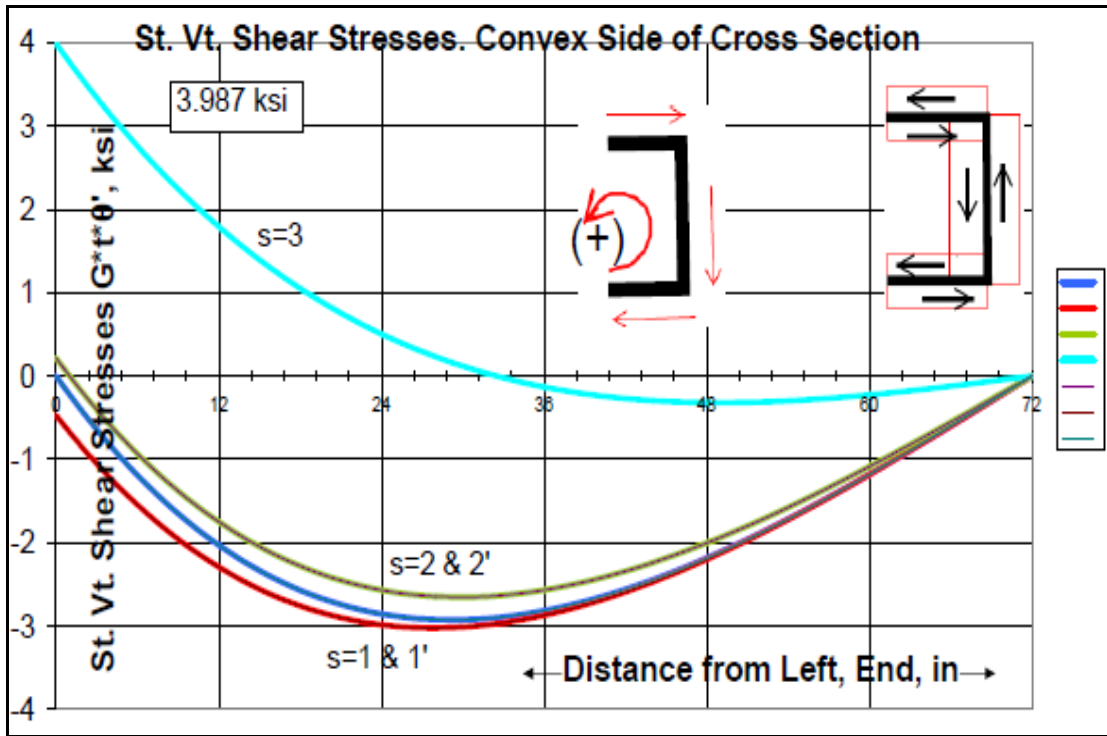
From summation of moments in the EBC:  $\Delta = (B1 + B2 - wL^2/2) / GJ = (813.73 + 353.83 - 0.555 \times 722/2) / 13776 = 0.01967 \sim 0.0195$  (BMTORSWP).



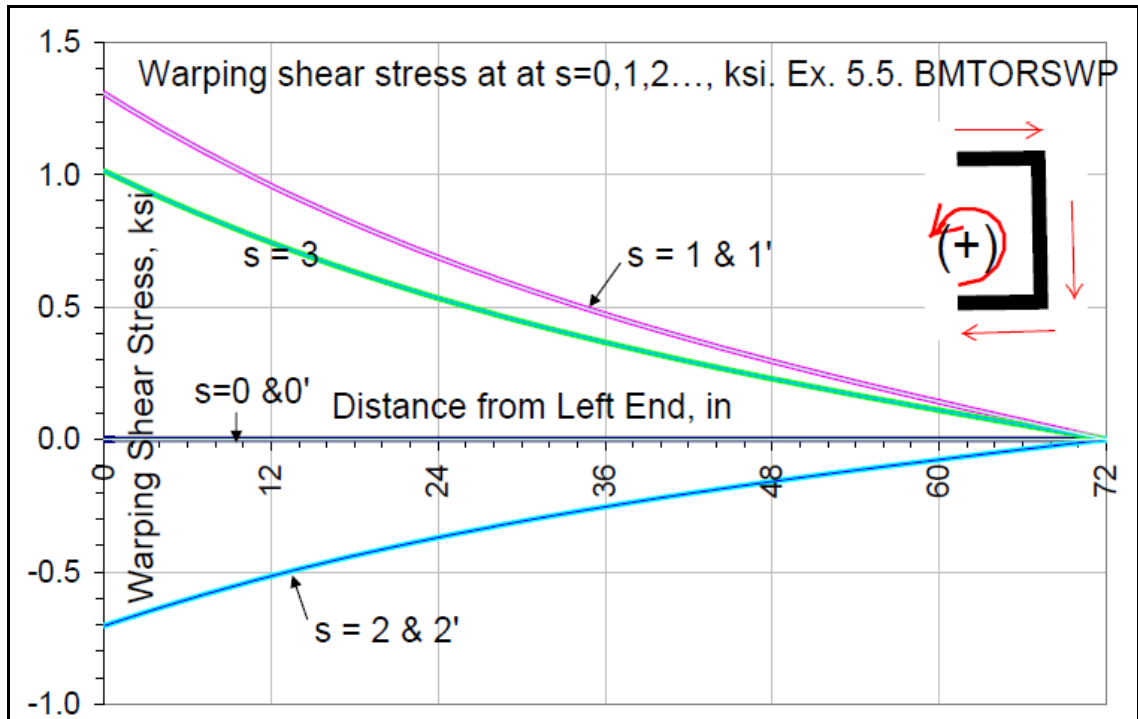
TWB Bimoment and BC Moment of Ex. DG 5.5 from BMTORSWP



St. Vt. Shear Stresses in Concave Side of Cross Section

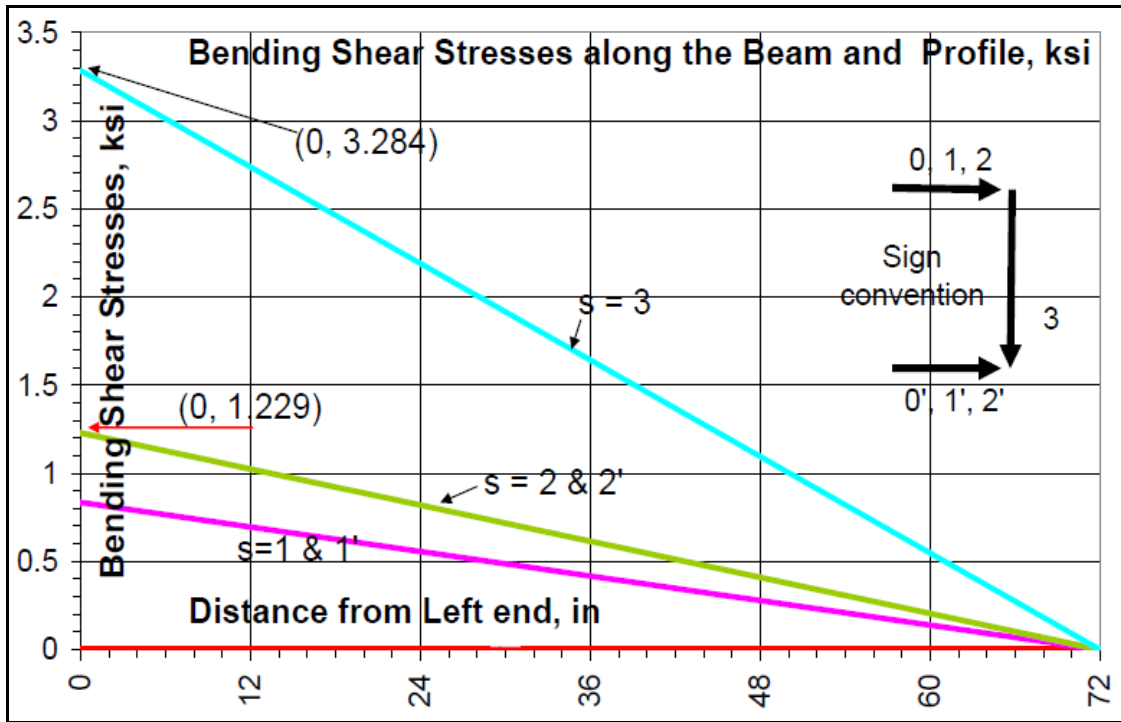


St. Vt. Shear Stresses in Convex Side of Cross Section



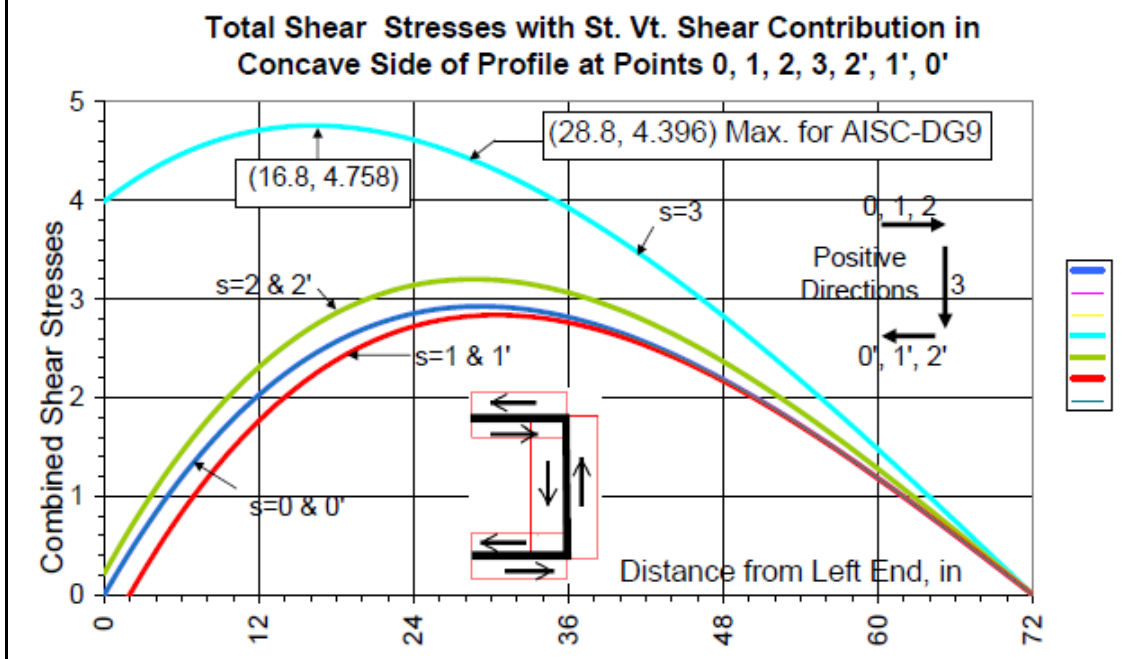
Warping Shear Stress at s = 0,1,2,3 of Ex. 5.5 from BMTORSWP



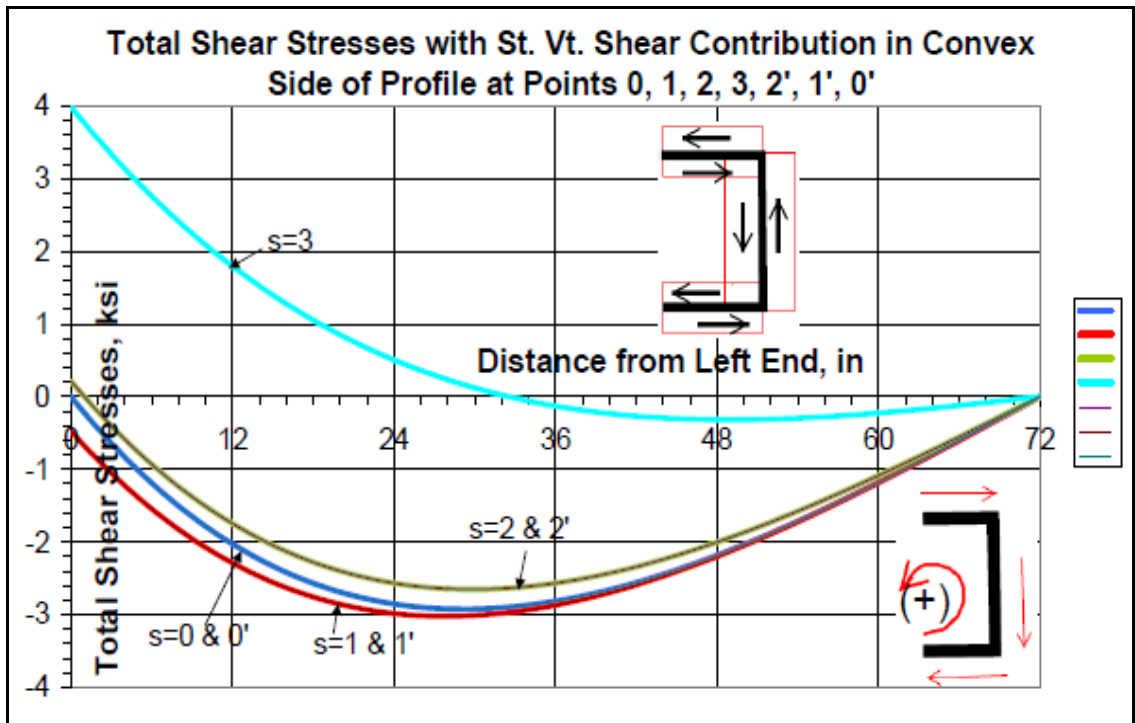


Bending Shear Stresses along the Beam and Profile

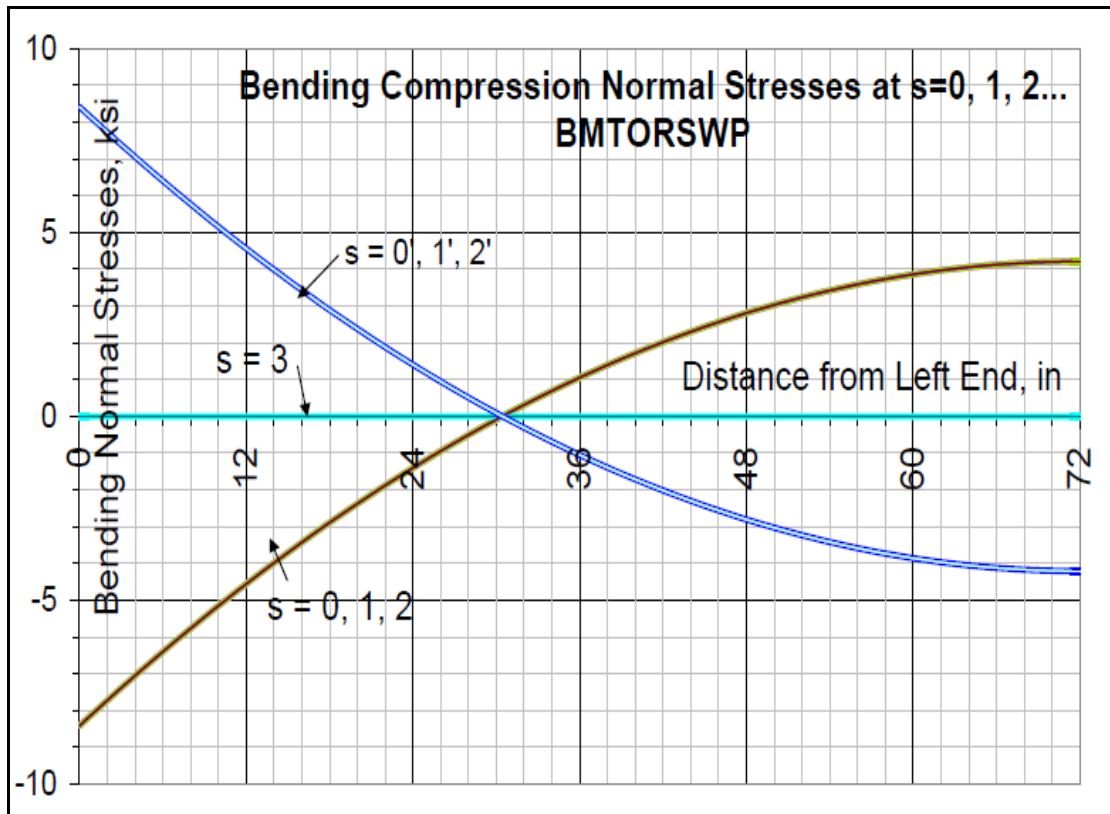
At 23.2in from the left end of the beam at points  $s = 2$  and  $2'$  of the profile occurs the highest combined shear stress unnoticed by the AISC DG9 and shown here:



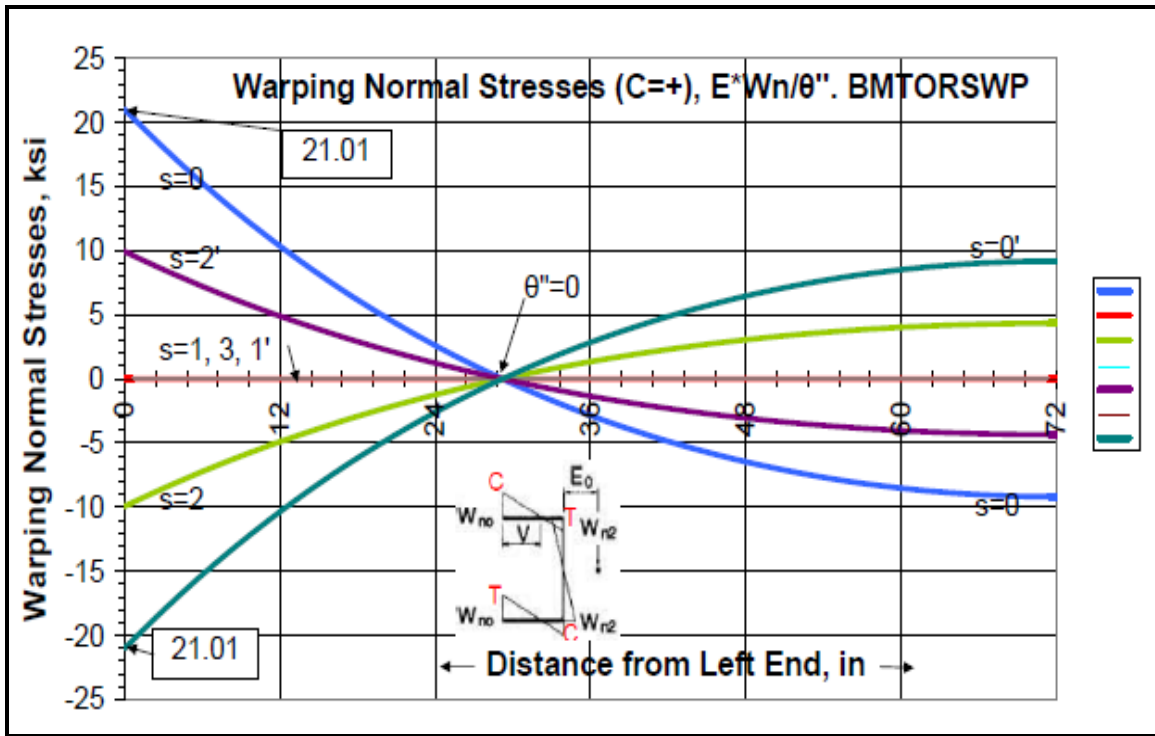
Total Shear Stresses in Concave Side of Profile



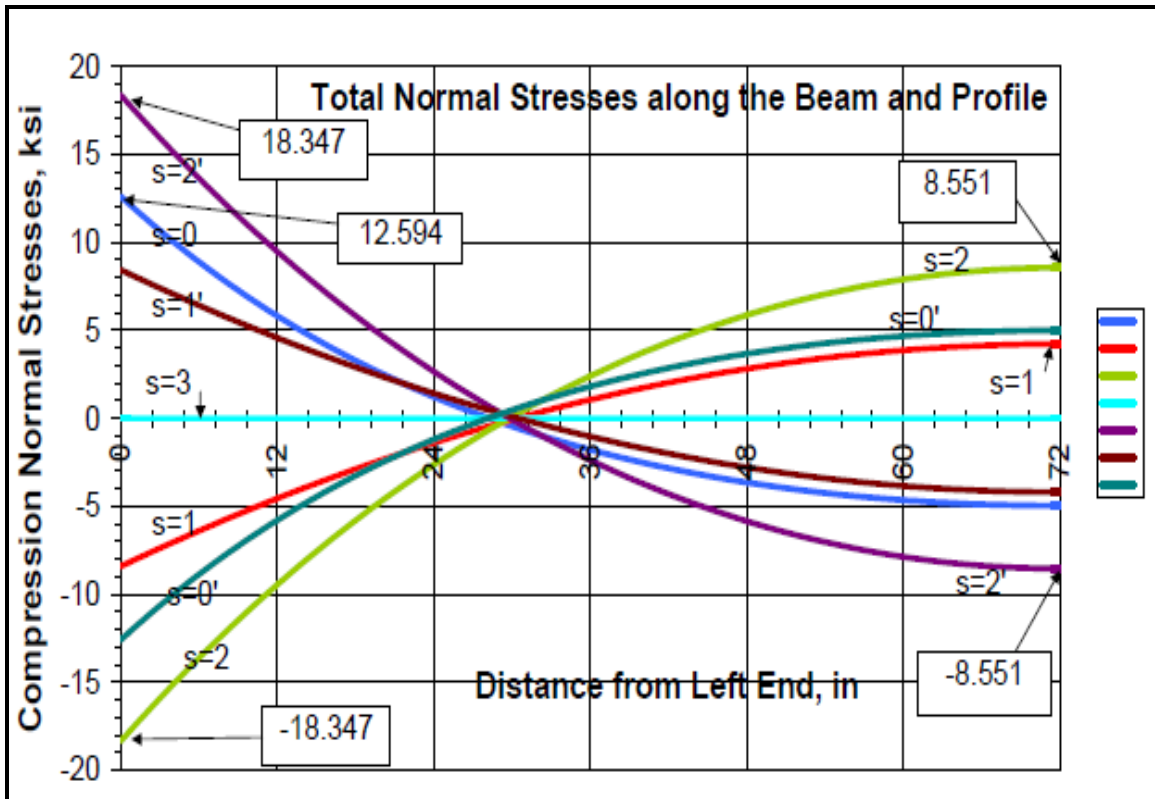
Total Shear Stresses in Convex Side of Profile



Bending Normal Stresses at s = 0, 1, 2, 3 from BMTORSWP



Warping Normal Stresses along Beam and Profile



Combined Normal Stresses along the Beam and Profile

### Numeric Checks

Rows reflect interest points along the axial coordinate, and columns reflect orderly the torque angle and 3 derivatives. Entry arrays begin from zero; thus  $\theta_{03}$  in the first row and fourth column is the 3<sup>rd</sup> derivative at the interest point 0

$$\theta = \begin{pmatrix} 0 & 8.609 \times 10^{-3} & 0.018 & 0 \\ 0 & 4.061 \times 10^{-4} & 0 & 0 \\ 3.147 \times 10^{-5} & 0 & -1.368 \times 10^{-5} & 3.147 \times 10^{-5} \\ -1.614 \times 10^{-6} & -7.422 \times 10^{-7} & 0 & -1.614 \times 10^{-6} \end{pmatrix} \quad \text{AISC}$$

$$\theta_{\text{Bmt}} := \begin{pmatrix} 0 & 8.3 \times 10^{-3} & 1.95 \times 10^{-2} & 0 \\ 0 & 4.176 \times 10^{-4} & 0 & 0 \\ 3.293 \times 10^{-5} & 2.825 \times 10^{-7} & -1.44 \times 10^{-5} & 3.293 \times 10^{-5} \\ -1.617 \times 10^{-6} & -7.372 \times 10^{-7} & -2.073 \times 10^{-11} & -1.617 \times 10^{-6} \end{pmatrix} \quad \text{BMTORSWP}$$

Torque Angles & its 3 Dimensionless Derivatives along the Span

**Shear stress due to pure torsion  $\tau_t = GJ\theta'$ . Pure torsion**

shear stresses appear in matrix rows containing points of interest 0, 1, 2, 3 of the profile. Each matrix column contains the defined points of interest along the axial coordinate. Points 0, 1 and 2 of the profile have the same pure torsion shear stress because they have the same thickness

$$\tau_{\text{tBmt},j,i} := G \cdot t_i \cdot \frac{\theta_{\text{Bmt},1,j}}{\text{in}} \quad \tau_{t,j,i} := G \cdot t_i \cdot \frac{\theta_{1,j}}{\text{in}}$$

$$\tau_{\text{tBmt}} = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 2.923 & 2.923 & 2.923 & 2.105 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix} \cdot \text{ksi} \quad \begin{array}{l} \text{BMTORSW at} \\ \text{support} \\ 0.2L \\ 0.5L \\ L \end{array}$$

$$\tau_t = \begin{pmatrix} 0 & 0 & 0 & 0 \\ 2.843 & 2.843 & 2.843 & 2.047 \\ 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix} \cdot \text{ksi} \quad \begin{array}{l} \text{AISC-DG9 at} \\ \text{support} \\ 0.2L \\ 0.5L \\ L \end{array}$$

Shear Stresses due to Pure Torsion along the Span and Profile

Shear stress from warping at point "s" of the profile. Warping shear stresses along one profile appear in matrix rows containing points of interest 0, 1, 2, 3 of the profile. Matrix columns contain the defined points of interest along the axial coordinate

$$\tau_{wBmt_{j,i}} := \frac{-E}{t_i} \cdot S_{w_i} \cdot \left( \theta_{Bmt_{3,j}} \cdot \frac{1}{in^3} \right) \quad \tau_{w_{j,i}} := \frac{-E}{t_i} \cdot S_{w_i} \cdot \left( \theta_{3,j} \cdot \frac{1}{in^3} \right)$$

$$\tau_{wBmt} = \begin{pmatrix} 0 & 1.306 & -1.013 & 0.703 \\ 0 & 0.595 & -0.462 & 0.321 \\ 0 & 0 & -0 & 0 \\ 0 & 1.306 & -1.013 & 0.703 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at  
support  
0.2L  
0.5L  
L

$$\tau_w = \begin{pmatrix} 0 & 1.303 & -1.011 & 0.702 \\ 0 & 0.599 & -0.465 & 0.323 \\ 0 & 0 & 0 & 0 \\ 0 & 1.303 & -1.011 & 0.702 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at  
support  
0.2L  
0.5L  
L

Shear Stresses due to Restrained Warping along the Span and Profile

Warping normal stresses  $\sigma_{ws} = E \cdot W_{ns} \cdot \theta$  along one profile appear in a matrix row containing points of interest 0, 1, 2, 3 of the profile. Each matrix column contains the defined points of interest along the axial coordinate.

$$\sigma_{w_{j,i}} := E \cdot W_{n_i} \cdot \left( \theta_{2,j} \cdot \frac{1}{in^2} \right) \quad \sigma_{wBmt_{j,i}} := E \cdot W_{n_i} \cdot \left( \theta_{Bmt_{2,j}} \cdot \frac{1}{in^2} \right)$$

$$\sigma_{wBmt} = \begin{pmatrix} 21.009 & 0 & -9.932 & 0 \\ 0.18 & 0 & -0.085 & 0 \\ -9.187 & 0 & 4.343 & 0 \\ 21.009 & 0 & -9.932 & 0 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at  
support  
0.2L  
0.5L  
L

$$\sigma_w = \begin{pmatrix} 20.078 & 0 & -9.491 & 0 \\ 0 & 0 & 0 & 0 \\ -8.729 & 0 & 4.127 & 0 \\ 20.078 & 0 & -9.491 & 0 \end{pmatrix} \cdot \text{ksi}$$

AISC-DG9 at  
support  
0.2L  
0.5L  
L

Normal Stresses due to Restrained Warping along the Span and Profile

service-load torque,  $T$ , where  $T = T_u / LF$ : and  $LF := \frac{W_u}{W} = 1.5$

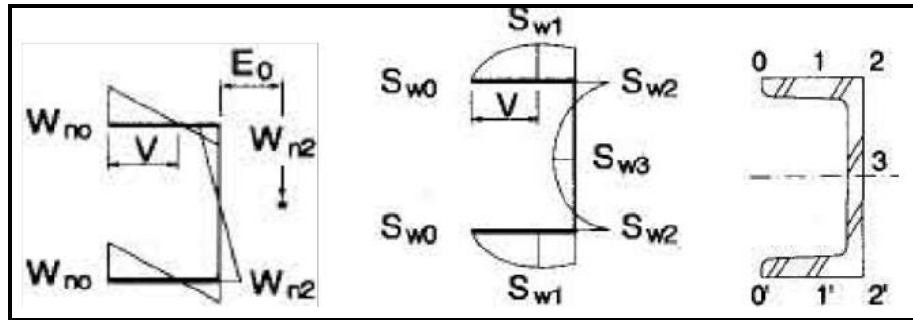
AISC

$$\frac{\max\left[\left(\theta^T\right)^{\langle 0 \rangle}\right]}{LF} = 1.230 \times 10^{-2} \cdot \text{rad}$$

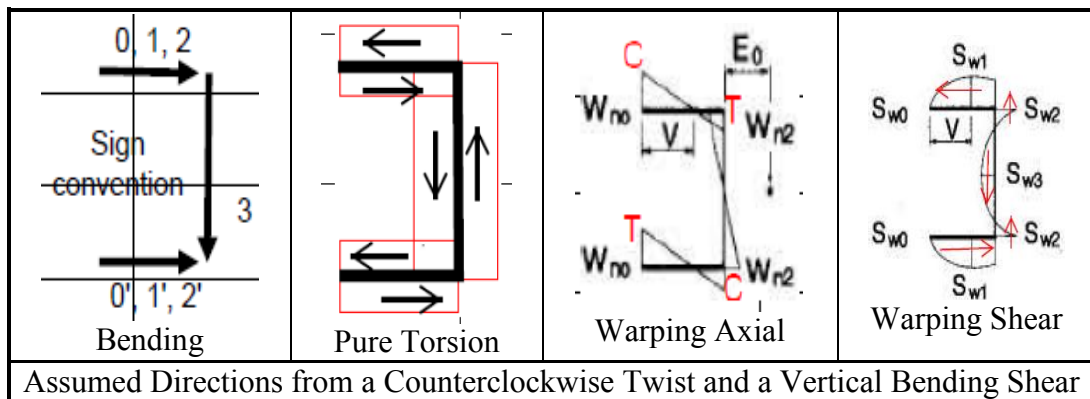
BMTORSW

$$\frac{\max\left[\left(\theta_{Bmt}^T\right)^{\langle 0 \rangle}\right]}{LF} = 1.300 \times 10^{-2} \cdot \text{rad}$$

Maximum Service Load Rotation



Profile Key Points for Stress Computations



Assumed Directions from a Counterclockwise Twist and a Vertical Bending Shear

Positive Directions

Location	Point	$\sigma_w$	$\sigma_b$	$f_{un}$
Support	0	20.1(C)	8.41(T)	11.7(C) 12.594
	1	0	8.41(T)	8.41(T) -8.416
	2	9.49(T)	8.41(T)	17.9(T) -18.347
	3	0	0	0 0.
Midspan	0	8.73(T)	4.20(C)	4.53(T)
	1	0	4.20(C)	4.20(C)
	2	4.13(C)	4.20(C)	8.33(C)
	3	0	0	0
$z/l = 0.20$	0	0	—	—
	1	0	—	—
	2	0	—	—
	3	0	—	—
Maximum				17.9(T) -18.347

Comparison of Total Normal Stresses along the Span and Profile

$$f_{unBmt} := \sigma_w Bmt + \sigma_b$$

$$f_{unBmt} = \begin{pmatrix} 12.594 & -8.416 & -18.347 & 0 \\ -0.156 & -0.337 & -0.422 & 0 \\ -4.979 & 4.208 & 8.551 & 0 \\ 12.594 & -8.416 & -18.347 & 0 \end{pmatrix} \cdot \text{ksi}$$

$$f_{un} := \sigma_w + \sigma_b$$

$$f_{un} = \begin{pmatrix} 11.662 & -8.416 & -17.907 & 0 \\ -0.337 & -0.337 & -0.337 & 0 \\ -4.522 & 4.208 & 8.334 & 0 \\ 11.662 & -8.416 & -17.907 & 0 \end{pmatrix} \cdot \text{ksi}$$

BMTORSW at support

0.2L

0.5L

L

AISC-DG9 at support

0.2L

0.5L

L

The maximum normal stress (tension) occurs at the support at point 2 in the flange. A discrepancy of 2.5% is noticed: 17.907 ksi and 18.347 ksi from DG9 and the software respectively. The AISC DG9 data have been recalculated to the third decimal place to be compared with BMTORSWP output calculated to the third decimal place.

Another Comparison of Total Normal Stresses along the Span and Profile

Location	Point	$\tau_t$	$\tau_w$	$\tau_b$	$f_{uv}$
Support	0	0	0	0	0
	1	0	1.30←	—	1.30←
	2	0	1.01←	1.23→	0.22→
	3	0	0.702↓	3.28↓	3.98↓
Midspan	0	0	0	0	0
	1	0	0	0	0
	2	0	0	0	0
	3	0	0	0	0
$z/l = 0.20$	0	2.84←→	0	0	2.84←→ 2.923
	1	2.84←→	0.599←	—	3.44← 3.158
	2	2.84←→	0.465←	0.740→	3.12→ 3.199
	3	2.05↑↓	0.323↓	1.98↓	4.35↓ 4.396
<b>Maximum</b>					<b>4.35↓ 4.396</b>

The maximum shear stress occurs at  $z/L = 0.20$  at point 3 in the web. A discrepancy of 1.3% is noticed: 4.340 ksi and 4.396 ksi from DG9 and the software respectively. The AISC DG9 data have been recalculated to the third decimal place to be compared with BMTORSWP output calculated to the third decimal place.

$$f_{uv} := \tau_t + \tau_w + \tau_b$$

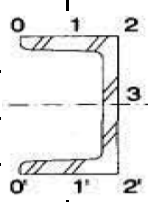
$$f_{uvBmt} := \tau_{tBmt} + \tau_{wBmt} + \tau_b$$

$$f_{uv} = \begin{pmatrix} 0 & 1.303 & 0.218 & 3.986 \\ 2.843 & 3.442 & 3.115 & 4.34 \\ 0 & 0 & 0 & 0 \\ 0 & 1.303 & -2.24 & -2.582 \end{pmatrix} \cdot \text{ksi} \quad \begin{matrix} \text{BMTORSW at} \\ \text{support} \\ 0.2L \\ 0.5L \\ L \end{matrix}$$

Comparison of Total Shear Stresses along the Span and Profile



Torque, Bimoment, Twist Angle and Derivatives

									
						G = 11200ksi			
				9		E = 29000ksi			
		9.6				Cw = 852in <sup>6</sup>			
		12							
Elmn	z	θ	Torq	Bmnt	z	θ'	z	θ''	
1									
	0	0	-39.96	813.7299	0	0.000E+00	0	3.293E-05	
	0.6	0	-39.63	789.9347	0.6	1.947E-05	0.6	3.197E-05	
	1.2	0	-39.29	766.4978	1.2	3.837E-05	1.2	3.102E-05	
	1.8	0.0001	-38.96	743.4146	1.8	5.670E-05	1.8	3.009E-05	
	2.4	0.0001	-38.63	720.6804	2.4	7.448E-05	2.4	2.917E-05	
	3	0.0001	-38.3	698.2907	3	9.170E-05	3	2.826E-05	
	3.6	0.0002	-37.96	676.2409	3.6	1.084E-04	3.6	2.737E-05	
	4.2	0.0003	-37.63	654.5267	4.2	1.246E-04	4.2	2.649E-05	
	4.8	0.0004	-37.3	633.1436	4.8	1.402E-04	4.8	2.563E-05	
	5.4	0.0004	-36.96	612.0874	5.4	1.553E-04	5.4	2.477E-05	
	6	0.0005	-36.63	591.354	6	1.699E-04	6	2.393E-05	
2									
	6	0.0005	-36.63	591.354	6	1.699E-04	6	2.393E-05	
	6.6	0.0006	-36.3	570.939	6.6	1.840E-04	6.6	2.311E-05	
	7.2	0.0008	-35.96	550.8384	7.2	1.976E-04	7.2	2.229E-05	
	7.8	0.0009	-35.63	531.0482	7.8	2.108E-04	7.8	2.149E-05	
	8.4	0.001	-35.3	511.5643	8.4	2.234E-04	8.4	2.070E-05	
	9	0.0011	-34.97	492.383	9	2.356E-04	9	1.993E-05	
	9.6	0.0013	-34.63	473.5003	9.6	2.474E-04	9.6	1.916E-05	
	10.2	0.0014	-34.3	454.9124	10.2	2.586E-04	10.2	1.841E-05	
	10.8	0.0016	-33.97	436.6157	10.8	2.695E-04	10.8	1.767E-05	
	11.4	0.0018	-33.63	418.6064	11.4	2.798E-04	11.4	1.694E-05	
	12	0.0019	-33.3	400.8809	12	2.898E-04	12	1.622E-05	
3									
	12	0.0019	-33.3	400.8809	12	2.898E-04	12	1.622E-05	
	13.6	0.0024	-32.41	354.9737	13.6	3.142E-04	13.6	1.437E-05	
	15.2	0.0029	-31.52	310.9942	15.2	3.358E-04	15.2	1.259E-05	
	16.8	0.0035	-30.64	268.8796	16.8	3.546E-04	16.8	1.088E-05	
	18.4	0.0041	-29.75	228.5698	18.4	3.707E-04	18.4	9.251E-06	
	20	0.0047	-28.86	190.0072	20	3.842E-04	20	7.690E-06	
	21.6	0.0053	-27.97	153.1369	21.6	3.953E-04	21.6	6.198E-06	
	23.2	0.0059	-27.08	117.9061	23.2	4.041E-04	23.2	4.772E-06	
	24.8	0.0066	-26.2	84.2646	24.8	4.106E-04	24.8	3.411E-06	
	26.4	0.0073	-25.31	52.1643	26.4	4.150E-04	26.4	2.111E-06	

(continued) Torque, Bimoment, Twist Angle and Derivatives

	28	0.0079	-24.42	21.5595	28	4.174E-04	28	8.720E-07
4								
	28	0.0079	-24.42	21.5595	28	4.174E-04	28	8.721E-07
	28.8	0.0083	-23.98	6.983	28.8	4.176E-04	28.8	2.825E-07
	29.6	0.0086	-23.53	-7.5935	29.6	4.178E-04	29.6	-3.072E-07
	31.2	0.0093	-22.64	-35.3365	31.2	4.164E-04	31.2	-1.430E-06
	32.8	0.0099	-21.76	-61.7089	32.8	4.133E-04	32.8	-2.497E-06
	34.4	0.0106	-20.87	-86.7484	34.4	4.085E-04	34.4	-3.511E-06
	36	0.0112	-19.98	-110.491	36	4.021E-04	36	-4.472E-06
	37.6	0.0119	-19.09	-132.97	37.6	3.942E-04	37.6	-5.382E-06
	39.2	0.0125	-18.2	-154.218	39.2	3.849E-04	39.2	-6.242E-06
	40.8	0.0131	-17.32	-174.265	40.8	3.743E-04	40.8	-7.053E-06
	42.4	0.0137	-16.43	-193.14	42.4	3.624E-04	42.4	-7.817E-06
	44	0.0143	-15.54	-210.87	44	3.493E-04	44	-8.535E-06
5								
	44	0.0143	-15.54	-210.87	44	3.493E-04	44	-8.535E-06
	45.6	0.0148	-14.65	-227.48	45.6	3.351E-04	45.6	-9.207E-06
	47.2	0.0153	-13.76	-242.993	47.2	3.198E-04	47.2	-9.834E-06
	48.8	0.0158	-12.88	-257.433	48.8	3.036E-04	48.8	-1.042E-05
	50.4	0.0163	-11.99	-270.818	50.4	2.865E-04	50.4	-1.096E-05
	52	0.0167	-11.1	-283.17	52	2.686E-04	52	-1.146E-05
	53.6	0.0172	-10.21	-294.505	53.6	2.499E-04	53.6	-1.192E-05
	55.2	0.0175	-9.324	-304.839	55.2	2.304E-04	55.2	-1.234E-05
	56.8	0.0179	-8.436	-314.188	56.8	2.104E-04	56.8	-1.272E-05
	58.4	0.0182	-7.548	-322.564	58.4	1.898E-04	58.4	-1.305E-05
	60	0.0185	-6.66	-329.979	60	1.686E-04	60	-1.336E-05
6								
	60	0.0185	-6.66	-329.979	60	1.686E-04	60	-1.336E-05
	60.6	0.0186	-6.327	-332.514	60.6	1.606E-04	60.6	-1.346E-05
	61.2	0.0187	-5.994	-334.917	61.2	1.525E-04	61.2	-1.355E-05
	61.8	0.0188	-5.661	-337.186	61.8	1.443E-04	61.8	-1.365E-05
	62.4	0.0189	-5.328	-339.324	62.4	1.361E-04	62.4	-1.373E-05
	63	0.0189	-4.995	-341.33	63	1.279E-04	63	-1.381E-05
	63.6	0.019	-4.662	-343.205	63.6	1.195E-04	63.6	-1.389E-05
	64.2	0.0191	-4.329	-344.948	64.2	1.112E-04	64.2	-1.396E-05
	64.8	0.0192	-3.996	-346.561	64.8	1.028E-04	64.8	-1.403E-05
	65.4	0.0192	-3.663	-348.044	65.4	9.436E-05	65.4	-1.409E-05
	66	0.0193	-3.33	-349.397	66	8.589E-05	66	-1.414E-05
7								
	66	0.0193	-3.33	-349.397	66	8.589E-05	66	-1.414E-05
	66.6	0.0193	-2.997	-350.621	66.6	7.739E-05	66.6	-1.419E-05

(continued) Torque, Bimoment, Twist Angle and Derivatives

67.2	0.0194	-2.664	-351.714	67.2	6.886E-05	67.2	-1.423E-05
67.8	0.0194	-2.331	-352.679	67.8	6.031E-05	67.8	-1.427E-05
68.4	0.0194	-1.998	-353.515	68.4	5.173E-05	68.4	-1.431E-05
69	0.0195	-1.665	-354.222	69	4.314E-05	69	-1.434E-05
69.6	0.0195	-1.332	-354.8	69.6	3.453E-05	69.6	-1.436E-05
70.2	0.0195	-0.999	-355.249	70.2	2.591E-05	70.2	-1.438E-05
70.8	0.0195	-0.666	-355.57	70.8	1.728E-05	70.8	-1.439E-05
71.4	0.0195	-0.333	-355.763	71.4	8.640E-06	71.4	-1.440E-05
72	0.0195	0	-355.827	72	0.000E+00	72	-1.440E-05

Table of Shear Stress due to Pure Torsion

Pure Torsion Shear Stress $G*t*\theta'$								
point	0	1	2	3	2'	1'	0'	
$t=$	0.625	0.625	0.625	0.450	0.625	0.625	0.625	
$\theta'''$	$z$							
-1.617E-06	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
-1.593E-06	0.6	0.136	0.136	0.136	0.098	0.136	0.136	0.136
-1.569E-06	1.2	0.269	0.269	0.269	0.193	0.269	0.269	0.269
-1.545E-06	1.8	0.397	0.397	0.397	0.286	0.397	0.397	0.397
-1.522E-06	2.4	0.521	0.521	0.521	0.375	0.521	0.521	0.521
-1.499E-06	3	0.642	0.642	0.642	0.462	0.642	0.642	0.642
-1.476E-06	3.6	0.759	0.759	0.759	0.546	0.759	0.759	0.759
-1.453E-06	4.2	0.872	0.872	0.872	0.628	0.872	0.872	0.872
-1.431E-06	4.8	0.981	0.981	0.981	0.707	0.981	0.981	0.981
-1.409E-06	5.4	1.087	1.087	1.087	0.783	1.087	1.087	1.087
-1.388E-06	6	1.189	1.189	1.189	0.856	1.189	1.189	1.189
-1.388E-06	6	1.189	1.189	1.189	0.856	1.189	1.189	1.189
-1.366E-06	6.6	1.288	1.288	1.288	0.927	1.288	1.288	1.288
-1.345E-06	7.2	1.383	1.383	1.383	0.996	1.383	1.383	1.383
-1.325E-06	7.8	1.476	1.476	1.476	1.062	1.476	1.476	1.476
-1.304E-06	8.4	1.564	1.564	1.564	1.126	1.564	1.564	1.564
-1.284E-06	9	1.649	1.649	1.649	1.187	1.649	1.649	1.649
-1.264E-06	9.6	1.732	1.732	1.732	1.247	1.732	1.732	1.732
-1.244E-06	10.2	1.810	1.810	1.810	1.303	1.810	1.810	1.810
-1.224E-06	10.8	1.887	1.887	1.887	1.358	1.887	1.887	1.887
-1.205E-06	11.4	1.959	1.959	1.959	1.410	1.959	1.959	1.959
-1.186E-06	12	2.029	2.029	2.029	1.461	2.029	2.029	2.029
-1.185E-06	12	2.029	2.029	2.029	1.461	2.029	2.029	2.029
-1.136E-06	13.6	2.199	2.199	2.199	1.584	2.199	2.199	2.199

(continued) Table of Shear Stress due to Pure Torsion

-1.089E-06	15.2	2.351	2.351	2.351	1.692	2.351	2.351	2.351
-1.042E-06	16.8	2.482	2.482	2.482	1.787	2.482	2.482	2.482
-9.974E-07	18.4	2.595	2.595	2.595	1.868	2.595	2.595	2.595
-9.538E-07	20	2.689	2.689	2.689	1.936	2.689	2.689	2.689
-9.116E-07	21.6	2.767	2.767	2.767	1.992	2.767	2.767	2.767
-8.707E-07	23.2	2.829	2.829	2.829	2.037	2.829	2.829	2.829
-8.312E-07	24.8	2.874	2.874	2.874	2.069	2.874	2.874	2.874
-7.931E-07	26.4	2.905	2.905	2.905	2.092	2.905	2.905	2.905
-7.563E-07	28	2.922	2.922	2.922	2.104	2.922	2.922	2.922
-7.551E-07	28	2.922	2.922	2.922	2.104	2.922	2.922	2.922
-7.37E-07	28.8	2.923	2.923	2.923	2.105	2.923	2.923	2.923
-7.193E-07	29.6	2.925	2.925	2.925	2.106	2.925	2.925	2.925
-6.843E-07	31.2	2.915	2.915	2.915	2.099	2.915	2.915	2.915
-6.502E-07	32.8	2.893	2.893	2.893	2.083	2.893	2.893	2.893
-6.169E-07	34.4	2.860	2.860	2.860	2.059	2.860	2.860	2.860
-5.845E-07	36	2.815	2.815	2.815	2.027	2.815	2.815	2.815
-5.528E-07	37.6	2.759	2.759	2.759	1.987	2.759	2.759	2.759
-5.221E-07	39.2	2.694	2.694	2.694	1.940	2.694	2.694	2.694
-4.921E-07	40.8	2.620	2.620	2.620	1.886	2.620	2.620	2.620
-4.630E-07	42.4	2.537	2.537	2.537	1.826	2.537	2.537	2.537
-4.347E-07	44	2.445	2.445	2.445	1.760	2.445	2.445	2.445
-4.338E-07	44	2.445	2.445	2.445	1.760	2.445	2.445	2.445
-4.061E-07	45.6	2.346	2.346	2.346	1.689	2.346	2.346	2.346
-3.788E-07	47.2	2.239	2.239	2.239	1.612	2.239	2.239	2.239
-3.519E-07	48.8	2.125	2.125	2.125	1.530	2.125	2.125	2.125
-3.255E-07	50.4	2.006	2.006	2.006	1.444	2.006	2.006	2.006
-2.995E-07	52	1.880	1.880	1.880	1.354	1.880	1.880	1.880
-2.739E-07	53.6	1.749	1.749	1.749	1.259	1.749	1.749	1.749
-2.488E-07	55.2	1.613	1.613	1.613	1.161	1.613	1.613	1.613
-2.241E-07	56.8	1.473	1.473	1.473	1.060	1.473	1.473	1.473
-1.998E-07	58.4	1.329	1.329	1.329	0.957	1.329	1.329	1.329
-1.760E-07	60	1.180	1.180	1.180	0.850	1.180	1.180	1.180
-1.755E-07	60	1.180	1.180	1.180	0.850	1.180	1.180	1.180
-1.665E-07	60.6	1.124	1.124	1.124	0.809	1.124	1.124	1.124
-1.576E-07	61.2	1.068	1.068	1.068	0.769	1.068	1.068	1.068
-1.486E-07	61.8	1.010	1.010	1.010	0.727	1.010	1.010	1.010
-1.397E-07	62.4	0.953	0.953	0.953	0.686	0.953	0.953	0.953
-1.309E-07	63	0.895	0.895	0.895	0.645	0.895	0.895	0.895
-1.220E-07	63.6	0.837	0.837	0.837	0.602	0.837	0.837	0.837

(continued) Table of Shear Stress due to Pure Torsion

-1.132E-07	64.2	0.778	0.778	0.778	0.560	0.778	0.778	0.778
-1.044E-07	64.8	0.720	0.720	0.720	0.518	0.720	0.720	0.720
-9.565E-08	65.4	0.661	0.661	0.661	0.476	0.661	0.661	0.661
-8.691E-08	66	0.601	0.601	0.601	0.433	0.601	0.601	0.601
-8.687E-08	66	0.601	0.601	0.601	0.433	0.601	0.601	0.601
-7.814E-08	66.6	0.542	0.542	0.542	0.390	0.542	0.542	0.542
-6.943E-08	67.2	0.482	0.482	0.482	0.347	0.482	0.482	0.482
-6.072E-08	67.8	0.422	0.422	0.422	0.304	0.422	0.422	0.422
-5.202E-08	68.4	0.362	0.362	0.362	0.261	0.362	0.362	0.362
-4.333E-08	69	0.302	0.302	0.302	0.217	0.302	0.302	0.302
-3.465E-08	69.6	0.242	0.242	0.242	0.174	0.242	0.242	0.242
-2.598E-08	70.2	0.181	0.181	0.181	0.131	0.181	0.181	0.181
-1.732E-08	70.8	0.121	0.121	0.121	0.087	0.121	0.121	0.121
-8.666E-09	71.4	0.060	0.060	0.060	0.044	0.060	0.060	0.060
-2.073E-11	72	0.000	0.000	0.000	0.000	0.000	0.000	0.000

Table of Warping Shear Stress Data

Warping shear stress at "s", - E*Sw*θ"/ t							
	0	1	2	3	2'	1'	0'
Sw	0.00	-17.40	-13.50	6.75	-13.50	-17.40	0.00
z							
0	0	-1.306	-1.013	0.703	-1.013	-1.306	0
0.6	0	-1.286	-0.998	0.693	-0.998	-1.286	0
1.2	0	-1.267	-0.983	0.683	-0.983	-1.267	0
1.8	0	-1.247	-0.968	0.672	-0.968	-1.247	0
2.4	0	-1.229	-0.953	0.662	-0.953	-1.229	0
3	0	-1.210	-0.939	0.652	-0.939	-1.210	0
3.6	0	-1.192	-0.925	0.642	-0.925	-1.192	0
4.2	0	-1.173	-0.910	0.632	-0.910	-1.173	0
4.8	0	-1.155	-0.896	0.622	-0.896	-1.155	0
5.4	0	-1.138	-0.883	0.613	-0.883	-1.138	0
6	0	-1.121	-0.869	0.604	-0.869	-1.121	0
6	0	-1.121	-0.869	0.604	-0.869	-1.121	0
6.6	0	-1.103	-0.856	0.594	-0.856	-1.103	0
7.2	0	-1.086	-0.843	0.585	-0.843	-1.086	0
7.8	0	-1.070	-0.830	0.576	-0.830	-1.070	0
8.4	0	-1.053	-0.817	0.567	-0.817	-1.053	0
9	0	-1.037	-0.804	0.559	-0.804	-1.037	0
9.6	0	-1.021	-0.792	0.550	-0.792	-1.021	0
10.2	0	-1.004	-0.779	0.541	-0.779	-1.004	0

(continued) Table of Warping Shear Stress Data

10.8	0	-0.988	-0.767	0.532	-0.767	-0.988	0
11.4	0	-0.973	-0.755	0.524	-0.755	-0.973	0
12	0	-0.958	-0.743	0.516	-0.743	-0.958	0
12	0	-0.957	-0.742	0.515	-0.742	-0.957	0
13.6	0	-0.917	-0.712	0.494	-0.712	-0.917	0
15.2	0	-0.879	-0.682	0.474	-0.682	-0.879	0
16.8	0	-0.841	-0.653	0.453	-0.653	-0.841	0
18.4	0	-0.805	-0.625	0.434	-0.625	-0.805	0
20	0	-0.770	-0.597	0.415	-0.597	-0.770	0
21.6	0	-0.736	-0.571	0.397	-0.571	-0.736	0
23.2	0	-0.703	-0.545	0.379	-0.545	-0.703	0
24.8	0	-0.671	-0.521	0.362	-0.521	-0.671	0
26.4	0	-0.640	-0.497	0.345	-0.497	-0.640	0
28	0	-0.611	-0.474	0.329	-0.474	-0.611	0
28	0	-0.610	-0.473	0.328	-0.473	-0.610	0
28.8	0	-0.595	-0.462	0.321	-0.462	-0.595	0
29.6	0	-0.581	-0.451	0.313	-0.451	-0.581	0
31.2	0	-0.552	-0.429	0.298	-0.429	-0.552	0
32.8	0	-0.525	-0.407	0.283	-0.407	-0.525	0
34.4	0	-0.498	-0.386	0.268	-0.386	-0.498	0
36	0	-0.472	-0.366	0.254	-0.366	-0.472	0
37.6	0	-0.446	-0.346	0.240	-0.346	-0.446	0
39.2	0	-0.422	-0.327	0.227	-0.327	-0.422	0
40.8	0	-0.397	-0.308	0.214	-0.308	-0.397	0
42.4	0	-0.374	-0.290	0.201	-0.290	-0.374	0
44	0	-0.351	-0.272	0.189	-0.272	-0.351	0
44	0	-0.350	-0.272	0.189	-0.272	-0.350	0
45.6	0	-0.328	-0.254	0.177	-0.254	-0.328	0
47.2	0	-0.306	-0.237	0.165	-0.237	-0.306	0
48.8	0	-0.284	-0.220	0.153	-0.220	-0.284	0
50.4	0	-0.263	-0.204	0.142	-0.204	-0.263	0
52	0	-0.242	-0.188	0.130	-0.188	-0.242	0
53.6	0	-0.221	-0.172	0.119	-0.172	-0.221	0
55.2	0	-0.201	-0.156	0.108	-0.156	-0.201	0
56.8	0	-0.181	-0.140	0.097	-0.140	-0.181	0
58.4	0	-0.161	-0.125	0.087	-0.125	-0.161	0
60	0	-0.142	-0.110	0.077	-0.110	-0.142	0
60	0	-0.142	-0.110	0.076	-0.110	-0.142	0
60.6	0	-0.134	-0.104	0.072	-0.104	-0.134	0

(continued) Table of Warping Shear Stress Data

61.2	0	-0.127	-0.099	0.069	-0.099	-0.127	0
61.8	0	-0.120	-0.093	0.065	-0.093	-0.120	0
62.4	0	-0.113	-0.088	0.061	-0.088	-0.113	0
63	0	-0.106	-0.082	0.057	-0.082	-0.106	0
63.6	0	-0.098	-0.076	0.053	-0.076	-0.098	0
64.2	0	-0.091	-0.071	0.049	-0.071	-0.091	0
64.8	0	-0.084	-0.065	0.045	-0.065	-0.084	0
65.4	0	-0.077	-0.060	0.042	-0.060	-0.077	0
66	0	-0.070	-0.054	0.038	-0.054	-0.070	0
66	0	-0.070	-0.054	0.038	-0.054	-0.070	0
66.6	0	-0.063	-0.049	0.034	-0.049	-0.063	0
67.2	0	-0.056	-0.043	0.030	-0.043	-0.056	0
67.8	0	-0.049	-0.038	0.026	-0.038	-0.049	0
68.4	0	-0.042	-0.033	0.023	-0.033	-0.042	0
69	0	-0.035	-0.027	0.019	-0.027	-0.035	0
69.6	0	-0.028	-0.022	0.015	-0.022	-0.028	0
70.2	0	-0.021	-0.016	0.011	-0.016	-0.021	0
70.8	0	-0.014	-0.011	0.008	-0.011	-0.014	0
71.4	0	-0.007	-0.005	0.004	-0.005	-0.007	0
72	0	0.000	0.000	0.000	0.000	0.000	0

Table of Warping Normal Stresses

Warping Normal Stresses, $E \cdot W_n / \theta''$							
	0	1	2	3	2'	1'	0'
$W_n$	22.00	0.00	-10.4	0.00	10.40	0.00	-22.00
$z$							
0	21.009	0	-9.932	0	9.932	0	-21.009
0.6	20.397	0	-9.642	0	9.642	0	-20.397
1.2	19.791	0	-9.356	0	9.356	0	-19.791
1.8	19.197	0	-9.075	0	9.075	0	-19.197
2.4	18.610	0	-8.798	0	8.798	0	-18.610
3	18.030	0	-8.523	0	8.523	0	-18.030
3.6	17.462	0	-8.255	0	8.255	0	-17.462
4.2	16.901	0	-7.989	0	7.989	0	-16.901
4.8	16.352	0	-7.730	0	7.730	0	-16.352
5.4	15.803	0	-7.471	0	7.471	0	-15.803
6	15.267	0	-7.217	0	7.217	0	-15.267
6	15.267	0	-7.217	0	7.217	0	-15.267
6.6	14.744	0	-6.970	0	6.970	0	-14.744
7.2	14.221	0	-6.723	0	6.723	0	-14.221

(continued)Table of Warping Normal Stresses

7.8	13.711	0	-6.481	0	6.481	0	-13.711
8.4	13.207	0	-6.243	0	6.243	0	-13.207
9	12.715	0	-6.011	0	6.011	0	-12.715
9.6	12.224	0	-5.779	0	5.779	0	-12.224
10.2	11.746	0	-5.552	0	5.552	0	-11.746
10.8	11.273	0	-5.329	0	5.329	0	-11.273
11.4	10.808	0	-5.109	0	5.109	0	-10.808
12	10.348	0	-4.892	0	4.892	0	-10.348
12	10.348	0	-4.892	0	4.892	0	-10.348
13.6	9.168	0	-4.334	0	4.334	0	-9.168
15.2	8.032	0	-3.797	0	3.797	0	-8.032
16.8	6.941	0	-3.281	0	3.281	0	-6.941
18.4	5.902	0	-2.790	0	2.790	0	-5.902
20	4.906	0	-2.319	0	2.319	0	-4.906
21.6	3.954	0	-1.869	0	1.869	0	-3.954
23.2	3.045	0	-1.439	0	1.439	0	-3.045
24.8	2.176	0	-1.029	0	1.029	0	-2.176
26.4	1.347	0	-0.637	0	0.637	0	-1.347
28	0.556	0	-0.263	0	0.263	0	-0.556
28	0.556	0	-0.263	0	0.263	0	-0.556
28.8	0.180	0	-0.085	0	0.085	0	-0.180
29.6	-0.196	0	0.093	0	-0.093	0	0.196
31.2	-0.912	0	0.431	0	-0.431	0	0.912
32.8	-1.593	0	0.753	0	-0.753	0	1.593
34.4	-2.240	0	1.059	0	-1.059	0	2.240
36	-2.853	0	1.349	0	-1.349	0	2.853
37.6	-3.434	0	1.623	0	-1.623	0	3.434
39.2	-3.982	0	1.883	0	-1.883	0	3.982
40.8	-4.500	0	2.127	0	-2.127	0	4.500
42.4	-4.987	0	2.358	0	-2.358	0	4.987
44	-5.445	0	2.574	0	-2.574	0	5.445
44	-5.445	0	2.574	0	-2.574	0	5.445
45.6	-5.874	0	2.777	0	-2.777	0	5.874
47.2	-6.274	0	2.966	0	-2.966	0	6.274
48.8	-6.648	0	3.143	0	-3.143	0	6.648
50.4	-6.992	0	3.306	0	-3.306	0	6.992
52	-7.311	0	3.456	0	-3.456	0	7.311
53.6	-7.605	0	3.595	0	-3.595	0	7.605
55.2	-7.873	0	3.722	0	-3.722	0	7.873



(continued)Table of Warping Normal Stresses

56.8	-8.115	0	3.836	0	-3.836	0	8.115
58.4	-8.326	0	3.936	0	-3.936	0	8.326
60	-8.524	0	4.029	0	-4.029	0	8.524
60	-8.524	0	4.029	0	-4.029	0	8.524
60.6	-8.587	0	4.060	0	-4.060	0	8.587
61.2	-8.645	0	4.087	0	-4.087	0	8.645
61.8	-8.709	0	4.117	0	-4.117	0	8.709
62.4	-8.760	0	4.141	0	-4.141	0	8.760
63	-8.811	0	4.165	0	-4.165	0	8.811
63.6	-8.862	0	4.189	0	-4.189	0	8.862
64.2	-8.906	0	4.210	0	-4.210	0	8.906
64.8	-8.951	0	4.231	0	-4.231	0	8.951
65.4	-8.989	0	4.250	0	-4.250	0	8.989
66	-9.021	0	4.265	0	-4.265	0	9.021
66	-9.021	0	4.265	0	-4.265	0	9.021
66.6	-9.053	0	4.280	0	-4.280	0	9.053
67.2	-9.079	0	4.292	0	-4.292	0	9.079
67.8	-9.104	0	4.304	0	-4.304	0	9.104
68.4	-9.130	0	4.316	0	-4.316	0	9.130
69	-9.149	0	4.325	0	-4.325	0	9.149
69.6	-9.162	0	4.331	0	-4.331	0	9.162
70.2	-9.174	0	4.337	0	-4.337	0	9.174
70.8	-9.181	0	4.340	0	-4.340	0	9.181
71.4	-9.187	0	4.343	0	-4.343	0	9.187
72	-9.187	0	4.343	0	-4.343	0	9.187

Table of Bending Normal Stresses

Bending Normal Stresses, M/ Sx								
		1	1	1	0	-1	-1	-1
Mus	z							
-518.40	0	-8.42	-8.42	-8.42	0.00	8.42	8.42	8.42
-505.49	0.6	-8.21	-8.21	-8.21	0	8.21	8.21	8.21
-492.70	1.2	-8.00	-8.00	-8.00	0	8.00	8.00	8.00
-480.01	1.8	-7.79	-7.79	-7.79	0	7.79	7.79	7.79
-467.42	2.4	-7.59	-7.59	-7.59	0	7.59	7.59	7.59
-454.95	3	-7.39	-7.39	-7.39	0	7.39	7.39	7.39
-442.58	3.6	-7.18	-7.18	-7.18	0	7.18	7.18	7.18
-430.33	4.2	-6.99	-6.99	-6.99	0	6.99	6.99	6.99
-418.18	4.8	-6.79	-6.79	-6.79	0	6.79	6.79	6.79
-406.13	5.4	-6.59	-6.59	-6.59	0	6.59	6.59	6.59
-394.20	6	-6.40	-6.40	-6.40	0	6.40	6.40	6.40
-394.20	6	-6.40	-6.40	-6.40	0	6.40	6.40	6.40
-382.37	6.6	-6.21	-6.21	-6.21	0	6.21	6.21	6.21
-370.66	7.2	-6.02	-6.02	-6.02	0	6.02	6.02	6.02
-359.05	7.8	-5.83	-5.83	-5.83	0	5.83	5.83	5.83
-347.54	8.4	-5.64	-5.64	-5.64	0	5.64	5.64	5.64
-336.15	9	-5.46	-5.46	-5.46	0	5.46	5.46	5.46
-324.86	9.6	-5.27	-5.27	-5.27	0	5.27	5.27	5.27
-313.69	10.2	-5.09	-5.09	-5.09	0	5.09	5.09	5.09
-302.62	10.8	-4.91	-4.91	-4.91	0	4.91	4.91	4.91
-291.65	11.4	-4.73	-4.73	-4.73	0	4.73	4.73	4.73
-280.80	12	-4.56	-4.56	-4.56	0	4.56	4.56	4.56
-280.80	12	-4.56	-4.56	-4.56	0	4.56	4.56	4.56
-252.38	13.6	-4.10	-4.10	-4.10	0	4.10	4.10	4.10
-224.74	15.2	-3.65	-3.65	-3.65	0	3.65	3.65	3.65
-197.86	16.8	-3.21	-3.21	-3.21	0	3.21	3.21	3.21
-171.74	18.4	-2.79	-2.79	-2.79	0	2.79	2.79	2.79
-146.40	20	-2.38	-2.38	-2.38	0	2.38	2.38	2.38
-121.82	21.6	-1.98	-1.98	-1.98	0	1.98	1.98	1.98
-98.02	23.2	-1.59	-1.59	-1.59	0	1.59	1.59	1.59
-74.98	24.8	-1.22	-1.22	-1.22	0	1.22	1.22	1.22
-52.70	26.4	-0.86	-0.86	-0.86	0	0.86	0.86	0.86
-31.20	28	-0.51	-0.51	-0.51	0	0.51	0.51	0.51
-31.20	28	-0.51	-0.51	-0.51	0	0.51	0.51	0.51
-20.74	28.8	-0.34	-0.34	-0.34	0	0.34	0.34	0.34
-10.46	29.6	-0.17	-0.17	-0.17	0	0.17	0.17	0.17

(continued) Table of Bending Normal Stresses

9.50	31.2	0.15	0.15	0.15	0	-0.15	-0.15	-0.15
28.70	32.8	0.47	0.47	0.47	0	-0.47	-0.47	-0.47
47.14	34.4	0.77	0.77	0.77	0	-0.77	-0.77	-0.77
64.80	36	1.05	1.05	1.05	0	-1.05	-1.05	-1.05
81.70	37.6	1.33	1.33	1.33	0	-1.33	-1.33	-1.33
97.82	39.2	1.59	1.59	1.59	0	-1.59	-1.59	-1.59
113.18	40.8	1.84	1.84	1.84	0	-1.84	-1.84	-1.84
127.78	42.4	2.07	2.07	2.07	0	-2.07	-2.07	-2.07
141.60	44	2.30	2.30	2.30	0	-2.30	-2.30	-2.30
141.60	44	2.30	2.30	2.30	0	-2.30	-2.30	-2.30
154.66	45.6	2.51	2.51	2.51	0	-2.51	-2.51	-2.51
166.94	47.2	2.71	2.71	2.71	0	-2.71	-2.71	-2.71
178.46	48.8	2.90	2.90	2.90	0	-2.90	-2.90	-2.90
189.22	50.4	3.07	3.07	3.07	0	-3.07	-3.07	-3.07
199.20	52	3.23	3.23	3.23	0	-3.23	-3.23	-3.23
208.42	53.6	3.38	3.38	3.38	0	-3.38	-3.38	-3.38
216.86	55.2	3.52	3.52	3.52	0	-3.52	-3.52	-3.52
224.54	56.8	3.65	3.65	3.65	0	-3.65	-3.65	-3.65
231.46	58.4	3.76	3.76	3.76	0	-3.76	-3.76	-3.76
237.60	60	3.86	3.86	3.86	0	-3.86	-3.86	-3.86
237.60	60	3.86	3.86	3.86	0	-3.86	-3.86	-3.86
239.71	60.6	3.89	3.89	3.89	0	-3.89	-3.89	-3.89
241.70	61.2	3.92	3.92	3.92	0	-3.92	-3.92	-3.92
243.59	61.8	3.95	3.95	3.95	0	-3.95	-3.95	-3.95
245.38	62.4	3.98	3.98	3.98	0	-3.98	-3.98	-3.98
247.05	63	4.01	4.01	4.01	0	-4.01	-4.01	-4.01
248.62	63.6	4.04	4.04	4.04	0	-4.04	-4.04	-4.04
250.07	64.2	4.06	4.06	4.06	0	-4.06	-4.06	-4.06
251.42	64.8	4.08	4.08	4.08	0	-4.08	-4.08	-4.08
252.67	65.4	4.10	4.10	4.10	0	-4.10	-4.10	-4.10
253.80	66	4.12	4.12	4.12	0	-4.12	-4.12	-4.12
253.80	66	4.12	4.12	4.12	0	-4.12	-4.12	-4.12
254.83	66.6	4.14	4.14	4.14	0	-4.14	-4.14	-4.14
255.74	67.2	4.15	4.15	4.15	0	-4.15	-4.15	-4.15
256.55	67.8	4.16	4.16	4.16	0	-4.16	-4.16	-4.16
257.26	68.4	4.18	4.18	4.18	0	-4.18	-4.18	-4.18
257.85	69	4.19	4.19	4.19	0	-4.19	-4.19	-4.19
258.34	69.60	4.19	4.19	4.19	0	-4.19	-4.19	-4.19
258.71	70.2	4.20	4.20	4.20	0	-4.20	-4.20	-4.20

(continued) Table of Bending Normal Stresses

258.98	70.8	4.20	4.20	4.20	0	-4.20	-4.20	-4.20
259.15	71.4	4.21	4.21	4.21	0	-4.21	-4.21	-4.21
259.20	72	4.21	4.21	4.21	0	-4.21	-4.21	-4.21

Table of Bending Shear Stresses

Vu=21.6 Bending Shear Stresses, $V*Q/ I*t$								
Ix 554								
	t	0.625	0.625	0.625	0.450	0.625	0.625	1
	Q	0	13.380	19.70	37.90	19.70	13.380	0.00
Shear	z							
21.60	0.0	0	0.835	1.229	3.284	1.229	0.835	0
21.42	0.6	0	0.828	1.219	3.256	1.219	0.828	0
21.24	1.2	0	0.821	1.208	3.229	1.208	0.821	0
21.06	1.8	0	0.814	1.198	3.202	1.198	0.814	0
20.88	2.4	0	0.807	1.188	3.174	1.188	0.807	0
20.70	3.0	0	0.800	1.178	3.147	1.178	0.800	0
20.52	3.6	0	0.793	1.167	3.120	1.167	0.793	0
20.34	4.2	0	0.786	1.157	3.092	1.157	0.786	0
20.16	4.8	0	0.779	1.147	3.065	1.147	0.779	0
19.98	5.4	0	0.772	1.137	3.037	1.137	0.772	0
19.80	6.0	0	0.765	1.127	3.010	1.127	0.765	0
19.80	6.0	0	0.765	1.127	3.010	1.127	0.765	0
19.62	6.6	0	0.758	1.116	2.983	1.116	0.758	0
19.44	7.2	0	0.751	1.106	2.955	1.106	0.751	0
19.26	7.8	0	0.744	1.096	2.928	1.096	0.744	0
19.08	8.4	0	0.737	1.086	2.901	1.086	0.737	0
18.90	9.0	0	0.730	1.075	2.873	1.075	0.730	0
18.72	9.6	0	0.723	1.065	2.846	1.065	0.723	0
18.54	10.2	0	0.716	1.055	2.819	1.055	0.716	0
18.36	10.8	0	0.709	1.045	2.791	1.045	0.709	0
18.18	11.4	0	0.703	1.034	2.764	1.034	0.703	0
18.00	12.0	0	0.696	1.024	2.736	1.024	0.696	0
18.00	12.0	0	0.696	1.024	2.736	1.024	0.696	0
17.52	13.6	0	0.677	0.997	2.663	0.997	0.677	0
17.04	15.2	0	0.658	0.969	2.591	0.969	0.658	0
16.56	16.8	0	0.640	0.942	2.518	0.942	0.640	0
16.08	18.4	0	0.621	0.915	2.445	0.915	0.621	0
15.60	20.0	0	0.603	0.888	2.372	0.888	0.603	0
15.12	21.6	0	0.584	0.860	2.299	0.860	0.584	0

(continued) Table of Bending Shear Stresses

14.64	23.20	0.566	0.833	2.226	0.833	0.566	0
14.16	24.80	0.547	0.806	2.153	0.806	0.547	0
13.68	26.40	0.529	0.778	2.080	0.778	0.529	0
13.20	28.00	0.510	0.751	2.007	0.751	0.510	0
13.20	28.00	0.510	0.751	2.007	0.751	0.510	0
12.96	28.80	0.501	0.737	1.970	0.737	0.501	0
12.72	29.60	0.492	0.724	1.934	0.724	0.492	0
12.24	31.20	0.473	0.696	1.861	0.696	0.473	0
11.76	32.80	0.454	0.669	1.788	0.669	0.454	0
11.28	34.40	0.436	0.642	1.715	0.642	0.436	0
10.80	36.00	0.417	0.614	1.642	0.614	0.417	0
10.32	37.60	0.399	0.587	1.569	0.587	0.399	0
9.84	39.20	0.380	0.560	1.496	0.560	0.380	0
9.36	40.80	0.362	0.533	1.423	0.533	0.362	0
8.88	42.40	0.343	0.505	1.350	0.505	0.343	0
8.40	44.00	0.325	0.478	1.277	0.478	0.325	0
8.40	44.00	0.325	0.478	1.277	0.478	0.325	0
7.92	45.60	0.306	0.451	1.204	0.451	0.306	0
7.44	47.20	0.288	0.423	1.131	0.423	0.288	0
6.96	48.80	0.269	0.396	1.058	0.396	0.269	0
6.48	50.40	0.250	0.369	0.985	0.369	0.250	0
6.00	52.00	0.232	0.341	0.912	0.341	0.232	0
5.52	53.60	0.213	0.314	0.839	0.314	0.213	0
5.04	55.20	0.195	0.287	0.766	0.287	0.195	0
4.56	56.80	0.176	0.259	0.693	0.259	0.176	0
4.08	58.40	0.158	0.232	0.620	0.232	0.158	0
3.60	60.00	0.139	0.205	0.547	0.205	0.139	0
3.60	60.00	0.139	0.205	0.547	0.205	0.139	0
3.42	60.60	0.132	0.195	0.520	0.195	0.132	0
3.24	61.20	0.125	0.184	0.493	0.184	0.125	0
3.06	61.80	0.118	0.174	0.465	0.174	0.118	0
2.88	62.40	0.111	0.164	0.438	0.164	0.111	0
2.70	63.00	0.104	0.154	0.410	0.154	0.104	0
2.52	63.60	0.097	0.143	0.383	0.143	0.097	0
2.34	64.20	0.090	0.133	0.356	0.133	0.090	0
2.16	64.80	0.083	0.123	0.328	0.123	0.083	0
1.98	65.40	0.077	0.113	0.301	0.113	0.077	0
1.80	66.00	0.070	0.102	0.274	0.102	0.070	0

(continued) Table of Bending Shear Stresses

1.80	66.00	0.070	0.102	0.274	0.102	0.070	0
1.62	66.60	0.063	0.092	0.246	0.092	0.063	0
1.44	67.20	0.056	0.082	0.219	0.082	0.056	0
1.26	67.80	0.049	0.072	0.192	0.072	0.049	0
1.08	68.40	0.042	0.061	0.164	0.061	0.042	0
0.90	69.00	0.035	0.051	0.137	0.051	0.035	0
0.72	69.60	0.028	0.041	0.109	0.041	0.028	0
0.54	70.20	0.021	0.031	0.082	0.031	0.021	0
0.36	70.80	0.014	0.020	0.055	0.020	0.014	0
0.18	71.40	0.007	0.010	0.027	0.010	0.007	0
0.00	72.00	0.000	0.000	0.000	0.000	0.000	0

Table of Total Normal Stresses

Combined Normal Stresses							
	0	1	2	3	2'	1'	0'
<b>z</b>							
<b>0</b>	12.594	-8.416	-18.347	0.0	18.347	8.416	-12.594
<b>0.6</b>	12.191	-8.206	-17.848	0	17.848	8.206	-12.191
<b>1.2</b>	11.792	-7.998	-17.354	0	17.354	7.998	-11.792
<b>1.8</b>	11.405	-7.792	-16.867	0	16.867	7.792	-11.405
<b>2.4</b>	11.022	-7.588	-16.386	0	16.386	7.588	-11.022
<b>3</b>	10.644	-7.386	-15.909	0	15.909	7.386	-10.644
<b>3.6</b>	10.277	-7.185	-15.440	0	15.440	7.185	-10.277
<b>4.2</b>	9.915	-6.986	-14.975	0	14.975	6.986	-9.915
<b>4.8</b>	9.563	-6.789	-14.519	0	14.519	6.789	-9.563
<b>5.4</b>	9.210	-6.593	-14.064	0	14.064	6.593	-9.210
<b>6</b>	8.868	-6.399	-13.617	0	13.617	6.399	-8.868
	0.000	0.000	0.000	0	0.000	0.000	0.000
<b>6</b>	8.868	-6.399	-13.617	0	13.617	6.399	-8.868
<b>6.6</b>	8.537	-6.207	-13.177	0	13.177	6.207	-8.537
<b>7.2</b>	8.204	-6.017	-12.740	0	12.740	6.017	-8.204
<b>7.8</b>	7.882	-5.829	-12.310	0	12.310	5.829	-7.882
<b>8.4</b>	7.565	-5.642	-11.885	0	11.885	5.642	-7.565
<b>9</b>	7.258	-5.457	-11.468	0	11.468	5.457	-7.258
<b>9.6</b>	6.950	-5.274	-11.052	0	11.052	5.274	-6.950
<b>10.2</b>	6.653	-5.092	-10.645	0	10.645	5.092	-6.653
<b>10.8</b>	6.361	-4.913	-10.242	0	10.242	4.913	-6.361
<b>11.4</b>	6.073	-4.735	-9.844	0	9.844	4.735	-6.073
<b>12</b>	5.790	-4.558	-9.450	0	9.450	4.558	-5.790
	0.000	0.000	0.000	0	0.000	0.000	0.000
<b>12</b>	5.790	-4.558	-9.450	0	9.450	4.558	-5.790
<b>13.6</b>	5.071	-4.097	-8.431	0	8.431	4.097	-5.071

(continued) Table of Total Normal Stresses

15.2	4.384	-3.648	-7.445	0	7.445	3.648	-4.384
16.8	3.729	-3.212	-6.493	0	6.493	3.212	-3.729
18.4	3.114	-2.788	-5.578	0	5.578	2.788	-3.114
20	2.530	-2.377	-4.696	0	4.696	2.377	-2.530
21.6	1.977	-1.978	-3.847	0	3.847	1.978	-1.977
23.2	1.453	-1.591	-3.030	0	3.030	1.591	-1.453
24.8	0.959	-1.217	-2.246	0	2.246	1.217	-0.959
26.4	0.491	-0.856	-1.492	0	1.492	0.856	-0.491
28	0.050	-0.506	-0.769	0	0.769	0.506	-0.050
	0.000	0.000	0.000	0	0.000	0.000	0.000
28	0.050	-0.506	-0.770	0	0.770	0.506	-0.050
28.8	-0.156	-0.337	-0.422	0	0.422	0.337	0.156
29.6	-0.366	-0.170	-0.077	0	0.077	0.170	0.366
31.2	-0.758	0.154	0.586	0	-0.586	-0.154	0.758
32.8	-1.127	0.466	1.219	0	-1.219	-0.466	1.127
34.4	-1.475	0.765	1.824	0	-1.824	-0.765	1.475
36	-1.801	1.052	2.401	0	-2.401	-1.052	1.801
37.6	-2.107	1.326	2.949	0	-2.949	-1.326	2.107
39.2	-2.394	1.588	3.471	0	-3.471	-1.588	2.394
40.8	-2.662	1.837	3.965	0	-3.965	-1.837	2.662
42.4	-2.913	2.074	4.432	0	-4.432	-2.074	2.913
44	-3.147	2.299	4.873	0	-4.873	-2.299	3.147
	0.000	0.000	0.000	0	0.000	0.000	0.000
44	-3.147	2.299	4.873	0	-4.873	-2.299	3.147
45.6	-3.363	2.511	5.287	0	-5.287	-2.511	3.363
47.2	-3.564	2.710	5.676	0	-5.676	-2.710	3.564
48.8	-3.751	2.897	6.040	0	-6.040	-2.897	3.751
50.4	-3.921	3.072	6.377	0	-6.377	-3.072	3.921
52	-4.078	3.234	6.690	0	-6.690	-3.234	4.078
53.6	-4.222	3.383	6.978	0	-6.978	-3.383	4.222
55.2	-4.352	3.521	7.242	0	-7.242	-3.521	4.352
56.8	-4.470	3.645	7.482	0	-7.482	-3.645	4.470
58.4	-4.568	3.757	7.693	0	-7.693	-3.757	4.568
60	-4.667	3.857	7.887	0	-7.887	-3.857	4.667
	0.000	0.000	0.000	0	0.000	0.000	0.000
60	-4.667	3.857	7.887	0	-7.887	-3.857	4.667
60.6	-4.696	3.891	7.951	0	-7.951	-3.891	4.696
61.2	-4.721	3.924	8.010	0	-8.010	-3.924	4.721
61.8	-4.754	3.954	8.071	0	-8.071	-3.954	4.754
62.4	-4.776	3.983	8.124	0	-8.124	-3.983	4.776
63	-4.800	4.011	8.176	0	-8.176	-4.011	4.800
63.6	-4.826	4.036	8.225	0	-8.225	-4.036	4.826

Table of Total Normal Stresses

64.2	-4.847	4.060	8.270	0	-8.270	-4.060	4.847
64.8	-4.870	4.082	8.313	0	-8.313	-4.082	4.870
65.4	-4.888	4.102	8.351	0	-8.351	-4.102	4.888
66	-4.901	4.120	8.385	0	-8.385	-4.120	4.901
	0.000	0.000	0.000	0	0.000	0.000	0.000
66	-4.901	4.120	8.385	0	-8.385	-4.120	4.901
66.6	-4.916	4.137	8.416	0	-8.416	-4.137	4.916
67.2	-4.927	4.152	8.443	0	-8.443	-4.152	4.927
67.8	-4.939	4.165	8.469	0	-8.469	-4.165	4.939
68.4	-4.954	4.176	8.492	0	-8.492	-4.176	4.954
69	-4.963	4.186	8.511	0	-8.511	-4.186	4.963
69.6	-4.968	4.194	8.525	0	-8.525	-4.194	4.968
70.2	-4.975	4.200	8.537	0	-8.537	-4.200	4.975
70.8	-4.977	4.204	8.544	0	-8.544	-4.204	4.977
71.4	-4.980	4.207	8.550	0	-8.550	-4.207	4.980
72	-4.979	4.208	8.551	0	-8.551	-4.208	4.979

Table of Combined Shear St. Vt.  $\tau$  positive

S=	0	1	2	3	2'	1'	0'
z							
0	0.000	-0.471	0.216	3.987	0.216	-0.471	0.000
0.6	0.136	-0.322	0.357	4.047	0.357	-0.322	0.136
1.2	0.269	-0.177	0.494	4.105	0.494	-0.177	0.269
1.8	0.397	-0.037	0.627	4.160	0.627	-0.037	0.397
2.4	0.521	0.099	0.756	4.212	0.756	0.099	0.521
3	0.642	0.232	0.881	4.261	0.881	0.232	0.642
3.6	0.759	0.360	1.002	4.308	1.002	0.360	0.759
4.2	0.872	0.485	1.119	4.352	1.119	0.485	0.872
4.8	0.981	0.605	1.232	4.394	1.232	0.605	0.981
5.4	1.087	0.722	1.341	4.433	1.341	0.722	1.087
6	1.189	0.834	1.446	4.470	1.446	0.834	1.189
6	1.189	0.834	1.446	4.470	1.446	0.834	1.189
6.6	1.288	0.943	1.549	4.504	1.549	0.943	1.288
7.2	1.383	1.049	1.647	4.536	1.647	1.049	1.383
7.8	1.476	1.150	1.741	4.567	1.741	1.150	1.476
8.4	1.564	1.248	1.833	4.594	1.833	1.248	1.564
9	1.649	1.343	1.920	4.619	1.920	1.343	1.649
9.6	1.732	1.435	2.005	4.643	2.005	1.435	1.732
10.2	1.810	1.522	2.086	4.663	2.086	1.522	1.810
10.8	1.887	1.608	2.164	4.682	2.164	1.608	1.887
11.4	1.959	1.688	2.238	4.698	2.238	1.688	1.959



(continued) Table of Combined Shear St. Vt.  $\tau$  positive

12	2.029	1.767	2.310	4.713	2.310	1.767	2.029
12	2.029	1.767	2.310	4.713	2.310	1.767	2.029
13.6	2.199	1.959	2.485	4.741	2.485	1.959	2.199
15.2	2.351	2.130	2.638	4.757	2.638	2.130	2.351
16.8	2.482	2.281	2.772	4.758	2.772	2.281	2.482
18.4	2.595	2.411	2.885	4.747	2.885	2.411	2.595
20	2.689	2.522	2.980	4.723	2.980	2.522	2.689
21.6	2.767	2.615	3.056	4.687	3.056	2.615	2.767
23.2	2.829	2.691	3.116	4.641	3.116	2.691	2.829
24.8	2.874	2.750	3.159	4.584	3.159	2.750	2.874
26.4	2.905	2.793	3.187	4.516	3.187	2.793	2.905
28	2.922	2.821	3.199	4.439	3.199	2.821	2.922
28	2.922	2.822	3.200	4.439	3.200	2.822	2.922
28.8	2.923	2.829	3.199	4.396	3.199	2.829	2.923
29.6	2.925	2.835	3.198	4.352	3.198	2.835	2.925
31.2	2.915	2.835	3.183	4.257	3.183	2.835	2.915
32.8	2.893	2.823	3.155	4.154	3.155	2.823	2.893
34.4	2.860	2.797	3.115	4.042	3.115	2.797	2.860
36	2.815	2.760	3.063	3.923	3.063	2.760	2.815
37.6	2.759	2.712	3.000	3.796	3.000	2.712	2.759
39.2	2.694	2.653	2.927	3.663	2.927	2.653	2.694
40.8	2.620	2.584	2.844	3.523	2.844	2.584	2.620
42.4	2.537	2.506	2.752	3.378	2.752	2.506	2.537
44	2.445	2.419	2.651	3.227	2.651	2.419	2.445
44	2.445	2.419	2.651	3.226	2.651	2.419	2.445
45.6	2.346	2.324	2.542	3.070	2.542	2.324	2.346
47.2	2.239	2.220	2.425	2.908	2.425	2.220	2.239
48.8	2.125	2.110	2.301	2.741	2.301	2.110	2.125
50.4	2.006	1.993	2.170	2.571	2.170	1.993	2.006
52	1.880	1.870	2.034	2.396	2.034	1.870	1.880
53.6	1.749	1.741	1.892	2.218	1.892	1.741	1.749
55.2	1.613	1.607	1.744	2.036	1.744	1.607	1.613
56.8	1.473	1.468	1.592	1.851	1.592	1.468	1.473
58.4	1.329	1.325	1.436	1.664	1.436	1.325	1.329
60	1.180	1.177	1.275	1.474	1.275	1.177	1.180
60	1.180	1.178	1.275	1.473	1.275	1.178	1.180
60.6	1.124	1.122	1.214	1.402	1.214	1.122	1.124
61.2	1.068	1.065	1.153	1.330	1.153	1.065	1.068

(continued) Table of Combined Shear St. Vt.  $\tau$  positive

61.8	1.010	1.008	1.091	1.257	1.091	1.008	1.010
62.4	0.953	0.951	1.029	1.185	1.029	0.951	0.953
63	0.895	0.894	0.967	1.112	0.967	0.894	0.895
63.6	0.837	0.835	0.903	1.038	0.903	0.835	0.837
64.2	0.778	0.777	0.841	0.965	0.841	0.777	0.778
64.8	0.720	0.719	0.777	0.892	0.777	0.719	0.720
65.4	0.661	0.660	0.713	0.818	0.713	0.660	0.661
66	0.601	0.601	0.649	0.744	0.649	0.601	0.601
66.6	0.542	0.541	0.585	0.670	0.585	0.541	0.542
67.2	0.482	0.482	0.520	0.596	0.520	0.482	0.482
67.8	0.422	0.422	0.456	0.522	0.456	0.422	0.422
68.4	0.362	0.362	0.391	0.448	0.391	0.362	0.362
69	0.302	0.302	0.326	0.373	0.326	0.302	0.302
69.6	0.242	0.242	0.261	0.299	0.261	0.242	0.242
70.2	0.181	0.181	0.196	0.224	0.196	0.181	0.181
70.8	0.121	0.121	0.131	0.149	0.131	0.121	0.121
71.4	0.060	0.060	0.065	0.075	0.065	0.060	0.060
72	0.000	0.000	0.000	0.000	0.000	0.000	0.000

Table of Total Shear Stress with Negative St. Vt.

S=	0	1	2	3	2'	1'	0'
z							
0	0.000	-0.471	0.216	3.987	0.216	-0.471	0.000
0.6	-0.136	-0.595	0.085	3.851	0.085	-0.595	-0.136
1.2	-0.269	-0.715	-0.043	3.718	-0.043	-0.715	-0.269
1.8	-0.397	-0.830	-0.166	3.588	-0.166	-0.830	-0.397
2.4	-0.521	-0.943	-0.287	3.461	-0.287	-0.943	-0.521
3	-0.642	-1.052	-0.403	3.337	-0.403	-1.052	-0.642
3.6	-0.759	-1.158	-0.516	3.215	-0.516	-1.158	-0.759
4.2	-0.872	-1.259	-0.625	3.096	-0.625	-1.259	-0.872
4.8	-0.981	-1.358	-0.731	2.981	-0.731	-1.358	-0.981
5.4	-1.087	-1.453	-0.833	2.868	-0.833	-1.453	-1.087
6	-1.189	-1.545	-0.932	2.758	-0.932	-1.545	-1.189
6	-1.189	-1.545	-0.932	2.758	-0.932	-1.545	-1.189
6.6	-1.288	-1.633	-1.027	2.650	-1.027	-1.633	-1.288
7.2	-1.383	-1.718	-1.120	2.545	-1.120	-1.718	-1.383
7.8	-1.476	-1.801	-1.210	2.442	-1.210	-1.801	-1.476
8.4	-1.564	-1.879	-1.295	2.342	-1.295	-1.879	-1.564
9	-1.649	-1.956	-1.378	2.244	-1.378	-1.956	-1.649
9.6	-1.732	-2.029	-1.458	2.149	-1.458	-2.029	-1.732
10.2	-1.810	-2.098	-1.535	2.056	-1.535	-2.098	-1.810

(continued) Table of Total Shear Stress with Negative St. Vt.

10.8	-1.887	-2.165	-1.609	1.965	-1.609	-2.165	-1.887
11.4	-1.959	-2.229	-1.679	1.878	-1.679	-2.229	-1.959
12	-2.029	-2.291	-1.747	1.792	-1.747	-2.291	-2.029
12	-2.029	-2.290	-1.747	1.791	-1.747	-2.290	-2.029
13.6	-2.199	-2.440	-1.914	1.574	-1.914	-2.440	-2.199
15.2	-2.351	-2.571	-2.063	1.372	-2.063	-2.571	-2.351
16.8	-2.482	-2.684	-2.193	1.184	-2.193	-2.684	-2.482
18.4	-2.595	-2.779	-2.305	1.010	-2.305	-2.779	-2.595
20	-2.689	-2.857	-2.399	0.850	-2.399	-2.857	-2.689
21.6	-2.767	-2.919	-2.478	0.703	-2.478	-2.919	-2.767
23.2	-2.829	-2.966	-2.541	0.568	-2.541	-2.966	-2.829
24.8	-2.874	-2.998	-2.589	0.445	-2.589	-2.998	-2.874
26.4	-2.905	-3.017	-2.623	0.333	-2.623	-3.017	-2.905
28	-2.922	-3.022	-2.645	0.232	-2.645	-3.022	-2.922
28	-2.922	-3.021	-2.644	0.232	-2.644	-3.021	-2.922
28.8	-2.923	-3.018	-2.648	0.186	-2.648	-3.018	-2.923
29.6	-2.925	-3.014	-2.651	0.141	-2.651	-3.014	-2.925
31.2	-2.915	-2.994	-2.647	0.060	-2.647	-2.994	-2.915
32.8	-2.893	-2.964	-2.631	-0.012	-2.631	-2.964	-2.893
34.4	-2.860	-2.922	-2.604	-0.076	-2.604	-2.922	-2.860
36	-2.815	-2.869	-2.566	-0.130	-2.566	-2.869	-2.815
37.6	-2.759	-2.807	-2.519	-0.177	-2.519	-2.807	-2.759
39.2	-2.694	-2.736	-2.461	-0.217	-2.461	-2.736	-2.694
40.8	-2.620	-2.656	-2.396	-0.249	-2.396	-2.656	-2.620
42.4	-2.537	-2.567	-2.322	-0.275	-2.322	-2.567	-2.537
44	-2.445	-2.471	-2.239	-0.294	-2.239	-2.471	-2.445
44	-2.445	-2.471	-2.239	-0.295	-2.239	-2.471	-2.445
45.6	-2.346	-2.368	-2.149	-0.308	-2.149	-2.368	-2.346
47.2	-2.239	-2.257	-2.053	-0.316	-2.053	-2.257	-2.239
48.8	-2.125	-2.140	-1.950	-0.319	-1.950	-2.140	-2.125
50.4	-2.006	-2.018	-1.841	-0.317	-1.841	-2.018	-2.006
52	-1.880	-1.890	-1.726	-0.311	-1.726	-1.890	-1.880
53.6	-1.749	-1.757	-1.607	-0.301	-1.607	-1.757	-1.749
55.2	-1.613	-1.619	-1.482	-0.287	-1.482	-1.619	-1.613
56.8	-1.473	-1.478	-1.354	-0.270	-1.354	-1.478	-1.473
58.4	-1.329	-1.332	-1.222	-0.249	-1.222	-1.332	-1.329
60	-1.180	-1.183	-1.086	-0.226	-1.086	-1.183	-1.180
60	-1.180	-1.183	-1.085	-0.226	-1.085	-1.183	-1.180
60.6	-1.124	-1.126	-1.034	-0.217	-1.034	-1.126	-1.124
61.2	-1.068	-1.070	-0.982	-0.207	-0.982	-1.070	-1.068

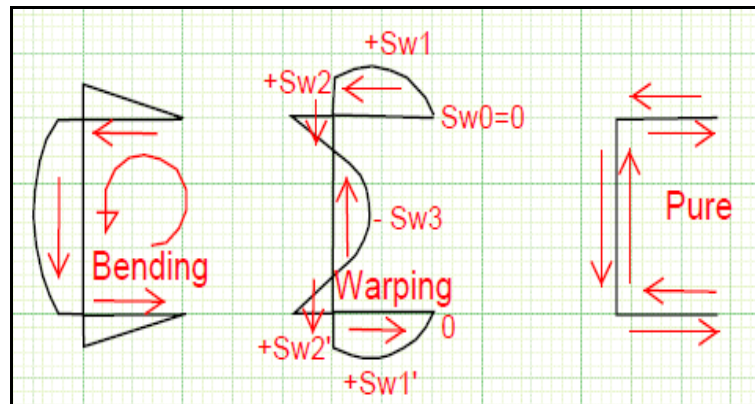
(continued) Table of Total Shear Stress with Negative St. Vt.

61.8	-1.010	-1.012	-0.929	-0.197	-0.929	-1.012	-1.010
62.4	-0.953	-0.954	-0.876	-0.187	-0.876	-0.954	-0.953
63	-0.895	-0.897	-0.824	-0.177	-0.824	-0.897	-0.895
63.6	-0.837	-0.838	-0.770	-0.166	-0.770	-0.838	-0.837
64.2	-0.778	-0.779	-0.716	-0.155	-0.716	-0.779	-0.778
64.8	-0.720	-0.720	-0.662	-0.144	-0.662	-0.720	-0.720
65.4	-0.661	-0.661	-0.608	-0.133	-0.608	-0.661	-0.661
66	-0.601	-0.602	-0.553	-0.121	-0.553	-0.602	-0.601
66	-0.601	-0.602	-0.553	-0.121	-0.553	-0.602	-0.601
66.6	-0.542	-0.542	-0.499	-0.110	-0.499	-0.542	-0.542
67.2	-0.482	-0.482	-0.444	-0.098	-0.444	-0.482	-0.482
67.8	-0.422	-0.423	-0.389	-0.086	-0.389	-0.423	-0.422
68.4	-0.362	-0.362	-0.333	-0.074	-0.333	-0.362	-0.362
69	-0.302	-0.302	-0.278	-0.062	-0.278	-0.302	-0.302
69.6	-0.242	-0.242	-0.222	-0.049	-0.222	-0.242	-0.242
70.2	-0.181	-0.181	-0.167	-0.037	-0.167	-0.181	-0.181
70.8	-0.121	-0.121	-0.111	-0.025	-0.111	-0.121	-0.121
71.4	-0.060	-0.061	-0.056	-0.012	-0.056	-0.061	-0.060
72	0.000	0.000	0.000	0.000	0.000	0.000	0.000

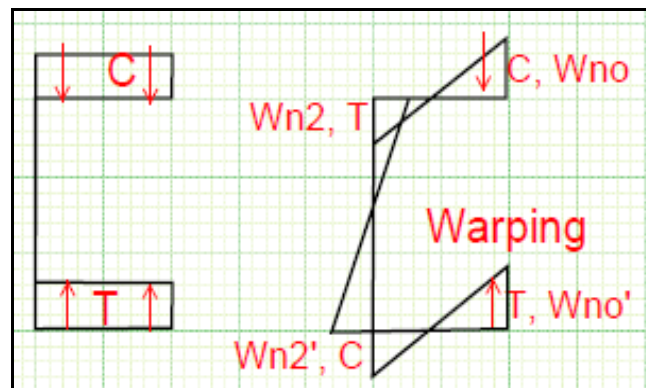
## APPENDIX G

### BOOTHBY'S TORSION PROBLEM

In the EBC of case study four, this material corresponds to the input model, input data, input forms, figures of output data with checks in notepad version, and excel processed output data and charts. Charts are presented containing both partial and combined stresses along interest points of the beam and cross section profile. Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP. Positive shear and axial stresses are assumed similar to those occurring at the cross section flanges and web when the beam undergoes bending as shown below.

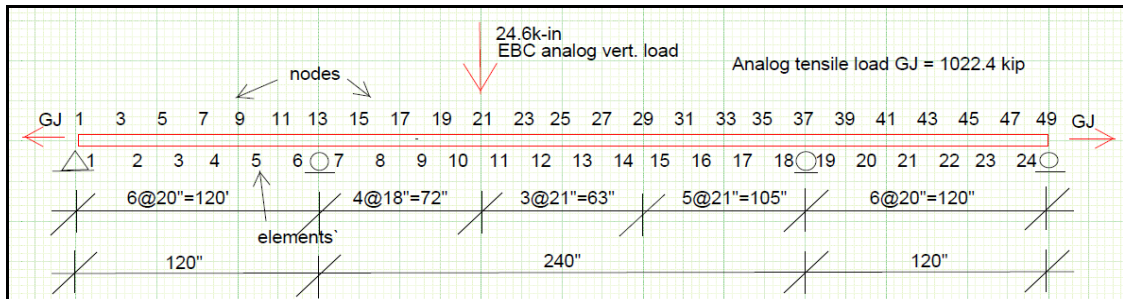


Convention for Shear Contributions



Convention for Normal Stresses Contributions

The analog EBC model used in the input form is shown together with the corresponding input notepad and input forms. The input form used for BMTORSWP was edited to be used in BMCOLDG without resorting to input forms for the bending problem. The purpose was to obtain output data for the bending solution in the exact cross section points where the warping analysis provided data. Afterwards, stresses were superposed and combined. Final results are shown in the charts in this appendix.



EBC Model for BMTORSWP (Boothby, 1984)

Bby  
Analysis of restrained warping for a 3-span beam, Boothby, AISC-EJ, 1984

24	49	4	0	9	0	1	0	29000.								
1	1	2	3	151.		1.		20.	E	-10022.4						
2	3	4	5	151.		1.		20.		-10022.4						
3	5	6	7	151.		1.		20.		-10022.4						
4	7	8	9	151.		1.		20.		-10022.4						
5	9	10	11	151.		1.		20.		-10022.4						
6	11	12	13	151.		1.		20.		-10022.4						
7	13	14	15	151.		1.		18.	Cw	-10022.4						
8	15	16	17	151.		1.		18.		-10022.4						
9	17	18	19	151.		1.		18.		-10022.4						
10	19	20	21	151.		1.		18.		-10022.4						
11	21	22	23	151.		1.		21.		-10022.4						
12	23	24	25	151.		1.		21.		-10022.4						
13	25	26	27	151.		1.		21.		-10022.4						
14	27	28	29	151.		1.		21.		-10022.4						
15	29	30	31	151.		1.		21.		-10022.4						
16	31	32	33	151.		1.		21.		-10022.4						
17	33	34	35	151.		1.		21.		-10022.4						
18	35	36	37	151.		1.		21.		-10022.4						
19	37	38	39	151.		1.		20.		-10022.4						
20	39	40	41	151.		1.		20.		-10022.4						
21	41	42	43	151.		1.		20.		-10022.4						
22	43	44	45	151.		1.		20.		-10022.4						
23	45	46	47	151.		1.		20.		-10022.4						
24	47	48	49	151.		1.		20.		-10022.4						
4	1	1	1	0	13	0	1	0	37	0	1	0	49	0	1	0
121	24.6															

KX=KY=1  
TWB torque ~ analog EBC vertical load

Input Notepad of the Torsional Problem for BMTORSWP

```

Bf2
Analysis of flexure for a 3-span beam, Boothy, AISC-EJ, 1984
24 49 4 0 9 0 1 0 29000.
 1  1  2  3 162.  1.  20.
 2  3  4  5 162.  1.  20.
 3  5  6  7 162.  1.  20.
 4  7  8  9 162.  1.  20.
 5  9 10 11 162.  1.  20.
 6 11 12 13 162.  1.  20.
 7 13 14 15 162.  1.  18.
 8 15 16 17 162.  1.  18.
 9 17 18 19 162.  1.  18.
10 19 20 21 162.  1.  18.
11 21 22 23 162.  1.  21.
12 23 24 25 162.  1.  21.
13 25 26 27 162.  1.  21.
14 27 28 29 162.  1.  21.
15 29 30 31 162.  1.  21.
16 31 32 33 162.  1.  21.
17 33 34 35 162.  1.  21.
18 35 36 37 162.  1.  21.
19 37 38 39 162.  1.  20.
20 39 40 41 162.  1.  20.
21 41 42 43 162.  1.  20.
22 43 44 45 162.  1.  20.
23 45 46 47 162.  1.  20.
24 47 48 49 162.  1.  20.
4 1  1  1 013 0 1 037 0 1 049 0 1 0
121 10.0

```

BMTORSWP note pad edited for BMCOLDGP and spreadsheet post-processing

Cross section inertia

An arbitrary value of cross section area

Load

Input Notepad of the Bending Problem for BMCOLDGP

Perfect coincidence was found in the bimoment diagrams. However the chart of torques by Boothby contains 2 mistakes. At left side of first interior support and right side of last interior support warping torques of 9.2 and 3.5) respectively were not detected by Boothby with corresponding errors of 700 and 300 per cent. See Chart Total and Warping Torque from BMTORSWP and by Boothby (1984.)

It was found that stresses calculations shown by Boothby correspond to the right side of the first interior support. That is why they are similar to calculations made from BMTORSWP output. Nevertheless, another mistake by Boothby was noticed regarding the sectorial coordinate at the flange-web corner. He used  $W_n2=6.75$  corresponding to a C15x50 shape instead of 5.02. See Chart Total and Warping Torque from BMTORSWP and Boothby.

## Program BMTORSWP Input Form

ALPHAMERIC DESCRIPTION OF THE JOB																																								FILE NAME:				
Analysis of restrained warping for a 3-span beam, Boothy, AISI-EJ, 1984																																												
NEL		NOD		NSUP		NSPD		JBW		NFX		NFY		NFZ		ELASTICITY (E11.4)																												
24		49		4		0		9		0		1		0		29000.																												
NE		NI		NC		NJ		Cw or Inert.				AREA		LENGTH		WTANG		SOIL NI		SOI																								
1		1		2		3		151.				1.		20.																														
2		3		4		5		151.				1.		SAME																														
3		5		6		7		151.				1.		SAME																														
4		7		8		9		151.				1.		SAME																														
5		9		10		11		151.				1.		SAME																														
6		11		12		13		151.				1.		SAME																														
7		13		14		15		151.				1.		18.																														
8		15		16		17		151.				1.		18.																														
9		17		18		19		151.				1.		18.																														
10		19		20		21		151.				1.		18.																														
11		21		22		23		151.				1.		21.																														
12		23		24		25		151.				1.		21.																														
13		25		26		27		151.				1.		21.																														
NF		N		KX		KY		KZ		N		KX		KY		KZ		N		KX		KY		K																				
4		1		1		1		0		13		0		1		0		37		0		1		0																				
NF		N		KD		SPRING						N		KD		SPRING						N		KD																				
NF		N		LOAD VALUE				LOAD VALUE				LOAD VALUE				LOAD VALUE				N		L																						
1		21		24.6																																								
keep going in the note pad until reaching element 24																																												
NF		ELEM		END		ROTATIONAL SPRING MODULUS				ELEM		END		ROTATIONAL SPRING MODULUS				ELEM		END		ROTATION SPRING MOD																						

Any value to avoid singularity

keep going in the note pad until reaching element 24

Input Form for BMTORSWP, Left Side

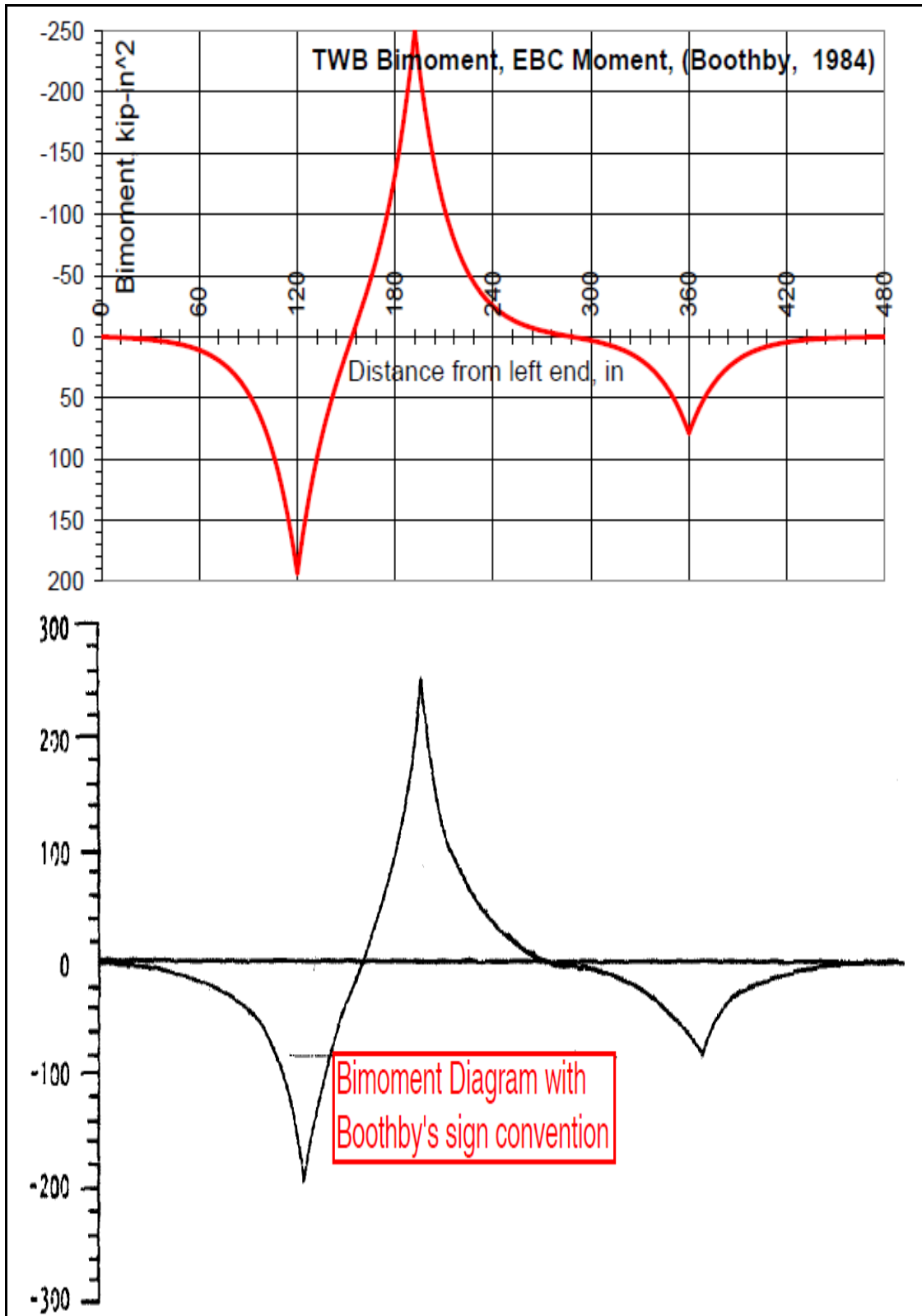


AUTHOR: B. DESCHAPELLES

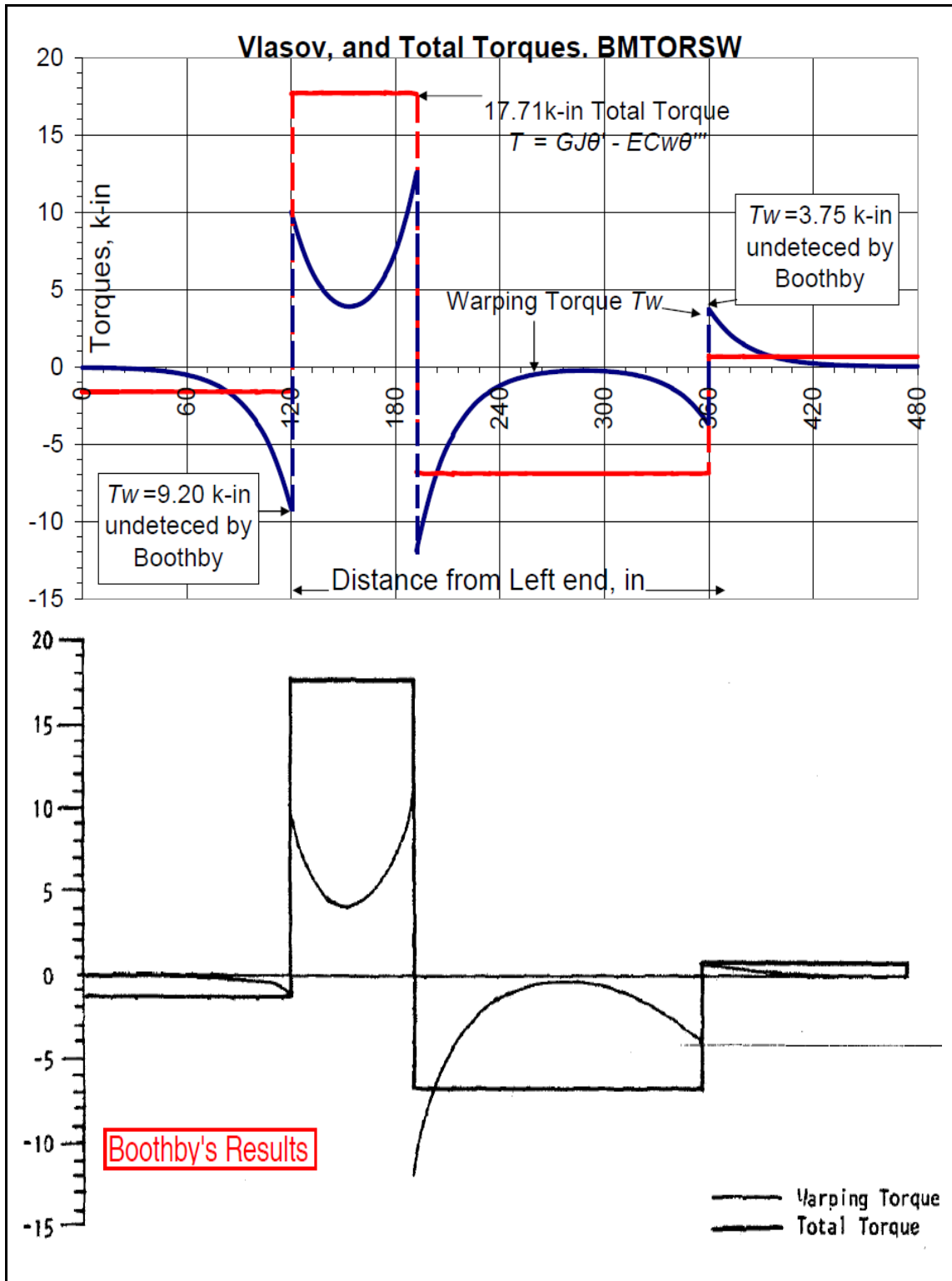
39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80
NAME:																																									
GENERAL INFORMATION																				PARTIAL RESTRAINT AT END ? 1 FOR YES 0 FOR NO																					
I	SOIL NJ	SOIL TI	SOIL TJ	ANGLE (°)	COMPR.	Wn ⊕ NI	Wn ⊕ NJ	1	2																																
					- 10 0 2 2																																				
					- 10 0 2 2																																				
					S																																				
					A																																				
					M																																				
					E																																				
					S																																				
					A																																				
					M																																				
					E																																				
KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	SUPPORTS																											
SPRING						N	KD	SPRING						SPRINGS AT SUPPORTS																											
N	LOAD VALUE				N	LOAD VALUE				N	LOAD VALUE				APPLIED NODAL FORCES																										
ROTATIONAL G MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				SPRINGS AT ELEM. END, 1 OR 2, IF ANY																									
39	40	41	42	43	44	45	46	47	48	49	50	51	52	53	54	55	56	57	58	59	60	61	62	63	64	65	66	67	68	69	70	71	72	73	74	75	76	77	78	79	80

Increase one field and reduce another for larger G\*J values

Input Form for BMTORSWP, Right Side



Match between TWB Bimoment from BMTORSWP and Boothby



Match between Total and Warping Torque from BMTORSWP and Boothby

BBYOUT

-----  
 YOU ARE USING COMPUTER PROGRAM BMTORSW, DEVELOPED BY DR. BERNARDO DESCHAPELLES

INPUT DATA FILE NAME IS = bby.txt

OUTPUT FILE NAME IS = bbyout.txt

STORAGE FILE FOR POST-PROCESSING WITH EXCEL = bbygrf.grf

-----  
 Analysis of restrained warping for a 3-span beam, Boothy, AISC-EJ, 1984

modulus of elasticity of the material= 29000. k/ft2

ELEM	nodes	inertia	length	distrib.	load	AXIAL	SOIL	NORMAL	MODULUS, Ksf	angle
	i	j	ft.4	ft	at i	at j	LOAD	1st END	2nd END	rad
1	1	3	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
2	3	5	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
3	5	7	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
4	7	9	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
5	9	11	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
6	11	13	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
7	13	15	18.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
8	15	17	18.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
9	17	19	18.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
10	19	21	18.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
11	21	23	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
12	23	25	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
13	25	27	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
14	27	29	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
15	29	31	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
16	31	33	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
17	33	35	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
18	35	37	21.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
19	37	39	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
20	39	41	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
21	41	43	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
22	43	45	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
23	45	47	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00
24	47	49	20.00	0.000	0.000	0.000	0.0	0.0	0.0	0.000 00

INPUT DATA RELATED TO THE 4 SUPPORTS  
 4 1 1 1 013 0 1 037 0 1 049 0 1 0

INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2  
 121 24.60

FINAL SOLUTION FOUND AFTER 1 ITERATIONS

Output of nodal displacements in reference to global axes

node	displ. along x or nonn1	displ. along y or nonn2	displ. around z or nonn 3	node	displ. along x or nonn1	displ. along y or nonn2	displ. around z or nonn 3
1	0.0000E+00	0.0000E+00	-0.1551E-03	2	0.0000E+00	-0.1546E-02	-0.2570E-03
3	0.0000E+00	-0.3082E-02	-0.1521E-03	4	0.0000E+00	-0.4574E-02	-0.2459E-03

			BBYOUT				
5	0.0000E+00	-0.6028E-02	-0.1404E-03	6	0.0000E+00	-0.7351E-02	-0.2125E-03
7	0.0000E+00	-0.8568E-02	-0.1084E-03	8	0.0000E+00	-0.9437E-02	-0.1240E-03
9	0.0000E+00	-0.1003E-01	-0.2456E-04	10	0.0000E+00	-0.9713E-02	0.1069E-03
11	0.0000E+00	-0.8679E-02	0.1940E-03	12	0.0000E+00	-0.5274E-02	0.7084E-03
13	0.0000E+00	0.0000E+00	0.7633E-03	14	0.0000E+00	0.8776E-02	0.1600E-02
15	0.0000E+00	0.1904E-01	0.1266E-02	16	0.0000E+00	0.3099E-01	0.2026E-02
17	0.0000E+00	0.4327E-01	0.1374E-02	18	0.0000E+00	0.5529E-01	0.1962E-02
19	0.0000E+00	0.6671E-01	0.1174E-02	20	0.0000E+00	0.7575E-01	0.1357E-02
21	0.0000E+00	0.8281E-01	0.5080E-03	22	0.0000E+00	0.8482E-01	0.9450E-04
23	0.0000E+00	0.8421E-01	-0.2502E-03	24	0.0000E+00	0.8036E-01	-0.7284E-03
25	0.0000E+00	0.7556E-01	-0.5273E-03	26	0.0000E+00	0.6958E-01	-0.1028E-02
27	0.0000E+00	0.6326E-01	-0.6274E-03	28	0.0000E+00	0.5652E-01	-0.1134E-02
29	0.0000E+00	0.4967E-01	-0.6604E-03	30	0.0000E+00	0.4270E-01	-0.1162E-02
31	0.0000E+00	0.3573E-01	-0.6623E-03	32	0.0000E+00	0.2884E-01	-0.1142E-02
33	0.0000E+00	0.2205E-01	-0.6352E-03	34	0.0000E+00	0.1560E-01	-0.1052E-02
35	0.0000E+00	0.9451E-02	-0.5495E-03	36	0.0000E+00	0.4315E-02	-0.7945E-03
37	0.0000E+00	0.0000E+00	-0.3111E-03	38	0.0000E+00	-0.2150E-02	-0.2888E-03
39	0.0000E+00	-0.3538E-02	-0.7910E-04	40	0.0000E+00	-0.3960E-02	-0.4359E-04
41	0.0000E+00	-0.4089E-02	0.1001E-04	42	0.0000E+00	-0.3847E-02	0.5056E-04
43	0.0000E+00	-0.3493E-02	0.4421E-04	44	0.0000E+00	-0.2996E-02	0.8662E-04
45	0.0000E+00	-0.2457E-02	0.5725E-04	46	0.0000E+00	-0.1865E-02	0.1002E-03

```

47 0.0000E+00 -0.1256E-02 0.6201E-04
BBYOUT
48 0.0000E+00 -0.6302E-03 0.1048E-03
49 0.0000E+00 0.0000E+00 0.6321E-04

```

-----  
OUTPUT OF SOIL REACTIONS,STRESSES AND TRANSVERSE DISPLACEMENTS  
-----

		DISPLACEMENTS IN INCIDENCES		
		1	2	3
ELEMENT	1	0.0000E+00	0.0000E+00	-0.15505E-03
NODE	1	0.0000E+00	-0.15459E-02	-0.25701E-03
NODE	2	0.0000E+00	-0.30821E-02	-0.15212E-03
NODE	3	0.0000E+00		
FORCES ACTING ALONG THE 9 DOF				
NODE	1	0.16135E+01	0.20670E-12	
NODE	2	0.15719E-12	0.28300E-12	
NODE	3	-0.16135E+01	0.13802E+01	

ELEMENT 1, FROM NODE 1, TO NODE 3 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61
bmom,kft	0.00	0.12	0.24	0.36	0.49	0.62	
tdisp,ft	0.00000	-0.00031	-0.00062	-0.00093	-0.00124	-0.00155	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END			

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61
bmom,kft	0.62	0.75	0.90	1.05	1.21	1.38	
tdisp,ft	-0.00155	-0.00186	-0.00216	-0.00247	-0.00278	-0.00308	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END			

		DISPLACEMENTS IN INCIDENCES		
		3	4	5
ELEMENT	2	0.0000E+00	-0.30821E-02	-0.15212E-03
NODE	3	0.0000E+00	-0.45744E-02	-0.24585E-03
NODE	4	0.0000E+00	-0.60282E-02	-0.14043E-03
NODE	5	0.0000E+00		
FORCES ACTING ALONG THE 9 DOF				
NODE	3	0.16135E+01	-0.13802E+01	
NODE	4	0.43340E-12	0.38510E-11	
NODE	5	-0.16135E+01	0.41233E+01	

ELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61
bmom,kft	1.38	1.56	1.76	1.98	2.21	2.46	
tdisp,ft	-0.00308	-0.00339	-0.00369	-0.00399	-0.00429	-0.00458	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END			

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000

BBYOUT						
shear,k	1.61	1.61	1.61	1.61	1.61	1.61
bmom,kft	2.46	2.74	3.04	3.37	3.73	4.12
tdisp,ft	-0.00458	-0.00488	-0.00517	-0.00546	-0.00575	-0.00603
axial,k	0.00 AT 1st END and			0.00 AT 2nd END		

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ELEMENT	3	DISPLACEMENTS IN INCIDENCES			5	6	7
NODE	5	0.00000E+00			-0.60282E-02	-0.14043E-03	
NODE	6	0.00000E+00			-0.73507E-02	-0.21250E-03	
NODE	7	0.00000E+00			-0.85681E-02	-0.10844E-03	
FORCES ACTING ALONG THE 9 DOF							
NODE	5	0.00000E+00			0.16135E+01	-0.41233E+01	
NODE	6	0.00000E+00			0.17847E-13	0.63055E-11	
NODE	7	0.00000E+00			-0.16135E+01	0.10938E+02	

ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span		span		span		span	
	0.0	0.1	0.2	0.3	0.4	0.5		
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61	
bmom,kft	4.12	4.55	5.03	5.55	6.12	6.74	6.74	
tdisp,ft	-0.00603	-0.00631	-0.00658	-0.00685	-0.00712	-0.00738	-0.00738	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END				

right half of span,at tenth points of length

	span		span		span		span	
	0.5	0.6	0.7	0.8	0.9	1.0		
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61	
bmom,kft	6.74	7.43	8.19	9.02	9.93	10.94	10.94	
tdisp,ft	-0.00738	-0.00763	-0.00788	-0.00812	-0.00835	-0.00857	-0.00857	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END				

---

ELEMENT	4	DISPLACEMENTS IN INCIDENCES			7	8	9
NODE	7	0.00000E+00			-0.85681E-02	-0.10844E-03	
NODE	8	0.00000E+00			-0.94369E-02	-0.12403E-03	
NODE	9	0.00000E+00			-0.10030E-01	-0.24560E-04	
FORCES ACTING ALONG THE 9 DOF							
NODE	7	0.00000E+00			0.16135E+01	-0.10938E+02	
NODE	8	0.00000E+00			-0.55385E-12	-0.58053E-11	
NODE	9	0.00000E+00			-0.16135E+01	0.28554E+02	

ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span		span		span		span	
	0.0	0.1	0.2	0.3	0.4	0.5		
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61	
bmom,kft	10.94	12.04	13.26	14.60	16.07	17.68	17.68	
tdisp,ft	-0.00857	-0.00878	-0.00898	-0.00917	-0.00934	-0.00950	-0.00950	
axial,k	0.00 AT 1st END and			0.00 AT 2nd END				

right half of span,at tenth points of length

	span		span		span		span	
	0.5	0.6	0.7	0.8	0.9	1.0		
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
shear,k	1.61	1.61	1.61	1.61	1.61	1.61	1.61	
bmom,kft	17.68	19.46	21.42	23.58	25.95	28.55	28.55	
tdisp,ft	-0.00950	-0.00965	-0.00978	-0.00988	-0.00997	-0.01003	-0.01003	

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axial,k      0.00 AT 1st END and      BBYOUT      0.00 AT 2nd END
-----
ELEMENT 5 DISPLACEMENTS IN INCIDENCES 9 10 11
  NODE 9 0.00000E+00 -0.10030E-01 -0.24560E-04
  NODE 10 0.00000E+00 -0.97135E-02 0.10692E-03
  NODE 11 0.00000E+00 -0.86789E-02 0.19405E-03
FORCES ACTING ALONG THE 9 DOF
  NODE 9 0.00000E+00 0.16135E+01 -0.28554E+02
  NODE 10 0.00000E+00 0.58212E-12 -0.21937E-11
  NODE 11 0.00000E+00 -0.16135E+01 0.74369E+02

ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 20.00 ft

left half of span,at tenth points of length
span span span span span span
0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.61 1.61 1.61 1.61 1.61 1.61
bmom,kft 28.55 31.42 34.58 38.06 41.88 46.09
tdisp,ft -0.01003 -0.01007 -0.01007 -0.01005 -0.00999 -0.00989
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.61 1.61 1.61 1.61 1.61 1.61
bmom,kft 46.09 50.72 55.81 61.41 67.58 74.37
tdisp,ft -0.00989 -0.00975 -0.00956 -0.00933 -0.00903 -0.00868
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13
  NODE 11 0.00000E+00 -0.86789E-02 0.19405E-03
  NODE 12 0.00000E+00 -0.52740E-02 0.70843E-03
  NODE 13 0.00000E+00 0.00000E+00 0.76326E-03
FORCES ACTING ALONG THE 9 DOF
  NODE 11 0.00000E+00 0.16135E+01 -0.74369E+02
  NODE 12 0.00000E+00 -0.11741E-13 0.28773E-11
  NODE 13 0.00000E+00 -0.16135E+01 0.19362E+03

ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 20.00 ft

left half of span,at tenth points of length
span span span span span span
0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.61 1.61 1.61 1.61 1.61 1.61
bmom,kft 74.37 81.84 90.05 99.10 109.05 120.00
tdisp,ft -0.00868 -0.00826 -0.00776 -0.00718 -0.00651 -0.00574
axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span span span span span span
0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.61 1.61 1.61 1.61 1.61 1.61
bmom,kft 120.00 132.05 145.31 159.90 175.95 193.62
tdisp,ft -0.00574 -0.00486 -0.00385 -0.00272 -0.00144 0.00000
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
(193.62+1.38)120=1.625 ~ 01.61
-----
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BBYOUT			
ELEMENT	7	DISPLACEMENTS IN INCIDENCES	13 14 15
NODE	13	0.00000E+00	0.00000E+00
NODE	14	0.00000E+00	0.87756E-02
NODE	15	0.00000E+00	0.19041E-01
FORCES ACTING ALONG THE 9 DOF			
NODE	13	0.00000E+00	-0.17698E+02
NODE	14	0.00000E+00	0.35749E-12
NODE	15	0.00000E+00	0.17698E+02
ELEMENT 7, FROM NODE 13, TO NODE 15 - LENGTH = 18.00 ft			
left half of span, at tenth points of length			
	span	span	span
	0.0	0.1	0.2
soil, k/ft	0.000	0.000	0.000
shear, k	-17.70	-17.70	-17.70
bmom, kft	193.62	176.23	160.15
tdisp, ft	0.00000	0.00144	0.00302
axial, k	0.00	AT 1st END and	
		span	span
		0.3	0.4
		0.000	0.000
		-17.70	-17.70
		145.25	131.44
		0.00471	0.00651
		0.00 AT 2nd END	
right half of span, at tenth points of length			
	span	span	span
	0.5	0.6	0.7
soil, k/ft	0.000	0.000	0.000
shear, k	-17.70	-17.70	-17.70
bmom, kft	118.60	106.63	95.46
tdisp, ft	0.00841	0.01039	0.01246
axial, k	0.00	AT 1st END and	
		span	span
		0.8	0.9
		0.000	0.000
		-17.70	-17.70
		85.00	75.17
		0.01459	0.01679
		0.00 AT 2nd END	
-----			
ELEMENT	8	DISPLACEMENTS IN INCIDENCES	15 16 17
NODE	15	0.00000E+00	0.19041E-01
NODE	16	0.00000E+00	0.30993E-01
NODE	17	0.00000E+00	0.43266E-01
FORCES ACTING ALONG THE 9 DOF			
NODE	15	0.00000E+00	-0.17698E+02
NODE	16	0.00000E+00	-0.22204E-12
NODE	17	0.00000E+00	0.17698E+02
ELEMENT 8, FROM NODE 15, TO NODE 17 - LENGTH = 18.00 ft			
left half of span, at tenth points of length			
	span	span	span
	0.0	0.1	0.2
soil, k/ft	0.000	0.000	0.000
shear, k	-17.70	-17.70	-17.70
bmom, kft	65.90	57.11	48.75
tdisp, ft	0.01904	0.02134	0.02369
axial, k	0.00	AT 1st END and	
		span	span
		0.3	0.4
		0.000	0.000
		-17.70	-17.70
		40.75	33.05
		0.02607	0.02848
		0.00 AT 2nd END	
right half of span, at tenth points of length			
	span	span	span
	0.5	0.6	0.7
soil, k/ft	0.000	0.000	0.000
shear, k	-17.70	-17.70	-17.70
bmom, kft	25.60	18.34	11.22
tdisp, ft	0.03091	0.03337	0.03583
axial, k	0.00	AT 1st END and	
		span	span
		0.8	0.9
		0.000	0.000
		-17.70	-17.70
		4.17	-2.84
		0.03831	0.04079
		0.00 AT 2nd END	
-----			
ELEMENT	9	DISPLACEMENTS IN INCIDENCES	17 18 19
NODE	17	0.00000E+00	0.43266E-01
NODE	18	0.00000E+00	0.55287E-01

```

          NODE 19      0.00000E+00      BBYOUT      0.66715E-01      0.11744E-02
FORCES ACTING ALONG THE 9 DOF
          NODE 17      0.00000E+00      -0.17698E+02      0.98709E+01
          NODE 18      0.00000E+00      -0.19454E-11      0.14981E-10
          NODE 19      0.00000E+00      0.17698E+02      -0.93421E+02

```

ELEMENT 9, FROM NODE 17, TO NODE 19 - LENGTH = 18.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k   -17.70     -17.70     -17.70     -17.70     -17.70     -17.70
bmom,kft  -9.87      -16.98     -24.21     -31.62     -39.27     -47.20
tdisp,ft  0.04327   0.04574   0.04819   0.05063   0.05305   0.05543
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k   -17.70     -17.70     -17.70     -17.70     -17.70     -17.70
bmom,kft  -47.20     -55.49     -64.19     -73.36     -83.08     -93.42
tdisp,ft  0.05543   0.05779   0.06010   0.06236   0.06457   0.06671
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

```

```

-----
ELEMENT 10 DISPLACEMENTS IN INCIDENCES 19 20 21
          NODE 19      0.00000E+00      0.66715E-01      0.11744E-02
          NODE 20      0.00000E+00      0.75752E-01      0.13574E-02
          NODE 21      0.00000E+00      0.82814E-01      0.50795E-03
FORCES ACTING ALONG THE 9 DOF
          NODE 19      0.00000E+00      -0.17698E+02      0.93421E+02
          NODE 20      0.00000E+00      -0.95102E-12     -0.30931E-11
          NODE 21      0.00000E+00      0.17698E+02      -0.25063E+03

```

ELEMENT 10, FROM NODE 19, TO NODE 21 - LENGTH = 18.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k   -17.70     -17.70     -17.70     -17.70     -17.70     -17.70
bmom,kft  -93.42     -104.45    -116.25    -128.92    -142.55    -157.23
tdisp,ft  0.06671   0.06879   0.07079   0.07271   0.07453   0.07624
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k   -17.70     -17.70     -17.70     -17.70     -17.70     -17.70
bmom,kft  -157.23    -173.08    -190.21    -208.75    -228.84    -250.63
tdisp,ft  0.07624   0.07784   0.07931   0.08064   0.08181   0.08281
axial,k   0.00 AT 1st END and 0.00 AT 2nd END

```

```

-----
ELEMENT 11 DISPLACEMENTS IN INCIDENCES 21 22 23
          NODE 21      0.00000E+00      0.82814E-01      0.50795E-03
          NODE 22      0.00000E+00      0.84816E-01      0.94498E-04
          NODE 23      0.00000E+00      0.84208E-01     -0.25021E-03
FORCES ACTING ALONG THE 9 DOF
          NODE 21      0.00000E+00      0.69021E+01      0.25063E+03

```

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Sum. of moments (17.7x72-193.62-250.63)/10022 = 0.08283 ~ 0.08281 OK

193.62 k-in from elem. 7, node 13

Output Page 7

```

BBYOUT
NODE 22      0.00000E+00      -0.28362E-11      -0.18572E-10
NODE 23      0.00000E+00      -0.69021E+01      -0.91715E+02

```

ELEMENT 11, FROM NODE 21, TO NODE 23 - LENGTH = 21.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft  -250.63    -226.67    -205.00    -185.40    -167.67    -151.64
tdisp,ft   0.08281    0.08376    0.08447    0.08498    0.08531    0.08546
axial,k    0.00 AT 1st END and      0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft  -151.64    -137.14    -124.02    -112.15    -101.42    -91.71
tdisp,ft   0.08546    0.08546    0.08532    0.08506    0.08469    0.08421
axial,k    0.00 AT 1st END and      0.00 AT 2nd END

```

```

-----
ELEMENT 12 DISPLACEMENTS IN INCIDENCES 23 24 25
NODE 23      0.00000E+00      0.84208E-01      -0.25021E-03
NODE 24      0.00000E+00      0.80362E-01      -0.72842E-03
NODE 25      0.00000E+00      0.75563E-01      -0.52733E-03
FORCES ACTING ALONG THE 9 DOF
NODE 23      0.00000E+00      0.69021E+01      0.91715E+02
NODE 24      0.00000E+00      0.22534E-11      -0.25101E-10
NODE 25      0.00000E+00      -0.69021E+01      -0.33418E+02

```

ELEMENT 12, FROM NODE 23, TO NODE 25 - LENGTH = 21.00 ft

```

left half of span,at tenth points of length
span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft  -91.71     -82.93     -74.99     -67.81     -61.31     -55.43
tdisp,ft   0.08421    0.08364    0.08298    0.08225    0.08146    0.08060
axial,k    0.00 AT 1st END and      0.00 AT 2nd END

```

```

right half of span,at tenth points of length
span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000      0.000      0.000      0.000      0.000      0.000
shear,k    6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft  -55.43     -50.10     -45.29     -40.93     -36.99     -33.42
tdisp,ft   0.08060    0.07968    0.07872    0.07771    0.07665    0.07556
axial,k    0.00 AT 1st END and      0.00 AT 2nd END

```

```

-----
ELEMENT 13 DISPLACEMENTS IN INCIDENCES 25 26 27
NODE 25      0.00000E+00      0.75563E-01      -0.52733E-03
NODE 26      0.00000E+00      0.69583E-01      -0.10283E-02
NODE 27      0.00000E+00      0.63259E-01      -0.62744E-03
FORCES ACTING ALONG THE 9 DOF
NODE 25      0.00000E+00      0.69021E+01      0.33418E+02
NODE 26      0.00000E+00      -0.30136E-11      -0.12480E-10
NODE 27      0.00000E+00      -0.69021E+01      -0.11786E+02

```

BBYOUT  
 ELEMENT 13, FROM NODE 25, TO NODE 27 - LENGTH = 21.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	6.90	6.90	6.90	6.90	6.90	6.90
bmom,kft	-33.42	-30.19	-27.26	-24.61	-22.20	-20.02
tdisp,ft	0.07556	0.07444	0.07328	0.07210	0.07090	0.06967
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

right half of span,at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	6.90	6.90	6.90	6.90	6.90	6.90
bmom,kft	-20.02	-18.05	-16.25	-14.62	-13.14	-11.79
tdisp,ft	0.06967	0.06842	0.06715	0.06587	0.06457	0.06326
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

-----  
 ELEMENT 14 DISPLACEMENTS IN INCIDENCES 27 28 29  
 NODE 27 0.00000E+00 0.63259E-01 -0.62744E-03  
 NODE 28 0.00000E+00 0.56519E-01 -0.11340E-02  
 NODE 29 0.00000E+00 0.49665E-01 -0.66038E-03  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 27 0.00000E+00 0.69021E+01 0.11786E+02  
 NODE 28 0.00000E+00 0.64125E-11 -0.84248E-10  
 NODE 29 0.00000E+00 -0.69021E+01 -0.30859E+01

ELEMENT 14, FROM NODE 27, TO NODE 29 - LENGTH = 21.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	6.90	6.90	6.90	6.90	6.90	6.90
bmom,kft	-11.79	-10.56	-9.43	-8.40	-7.46	-6.59
tdisp,ft	0.06326	0.06194	0.06060	0.05926	0.05791	0.05655
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

right half of span,at tenth points of length

	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	6.90	6.90	6.90	6.90	6.90	6.90
bmom,kft	-6.59	-5.78	-5.04	-4.35	-3.70	-3.09
tdisp,ft	0.05655	0.05518	0.05381	0.05243	0.05105	0.04967
axial,k	0.00	AT 1st END and		0.00	AT 2nd END	

-----  
 ELEMENT 15 DISPLACEMENTS IN INCIDENCES 29 30 31  
 NODE 29 0.00000E+00 0.49665E-01 -0.66038E-03  
 NODE 30 0.00000E+00 0.42702E-01 -0.11617E-02  
 NODE 31 0.00000E+00 0.35733E-01 -0.66228E-03  
 FORCES ACTING ALONG THE 9 DOF  
 NODE 29 0.00000E+00 0.69021E+01 0.30859E+01  
 NODE 30 0.00000E+00 0.46169E-12 0.54601E-11  
 NODE 31 0.00000E+00 -0.69021E+01 0.22290E+01

ELEMENT 15, FROM NODE 29, TO NODE 31 - LENGTH = 21.00 ft

left half of span,at tenth points of length

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```

BBYOUT
      span      span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000  0.000  0.000  0.000  0.000  0.000
shear,k    6.90   6.90   6.90   6.90   6.90   6.90
bmom,kft  -3.09  -2.51  -1.95  -1.41  -0.89  -0.38
tdisp,ft  0.04967 0.04828 0.04689 0.04549 0.04410 0.04270
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
      span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000  0.000  0.000  0.000  0.000  0.000
shear,k    6.90   6.90   6.90   6.90   6.90   6.90
bmom,kft  -0.38   0.13   0.64   1.16   1.68   2.23
tdisp,ft  0.04270 0.04131 0.03991 0.03852 0.03713 0.03573
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 16 DISPLACEMENTS IN INCIDENCES 31 32 33
  NODE 31 0.00000E+00 0.35733E-01 -0.66228E-03
  NODE 32 0.00000E+00 0.28843E-01 -0.11417E-02
  NODE 33 0.00000E+00 0.22046E-01 -0.63522E-03
FORCES ACTING ALONG THE 9 DOF
  NODE 31 0.00000E+00 0.69021E+01 -0.22290E+01
  NODE 32 0.00000E+00 -0.14607E-11 -0.14266E-10
  NODE 33 0.00000E+00 -0.69021E+01 0.99895E+01

ELEMENT 16, FROM NODE 31, TO NODE 33 - LENGTH = 21.00 ft

left half of span,at tenth points of length
      span      span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000  0.000  0.000  0.000  0.000  0.000
shear,k    6.90   6.90   6.90   6.90   6.90   6.90
bmom,kft  2.23   2.80   3.39   4.02   4.69   5.41
tdisp,ft  0.03573 0.03434 0.03296 0.03157 0.03019 0.02882
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
      span      span      span      span      span      span
0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft 0.000  0.000  0.000  0.000  0.000  0.000
shear,k    6.90   6.90   6.90   6.90   6.90   6.90
bmom,kft  5.41   6.18   7.02   7.93   8.91   9.99
tdisp,ft  0.02882 0.02745 0.02609 0.02473 0.02338 0.02205
axial,k    0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 17 DISPLACEMENTS IN INCIDENCES 33 34 35
  NODE 33 0.00000E+00 0.22046E-01 -0.63522E-03
  NODE 34 0.00000E+00 0.15601E-01 -0.10521E-02
  NODE 35 0.00000E+00 0.94514E-02 -0.54951E-03
FORCES ACTING ALONG THE 9 DOF
  NODE 33 0.00000E+00 0.69021E+01 -0.99895E+01
  NODE 34 0.00000E+00 -0.39885E-13 0.82379E-11
  NODE 35 0.00000E+00 -0.69021E+01 0.28710E+02

ELEMENT 17, FROM NODE 33, TO NODE 35 - LENGTH = 21.00 ft

left half of span,at tenth points of length
      span      span      span      span      span      span
0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft 0.000  0.000  0.000  0.000  0.000  0.000

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shear,k      6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft    9.99      11.17     12.46     13.87     15.43     17.14
tdisp,ft    0.02205   0.02072   0.01940   0.01809   0.01680   0.01553
axial,k      0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span        span        span        span        span        span
0.5         0.6         0.7         0.8         0.9         1.0
soil,k/ft   0.000      0.000      0.000      0.000      0.000      0.000
shear,k     6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft    17.14     19.03     21.10     23.39     25.92     28.71
tdisp,ft    0.01553   0.01427   0.01303   0.01181   0.01062   0.00945
axial,k     0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 18 DISPLACEMENTS IN INCIDENCES 35 36 37
NODE 35 0.00000E+00 0.94514E-02 -0.54951E-03
NODE 36 0.00000E+00 0.43154E-02 -0.79447E-03
NODE 37 0.00000E+00 0.00000E+00 -0.31114E-03
FORCES ACTING ALONG THE 9 DOF
NODE 35 0.00000E+00 0.69021E+01 -0.28710E+02
NODE 36 0.00000E+00 0.61062E-15 -0.29798E-11
NODE 37 0.00000E+00 -0.69021E+01 0.78929E+02

ELEMENT 18, FROM NODE 35, TO NODE 37 - LENGTH = 21.00 ft

left half of span,at tenth points of length
span        span        span        span        span        span
0.0         0.1         0.2         0.3         0.4         0.5
soil,k/ft   0.000      0.000      0.000      0.000      0.000      0.000
shear,k     6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft    28.71     31.79     35.19     38.94     43.09     47.68
tdisp,ft    0.00945   0.00831   0.00721   0.00613   0.00510   0.00411
axial,k     0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length
span        span        span        span        span        span
0.5         0.6         0.7         0.8         0.9         1.0
soil,k/ft   0.000      0.000      0.000      0.000      0.000      0.000
shear,k     6.90      6.90      6.90      6.90      6.90      6.90
bmom,kft    47.68     52.74     58.34     64.53     71.37     78.93
tdisp,ft    0.00411   0.00317   0.00228   0.00146   0.00069   0.00000
axial,k     0.00 AT 1st END and 0.00 AT 2nd END

-----
ELEMENT 19 DISPLACEMENTS IN INCIDENCES 37 38 39
NODE 37 0.00000E+00 0.00000E+00 -0.31114E-03
NODE 38 0.00000E+00 -0.21499E-02 -0.28879E-03
NODE 39 0.00000E+00 -0.35379E-02 -0.79102E-04
FORCES ACTING ALONG THE 9 DOF
NODE 37 0.00000E+00 -0.65774E+00 -0.78929E+02
NODE 38 0.00000E+00 0.82531E-13 0.65675E-12
NODE 39 0.00000E+00 0.65774E+00 0.30316E+02

ELEMENT 19, FROM NODE 37, TO NODE 39 - LENGTH = 20.00 ft

left half of span,at tenth points of length
span        span        span        span        span        span
0.0         0.1         0.2         0.3         0.4         0.5
soil,k/ft   0.000      0.000      0.000      0.000      0.000      0.000
shear,k     -0.66     -0.66     -0.66     -0.66     -0.66     -0.66
bmom,kft    78.93     71.73     65.18     59.23     53.83     48.92
tdisp,ft    0.00000   -0.00059  -0.00111  -0.00157  -0.00198  -0.00234

```

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48.92/120=0.41 ~ 0.66

axial,k 0.00 AT 1st END and BBYOUT 0.00 AT 2nd END

right half of span,at tenth points of length

	span 0.5	span 0.6	span 0.7	span 0.8	span 0.9	span 1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	48.92	44.45	40.40	36.71	33.36	30.32
tdisp,ft	-0.00234	-0.00265	-0.00293	-0.00316	-0.00337	-0.00354

axial,k 0.00 AT 1st END and 0.00 AT 2nd END

-----

ELEMENT	20	DISPLACEMENTS IN	INCIDENCES	39	40	41
NODE	39	0.00000E+00		-0.35379E-02		-0.79102E-04
NODE	40	0.00000E+00		-0.39596E-02		-0.43587E-04
NODE	41	0.00000E+00		-0.40888E-02		0.10012E-04
FORCES ACTING ALONG THE 9 DOF						
NODE	39	0.00000E+00		-0.65774E+00		-0.30316E+02
NODE	40	0.00000E+00		-0.37181E-12		-0.12364E-11
NODE	41	0.00000E+00		0.65774E+00		0.11640E+02

ELEMENT 20, FROM NODE 39, TO NODE 41 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span 0.0	span 0.1	span 0.2	span 0.3	span 0.4	span 0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	30.32	27.55	25.03	22.75	20.67	18.79
tdisp,ft	-0.00354	-0.00368	-0.00380	-0.00390	-0.00397	-0.00403

axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length

	span 0.5	span 0.6	span 0.7	span 0.8	span 0.9	span 1.0
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	18.79	17.07	15.51	14.10	12.81	11.64
tdisp,ft	-0.00403	-0.00407	-0.00410	-0.00411	-0.00410	-0.00409

axial,k 0.00 AT 1st END and 0.00 AT 2nd END

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ELEMENT	21	DISPLACEMENTS IN	INCIDENCES	41	42	43
NODE	41	0.00000E+00		-0.40888E-02		0.10012E-04
NODE	42	0.00000E+00		-0.38469E-02		0.50560E-04
NODE	43	0.00000E+00		-0.34927E-02		0.44206E-04
FORCES ACTING ALONG THE 9 DOF						
NODE	41	0.00000E+00		-0.65774E+00		-0.11640E+02
NODE	42	0.00000E+00		-0.18067E-12		0.41511E-11
NODE	43	0.00000E+00		0.65774E+00		0.44588E+01

ELEMENT 21, FROM NODE 41, TO NODE 43 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span 0.0	span 0.1	span 0.2	span 0.3	span 0.4	span 0.5
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	11.64	10.58	9.61	8.73	7.93	7.21
tdisp,ft	-0.00409	-0.00406	-0.00403	-0.00399	-0.00393	-0.00387

axial,k 0.00 AT 1st END and 0.00 AT 2nd END

right half of span,at tenth points of length

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	BBYOUT						
	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	7.21	6.55	5.95	5.40	4.91	4.46	
tdisp,ft	-0.00387	-0.00381	-0.00374	-0.00366	-0.00358	-0.00349	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		

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ELEMENT 22 DISPLACEMENTS IN INCIDENCES 43 44 45
  NODE 43 0.00000E+00 -0.34927E-02 0.44206E-04
  NODE 44 0.00000E+00 -0.29965E-02 0.86625E-04
  NODE 45 0.00000E+00 -0.24574E-02 0.57247E-04
FORCES ACTING ALONG THE 9 DOF
  NODE 43 0.00000E+00 -0.65774E+00 -0.44588E+01
  NODE 44 0.00000E+00 -0.31086E-14 0.35572E-11
  NODE 45 0.00000E+00 0.65774E+00 0.16808E+01

```

ELEMENT 22, FROM NODE 43, TO NODE 45 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	4.46	4.05	3.68	3.34	3.03	2.75	
tdisp,ft	-0.00349	-0.00340	-0.00331	-0.00321	-0.00311	-0.00301	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	2.75	2.49	2.26	2.05	1.86	1.68	
tdisp,ft	-0.00301	-0.00290	-0.00279	-0.00268	-0.00257	-0.00246	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		

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ELEMENT 23 DISPLACEMENTS IN INCIDENCES 45 46 47
  NODE 45 0.00000E+00 -0.24574E-02 0.57247E-04
  NODE 46 0.00000E+00 -0.18647E-02 0.10022E-03
  NODE 47 0.00000E+00 -0.12564E-02 0.62012E-04
FORCES ACTING ALONG THE 9 DOF
  NODE 45 0.00000E+00 -0.65774E+00 -0.16808E+01
  NODE 46 0.00000E+00 0.49689E-13 -0.59464E-12
  NODE 47 0.00000E+00 0.65774E+00 0.56262E+00

```

ELEMENT 23, FROM NODE 45, TO NODE 47 - LENGTH = 20.00 ft

left half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	1.68	1.52	1.37	1.24	1.12	1.00	
tdisp,ft	-0.00246	-0.00234	-0.00223	-0.00211	-0.00199	-0.00187	
axial,k	0.00	AT 1st END and		0.00	AT 2nd END		

right half of span,at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000

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shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	1.00	0.90	0.81	0.72	0.64	0.56	0.56
tdisp,ft	-0.00187	-0.00175	-0.00163	-0.00150	-0.00138	-0.00126	-0.00126
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

-----

ELEMENT	24	DISPLACEMENTS IN INCIDENCES	47	48	49	
NODE	47	0.00000E+00	-0.12564E-02			0.62012E-04
NODE	48	0.00000E+00	-0.63016E-03			0.10477E-03
NODE	49	0.00000E+00	0.00000E+00			0.63207E-04
FORCES ACTING ALONG THE 9 DOF						
NODE	47	0.00000E+00	-0.65774E+00			-0.56262E+00
NODE	48	0.00000E+00	0.27485E-13			0.29610E-12
NODE	49	0.00000E+00	0.65774E+00			0.26197E-12

ELEMENT 24, FROM NODE 47, TO NODE 49 - LENGTH = 20.00 ft

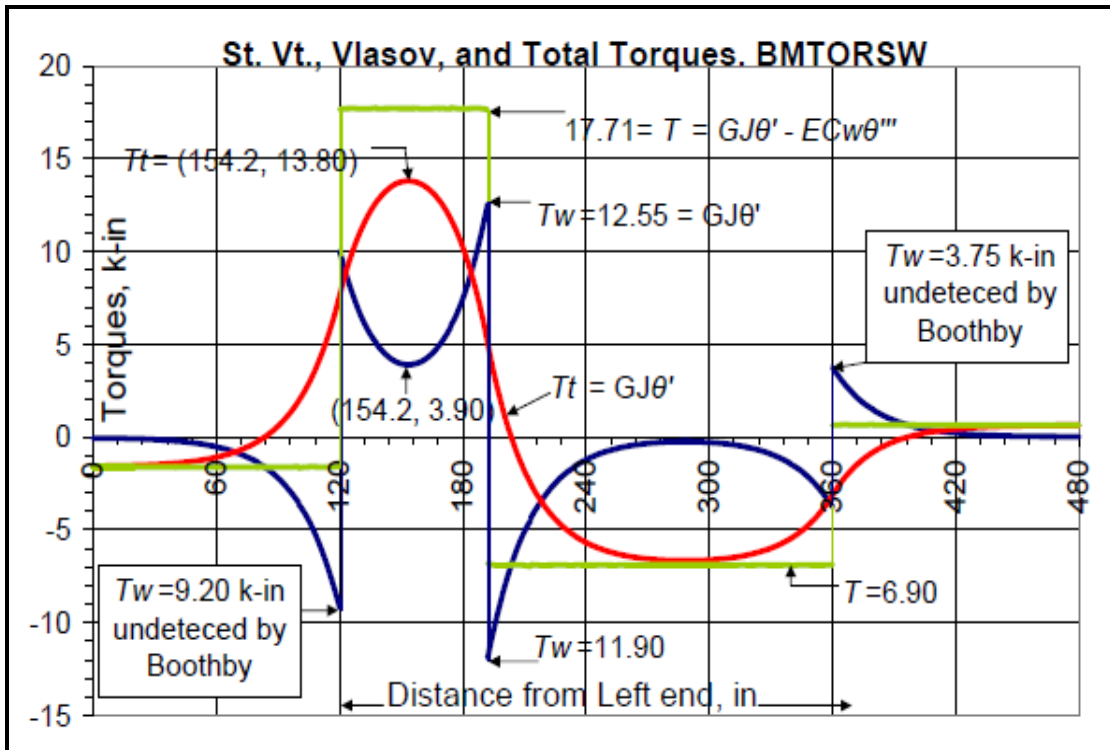
left half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	0.56	0.49	0.43	0.37	0.31	0.25	0.25
tdisp,ft	-0.00126	-0.00113	-0.00101	-0.00088	-0.00076	-0.00063	-0.00063
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

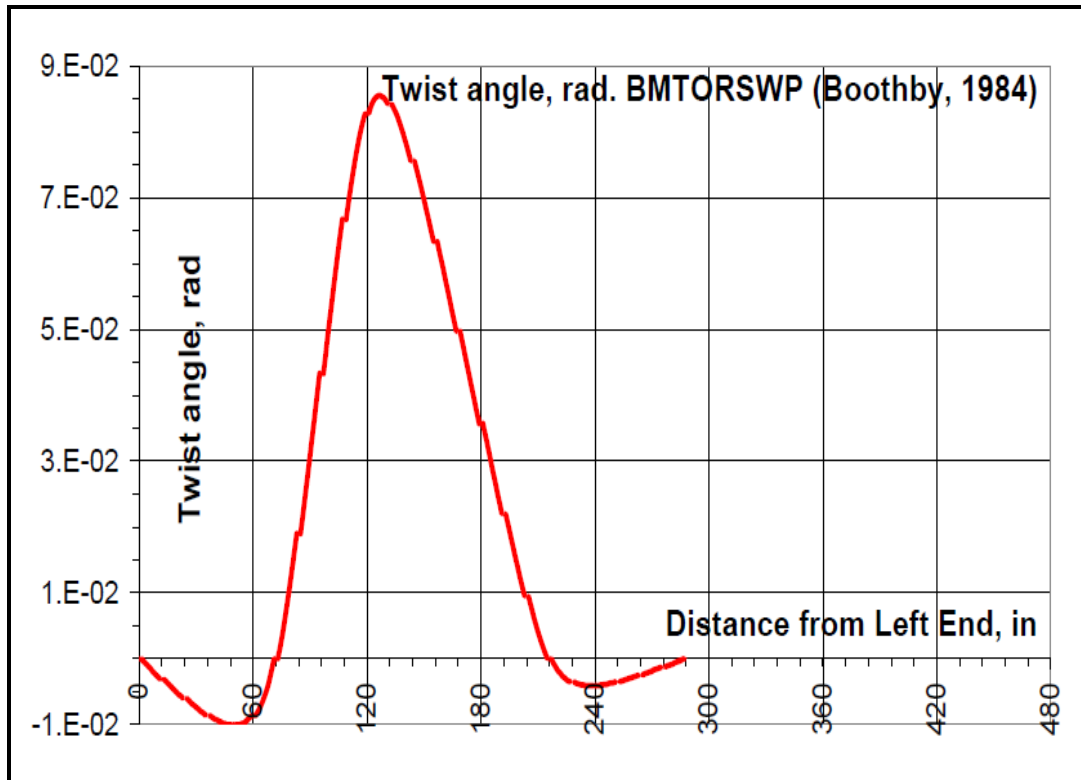
right half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil,k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear,k	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66	-0.66
bmom,kft	0.25	0.20	0.15	0.10	0.05	0.00	0.00
tdisp,ft	-0.00063	-0.00051	-0.00038	-0.00025	-0.00013	0.00000	0.00000
axial,k	0.00	AT 1st END	and	0.00	AT 2nd END		

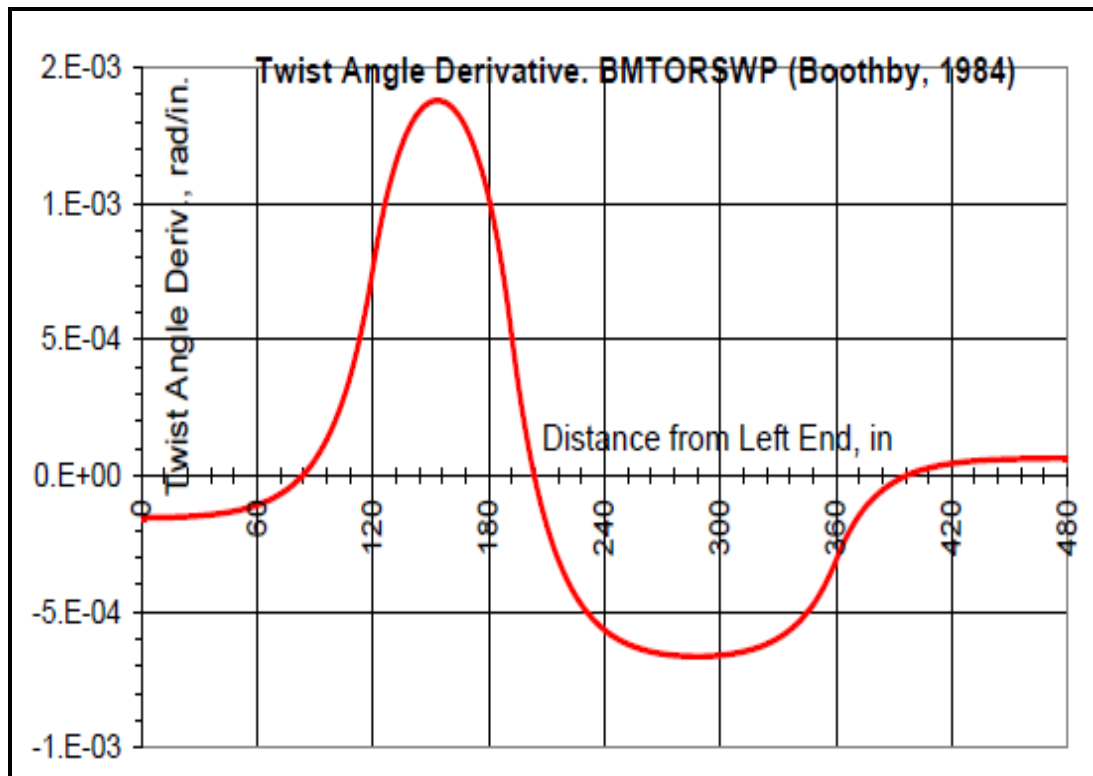
Output Final Page



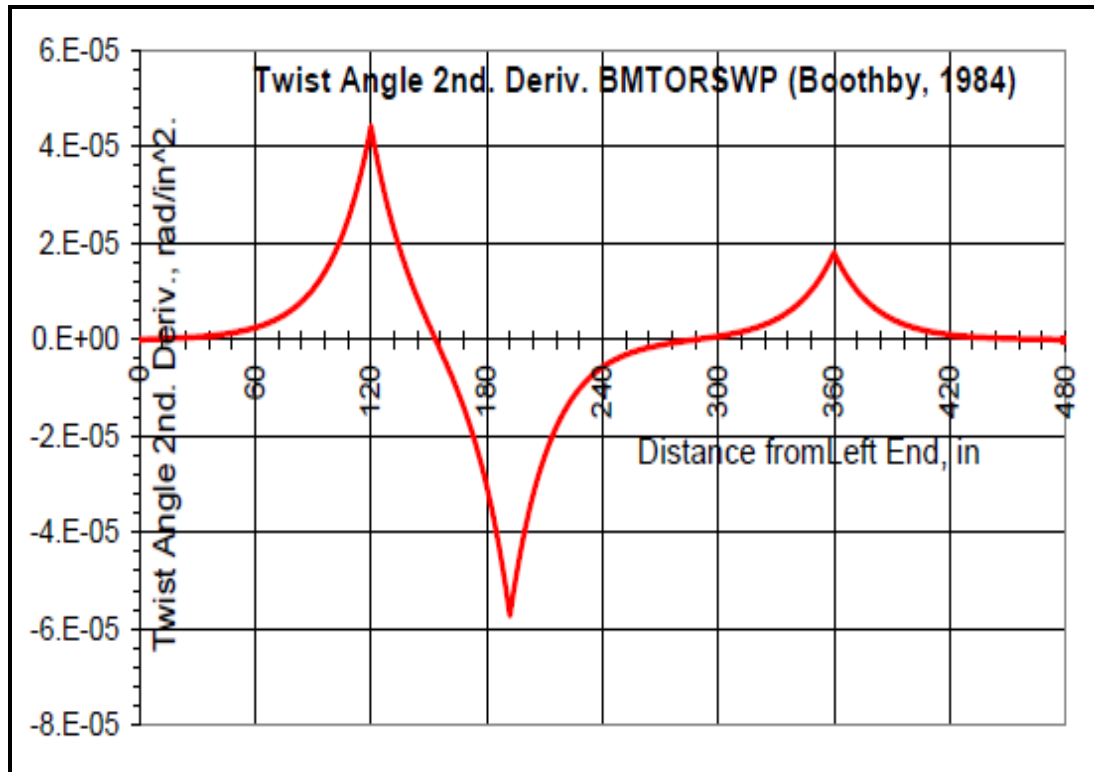
Uniform, Non Uniform and Total Torques from BMTORSWP



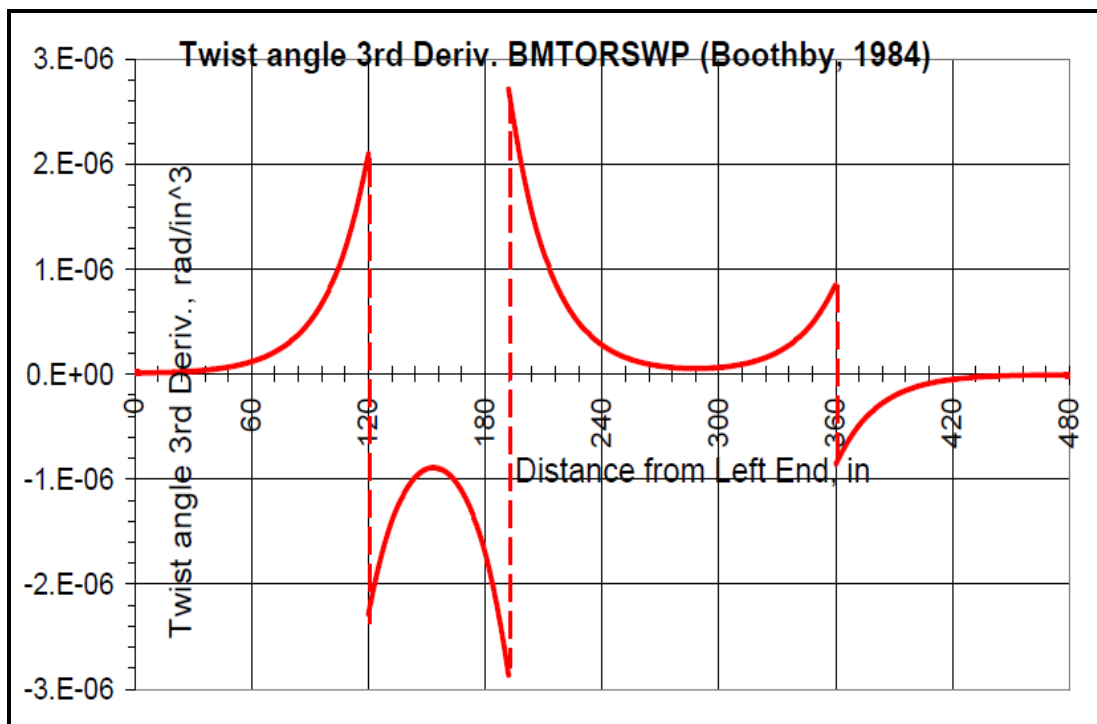
Torque Angle from BMTORSWP



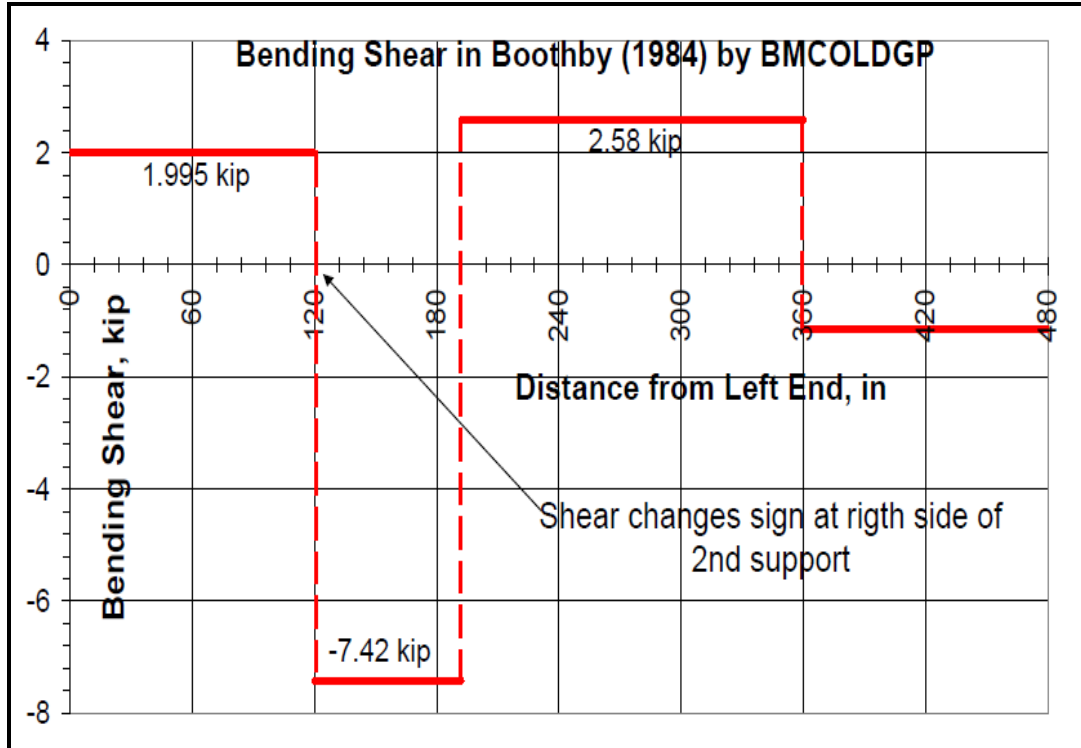
Torque Angle 1<sup>st</sup> Derivative



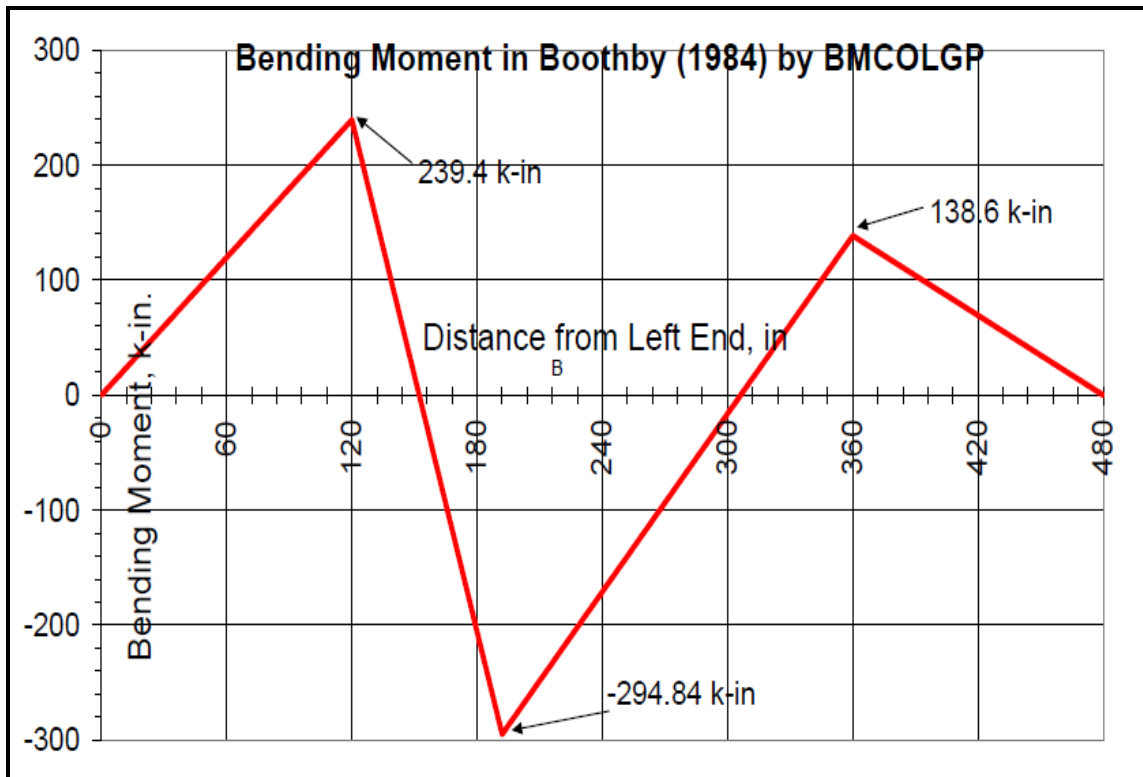
Torque Angle 2<sup>nd</sup> Derivative



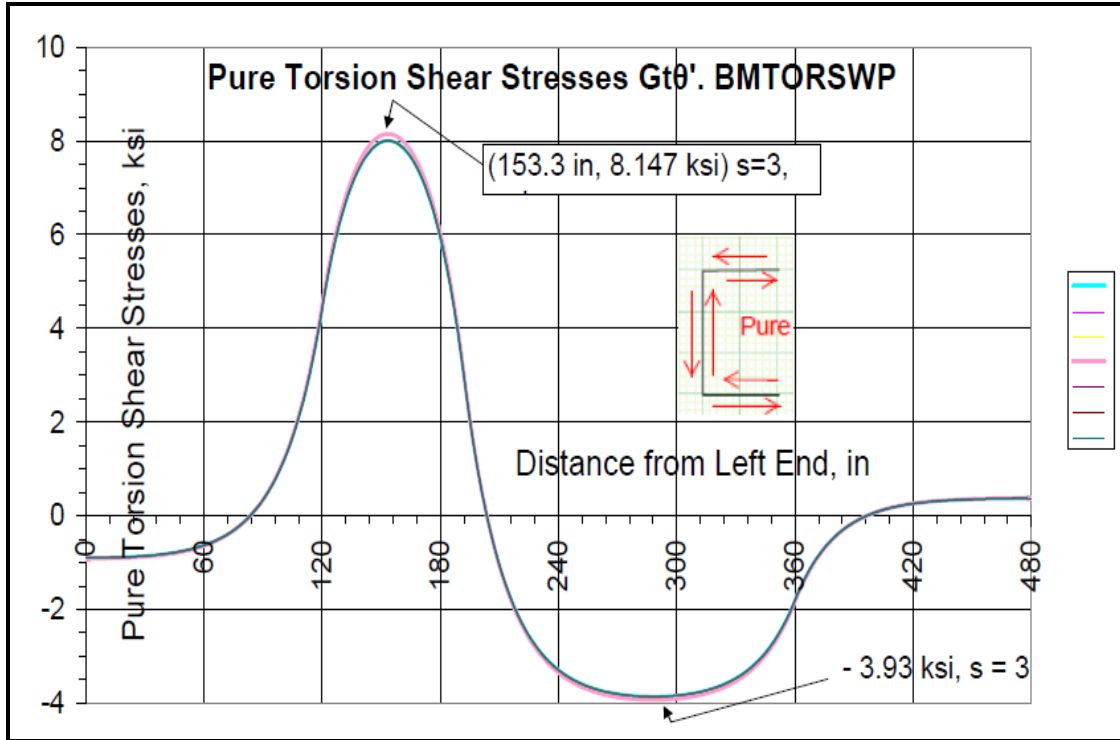
Torque Angle 3<sup>rd</sup> Derivative



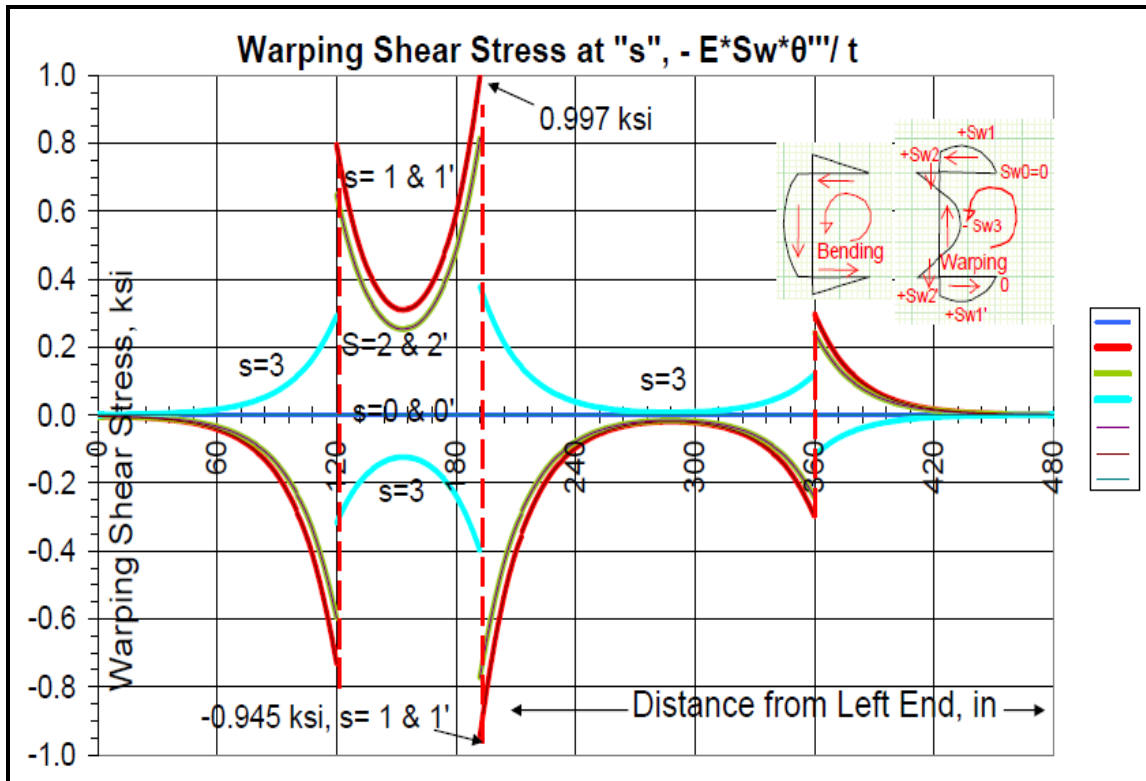
Bending Shear from BMCOLDGP



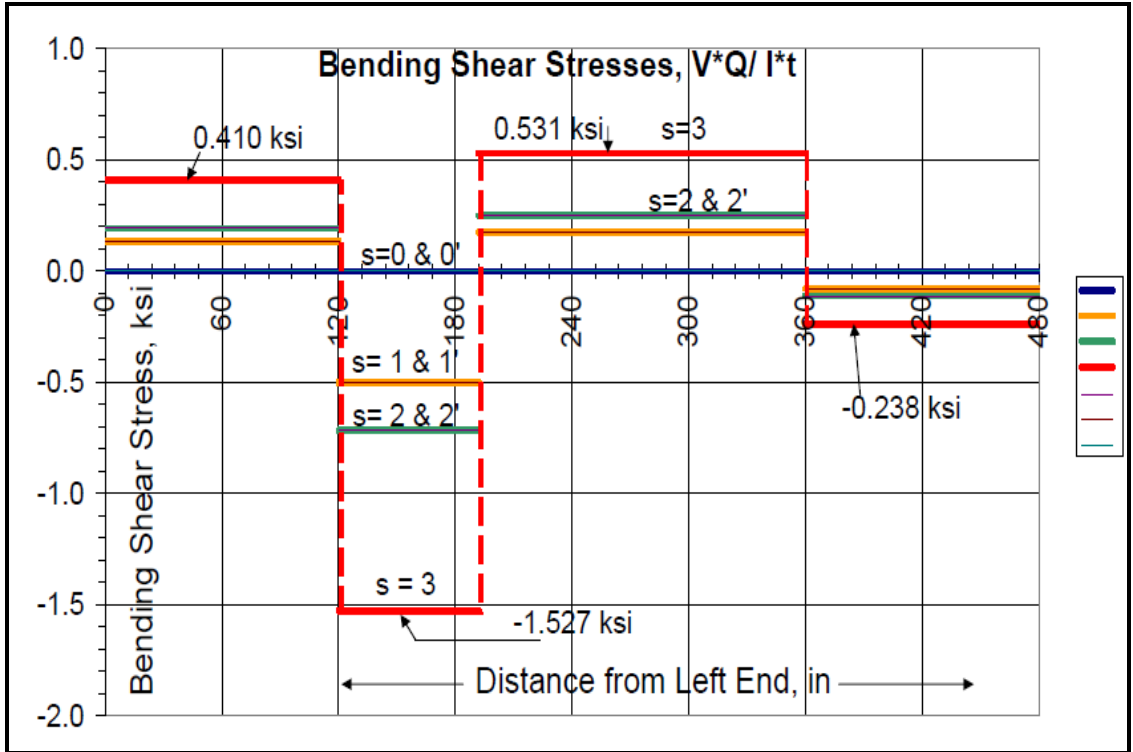
Bending Moment from BMCOLDGP



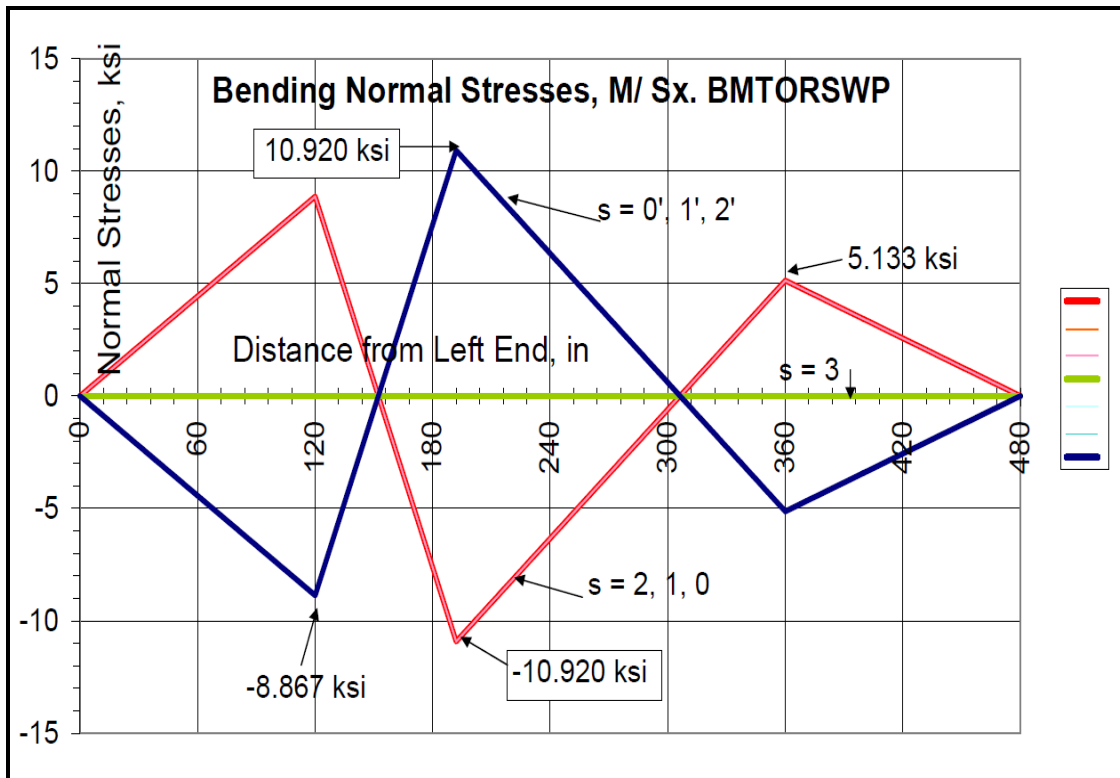
Pure Torsion Shear Stresses at  $s=0, 1, 2, 3, 2', 1'$  and  $0'$



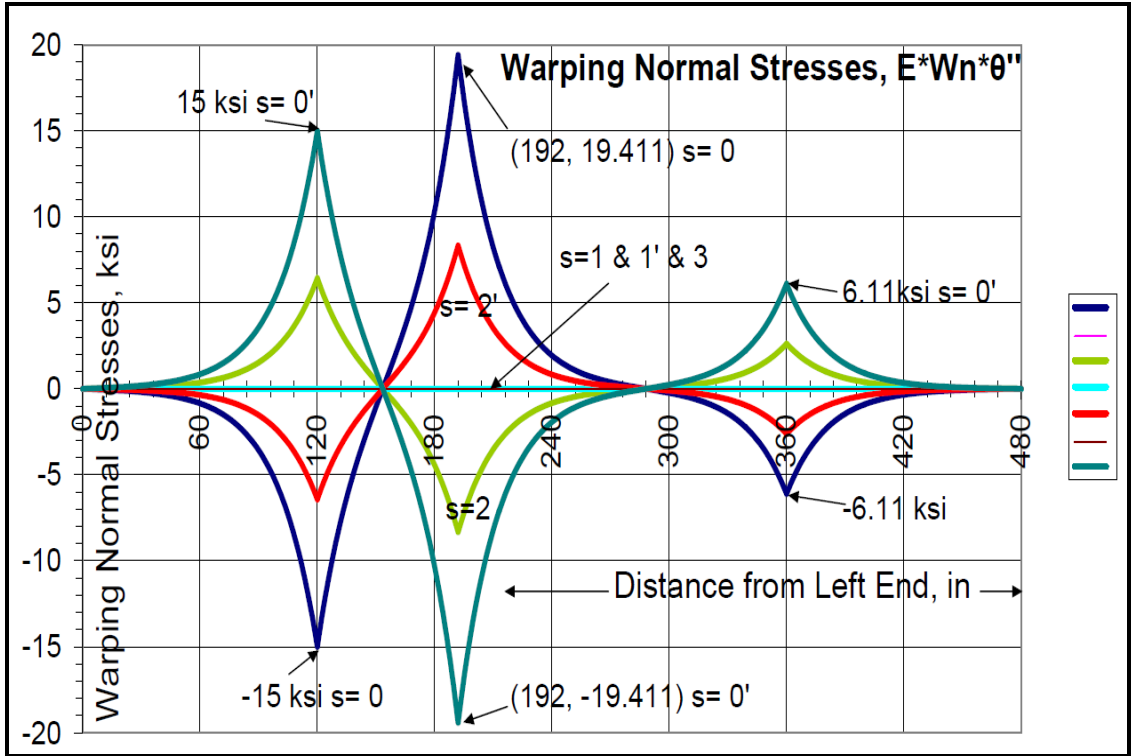
Warping Shear Stresses along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')



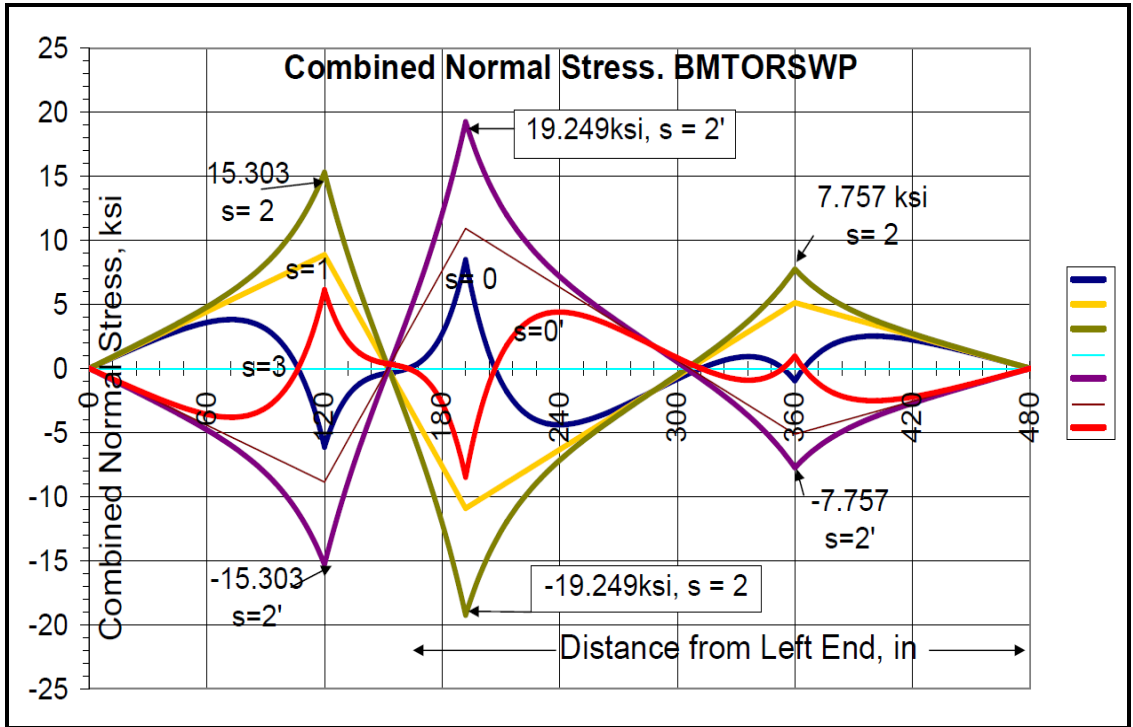
Bending Shear Stresses along the Spans and Profile Points, Legend in Order



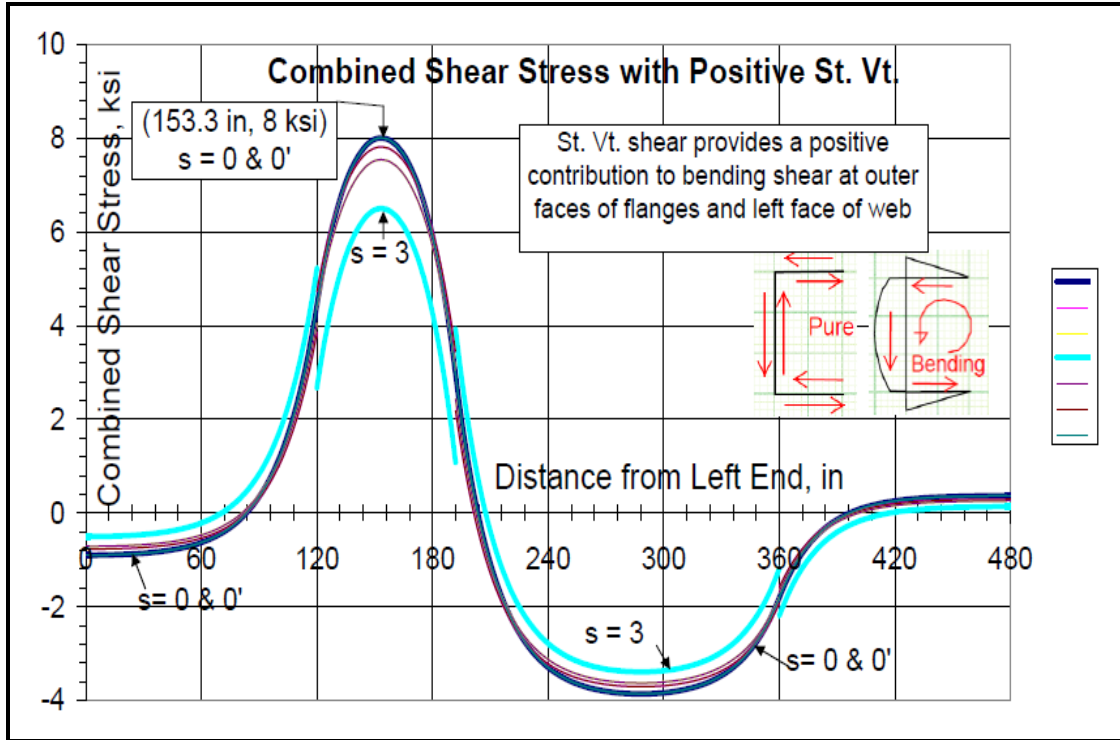
Bending Normal Stresses along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')



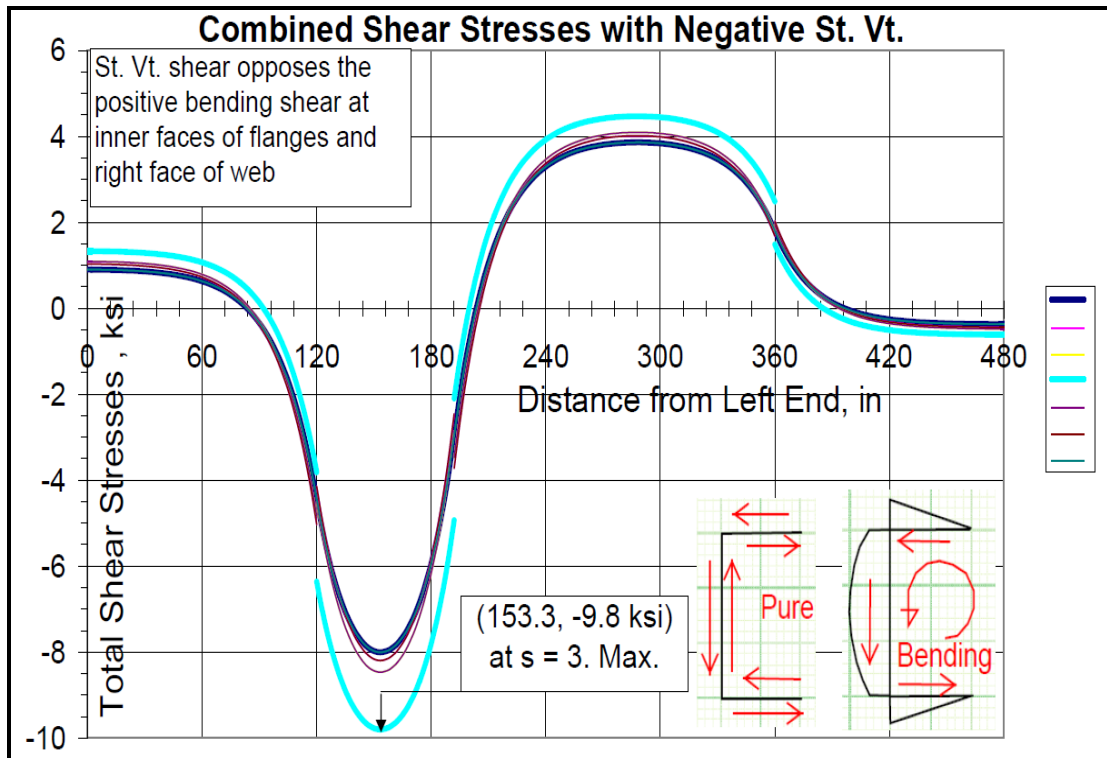
Warping Normal Stresses along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')



Combined Normal Stress along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')

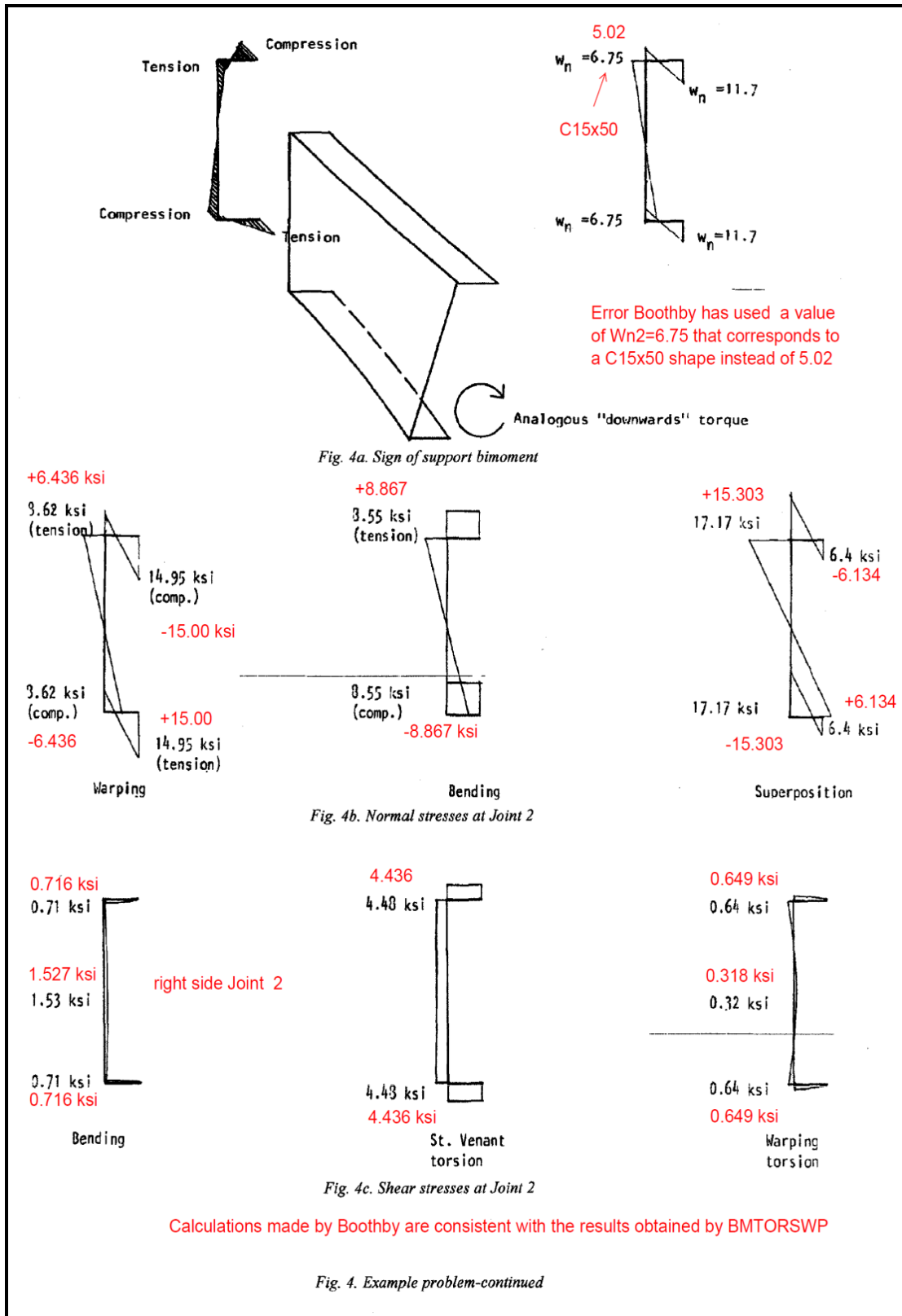


First Case of Combined Shear Stress along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')

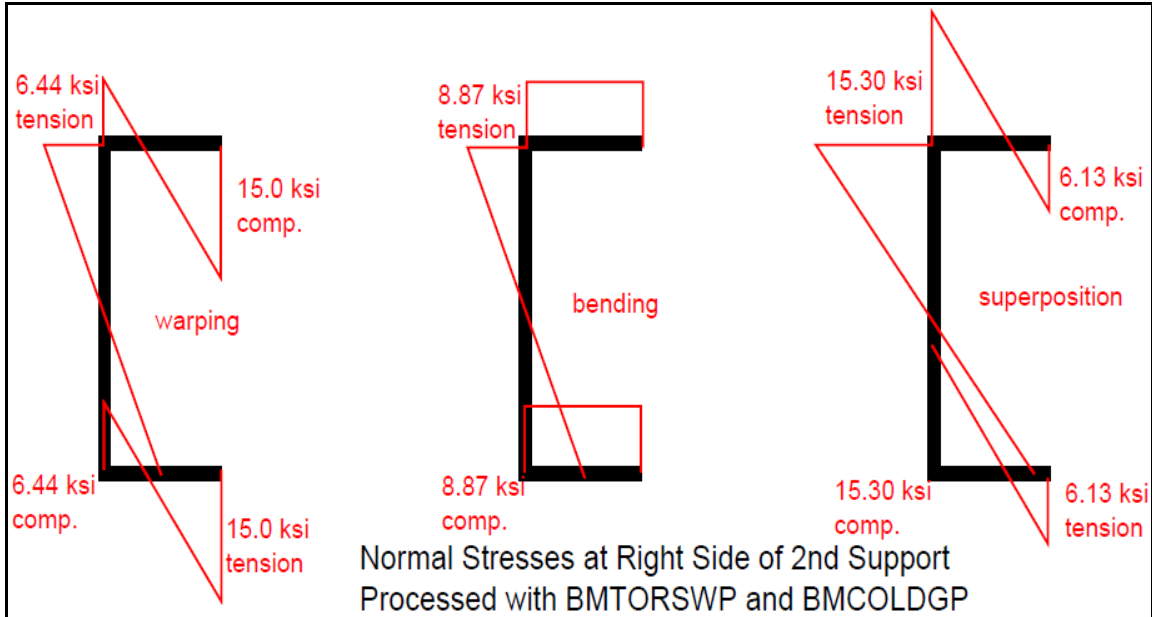


Second Case of Combined Shear Stress along the Spans and Profile Points, Legend in Order (0, 1, 2, 3, 2', 1', 0')

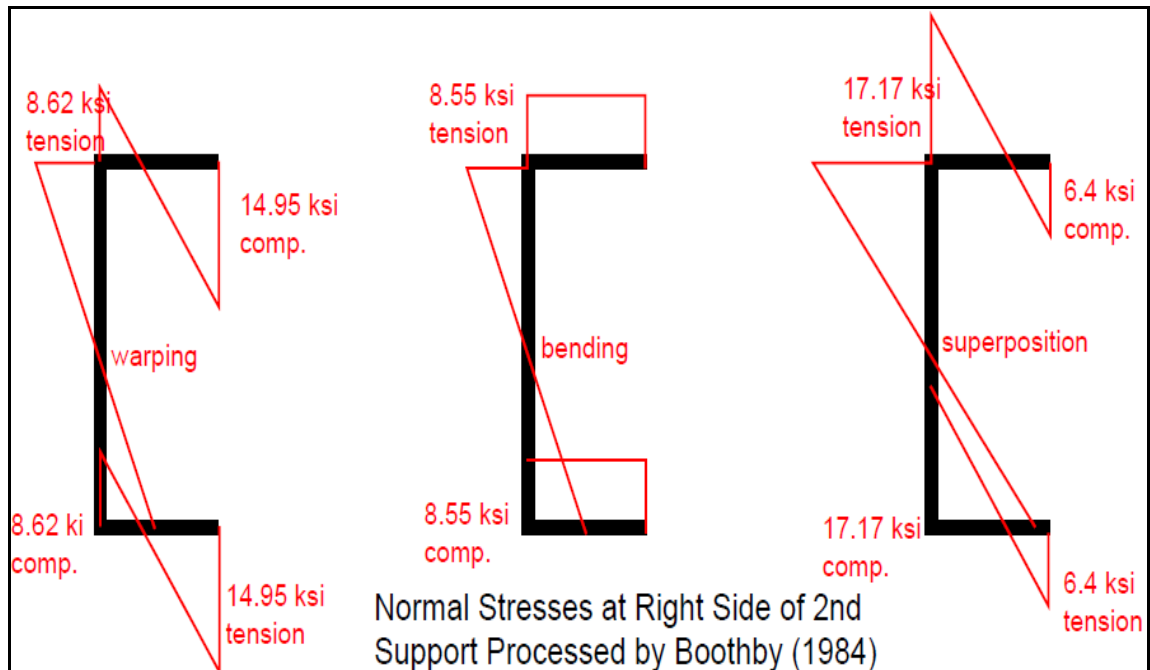




BMTORSWP and Boothby's Calculations for Right Side of First Interior Support



Another View of the Normal Stresses at Right Side of 2nd Support Processed with BMTORSWP and BMCOLDGP



Normal Stresses at Right Side of 2<sup>nd</sup> Support Processed by Boothby (1984)

The error committed by Boothby for the sectorial coordinate at the flange-web corner is illustrated in the previous page. Regarding shear stresses, no discrepancies were found as shown in the same page.

Torques and Torque Angles and Derivatives and Bimoment									
z	$T_w$	$T_t$	$T$	El	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
	$-EC_w\theta''' + GJ\theta' = T$			1					Bim
0.0	-0.060	-1.554	-1.614		0.00E+00	-1.551E-04	-6.305E-11	1.365E-08	0.00
2.0	-0.060	-1.553	-1.613		-3.00E-04	-1.550E-04	2.723E-08	1.367E-08	0.12
4.0	-0.061	-1.552	-1.613		-6.00E-04	-1.549E-04	5.471E-08	1.384E-08	0.24
6.0	-0.062	-1.551	-1.613		-9.00E-04	-1.548E-04	8.267E-08	1.414E-08	0.36
8.0	-0.064	-1.549	-1.613		-1.20E-03	-1.546E-04	1.114E-07	1.459E-08	0.49
10.0	-0.066	-1.547	-1.614		-1.50E-03	-1.544E-04	1.411E-07	1.517E-08	0.62
12.0	-0.070	-1.543	-1.613		-1.90E-03	-1.540E-04	1.721E-07	1.590E-08	0.75
14.0	-0.073	-1.540	-1.614		-2.20E-03	-1.537E-04	2.048E-07	1.677E-08	0.90
16.0	-0.078	-1.535	-1.613		-2.50E-03	-1.532E-04	2.393E-07	1.777E-08	1.05
18.0	-0.083	-1.530	-1.613		-2.80E-03	-1.527E-04	2.760E-07	1.892E-08	1.21
20.0	-0.088	-1.524	-1.613		-3.10E-03	-1.521E-04	3.151E-07	2.021E-08	1.38
				2					
20.0	-0.090	-1.524	-1.614		-3.10E-03	-1.521E-04	3.149E-07	2.056E-08	1.38
22.0	-0.096	-1.518	-1.614		-3.40E-03	-1.515E-04	3.573E-07	2.192E-08	1.56
24.0	-0.103	-1.510	-1.614		-3.70E-03	-1.507E-04	4.028E-07	2.356E-08	1.76
26.0	-0.112	-1.501	-1.613		-4.00E-03	-1.498E-04	4.518E-07	2.548E-08	1.98
28.0	-0.121	-1.492	-1.614		-4.30E-03	-1.489E-04	5.049E-07	2.768E-08	2.21
30.0	-0.132	-1.481	-1.613		-4.60E-03	-1.478E-04	5.626E-07	3.015E-08	2.46
32.0	-0.144	-1.469	-1.613		-4.90E-03	-1.466E-04	6.257E-07	3.291E-08	2.74
34.0	-0.157	-1.456	-1.614		-5.20E-03	-1.453E-04	6.945E-07	3.595E-08	3.04
36.0	-0.172	-1.441	-1.613		-5.50E-03	-1.438E-04	7.697E-07	3.927E-08	3.37
38.0	-0.188	-1.425	-1.613		-5.70E-03	-1.422E-04	8.517E-07	4.286E-08	3.73
40.0	-0.205	-1.407	-1.612		-6.00E-03	-1.404E-04	9.413E-07	4.674E-08	4.12
				3					
40.0	-0.209	-1.407	-1.616		-6.00E-03	-1.404E-04	9.409E-07	4.777E-08	4.12
42.0	-0.227	-1.388	-1.615		-6.30E-03	-1.385E-04	1.040E-06	5.181E-08	4.55
44.0	-0.248	-1.366	-1.614		-6.60E-03	-1.363E-04	1.149E-06	5.654E-08	5.03
46.0	-0.271	-1.342	-1.613		-6.90E-03	-1.339E-04	1.267E-06	6.197E-08	5.55
48.0	-0.298	-1.315	-1.613		-7.10E-03	-1.312E-04	1.397E-06	6.810E-08	6.12
50.0	-0.328	-1.286	-1.614		-7.40E-03	-1.283E-04	1.540E-06	7.491E-08	6.74
52.0	-0.361	-1.253	-1.614		-7.60E-03	-1.250E-04	1.697E-06	8.243E-08	7.43
54.0	-0.397	-1.218	-1.615		-7.90E-03	-1.215E-04	1.870E-06	9.063E-08	8.19
56.0	-0.436	-1.178	-1.614		-8.10E-03	-1.175E-04	2.060E-06	9.954E-08	9.02
58.0	-0.478	-1.135	-1.612		-8.30E-03	-1.132E-04	2.269E-06	1.091E-07	9.93
60.0	-0.523	-1.086	-1.609		-8.60E-03	-1.084E-04	2.497E-06	1.194E-07	10.94
				4					
60.0	-0.535	-1.086	-1.622		-8.60E-03	-1.084E-04	2.496E-06	1.222E-07	10.94
62.0	-0.582	-1.034	-1.616		-8.80E-03	-1.032E-04	2.751E-06	1.329E-07	12.04
64.0	-0.637	-0.976	-1.613		-9.00E-03	-9.742E-05	3.029E-06	1.454E-07	13.26
66.0	-0.699	-0.913	-1.612		-9.20E-03	-9.107E-05	3.333E-06	1.597E-07	14.60
68.0	-0.770	-0.843	-1.612		-9.30E-03	-8.407E-05	3.668E-06	1.758E-07	16.07

(continued) Torques and Torque Angles and Derivatives and Bimoment

z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
70.0	-0.848	-0.765	-1.614		-9.50E-03	-7.637E-05	4.038E-06	1.937E-07	17.68
72.0	-0.934	-0.681	-1.615		-9.60E-03	-6.790E-05	4.444E-06	2.133E-07	19.46
74.0	-1.028	-0.587	-1.615		-9.80E-03	-5.857E-05	4.892E-06	2.348E-07	21.42
76.0	-1.130	-0.484	-1.614		-9.90E-03	-4.830E-05	5.385E-06	2.581E-07	23.58
78.0	-1.240	-0.371	-1.611		-1.00E-02	-3.700E-05	5.926E-06	2.832E-07	25.95
80.0	-1.357	-0.246	-1.604		-1.00E-02	-2.456E-05	6.519E-06	3.100E-07	28.55
				<b>5</b>					
80.0	-1.389	-0.246	-1.635		-1.00E-02	-2.456E-05	6.516E-06	3.172E-07	28.55
82.0	-1.511	-0.109	-1.620		-1.01E-02	-1.088E-05	7.177E-06	3.451E-07	31.42
84.0	-1.654	0.042	-1.612		-1.01E-02	4.189E-06	7.899E-06	3.777E-07	34.58
86.0	-1.817	0.208	-1.609		-1.00E-02	2.077E-05	8.691E-06	4.150E-07	38.06
88.0	-2.001	0.391	-1.610		-1.00E-02	3.901E-05	9.563E-06	4.570E-07	41.88
90.0	-2.205	0.592	-1.613		-9.90E-03	5.908E-05	1.052E-05	5.036E-07	46.09
92.0	-2.430	0.813	-1.616		-9.80E-03	8.116E-05	1.158E-05	5.549E-07	50.72
94.0	-2.675	1.057	-1.618		-9.60E-03	1.055E-04	1.275E-05	6.109E-07	55.81
96.0	-2.940	1.325	-1.616		-9.30E-03	1.322E-04	1.403E-05	6.715E-07	61.41
98.0	-3.227	1.621	-1.606		-9.00E-03	1.617E-04	1.543E-05	7.369E-07	67.58
100.0	-3.533	1.944	-1.589		-8.70E-03	1.940E-04	1.698E-05	8.068E-07	74.37
				<b>6</b>					
100.0	-3.614	1.944	-1.670		-8.70E-03	1.940E-04	1.697E-05	8.254E-07	74.37
102.0	-3.933	2.302	-1.631		-8.30E-03	2.297E-04	1.869E-05	8.982E-07	81.84
104.0	-4.305	2.695	-1.610		-7.80E-03	2.689E-04	2.057E-05	9.831E-07	90.05
106.0	-4.729	3.128	-1.601		-7.20E-03	3.121E-04	2.263E-05	1.080E-06	99.10
108.0	-5.211	3.604	-1.607		-6.50E-03	3.596E-04	2.490E-05	1.190E-06	109.05
110.0	-5.741	4.127	-1.614		-5.70E-03	4.118E-04	2.740E-05	1.311E-06	120.00
112.0	-6.328	4.704	-1.624		-4.90E-03	4.693E-04	3.015E-05	1.445E-06	132.05
114.0	-6.963	5.338	-1.625		-3.90E-03	5.326E-04	3.318E-05	1.590E-06	145.31
116.0	-7.654	6.036	-1.618		-2.70E-03	6.023E-04	3.652E-05	1.748E-06	159.90
118.0	-8.399	6.804	-1.595		-1.40E-03	6.789E-04	4.018E-05	1.918E-06	175.95
120.0	-9.196	7.650	-1.546		0.00E+00	7.633E-04	4.420E-05	2.100E-06	193.62
				<b>7</b>					
120.0	10.002	7.650	17.652		0.00E+00	7.633E-04	4.421E-05	-2.284E-06	193.62
121.8	9.275	8.411	17.686		1.40E-03	8.392E-04	4.025E-05	-2.118E-06	176.23
123.6	8.596	9.103	17.699		3.00E-03	9.083E-04	3.658E-05	-1.963E-06	160.15
125.4	7.974	9.733	17.707		4.70E-03	9.711E-04	3.317E-05	-1.821E-06	145.25
127.2	7.401	10.303	17.704		6.50E-03	1.028E-03	3.001E-05	-1.690E-06	131.44
129.0	6.879	10.814	17.694		8.40E-03	1.079E-03	2.708E-05	-1.571E-06	118.60
130.8	6.411	11.275	17.686		1.04E-02	1.125E-03	2.435E-05	-1.464E-06	106.63
132.6	5.995	11.696	17.691		1.25E-02	1.167E-03	2.180E-05	-1.369E-06	95.46
134.4	5.627	12.067	17.694		1.46E-02	1.204E-03	1.941E-05	-1.285E-06	85.00
136.2	5.312	12.398	17.709		1.68E-02	1.237E-03	1.717E-05	-1.213E-06	75.17
138.0	5.049	12.688	17.737		1.90E-02	1.266E-03	1.504E-05	-1.153E-06	65.90

(continued) Torques and Torque Angles and Derivatives and Bimoment

z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
				<b>8</b>					
138.0	4.996	12.688	17.685		1.90E-02	1.266E-03	1.505E-05	-1.141E-06	65.90
139.8	4.756	12.939	17.695		2.13E-02	1.291E-03	1.304E-05	-1.086E-06	57.11
141.6	4.541	13.159	17.700		2.37E-02	1.313E-03	1.113E-05	-1.037E-06	48.75
143.4	4.357	13.340	17.696		2.61E-02	1.331E-03	9.306E-06	-9.949E-07	40.75
145.2	4.204	13.500	17.704		2.85E-02	1.347E-03	7.548E-06	-9.600E-07	33.05
147.0	4.082	13.620	17.702		3.09E-02	1.359E-03	5.846E-06	-9.321E-07	25.60
148.8	3.990	13.711	17.701		3.34E-02	1.368E-03	4.188E-06	-9.112E-07	18.34
150.6	3.929	13.771	17.700		3.58E-02	1.374E-03	2.561E-06	-8.973E-07	11.22
150.6	3.929	13.771	17.700		3.58E-02	1.374E-03	2.561E-06	-8.973E-07	11.22
152.4	3.899	13.801	17.699		3.83E-02	1.377E-03	9.537E-07	-8.903E-07	4.17
154.2	3.899	13.801	17.699		4.08E-02	1.377E-03	-6.479E-07	-8.903E-07	-2.84
156.0	3.930	13.771	17.700		4.33E-02	1.374E-03	-2.256E-06	-8.974E-07	-9.87
				<b>9</b>					
156.0	3.937	13.771	17.708		4.33E-02	1.374E-03	-2.251E-06	-8.991E-07	-9.87
157.8	3.983	13.721	17.703		4.57E-02	1.369E-03	-3.877E-06	-9.095E-07	-16.98
159.6	4.062	13.630	17.692		4.82E-02	1.360E-03	-5.530E-06	-9.275E-07	-24.21
161.4	4.175	13.520	17.695		5.06E-02	1.349E-03	-7.221E-06	-9.533E-07	-31.62
163.2	4.321	13.370	17.691		5.30E-02	1.334E-03	-8.966E-06	-9.867E-07	-39.27
165.0	4.502	13.200	17.701		5.54E-02	1.317E-03	-1.078E-05	-1.028E-06	-47.20
166.8	4.716	12.989	17.705		5.78E-02	1.296E-03	-1.267E-05	-1.077E-06	-55.49
168.6	4.961	12.738	17.700		6.01E-02	1.271E-03	-1.466E-05	-1.133E-06	-64.19
170.4	5.246	12.458	17.704		6.24E-02	1.243E-03	-1.676E-05	-1.198E-06	-73.36
172.2	5.557	12.137	17.694		6.46E-02	1.211E-03	-1.897E-05	-1.269E-06	-83.08
174.0	5.907	11.766	17.674		6.67E-02	1.174E-03	-2.133E-05	-1.349E-06	-93.42
				<b>10</b>					
174.0	5.982	11.766	17.748		6.67E-02	1.174E-03	-2.132E-05	-1.366E-06	-93.42
175.8	6.350	11.365	17.715		6.88E-02	1.134E-03	-2.386E-05	-1.450E-06	-104.45
177.6	6.787	10.904	17.692		7.08E-02	1.088E-03	-2.655E-05	-1.550E-06	-116.25
179.4	7.282	10.403	17.686		7.27E-02	1.038E-03	-2.944E-05	-1.663E-06	-128.92
181.2	7.843	9.845	17.688		7.45E-02	9.823E-04	-3.255E-05	-1.791E-06	-142.55
183.0	8.469	9.228	17.697		7.62E-02	9.207E-04	-3.590E-05	-1.934E-06	-157.23
184.8	9.156	8.548	17.705		7.78E-02	8.529E-04	-3.952E-05	-2.091E-06	-173.08
186.6	9.910	7.800	17.710		7.93E-02	7.783E-04	-4.344E-05	-2.263E-06	-190.21
188.4	10.724	6.979	17.703		8.06E-02	6.963E-04	-4.768E-05	-2.449E-06	-208.75
190.2	11.604	6.078	17.682		8.18E-02	6.064E-04	-5.226E-05	-2.650E-06	-228.84
192.0	12.546	5.091	17.637		8.28E-02	5.080E-04	-5.721E-05	-2.865E-06	-250.64
				<b>11</b>					
192.0	-11.898	5.091	-6.806		8.28E-02	5.080E-04	-5.721E-05	2.717E-06	-250.64
194.1	-10.821	3.945	-6.876		8.38E-02	3.936E-04	-5.177E-05	2.471E-06	-226.67
196.2	-9.818	2.909	-6.909		8.45E-02	2.902E-04	-4.682E-05	2.242E-06	-205.00
198.3	-8.889	1.970	-6.919		8.50E-02	1.966E-04	-4.234E-05	2.030E-06	-185.40

(continued) Torques and Torque Angles and Derivatives and Bimoment									
z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
200.4	-8.035	1.123	-6.913		8.53E-02	1.120E-04	-3.828E-05	1.835E-06	-167.67
202.5	-7.256	0.356	-6.900		8.55E-02	3.556E-05	-3.462E-05	1.657E-06	-151.64
204.6	-6.551	-0.337	-6.888		8.55E-02	-3.361E-05	-3.131E-05	1.496E-06	-137.14
206.7	-5.920	-0.964	-6.884		8.53E-02	-9.617E-05	-2.832E-05	1.352E-06	-124.02
208.8	-5.364	-1.531	-6.896		8.51E-02	-1.528E-04	-2.562E-05	1.225E-06	-112.15
210.9	-4.883	-2.045	-6.927		8.47E-02	-2.040E-04	-2.317E-05	1.115E-06	-101.42
213.0	-4.475	-2.508	-6.983		8.42E-02	-2.502E-04	-2.092E-05	1.022E-06	-91.71
				<b>12</b>					
213.0	-4.359	-2.508	-6.866		8.42E-02	-2.502E-04	-2.094E-05	9.954E-07	-91.71
215.1	-3.965	-2.928	-6.893		8.36E-02	-2.921E-04	-1.894E-05	9.055E-07	-82.93
217.2	-3.599	-3.306	-6.905		8.30E-02	-3.299E-04	-1.713E-05	8.218E-07	-74.99
219.3	-3.259	-3.649	-6.908		8.23E-02	-3.641E-04	-1.549E-05	7.443E-07	-67.81
221.4	-2.947	-3.959	-6.906		8.15E-02	-3.950E-04	-1.400E-05	6.730E-07	-61.31
223.5	-2.662	-4.239	-6.902		8.06E-02	-4.230E-04	-1.265E-05	6.080E-07	-55.43
225.6	-2.405	-4.493	-6.898		7.97E-02	-4.483E-04	-1.144E-05	5.491E-07	-50.10
227.7	-2.174	-4.722	-6.896		7.87E-02	-4.711E-04	-1.034E-05	4.965E-07	-45.29
229.8	-1.971	-4.929	-6.900		7.77E-02	-4.918E-04	-9.350E-06	4.502E-07	-40.93
231.9	-1.795	-5.116	-6.912		7.67E-02	-5.105E-04	-8.448E-06	4.100E-07	-36.99
234.0	-1.647	-5.285	-6.932		7.56E-02	-5.273E-04	-7.624E-06	3.761E-07	-33.42
				<b>13</b>					
234.0	-1.604	-5.285	-6.889		7.56E-02	-5.273E-04	-7.629E-06	3.663E-07	-33.42
236.1	-1.461	-5.438	-6.899		7.44E-02	-5.426E-04	-6.894E-06	3.336E-07	-30.19
238.2	-1.327	-5.575	-6.903		7.33E-02	-5.563E-04	-6.226E-06	3.031E-07	-27.26
240.3	-1.204	-5.701	-6.905		7.21E-02	-5.688E-04	-5.619E-06	2.750E-07	-24.61
242.4	-1.091	-5.813	-6.904		7.09E-02	-5.800E-04	-5.069E-06	2.491E-07	-22.20
244.5	-0.988	-5.914	-6.902		6.97E-02	-5.901E-04	-4.571E-06	2.256E-07	-20.02
246.6	-0.895	-6.005	-6.900		6.84E-02	-5.992E-04	-4.120E-06	2.044E-07	-18.05
248.7	-0.812	-6.088	-6.900		6.72E-02	-6.074E-04	-3.711E-06	1.855E-07	-16.25
250.8	-0.740	-6.162	-6.901		6.59E-02	-6.148E-04	-3.340E-06	1.689E-07	-14.62
252.9	-0.677	-6.229	-6.906		6.46E-02	-6.215E-04	-3.001E-06	1.546E-07	-13.14
255.0	-0.624	-6.288	-6.912		6.33E-02	-6.274E-04	-2.689E-06	1.426E-07	-11.79
				<b>14</b>					
255.0	-0.610	-6.288	-6.898		6.33E-02	-6.274E-04	-2.691E-06	1.392E-07	-11.79
257.1	-0.559	-6.342	-6.901		6.19E-02	-6.328E-04	-2.411E-06	1.276E-07	-10.56
259.2	-0.512	-6.390	-6.903		6.06E-02	-6.376E-04	-2.154E-06	1.170E-07	-9.43
261.3	-0.470	-6.433	-6.903		5.93E-02	-6.419E-04	-1.919E-06	1.073E-07	-8.40
263.4	-0.432	-6.471	-6.903		5.79E-02	-6.457E-04	-1.703E-06	9.856E-08	-7.46
265.5	-0.397	-6.505	-6.902		5.65E-02	-6.490E-04	-1.504E-06	9.074E-08	-6.59
267.6	-0.367	-6.535	-6.902		5.52E-02	-6.520E-04	-1.321E-06	8.384E-08	-5.78
269.7	-0.341	-6.561	-6.902		5.38E-02	-6.546E-04	-1.151E-06	7.787E-08	-5.04
271.8	-0.319	-6.583	-6.902		5.24E-02	-6.568E-04	-9.931E-07	7.283E-08	-4.35

continued) Torques and Torque Angles and Derivatives and Bimoment

z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
273.9	-0.301	-6.603	-6.904		5.11E-02	-6.588E-04	-8.446E-07	6.872E-08	-3.70
276.0	-0.287	-6.619	-6.906		4.97E-02	-6.604E-04	-7.038E-07	6.554E-08	-3.09
				<b>15</b>					
276.0	-0.283	-6.619	-6.902		4.97E-02	-6.604E-04	-7.046E-07	6.464E-08	-3.09
278.1	-0.270	-6.632	-6.902		4.83E-02	-6.617E-04	-5.721E-07	6.166E-08	-2.51
280.2	-0.259	-6.643	-6.902		4.69E-02	-6.628E-04	-4.452E-07	5.925E-08	-1.95
282.3	-0.251	-6.651	-6.902		4.55E-02	-6.636E-04	-3.228E-07	5.741E-08	-1.41
284.4	-0.246	-6.656	-6.902		4.41E-02	-6.641E-04	-2.037E-07	5.614E-08	-0.89
286.5	-0.243	-6.659	-6.902		4.27E-02	-6.644E-04	-8.665E-08	5.543E-08	-0.38
288.6	-0.242	-6.660	-6.902		4.13E-02	-6.645E-04	2.950E-08	5.529E-08	0.13
290.7	-0.244	-6.658	-6.902		3.99E-02	-6.643E-04	1.460E-07	5.571E-08	0.64
292.8	-0.248	-6.654	-6.902		3.85E-02	-6.639E-04	2.639E-07	5.671E-08	1.16
294.9	-0.255	-6.647	-6.902		3.71E-02	-6.632E-04	3.845E-07	5.827E-08	1.68
297.0	-0.264	-6.638	-6.902		3.57E-02	-6.623E-04	5.090E-07	6.040E-08	2.23
				<b>16</b>					
297.0	-0.267	-6.638	-6.905		3.57E-02	-6.623E-04	5.083E-07	6.105E-08	2.23
299.1	-0.277	-6.626	-6.903		3.43E-02	-6.611E-04	6.388E-07	6.337E-08	2.80
301.2	-0.291	-6.611	-6.902		3.30E-02	-6.596E-04	7.751E-07	6.652E-08	3.39
303.3	-0.309	-6.593	-6.901		3.16E-02	-6.578E-04	9.188E-07	7.049E-08	4.02
305.4	-0.330	-6.572	-6.901		3.02E-02	-6.557E-04	1.072E-06	7.530E-08	4.69
307.5	-0.354	-6.548	-6.902		2.88E-02	-6.533E-04	1.236E-06	8.093E-08	5.41
309.6	-0.383	-6.520	-6.902		2.75E-02	-6.505E-04	1.412E-06	8.739E-08	6.18
311.7	-0.415	-6.489	-6.903		2.61E-02	-6.474E-04	1.603E-06	9.469E-08	7.02
313.8	-0.450	-6.452	-6.903		2.47E-02	-6.438E-04	1.810E-06	1.028E-07	7.93
315.9	-0.490	-6.411	-6.901		2.34E-02	-6.397E-04	2.036E-06	1.118E-07	8.91
318.0	-0.532	-6.366	-6.898		2.20E-02	-6.352E-04	2.280E-06	1.215E-07	9.99
				<b>17</b>					
318.0	-0.545	-6.366	-6.911		2.20E-02	-6.352E-04	2.279E-06	1.244E-07	9.99
320.1	-0.589	-6.315	-6.905		2.07E-02	-6.301E-04	2.551E-06	1.346E-07	11.17
322.2	-0.643	-6.259	-6.902		1.94E-02	-6.245E-04	2.846E-06	1.468E-07	12.46
324.3	-0.705	-6.196	-6.900		1.81E-02	-6.182E-04	3.168E-06	1.609E-07	13.87
326.4	-0.776	-6.126	-6.901		1.68E-02	-6.112E-04	3.523E-06	1.771E-07	15.43
328.5	-0.855	-6.048	-6.902		1.55E-02	-6.034E-04	3.913E-06	1.952E-07	17.14
330.6	-0.943	-5.960	-6.904		1.43E-02	-5.947E-04	4.344E-06	2.154E-07	19.03
332.7	-1.040	-5.864	-6.904		1.30E-02	-5.851E-04	4.820E-06	2.375E-07	21.10
334.8	-1.146	-5.757	-6.903		1.18E-02	-5.744E-04	5.343E-06	2.617E-07	23.39
336.9	-1.260	-5.639	-6.899		1.06E-02	-5.626E-04	5.920E-06	2.878E-07	25.92
339.0	-1.384	-5.507	-6.891		9.50E-03	-5.495E-04	6.554E-06	3.160E-07	28.71
				<b>18</b>					
339.0	-1.421	-5.507	-6.928		9.50E-03	-5.495E-04	6.550E-06	3.244E-07	28.71
341.1	-1.548	-5.362	-6.910		8.30E-03	-5.350E-04	7.261E-06	3.535E-07	31.79
343.2	-1.699	-5.202	-6.901		7.20E-03	-5.190E-04	8.038E-06	3.880E-07	35.19

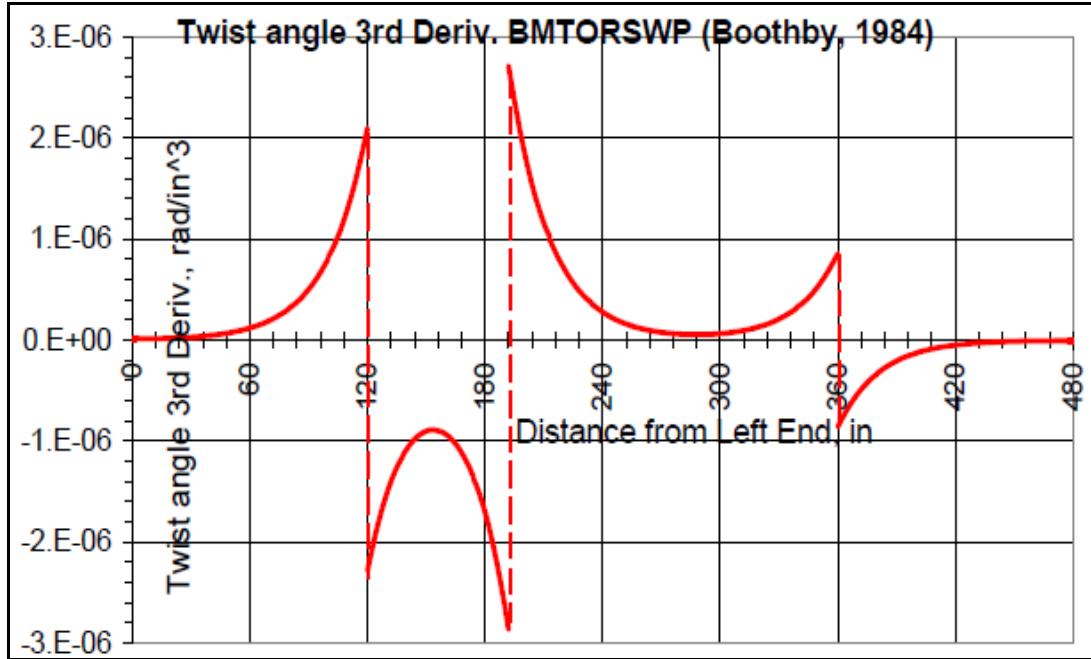
## (continued) Torques and Torque Angles and Derivatives and Bimoment

z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
345.3	-1.874	-5.023	-6.897		6.10E-03	-5.012E-04	8.894E-06	4.279E-07	38.94
347.4	-2.072	-4.827	-6.898		5.10E-03	-4.816E-04	9.839E-06	4.731E-07	43.09
349.5	-2.293	-4.608	-6.902		4.10E-03	-4.598E-04	1.089E-05	5.237E-07	47.68
351.6	-2.539	-4.368	-6.906		3.20E-03	-4.358E-04	1.204E-05	5.797E-07	52.74
353.7	-2.807	-4.100	-6.907		2.30E-03	-4.091E-04	1.332E-05	6.410E-07	58.34
355.8	-3.099	-3.806	-6.905		1.50E-03	-3.797E-04	1.474E-05	7.077E-07	64.53
357.9	-3.414	-3.479	-6.893		7.00E-04	-3.471E-04	1.630E-05	7.797E-07	71.37
360.0	-3.753	-3.118	-6.871		0.00E+00	-3.111E-04	1.802E-05	8.571E-07	78.93
				19					
360.0	3.749	-3.118	0.631		0.00E+00	-3.111E-04	1.802E-05	-8.562E-07	78.93
362.0	3.424	-2.774	0.650		-6.00E-04	-2.768E-04	1.638E-05	-7.819E-07	71.73
364.0	3.120	-2.460	0.660		-1.10E-03	-2.455E-04	1.489E-05	-7.126E-07	65.18
366.0	2.838	-2.176	0.663		-1.60E-03	-2.171E-04	1.353E-05	-6.482E-07	59.23
368.0	2.578	-1.917	0.661		-2.00E-03	-1.913E-04	1.229E-05	-5.888E-07	53.83
370.0	2.340	-1.683	0.657		-2.30E-03	-1.679E-04	1.117E-05	-5.344E-07	48.92
372.0	2.123	-1.469	0.654		-2.70E-03	-1.466E-04	1.015E-05	-4.849E-07	44.45
374.0	1.929	-1.275	0.654		-2.90E-03	-1.272E-04	9.226E-06	-4.404E-07	40.40
376.0	1.755	-1.098	0.657		-3.20E-03	-1.096E-04	8.385E-06	-4.008E-07	36.71
378.0	1.603	-0.938	0.665		-3.40E-03	-9.363E-05	7.619E-06	-3.661E-07	33.36
380.0	1.474	-0.793	0.681		-3.50E-03	-7.910E-05	6.918E-06	-3.365E-07	30.32
				20					
380.0	1.440	-0.793	0.647		-3.50E-03	-7.910E-05	6.921E-06	-3.289E-07	30.32
382.0	1.315	-0.660	0.655		-3.70E-03	-6.590E-05	6.292E-06	-3.004E-07	27.55
384.0	1.199	-0.540	0.658		-3.80E-03	-5.390E-05	5.718E-06	-2.737E-07	25.03
386.0	1.090	-0.431	0.660		-3.90E-03	-4.299E-05	5.195E-06	-2.490E-07	22.75
388.0	0.991	-0.332	0.659		-4.00E-03	-3.308E-05	4.721E-06	-2.262E-07	20.67
390.0	0.899	-0.241	0.658		-4.00E-03	-2.408E-05	4.289E-06	-2.053E-07	18.79
392.0	0.816	-0.159	0.656		-4.10E-03	-1.590E-05	3.898E-06	-1.863E-07	17.07
394.0	0.741	-0.085	0.656		-4.10E-03	-8.466E-06	3.543E-06	-1.692E-07	15.51
396.0	0.674	-0.017	0.657		-4.10E-03	-1.708E-06	3.220E-06	-1.540E-07	14.10
398.0	0.616	0.044	0.661		-4.10E-03	4.434E-06	2.926E-06	-1.407E-07	12.81
400.0	0.566	0.100	0.667		-4.10E-03	1.001E-05	2.656E-06	-1.293E-07	11.64
				21					
400.0	0.554	0.100	0.654		-4.10E-03	1.001E-05	2.657E-06	-1.264E-07	11.64
402.0	0.505	0.151	0.656		-4.10E-03	1.508E-05	2.416E-06	-1.154E-07	10.58
404.0	0.461	0.197	0.658		-4.00E-03	1.969E-05	2.195E-06	-1.052E-07	9.61
406.0	0.419	0.239	0.658		-4.00E-03	2.387E-05	1.994E-06	-9.572E-08	8.73
408.0	0.381	0.277	0.658		-3.90E-03	2.768E-05	1.812E-06	-8.697E-08	7.93
410.0	0.346	0.312	0.658		-3.90E-03	3.113E-05	1.646E-06	-7.894E-08	7.21
412.0	0.314	0.343	0.657		-3.80E-03	3.427E-05	1.495E-06	-7.165E-08	6.55
414.0	0.285	0.372	0.657		-3.70E-03	3.712E-05	1.359E-06	-6.509E-08	5.95
416.0	0.259	0.398	0.657		-3.70E-03	3.971E-05	1.235E-06	-5.926E-08	5.40

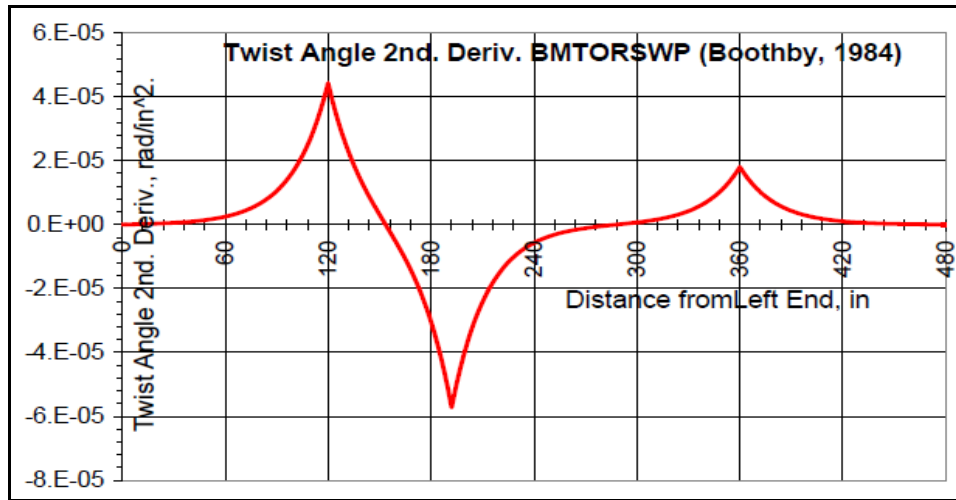


(continued) Torques and Torque Angles and Derivatives and Bimoment

z	$T_w$	$T_t$	$T$	Elm	$\theta$	$\theta'$	$\theta''$	$\theta'''$	B(z)
418.0	0.237	0.422	0.659		-3.60E-03	4.207E-05	1.121E-06	-5.416E-08	4.91
420.0	0.218	0.443	0.661		-3.50E-03	4.421E-05	1.017E-06	-4.980E-08	4.46
				<b>22</b>					
420.0	0.213	0.443	0.656		-3.50E-03	4.421E-05	1.018E-06	-4.868E-08	4.46
422.0	0.195	0.463	0.657		-3.40E-03	4.615E-05	9.248E-07	-4.449E-08	4.05
424.0	0.178	0.480	0.658		-3.30E-03	4.791E-05	8.398E-07	-4.058E-08	3.68
426.0	0.162	0.496	0.658		-3.20E-03	4.951E-05	7.623E-07	-3.695E-08	3.34
428.0	0.147	0.511	0.658		-3.10E-03	5.096E-05	6.918E-07	-3.360E-08	3.03
430.0	0.134	0.524	0.658		-3.00E-03	5.228E-05	6.277E-07	-3.054E-08	2.75
432.0	0.122	0.536	0.658		-2.90E-03	5.348E-05	5.694E-07	-2.776E-08	2.49
434.0	0.111	0.547	0.657		-2.80E-03	5.456E-05	5.165E-07	-2.526E-08	2.26
436.0	0.101	0.557	0.658		-2.70E-03	5.555E-05	4.682E-07	-2.305E-08	2.05
438.0	0.092	0.566	0.658		-2.60E-03	5.644E-05	4.241E-07	-2.112E-08	1.86
440.0	0.085	0.574	0.659		-2.50E-03	5.725E-05	3.835E-07	-1.947E-08	1.68
				<b>23</b>					
440.0	0.083	0.574	0.657		-2.50E-03	5.725E-05	3.837E-07	-1.905E-08	1.68
442.0	0.077	0.581	0.658		-2.30E-03	5.798E-05	3.472E-07	-1.747E-08	1.52
444.0	0.070	0.588	0.658		-2.20E-03	5.864E-05	3.137E-07	-1.601E-08	1.37
446.0	0.064	0.594	0.658		-2.10E-03	5.923E-05	2.831E-07	-1.465E-08	1.24
448.0	0.059	0.599	0.658		-2.00E-03	5.977E-05	2.550E-07	-1.342E-08	1.12
450.0	0.054	0.604	0.658		-1.90E-03	6.026E-05	2.294E-07	-1.229E-08	1.00
452.0	0.049	0.608	0.658		-1.70E-03	6.069E-05	2.058E-07	-1.128E-08	0.90
454.0	0.045	0.612	0.658		-1.60E-03	6.108E-05	1.842E-07	-1.039E-08	0.81
456.0	0.042	0.616	0.658		-1.50E-03	6.143E-05	1.642E-07	-9.603E-09	0.72
458.0	0.039	0.619	0.658		-1.40E-03	6.174E-05	1.457E-07	-8.935E-09	0.64
460.0	0.037	0.621	0.658		-1.30E-03	6.201E-05	1.284E-07	-8.380E-09	0.56
				<b>24</b>					
460.0	0.036	0.621	0.658		-1.30E-03	6.201E-05	1.284E-07	-8.240E-09	0.56
462.0	0.034	0.624	0.658		-1.10E-03	6.225E-05	1.125E-07	-7.714E-09	0.49
464.0	0.032	0.626	0.658		-1.00E-03	6.246E-05	9.755E-08	-7.246E-09	0.43
466.0	0.030	0.628	0.658		-9.00E-04	6.264E-05	8.348E-08	-6.835E-09	0.37
468.0	0.028	0.629	0.658		-8.00E-04	6.280E-05	7.018E-08	-6.481E-09	0.31
470.0	0.027	0.631	0.658		-6.00E-04	6.292E-05	5.752E-08	-6.185E-09	0.25
472.0	0.026	0.632	0.658		-5.00E-04	6.303E-05	4.540E-08	-5.946E-09	0.20
474.0	0.025	0.633	0.658		-4.00E-04	6.311E-05	3.370E-08	-5.764E-09	0.15
476.0	0.025	0.633	0.658		-3.00E-04	6.316E-05	2.230E-08	-5.640E-09	0.10
478.0	0.024	0.633	0.658		-1.00E-04	6.320E-05	1.110E-08	-5.573E-09	0.05
480.0	0.024	0.634	0.658		0.00E+00	6.321E-05	-2.570E-11	-5.563E-09	0.00



Third Derivative of Torque Angle



Second Derivative of Torque Angle

St. Vt. and Warping Shear Stresses

Pure Torsion Shear Stresses $G \cdot t \cdot \theta'$							Warping shear stress at "s", $- E \cdot S_w \cdot \theta''' / t$							
"t" at XS s= 1, 2, 3, 2', 1', 0'							axis	"s" and $S_w$						
0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
0.501	0.501	0.501	0.510	0.501	0.501	0.501	axial	0	6.01	4.91	-2.45	4.91	6.01	0
-0.90	-0.90	-0.90	-0.92	-0.90	-0.90	-0.90	0.0	0	0.00	0.00	0.00	0.00	0.00	0
-0.90	-0.90	-0.90	-0.92	-0.90	-0.90	-0.90	2.0	0	0.00	0.00	0.00	0.00	0.00	0
-0.90	-0.90	-0.90	-0.92	-0.90	-0.90	-0.90	4.0	0	0.00	0.00	0.00	0.00	0.00	0
-0.90	-0.90	-0.90	-0.92	-0.90	-0.90	-0.90	6.0	0	0.00	0.00	0.00	0.00	0.00	0
-0.90	-0.90	-0.90	-0.91	-0.90	-0.90	-0.90	8.0	0	-0.01	0.00	0.00	0.00	-0.01	0
-0.90	-0.90	-0.90	-0.91	-0.90	-0.90	-0.90	10.0	0	-0.01	0.00	0.00	0.00	-0.01	0
-0.89	-0.89	-0.89	-0.91	-0.89	-0.89	-0.89	12.0	0	-0.01	0.00	0.00	0.00	-0.01	0
-0.89	-0.89	-0.89	-0.91	-0.89	-0.89	-0.89	14.0	0	-0.01	0.00	0.00	0.00	-0.01	0
-0.89	-0.89	-0.89	-0.91	-0.89	-0.89	-0.89	16.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.89	-0.89	-0.89	-0.90	-0.89	-0.89	-0.89	18.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.88	-0.88	-0.88	-0.90	-0.88	-0.88	-0.88	20.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-0.88	-0.88	-0.88	-0.90	-0.88	-0.88	-0.88	20.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.88	-0.88	-0.88	-0.90	-0.88	-0.88	-0.88	22.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.88	-0.88	-0.88	-0.89	-0.88	-0.88	-0.88	24.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.87	-0.87	-0.87	-0.89	-0.87	-0.87	-0.87	26.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.87	-0.87	-0.87	-0.88	-0.87	-0.87	-0.87	28.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.86	-0.86	-0.86	-0.87	-0.86	-0.86	-0.86	30.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.85	-0.85	-0.85	-0.87	-0.85	-0.85	-0.85	32.0	0	-0.01	-0.01	0.00	-0.01	-0.01	0
-0.84	-0.84	-0.84	-0.86	-0.84	-0.84	-0.84	34.0	0	-0.01	-0.01	0.01	-0.01	-0.01	0
-0.84	-0.84	-0.84	-0.85	-0.84	-0.84	-0.84	36.0	0	-0.01	-0.01	0.01	-0.01	-0.01	0
-0.83	-0.83	-0.83	-0.84	-0.83	-0.83	-0.83	38.0	0	-0.01	-0.01	0.01	-0.01	-0.01	0
-0.82	-0.82	-0.82	-0.83	-0.82	-0.82	-0.82	40.0	0	-0.02	-0.01	0.01	-0.01	-0.02	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-0.82	-0.82	-0.82	-0.83	-0.82	-0.82	-0.82	40.0	0	-0.02	-0.01	0.01	-0.01	-0.02	0
-0.80	-0.80	-0.80	-0.82	-0.80	-0.80	-0.80	42.0	0	-0.02	-0.01	0.01	-0.01	-0.02	0
-0.79	-0.79	-0.79	-0.81	-0.79	-0.79	-0.79	44.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-0.78	-0.78	-0.78	-0.79	-0.78	-0.78	-0.78	46.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-0.76	-0.76	-0.76	-0.78	-0.76	-0.76	-0.76	48.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-0.75	-0.75	-0.75	-0.76	-0.75	-0.75	-0.75	50.0	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-0.73	-0.73	-0.73	-0.74	-0.73	-0.73	-0.73	52.0	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-0.71	-0.71	-0.71	-0.72	-0.71	-0.71	-0.71	54.0	0	-0.03	-0.03	0.01	-0.03	-0.03	0
-0.68	-0.68	-0.68	-0.70	-0.68	-0.68	-0.68	56.0	0	-0.03	-0.03	0.01	-0.03	-0.03	0
-0.66	-0.66	-0.66	-0.67	-0.66	-0.66	-0.66	58.0	0	-0.04	-0.03	0.02	-0.03	-0.04	0
-0.63	-0.63	-0.63	-0.64	-0.63	-0.63	-0.63	60.0	0	-0.04	-0.03	0.02	-0.03	-0.04	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-0.63	-0.63	-0.63	-0.64	-0.63	-0.63	-0.63	60.0	0	-0.04	-0.03	0.02	-0.03	-0.04	0
-0.60	-0.60	-0.60	-0.61	-0.60	-0.60	-0.60	62.0	0	-0.05	-0.04	0.02	-0.04	-0.05	0
-0.57	-0.57	-0.57	-0.58	-0.57	-0.57	-0.57	64.0	0	-0.05	-0.04	0.02	-0.04	-0.05	0

0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
-0.53	-0.53	-0.53	-0.54	-0.53	-0.53	-0.53	66.0	0	-0.06	-0.05	0.02	-0.05	-0.06	0
-0.49	-0.49	-0.49	-0.50	-0.49	-0.49	-0.49	68.0	0	-0.06	-0.05	0.02	-0.05	-0.06	0
-0.44	-0.44	-0.44	-0.45	-0.44	-0.44	-0.44	70.0	0	-0.07	-0.06	0.03	-0.06	-0.07	0
-0.39	-0.39	-0.39	-0.40	-0.39	-0.39	-0.39	72.0	0	-0.07	-0.06	0.03	-0.06	-0.07	0
-0.34	-0.34	-0.34	-0.35	-0.34	-0.34	-0.34	74.0	0	-0.08	-0.07	0.03	-0.07	-0.08	0
-0.28	-0.28	-0.28	-0.29	-0.28	-0.28	-0.28	76.0	0	-0.09	-0.07	0.04	-0.07	-0.09	0
-0.22	-0.22	-0.22	-0.22	-0.22	-0.22	-0.22	78.0	0	-0.10	-0.08	0.04	-0.08	-0.10	0
-0.14	-0.14	-0.14	-0.15	-0.14	-0.14	-0.14	80.0	0	-0.11	-0.09	0.04	-0.09	-0.11	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-0.14	-0.14	-0.14	-0.15	-0.14	-0.14	-0.14	80.0	0	-0.11	-0.09	0.04	-0.09	-0.11	0
-0.06	-0.06	-0.06	-0.06	-0.06	-0.06	-0.06	82.0	0	-0.12	-0.10	0.05	-0.10	-0.12	0
0.02	0.02	0.02	0.02	0.02	0.02	0.02	84.0	0	-0.13	-0.11	0.05	-0.11	-0.13	0
0.12	0.12	0.12	0.12	0.12	0.12	0.12	86.0	0	-0.14	-0.12	0.06	-0.12	-0.14	0
0.23	0.23	0.23	0.23	0.23	0.23	0.23	88.0	0	-0.16	-0.13	0.06	-0.13	-0.16	0
0.34	0.34	0.34	0.35	0.34	0.34	0.34	90.0	0	-0.18	-0.14	0.07	-0.14	-0.18	0
0.47	0.47	0.47	0.48	0.47	0.47	0.47	92.0	0	-0.19	-0.16	0.08	-0.16	-0.19	0
0.61	0.61	0.61	0.62	0.61	0.61	0.61	94.0	0	-0.21	-0.17	0.09	-0.17	-0.21	0
0.77	0.77	0.77	0.78	0.77	0.77	0.77	96.0	0	-0.23	-0.19	0.09	-0.19	-0.23	0
0.94	0.94	0.94	0.96	0.94	0.94	0.94	98.0	0	-0.26	-0.21	0.10	-0.21	-0.26	0
1.13	1.13	1.13	1.15	1.13	1.13	1.13	100.0	0	-0.28	-0.23	0.11	-0.23	-0.28	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
1.13	1.13	1.13	1.15	1.13	1.13	1.13	100.0	0	-0.29	-0.23	0.11	-0.23	-0.29	0
1.33	1.33	1.33	1.36	1.33	1.33	1.33	102.0	0	-0.31	-0.26	0.13	-0.26	-0.31	0
1.56	1.56	1.56	1.59	1.56	1.56	1.56	104.0	0	-0.34	-0.28	0.14	-0.28	-0.34	0
1.81	1.81	1.81	1.85	1.81	1.81	1.81	106.0	0	-0.38	-0.31	0.15	-0.31	-0.38	0
2.09	2.09	2.09	2.13	2.09	2.09	2.09	108.0	0	-0.41	-0.34	0.17	-0.34	-0.41	0
2.39	2.39	2.39	2.44	2.39	2.39	2.39	110.0	0	-0.46	-0.37	0.18	-0.37	-0.46	0
2.73	2.73	2.73	2.78	2.73	2.73	2.73	112.0	0	-0.50	-0.41	0.20	-0.41	-0.50	0
3.10	3.10	3.10	3.15	3.10	3.10	3.10	114.0	0	-0.55	-0.45	0.22	-0.45	-0.55	0
3.50	3.50	3.50	3.56	3.50	3.50	3.50	116.0	0	-0.61	-0.50	0.24	-0.50	-0.61	0
3.95	3.95	3.95	4.02	3.95	3.95	3.95	118.0	0	-0.67	-0.55	0.27	-0.55	-0.67	0
4.44	4.44	4.44	4.52	4.44	4.44	4.44	120.0	0	-0.73	-0.60	0.29	-0.60	-0.73	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
4.44	4.44	4.44	4.52	4.44	4.44	4.44	120.0	0	0.79	0.65	-0.32	0.65	0.79	0
4.88	4.88	4.88	4.96	4.88	4.88	4.88	121.8	0	0.74	0.60	-0.30	0.60	0.74	0
5.28	5.28	5.28	5.37	5.28	5.28	5.28	123.6	0	0.68	0.56	-0.27	0.56	0.68	0
5.64	5.64	5.64	5.75	5.64	5.64	5.64	125.4	0	0.63	0.52	-0.25	0.52	0.63	0
5.97	5.97	5.97	6.08	5.97	5.97	5.97	127.2	0	0.59	0.48	-0.24	0.48	0.59	0
6.27	6.27	6.27	6.38	6.27	6.27	6.27	129.0	0	0.55	0.45	-0.22	0.45	0.55	0
6.54	6.54	6.54	6.66	6.54	6.54	6.54	130.8	0	0.51	0.42	-0.20	0.42	0.51	0
6.78	6.78	6.78	6.90	6.78	6.78	6.78	132.6	0	0.48	0.39	-0.19	0.39	0.48	0
7.00	7.00	7.00	7.12	7.00	7.00	7.00	134.4	0	0.45	0.37	-0.18	0.37	0.45	0
7.19	7.19	7.19	7.32	7.19	7.19	7.19	136.2	0	0.42	0.34	-0.17	0.34	0.42	0
7.36	7.36	7.36	7.49	7.36	7.36	7.36	138.0	0	0.40	0.33	-0.16	0.33	0.40	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0

0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
7.36	7.36	7.36	7.49	7.36	7.36	7.36	138.0	0	0.40	0.32	-0.16	0.32	0.40	0
7.50	7.50	7.50	7.64	7.50	7.50	7.50	139.8	0	0.38	0.31	-0.15	0.31	0.38	0
7.63	7.63	7.63	7.77	7.63	7.63	7.63	141.6	0	0.36	0.29	-0.14	0.29	0.36	0
7.74	7.74	7.74	7.87	7.74	7.74	7.74	143.4	0	0.35	0.28	-0.14	0.28	0.35	0
7.83	7.83	7.83	7.97	7.83	7.83	7.83	145.2	0	0.33	0.27	-0.13	0.27	0.33	0
7.90	7.90	7.90	8.04	7.90	7.90	7.90	147.0	0	0.32	0.26	-0.13	0.26	0.32	0
7.95	7.95	7.95	8.09	7.95	7.95	7.95	148.8	0	0.32	0.26	-0.13	0.26	0.32	0
7.99	7.99	7.99	8.13	7.99	7.99	7.99	150.6	0	0.31	0.26	-0.13	0.26	0.31	0
8.00	8.00	8.00	8.15	8.00	8.00	8.00	152.4	0	0.31	0.25	-0.12	0.25	0.31	0
8.00	8.00	8.00	8.15	8.00	8.00	8.00	154.2	0	0.31	0.25	-0.12	0.25	0.31	0
7.99	7.99	7.99	8.13	7.99	7.99	7.99	156.0	0	0.31	0.26	-0.13	0.26	0.31	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
7.99	7.99	7.99	8.13	7.99	7.99	7.99	156.0	0	0.31	0.26	-0.13	0.26	0.31	0
7.96	7.96	7.96	8.10	7.96	7.96	7.96	157.8	0	0.32	0.26	-0.13	0.26	0.32	0
7.90	7.90	7.90	8.05	7.90	7.90	7.90	159.6	0	0.32	0.26	-0.13	0.26	0.32	0
7.84	7.84	7.84	7.98	7.84	7.84	7.84	161.4	0	0.33	0.27	-0.13	0.27	0.33	0
7.75	7.75	7.75	7.89	7.75	7.75	7.75	163.2	0	0.34	0.28	-0.14	0.28	0.34	0
7.65	7.65	7.65	7.79	7.65	7.65	7.65	165.0	0	0.36	0.29	-0.14	0.29	0.36	0
7.53	7.53	7.53	7.67	7.53	7.53	7.53	166.8	0	0.37	0.31	-0.15	0.31	0.37	0
7.39	7.39	7.39	7.52	7.39	7.39	7.39	168.6	0	0.39	0.32	-0.16	0.32	0.39	0
7.22	7.22	7.22	7.35	7.22	7.22	7.22	170.4	0	0.42	0.34	-0.17	0.34	0.42	0
7.04	7.04	7.04	7.16	7.04	7.04	7.04	172.2	0	0.44	0.36	-0.18	0.36	0.44	0
6.82	6.82	6.82	6.95	6.82	6.82	6.82	174.0	0	0.47	0.38	-0.19	0.38	0.47	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
6.82	6.82	6.82	6.95	6.82	6.82	6.82	174.0	0	0.48	0.39	-0.19	0.39	0.48	0
6.59	6.59	6.59	6.71	6.59	6.59	6.59	175.8	0	0.50	0.41	-0.20	0.41	0.50	0
6.32	6.32	6.32	6.44	6.32	6.32	6.32	177.6	0	0.54	0.44	-0.22	0.44	0.54	0
6.03	6.03	6.03	6.14	6.03	6.03	6.03	179.4	0	0.58	0.47	-0.23	0.47	0.58	0
5.71	5.71	5.71	5.81	5.71	5.71	5.71	181.2	0	0.62	0.51	-0.25	0.51	0.62	0
5.35	5.35	5.35	5.45	5.35	5.35	5.35	183.0	0	0.67	0.55	-0.27	0.55	0.67	0
4.96	4.96	4.96	5.05	4.96	4.96	4.96	184.8	0	0.73	0.59	-0.29	0.59	0.73	0
4.52	4.52	4.52	4.60	4.52	4.52	4.52	186.6	0	0.79	0.64	-0.32	0.64	0.79	0
4.05	4.05	4.05	4.12	4.05	4.05	4.05	188.4	0	0.85	0.70	-0.34	0.70	0.85	0
3.52	3.52	3.52	3.59	3.52	3.52	3.52	190.2	0	0.92	0.75	-0.37	0.75	0.92	0
2.95	2.95	2.95	3.01	2.95	2.95	2.95	192.0	0	1.00	0.81	-0.40	0.81	1.00	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
2.95	2.95	2.95	3.01	2.95	2.95	2.95	192.0	0	-0.95	-0.77	0.38	-0.77	-0.95	0
2.29	2.29	2.29	2.33	2.29	2.29	2.29	194.1	0	-0.86	-0.70	0.34	-0.70	-0.86	0
1.69	1.69	1.69	1.72	1.69	1.69	1.69	196.2	0	-0.78	-0.64	0.31	-0.64	-0.78	0
1.14	1.14	1.14	1.16	1.14	1.14	1.14	198.3	0	-0.71	-0.58	0.28	-0.58	-0.71	0
0.65	0.65	0.65	0.66	0.65	0.65	0.65	200.4	0	-0.64	-0.52	0.26	-0.52	-0.64	0
0.21	0.21	0.21	0.21	0.21	0.21	0.21	202.5	0	-0.58	-0.47	0.23	-0.47	-0.58	0
-0.20	-0.20	-0.20	-0.20	-0.20	-0.20	-0.20	204.6	0	-0.52	-0.43	0.21	-0.43	-0.52	0
-0.56	-0.56	-0.56	-0.57	-0.56	-0.56	-0.56	206.7	0	-0.47	-0.38	0.19	-0.38	-0.47	0
-0.89	-0.89	-0.89	-0.90	-0.89	-0.89	-0.89	208.8	0	-0.43	-0.35	0.17	-0.35	-0.43	0

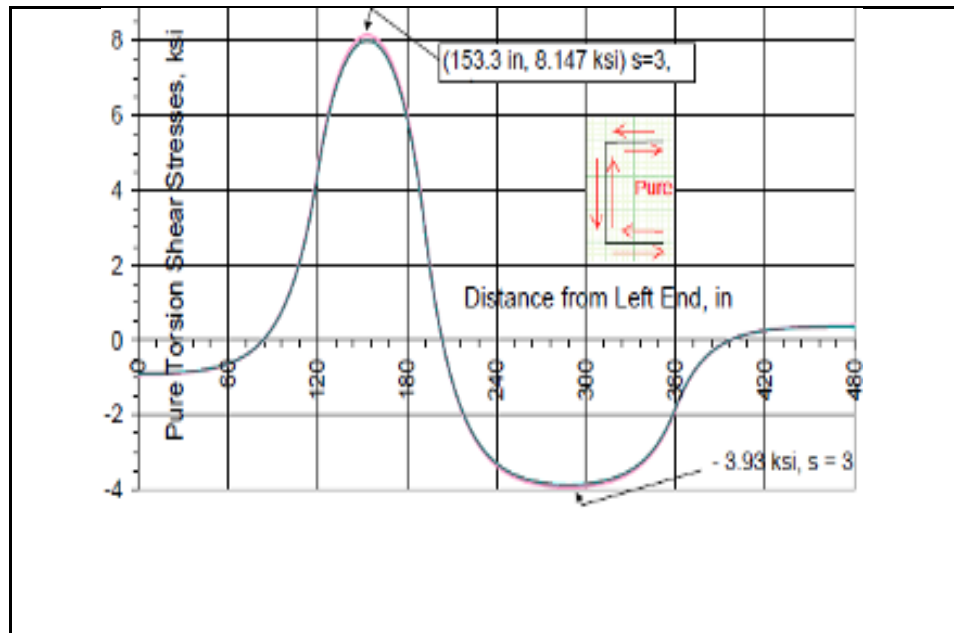
0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
-1.19	-1.19	-1.19	-1.21	-1.19	-1.19	-1.19	210.9	0	-0.39	-0.32	0.16	-0.32	-0.39	0
-1.45	-1.45	-1.45	-1.48	-1.45	-1.45	-1.45	213.0	0	-0.36	-0.29	0.14	-0.29	-0.36	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-1.45	-1.45	-1.45	-1.48	-1.45	-1.45	-1.45	213.0	0	-0.35	-0.28	0.14	-0.28	-0.35	0
-1.70	-1.70	-1.70	-1.73	-1.70	-1.70	-1.70	215.1	0	-0.32	-0.26	0.13	-0.26	-0.32	0
-1.92	-1.92	-1.92	-1.95	-1.92	-1.92	-1.92	217.2	0	-0.29	-0.23	0.11	-0.23	-0.29	0
-2.12	-2.12	-2.12	-2.15	-2.12	-2.12	-2.12	219.3	0	-0.26	-0.21	0.10	-0.21	-0.26	0
-2.30	-2.30	-2.30	-2.34	-2.30	-2.30	-2.30	221.4	0	-0.23	-0.19	0.09	-0.19	-0.23	0
-2.46	-2.46	-2.46	-2.50	-2.46	-2.46	-2.46	223.5	0	-0.21	-0.17	0.08	-0.17	-0.21	0
-2.61	-2.61	-2.61	-2.65	-2.61	-2.61	-2.61	225.6	0	-0.19	-0.16	0.08	-0.16	-0.19	0
-2.74	-2.74	-2.74	-2.79	-2.74	-2.74	-2.74	227.7	0	-0.17	-0.14	0.07	-0.14	-0.17	0
-2.86	-2.86	-2.86	-2.91	-2.86	-2.86	-2.86	229.8	0	-0.16	-0.13	0.06	-0.13	-0.16	0
-2.97	-2.97	-2.97	-3.02	-2.97	-2.97	-2.97	231.9	0	-0.14	-0.12	0.06	-0.12	-0.14	0
-3.06	-3.06	-3.06	-3.12	-3.06	-3.06	-3.06	234.0	0	-0.13	-0.11	0.05	-0.11	-0.13	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.06	-3.06	-3.06	-3.12	-3.06	-3.06	-3.06	234.0	0	-0.13	-0.10	0.05	-0.10	-0.13	0
-3.15	-3.15	-3.15	-3.21	-3.15	-3.15	-3.15	236.1	0	-0.12	-0.09	0.05	-0.09	-0.12	0
-3.23	-3.23	-3.23	-3.29	-3.23	-3.23	-3.23	238.2	0	-0.11	-0.09	0.04	-0.09	-0.11	0
-3.31	-3.31	-3.31	-3.37	-3.31	-3.31	-3.31	240.3	0	-0.10	-0.08	0.04	-0.08	-0.10	0
-3.37	-3.37	-3.37	-3.43	-3.37	-3.37	-3.37	242.4	0	-0.09	-0.07	0.03	-0.07	-0.09	0
-3.43	-3.43	-3.43	-3.49	-3.43	-3.43	-3.43	244.5	0	-0.08	-0.06	0.03	-0.06	-0.08	0
-3.48	-3.48	-3.48	-3.54	-3.48	-3.48	-3.48	246.6	0	-0.07	-0.06	0.03	-0.06	-0.07	0
-3.53	-3.53	-3.53	-3.59	-3.53	-3.53	-3.53	248.7	0	-0.06	-0.05	0.03	-0.05	-0.06	0
-3.57	-3.57	-3.57	-3.64	-3.57	-3.57	-3.57	250.8	0	-0.06	-0.05	0.02	-0.05	-0.06	0
-3.61	-3.61	-3.61	-3.68	-3.61	-3.61	-3.61	252.9	0	-0.05	-0.04	0.02	-0.04	-0.05	0
-3.65	-3.65	-3.65	-3.71	-3.65	-3.65	-3.65	255.0	0	-0.05	-0.04	0.02	-0.04	-0.05	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.65	-3.65	-3.65	-3.71	-3.65	-3.65	-3.65	255.0	0	-0.05	-0.04	0.02	-0.04	-0.05	0
-3.68	-3.68	-3.68	-3.74	-3.68	-3.68	-3.68	257.1	0	-0.04	-0.04	0.02	-0.04	-0.04	0
-3.71	-3.71	-3.71	-3.77	-3.71	-3.71	-3.71	259.2	0	-0.04	-0.03	0.02	-0.03	-0.04	0
-3.73	-3.73	-3.73	-3.80	-3.73	-3.73	-3.73	261.3	0	-0.04	-0.03	0.01	-0.03	-0.04	0
-3.75	-3.75	-3.75	-3.82	-3.75	-3.75	-3.75	263.4	0	-0.03	-0.03	0.01	-0.03	-0.03	0
-3.77	-3.77	-3.77	-3.84	-3.77	-3.77	-3.77	265.5	0	-0.03	-0.03	0.01	-0.03	-0.03	0
-3.79	-3.79	-3.79	-3.86	-3.79	-3.79	-3.79	267.6	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.80	-3.80	-3.80	-3.87	-3.80	-3.80	-3.80	269.7	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.82	-3.82	-3.82	-3.89	-3.82	-3.82	-3.82	271.8	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.83	-3.83	-3.83	-3.90	-3.83	-3.83	-3.83	273.9	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.84	-3.84	-3.84	-3.91	-3.84	-3.84	-3.84	276.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.84	-3.84	-3.84	-3.91	-3.84	-3.84	-3.84	276.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.85	-3.85	-3.85	-3.91	-3.85	-3.85	-3.85	278.1	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.85	-3.85	-3.85	-3.92	-3.85	-3.85	-3.85	280.2	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	282.3	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	284.4	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	286.5	0	-0.02	-0.02	0.01	-0.02	-0.02	0

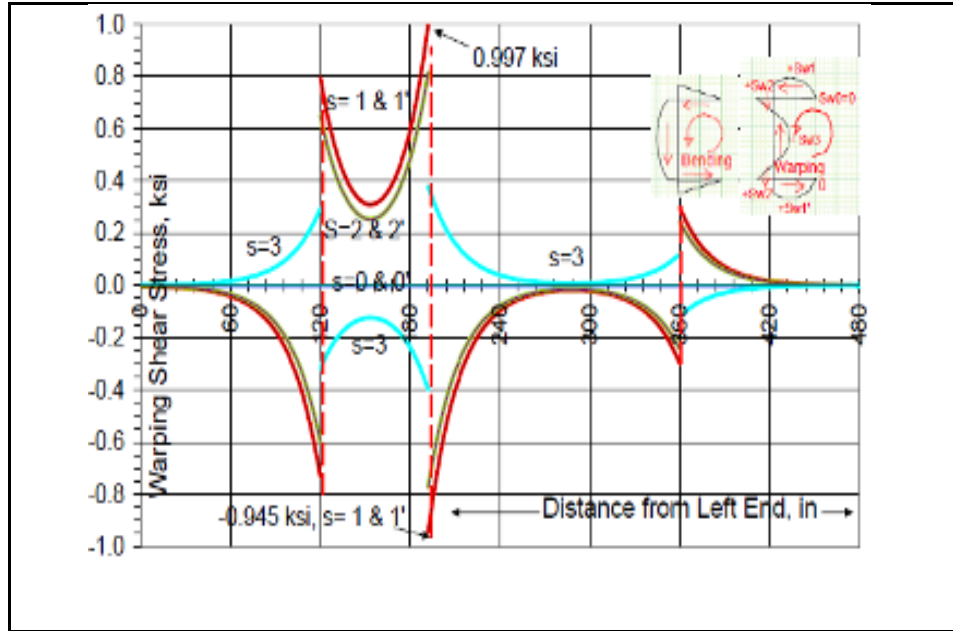
0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	288.6	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	290.7	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.86	-3.86	-3.86	-3.93	-3.86	-3.86	-3.86	292.8	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.85	-3.85	-3.85	-3.92	-3.85	-3.85	-3.85	294.9	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.85	-3.85	-3.85	-3.92	-3.85	-3.85	-3.85	297.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.85	-3.85	-3.85	-3.92	-3.85	-3.85	-3.85	297.0	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.84	-3.84	-3.84	-3.91	-3.84	-3.84	-3.84	299.1	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.83	-3.83	-3.83	-3.90	-3.83	-3.83	-3.83	301.2	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.82	-3.82	-3.82	-3.89	-3.82	-3.82	-3.82	303.3	0	-0.02	-0.02	0.01	-0.02	-0.02	0
-3.81	-3.81	-3.81	-3.88	-3.81	-3.81	-3.81	305.4	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.80	-3.80	-3.80	-3.86	-3.80	-3.80	-3.80	307.5	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.78	-3.78	-3.78	-3.85	-3.78	-3.78	-3.78	309.6	0	-0.03	-0.02	0.01	-0.02	-0.03	0
-3.76	-3.76	-3.76	-3.83	-3.76	-3.76	-3.76	311.7	0	-0.03	-0.03	0.01	-0.03	-0.03	0
-3.74	-3.74	-3.74	-3.81	-3.74	-3.74	-3.74	313.8	0	-0.04	-0.03	0.01	-0.03	-0.04	0
-3.72	-3.72	-3.72	-3.78	-3.72	-3.72	-3.72	315.9	0	-0.04	-0.03	0.02	-0.03	-0.04	0
-3.69	-3.69	-3.69	-3.76	-3.69	-3.69	-3.69	318.0	0	-0.04	-0.03	0.02	-0.03	-0.04	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.69	-3.69	-3.69	-3.76	-3.69	-3.69	-3.69	318.0	0	-0.04	-0.04	0.02	-0.04	-0.04	0
-3.66	-3.66	-3.66	-3.73	-3.66	-3.66	-3.66	320.1	0	-0.05	-0.04	0.02	-0.04	-0.05	0
-3.63	-3.63	-3.63	-3.69	-3.63	-3.63	-3.63	322.2	0	-0.05	-0.04	0.02	-0.04	-0.05	0
-3.59	-3.59	-3.59	-3.66	-3.59	-3.59	-3.59	324.3	0	-0.06	-0.05	0.02	-0.05	-0.06	0
-3.55	-3.55	-3.55	-3.62	-3.55	-3.55	-3.55	326.4	0	-0.06	-0.05	0.02	-0.05	-0.06	0
-3.51	-3.51	-3.51	-3.57	-3.51	-3.51	-3.51	328.5	0	-0.07	-0.06	0.03	-0.06	-0.07	0
-3.46	-3.46	-3.46	-3.52	-3.46	-3.46	-3.46	330.6	0	-0.07	-0.06	0.03	-0.06	-0.07	0
-3.40	-3.40	-3.40	-3.46	-3.40	-3.40	-3.40	332.7	0	-0.08	-0.07	0.03	-0.07	-0.08	0
-3.34	-3.34	-3.34	-3.40	-3.34	-3.34	-3.34	334.8	0	-0.09	-0.07	0.04	-0.07	-0.09	0
-3.27	-3.27	-3.27	-3.33	-3.27	-3.27	-3.27	336.9	0	-0.10	-0.08	0.04	-0.08	-0.10	0
-3.19	-3.19	-3.19	-3.25	-3.19	-3.19	-3.19	339.0	0	-0.11	-0.09	0.04	-0.09	-0.11	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-3.19	-3.19	-3.19	-3.25	-3.19	-3.19	-3.19	339.0	0	-0.11	-0.09	0.05	-0.09	-0.11	0
-3.11	-3.11	-3.11	-3.17	-3.11	-3.11	-3.11	341.1	0	-0.12	-0.10	0.05	-0.10	-0.12	0
-3.02	-3.02	-3.02	-3.07	-3.02	-3.02	-3.02	343.2	0	-0.13	-0.11	0.05	-0.11	-0.13	0
-2.91	-2.91	-2.91	-2.97	-2.91	-2.91	-2.91	345.3	0	-0.15	-0.12	0.06	-0.12	-0.15	0
-2.80	-2.80	-2.80	-2.85	-2.80	-2.80	-2.80	347.4	0	-0.16	-0.13	0.07	-0.13	-0.16	0
-2.67	-2.67	-2.67	-2.72	-2.67	-2.67	-2.67	349.5	0	-0.18	-0.15	0.07	-0.15	-0.18	0
-2.53	-2.53	-2.53	-2.58	-2.53	-2.53	-2.53	351.6	0	-0.20	-0.16	0.08	-0.16	-0.20	0
-2.38	-2.38	-2.38	-2.42	-2.38	-2.38	-2.38	353.7	0	-0.22	-0.18	0.09	-0.18	-0.22	0
-2.21	-2.21	-2.21	-2.25	-2.21	-2.21	-2.21	355.8	0	-0.25	-0.20	0.10	-0.20	-0.25	0
-2.02	-2.02	-2.02	-2.05	-2.02	-2.02	-2.02	357.9	0	-0.27	-0.22	0.11	-0.22	-0.27	0
-1.81	-1.81	-1.81	-1.84	-1.81	-1.81	-1.81	360.0	0	-0.30	-0.24	0.12	-0.24	-0.30	0
				0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-1.81	-1.81	-1.81	-1.84	-1.81	-1.81	-1.81	360.0	0	0.30	0.24	-0.12	0.24	0.30	0
-1.61	-1.61	-1.61	-1.64	-1.61	-1.61	-1.61	362.0	0	0.27	0.22	-0.11	0.22	0.27	0
-1.43	-1.43	-1.43	-1.45	-1.43	-1.43	-1.43	364.0	0	0.25	0.20	-0.10	0.20	0.25	0

0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
-1.26	-1.26	-1.26	-1.28	-1.26	-1.26	-1.26	366.0	0	0.23	0.18	-0.09	0.18	0.23	0
-1.11	-1.11	-1.11	-1.13	-1.11	-1.11	-1.11	368.0	0	0.20	0.17	-0.08	0.17	0.20	0
-0.98	-0.98	-0.98	-0.99	-0.98	-0.98	-0.98	370.0	0	0.19	0.15	-0.07	0.15	0.19	0
-0.85	-0.85	-0.85	-0.87	-0.85	-0.85	-0.85	372.0	0	0.17	0.14	-0.07	0.14	0.17	0
-0.74	-0.74	-0.74	-0.75	-0.74	-0.74	-0.74	374.0	0	0.15	0.13	-0.06	0.13	0.15	0
-0.64	-0.64	-0.64	-0.65	-0.64	-0.64	-0.64	376.0	0	0.14	0.11	-0.06	0.11	0.14	0
-0.54	-0.54	-0.54	-0.55	-0.54	-0.54	-0.54	378.0	0	0.13	0.10	-0.05	0.10	0.13	0
-0.46	-0.46	-0.46	-0.47	-0.46	-0.46	-0.46	380.0	0	0.12	0.10	-0.05	0.10	0.12	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
-0.46	-0.46	-0.46	-0.47	-0.46	-0.46	-0.46	380.0	0	0.11	0.09	-0.05	0.09	0.11	0
-0.38	-0.38	-0.38	-0.39	-0.38	-0.38	-0.38	382.0	0	0.10	0.09	-0.04	0.09	0.10	0
-0.31	-0.31	-0.31	-0.32	-0.31	-0.31	-0.31	384.0	0	0.10	0.08	-0.04	0.08	0.10	0
-0.25	-0.25	-0.25	-0.25	-0.25	-0.25	-0.25	386.0	0	0.09	0.07	-0.03	0.07	0.09	0
-0.19	-0.19	-0.19	-0.20	-0.19	-0.19	-0.19	388.0	0	0.08	0.06	-0.03	0.06	0.08	0
-0.14	-0.14	-0.14	-0.14	-0.14	-0.14	-0.14	390.0	0	0.07	0.06	-0.03	0.06	0.07	0
-0.09	-0.09	-0.09	-0.09	-0.09	-0.09	-0.09	392.0	0	0.06	0.05	-0.03	0.05	0.06	0
-0.05	-0.05	-0.05	-0.05	-0.05	-0.05	-0.05	394.0	0	0.06	0.05	-0.02	0.05	0.06	0
-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	396.0	0	0.05	0.04	-0.02	0.04	0.05	0
0.03	0.03	0.03	0.03	0.03	0.03	0.03	398.0	0	0.05	0.04	-0.02	0.04	0.05	0
0.06	0.06	0.06	0.06	0.06	0.06	0.06	400.0	0	0.04	0.04	-0.02	0.04	0.04	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
0.06	0.06	0.06	0.06	0.06	0.06	0.06	400.0	0	0.04	0.04	-0.02	0.04	0.04	0
0.09	0.09	0.09	0.09	0.09	0.09	0.09	402.0	0	0.04	0.03	-0.02	0.03	0.04	0
0.11	0.11	0.11	0.12	0.11	0.11	0.11	404.0	0	0.04	0.03	-0.01	0.03	0.04	0
0.14	0.14	0.14	0.14	0.14	0.14	0.14	406.0	0	0.03	0.03	-0.01	0.03	0.03	0
0.16	0.16	0.16	0.16	0.16	0.16	0.16	408.0	0	0.03	0.02	-0.01	0.02	0.03	0
0.18	0.18	0.18	0.18	0.18	0.18	0.18	410.0	0	0.03	0.02	-0.01	0.02	0.03	0
0.20	0.20	0.20	0.20	0.20	0.20	0.20	412.0	0	0.02	0.02	-0.01	0.02	0.02	0
0.22	0.22	0.22	0.22	0.22	0.22	0.22	414.0	0	0.02	0.02	-0.01	0.02	0.02	0
0.23	0.23	0.23	0.23	0.23	0.23	0.23	416.0	0	0.02	0.02	-0.01	0.02	0.02	0
0.24	0.24	0.24	0.25	0.24	0.24	0.24	418.0	0	0.02	0.02	-0.01	0.02	0.02	0
0.26	0.26	0.26	0.26	0.26	0.26	0.26	420.0	0	0.02	0.01	-0.01	0.01	0.02	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
0.26	0.26	0.26	0.26	0.26	0.26	0.26	420.0	0	0.02	0.01	-0.01	0.01	0.02	0
0.27	0.27	0.27	0.27	0.27	0.27	0.27	422.0	0	0.02	0.01	-0.01	0.01	0.02	0
0.28	0.28	0.28	0.28	0.28	0.28	0.28	424.0	0	0.01	0.01	-0.01	0.01	0.01	0
0.29	0.29	0.29	0.29	0.29	0.29	0.29	426.0	0	0.01	0.01	-0.01	0.01	0.01	0
0.30	0.30	0.30	0.30	0.30	0.30	0.30	428.0	0	0.01	0.01	0.00	0.01	0.01	0
0.30	0.30	0.30	0.31	0.30	0.30	0.30	430.0	0	0.01	0.01	0.00	0.01	0.01	0
0.31	0.31	0.31	0.32	0.31	0.31	0.31	432.0	0	0.01	0.01	0.00	0.01	0.01	0
0.32	0.32	0.32	0.32	0.32	0.32	0.32	434.0	0	0.01	0.01	0.00	0.01	0.01	0
0.32	0.32	0.32	0.33	0.32	0.32	0.32	436.0	0	0.01	0.01	0.00	0.01	0.01	0
0.33	0.33	0.33	0.33	0.33	0.33	0.33	438.0	0	0.01	0.01	0.00	0.01	0.01	0
0.33	0.33	0.33	0.34	0.33	0.33	0.33	440.0	0	0.01	0.01	0.00	0.01	0.01	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0



0	1	2	3	2'	1'	0'	z	0	1	2	3	2'	1'	0'
0.33	0.33	0.33	0.34	0.33	0.33	0.33	440.0	0	0.01	0.01	0.00	0.01	0.01	0
0.34	0.34	0.34	0.34	0.34	0.34	0.34	442.0	0	0.01	0.00	0.00	0.00	0.01	0
0.34	0.34	0.34	0.35	0.34	0.34	0.34	444.0	0	0.01	0.00	0.00	0.00	0.01	0
0.34	0.34	0.34	0.35	0.34	0.34	0.34	446.0	0	0.01	0.00	0.00	0.00	0.01	0
0.35	0.35	0.35	0.35	0.35	0.35	0.35	448.0	0	0.00	0.00	0.00	0.00	0.00	0
0.35	0.35	0.35	0.36	0.35	0.35	0.35	450.0	0	0.00	0.00	0.00	0.00	0.00	0
0.35	0.35	0.35	0.36	0.35	0.35	0.35	452.0	0	0.00	0.00	0.00	0.00	0.00	0
0.35	0.35	0.35	0.36	0.35	0.35	0.35	454.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.36	0.36	0.36	0.36	456.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	458.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	460.0	0	0.00	0.00	0.00	0.00	0.00	0
0.00	0.00	0.00	0.00	0.00	0.00	0.00		0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	460.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	462.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	464.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	466.0	0	0.00	0.00	0.00	0.00	0.00	0
0.36	0.36	0.36	0.37	0.36	0.36	0.36	468.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	470.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	472.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	474.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	476.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	478.0	0	0.00	0.00	0.00	0.00	0.00	0
0.37	0.37	0.37	0.37	0.37	0.37	0.37	480.0	0	0.00	0.00	0.00	0.00	0.00	0





Pure and Warping Torsion Shear Stresses

Values for Bending Shear Stress

s	0	1	2	3	2'	1'	0'
t	0.501	0.501	0.501	0.510	0.501	0.501	0.501
Q	0.000	5.479	7.830	17.000	7.830	5.479	0.000
	$S_x = 27 \text{ in}^3$						$I_x = 162.0$

Table of Bending Stresses

z	Shear Stresses, $V*Q/ I*t$							Normal Stresses, $M/ S_x$						
	0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
0.00	0.00	0.13	0.19	0.41	0.19	0.13	0	0.00	0.00	0.00	0	0.00	0.00	0.00
2.00		0.13	0.19	0.41	0.19	0.13	0	0.15	0.15	0.15	0	-0.15	-0.15	-0.15
4.00		0.13	0.19	0.41	0.19	0.13	0	0.30	0.30	0.30	0	-0.30	-0.30	-0.30
6.00		0.13	0.19	0.41	0.19	0.13	0	0.44	0.44	0.44	0	-0.44	-0.44	-0.44
8.00		0.13	0.19	0.41	0.19	0.13	0	0.59	0.59	0.59	0	-0.59	-0.59	-0.59
10.00		0.13	0.19	0.41	0.19	0.13	0	0.74	0.74	0.74	0	-0.74	-0.74	-0.74
12.00		0.13	0.19	0.41	0.19	0.13	0	0.89	0.89	0.89	0	-0.89	-0.89	-0.89
14.00		0.13	0.19	0.41	0.19	0.13	0	1.03	1.03	1.03	0	-1.03	-1.03	-1.03
16.00		0.13	0.19	0.41	0.19	0.13	0	1.18	1.18	1.18	0	-1.18	-1.18	-1.18
18.00		0.13	0.19	0.41	0.19	0.13	0	1.33	1.33	1.33	0	-1.33	-1.33	-1.33
20.00		0.13	0.19	0.41	0.19	0.13	0	1.48	1.48	1.48	0	-1.48	-1.48	-1.48
20.00	0	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00
20.00		0.13	0.19	0.41	0.19	0.13	0	1.48	1.48	1.48	0	-1.48	-1.48	-1.48
22.00		0.13	0.19	0.41	0.19	0.13	0	1.63	1.63	1.63	0	-1.63	-1.63	-1.63
24.00		0.13	0.19	0.41	0.19	0.13	0	1.77	1.77	1.77	0	-1.77	-1.77	-1.77

26.00	0.13	0.19	0.41	0.19	0.13	0	1.92	1.92	1.92	0	-1.92	-1.92	-1.92
28.00	0.13	0.19	0.41	0.19	0.13	0	2.07	2.07	2.07	0	-2.07	-2.07	-2.07
30.00	0.13	0.19	0.41	0.19	0.13	0	2.22	2.22	2.22	0	-2.22	-2.22	-2.22
32.00	0.13	0.19	0.41	0.19	0.13	0	2.36	2.36	2.36	0	-2.36	-2.36	-2.36
34.00	0.13	0.19	0.41	0.19	0.13	0	2.51	2.51	2.51	0	-2.51	-2.51	-2.51
36.00	0.13	0.19	0.41	0.19	0.13	0	2.66	2.66	2.66	0	-2.66	-2.66	-2.66
38.00	0.13	0.19	0.41	0.19	0.13	0	2.81	2.81	2.81	0	-2.81	-2.81	-2.81
40.00	0.13	0.19	0.41	0.19	0.13	0	2.96	2.96	2.96	0	-2.96	-2.96	-2.96
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00
40.00	0.13	0.19	0.41	0.19	0.13	0	2.96	2.96	2.96	0	-2.96	-2.96	-2.96
42.00	0.13	0.19	0.41	0.19	0.13	0	3.10	3.10	3.10	0	-3.10	-3.10	-3.10
44.00	0.13	0.19	0.41	0.19	0.13	0	3.25	3.25	3.25	0	-3.25	-3.25	-3.25
46.00	0.13	0.19	0.41	0.19	0.13	0	3.40	3.40	3.40	0	-3.40	-3.40	-3.40
48.00	0.13	0.19	0.41	0.19	0.13	0	3.55	3.55	3.55	0	-3.55	-3.55	-3.55
50.00	0.13	0.19	0.41	0.19	0.13	0	3.69	3.69	3.69	0	-3.69	-3.69	-3.69
52.00	0.13	0.19	0.41	0.19	0.13	0	3.84	3.84	3.84	0	-3.84	-3.84	-3.84
54.00	0.13	0.19	0.41	0.19	0.13	0	3.99	3.99	3.99	0	-3.99	-3.99	-3.99
56.00	0.13	0.19	0.41	0.19	0.13	0	4.14	4.14	4.14	0	-4.14	-4.14	-4.14
58.00	0.13	0.19	0.41	0.19	0.13	0	4.29	4.29	4.29	0	-4.29	-4.29	-4.29
60.00	0.13	0.19	0.41	0.19	0.13	0	4.43	4.43	4.43	0	-4.43	-4.43	-4.43
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00
	Shear Stresses, $V*Q/I*t$						Normal Stresses, $M/Sx$						
“s”=0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
z													
60.00	0.13	0.19	0.41	0.19	0.13	0	4.43	4.43	4.43	0	-4.43	-4.43	-4.43
62.00	0.13	0.19	0.41	0.19	0.13	0	4.58	4.58	4.58	0	-4.58	-4.58	-4.58
64.00	0.13	0.19	0.41	0.19	0.13	0	4.73	4.73	4.73	0	-4.73	-4.73	-4.73
66.00	0.13	0.19	0.41	0.19	0.13	0	4.88	4.88	4.88	0	-4.88	-4.88	-4.88
68.00	0.13	0.19	0.41	0.19	0.13	0	5.02	5.02	5.02	0	-5.02	-5.02	-5.02
70.00	0.13	0.19	0.41	0.19	0.13	0	5.17	5.17	5.17	0	-5.17	-5.17	-5.17
72.00	0.13	0.19	0.41	0.19	0.13	0	5.32	5.32	5.32	0	-5.32	-5.32	-5.32
74.00	0.13	0.19	0.41	0.19	0.13	0	5.47	5.47	5.47	0	-5.47	-5.47	-5.47
76.00	0.13	0.19	0.41	0.19	0.13	0	5.62	5.62	5.62	0	-5.62	-5.62	-5.62
78.00	0.13	0.19	0.41	0.19	0.13	0	5.76	5.76	5.76	0	-5.76	-5.76	-5.76
80.00	0.13	0.19	0.41	0.19	0.13	0	5.91	5.91	5.91	0	-5.91	-5.91	-5.91
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00
80.00	0.13	0.19	0.41	0.19	0.13	0	5.91	5.91	5.91	0	-5.91	-5.91	-5.91
82.00	0.13	0.19	0.41	0.19	0.13	0	6.06	6.06	6.06	0	-6.06	-6.06	-6.06
84.00	0.13	0.19	0.41	0.19	0.13	0	6.21	6.21	6.21	0	-6.21	-6.21	-6.21
86.00	0.13	0.19	0.41	0.19	0.13	0	6.35	6.35	6.35	0	-6.35	-6.35	-6.35
88.00	0.13	0.19	0.41	0.19	0.13	0	6.50	6.50	6.50	0	-6.50	-6.50	-6.50
90.00	0.13	0.19	0.41	0.19	0.13	0	6.65	6.65	6.65	0	-6.65	-6.65	-6.65
92.00	0.13	0.19	0.41	0.19	0.13	0	6.80	6.80	6.80	0	-6.80	-6.80	-6.80
94.00	0.13	0.19	0.41	0.19	0.13	0	6.95	6.95	6.95	0	-6.95	-6.95	-6.95

96.00	0.13	0.19	0.41	0.19	0.13	0	7.09	7.09	7.09	0	-7.09	-7.09	-7.09
98.00	0.13	0.19	0.41	0.19	0.13	0	7.24	7.24	7.24	0	-7.24	-7.24	-7.24
100.00	0.13	0.19	0.41	0.19	0.13	0	7.39	7.39	7.39	0	-7.39	-7.39	-7.39
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00
100.00	0.13	0.19	0.41	0.19	0.13	0	7.39	7.39	7.39	0	-7.39	-7.39	-7.39
102.00	0.13	0.19	0.41	0.19	0.13	0	7.54	7.54	7.54	0	-7.54	-7.54	-7.54
104.00	0.13	0.19	0.41	0.19	0.13	0	7.68	7.68	7.68	0	-7.68	-7.68	-7.68
106.00	0.13	0.19	0.41	0.19	0.13	0	7.83	7.83	7.83	0	-7.83	-7.83	-7.83
108.00	0.13	0.19	0.41	0.19	0.13	0	7.98	7.98	7.98	0	-7.98	-7.98	-7.98
110.00	0.13	0.19	0.41	0.19	0.13	0	8.13	8.13	8.13	0	-8.13	-8.13	-8.13
112.00	0.13	0.19	0.41	0.19	0.13	0	8.28	8.28	8.28	0	-8.28	-8.28	-8.28
114.00	0.13	0.19	0.41	0.19	0.13	0	8.42	8.42	8.42	0	-8.42	-8.42	-8.42
116.00	0.13	0.19	0.41	0.19	0.13	0	8.57	8.57	8.57	0	-8.57	-8.57	-8.57
118.00	0.13	0.19	0.41	0.19	0.13	0	8.72	8.72	8.72	0	-8.72	-8.72	-8.72
120.00	0.13	0.19	0.41	0.19	0.13	0	8.87	8.87	8.87	0	-8.87	-8.87	-8.87
				0.00	0.00	0					0.00	0.00	0.00
120.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	8.87	8.87	8.87	0	-8.87	-8.87	-8.87
121.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	8.37	8.37	8.37	0	-8.37	-8.37	-8.37
123.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	7.88	7.88	7.88	0	-7.88	-7.88	-7.88
125.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	7.38	7.38	7.38	0	-7.38	-7.38	-7.38
127.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	6.89	6.89	6.89	0	-6.89	-6.89	-6.89
129.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	6.39	6.39	6.39	0	-6.39	-6.39	-6.39
130.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	5.90	5.90	5.90	0	-5.90	-5.90	-5.90
	Shear Stresses, V*Q/ I*t							Normal Stresses, M/ Sx					
“s”=0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
z													
132.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	5.40	5.40	5.40	0	-5.40	-5.40	-5.40
134.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	4.91	4.91	4.91	0	-4.91	-4.91	-4.91
136.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	4.41	4.41	4.41	0	-4.41	-4.41	-4.41
138.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	3.92	3.92	3.92	0	-3.92	-3.92	-3.92
				0.00	0.00	0					0.00	0.00	0.00
138.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	3.92	3.92	3.92	0	-3.92	-3.92	-3.92
139.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	3.43	3.43	3.43	0	-3.43	-3.43	-3.43
141.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	2.93	2.93	2.93	0	-2.93	-2.93	-2.93
143.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	2.44	2.44	2.44	0	-2.44	-2.44	-2.44
145.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	1.94	1.94	1.94	0	-1.94	-1.94	-1.94
147.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	1.45	1.45	1.45	0	-1.45	-1.45	-1.45
148.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	0.95	0.95	0.95	0	-0.95	-0.95	-0.95
150.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	0.46	0.46	0.46	0	-0.46	-0.46	-0.46
152.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	-0.04	-0.04	-0.04	0	0.04	0.04	0.04
154.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	-0.53	-0.53	-0.53	0	0.53	0.53	0.53
156.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-1.03	-1.03	-1.03	0	1.03	1.03	1.03
				0.00	0.00	0					0.00	0.00	0.00
156.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-1.03	-1.03	-1.03	0	1.03	1.03	1.03

157.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	-1.52	-1.52	-1.52	0	1.52	1.52	1.52
159.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	-2.02	-2.02	-2.02	0	2.02	2.02	2.02
161.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	-2.51	-2.51	-2.51	0	2.51	2.51	2.51
163.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	-3.01	-3.01	-3.01	0	3.01	3.01	3.01
165.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-3.50	-3.50	-3.50	0	3.50	3.50	3.50
166.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	-3.99	-3.99	-3.99	0	3.99	3.99	3.99
168.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	-4.49	-4.49	-4.49	0	4.49	4.49	4.49
170.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	-4.98	-4.98	-4.98	0	4.98	4.98	4.98
172.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	-5.48	-5.48	-5.48	0	5.48	5.48	5.48
174.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-5.97	-5.97	-5.97	0	5.97	5.97	5.97
				0.00	0.00	0					0.00	0.00	0.00
174.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-5.97	-5.97	-5.97	0	5.97	5.97	5.97
175.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	-6.47	-6.47	-6.47	0	6.47	6.47	6.47
177.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	-6.96	-6.96	-6.96	0	6.96	6.96	6.96
179.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	-7.46	-7.46	-7.46	0	7.46	7.46	7.46
181.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	-7.95	-7.95	-7.95	0	7.95	7.95	7.95
183.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-8.45	-8.45	-8.45	0	8.45	8.45	8.45
184.80	-0.50	-0.72	-1.53	-0.72	-0.50	0	-8.94	-8.94	-8.94	0	8.94	8.94	8.94
186.60	-0.50	-0.72	-1.53	-0.72	-0.50	0	-9.44	-9.44	-9.44	0	9.44	9.44	9.44
188.40	-0.50	-0.72	-1.53	-0.72	-0.50	0	-9.93	-9.93	-9.93	0	9.93	9.93	9.93
190.20	-0.50	-0.72	-1.53	-0.72	-0.50	0	-10.43	-10.43	-10.43	0	10.43	10.43	10.43
192.00	-0.50	-0.72	-1.53	-0.72	-0.50	0	-10.92	-10.92	-10.92	0	10.92	10.92	10.92
				0.00	0.00	0					0.00	0.00	0.00
192.00	0.17	0.25	0.53	0.25	0.17	0	-10.92	-10.92	-10.92	0	10.92	10.92	10.92
194.10	0.17	0.25	0.53	0.25	0.17	0	-10.72	-10.72	-10.72	0	10.72	10.72	10.72
	Shear Stresses, $V*Q/ I*t$						Normal Stresses, $M/ Sx$						
“s”=0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
Z													
196.20	0.17	0.25	0.53	0.25	0.17	0	-10.52	-10.52	-10.52	0	10.52	10.52	10.52
198.30	0.17	0.25	0.53	0.25	0.17	0	-10.32	-10.32	-10.32	0	10.32	10.32	10.32
200.40	0.17	0.25	0.53	0.25	0.17	0	-10.12	-10.12	-10.12	0	10.12	10.12	10.12
202.50	0.17	0.25	0.53	0.25	0.17	0	-9.92	-9.92	-9.92	0	9.92	9.92	9.92
204.60	0.17	0.25	0.53	0.25	0.17	0	-9.72	-9.72	-9.72	0	9.72	9.72	9.72
206.70	0.17	0.25	0.53	0.25	0.17	0	-9.52	-9.52	-9.52	0	9.52	9.52	9.52
208.80	0.17	0.25	0.53	0.25	0.17	0	-9.31	-9.31	-9.31	0	9.31	9.31	9.31
210.90	0.17	0.25	0.53	0.25	0.17	0	-9.11	-9.11	-9.11	0	9.11	9.11	9.11
213.00	0.17	0.25	0.53	0.25	0.17	0	-8.91	-8.91	-8.91	0	8.91	8.91	8.91
	0	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00
213.00	0.17	0.25	0.53	0.25	0.17	0	-8.91	-8.91	-8.91	0	8.91	8.91	8.91
215.10	0.17	0.25	0.53	0.25	0.17	0	-8.71	-8.71	-8.71	0	8.71	8.71	8.71
217.20	0.17	0.25	0.53	0.25	0.17	0	-8.51	-8.51	-8.51	0	8.51	8.51	8.51
219.30	0.17	0.25	0.53	0.25	0.17	0	-8.31	-8.31	-8.31	0	8.31	8.31	8.31
221.40	0.17	0.25	0.53	0.25	0.17	0	-8.11	-8.11	-8.11	0	8.11	8.11	8.11
223.50	0.17	0.25	0.53	0.25	0.17	0	-7.91	-7.91	-7.91	0	7.91	7.91	7.91

225.60	0.17	0.25	0.53	0.25	0.17	0	-7.71	-7.71	-7.71	0	7.71	7.71	7.71	
227.70	0.17	0.25	0.53	0.25	0.17	0	-7.51	-7.51	-7.51	0	7.51	7.51	7.51	
229.80	0.17	0.25	0.53	0.25	0.17	0	-7.31	-7.31	-7.31	0	7.31	7.31	7.31	
231.90	0.17	0.25	0.53	0.25	0.17	0	-7.11	-7.11	-7.11	0	7.11	7.11	7.11	
234.00	0.17	0.25	0.53	0.25	0.17	0	-6.91	-6.91	-6.91	0	6.91	6.91	6.91	
	0	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00
234.00	0.17	0.25	0.53	0.25	0.17	0	-6.91	-6.91	-6.91	0	6.91	6.91	6.91	
236.10	0.17	0.25	0.53	0.25	0.17	0	-6.71	-6.71	-6.71	0	6.71	6.71	6.71	
238.20	0.17	0.25	0.53	0.25	0.17	0	-6.51	-6.51	-6.51	0	6.51	6.51	6.51	
240.30	0.17	0.25	0.53	0.25	0.17	0	-6.30	-6.30	-6.30	0	6.30	6.30	6.30	
242.40	0.17	0.25	0.53	0.25	0.17	0	-6.10	-6.10	-6.10	0	6.10	6.10	6.10	
244.50	0.17	0.25	0.53	0.25	0.17	0	-5.90	-5.90	-5.90	0	5.90	5.90	5.90	
246.60	0.17	0.25	0.53	0.25	0.17	0	-5.70	-5.70	-5.70	0	5.70	5.70	5.70	
248.70	0.17	0.25	0.53	0.25	0.17	0	-5.50	-5.50	-5.50	0	5.50	5.50	5.50	
250.80	0.17	0.25	0.53	0.25	0.17	0	-5.30	-5.30	-5.30	0	5.30	5.30	5.30	
252.90	0.17	0.25	0.53	0.25	0.17	0	-5.10	-5.10	-5.10	0	5.10	5.10	5.10	
255.00	0.17	0.25	0.53	0.25	0.17	0	-4.90	-4.90	-4.90	0	4.90	4.90	4.90	
	0	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00
255.00	0.17	0.25	0.53	0.25	0.17	0	-4.90	-4.90	-4.90	0	4.90	4.90	4.90	
257.10	0.17	0.25	0.53	0.25	0.17	0	-4.70	-4.70	-4.70	0	4.70	4.70	4.70	
259.20	0.17	0.25	0.53	0.25	0.17	0	-4.50	-4.50	-4.50	0	4.50	4.50	4.50	
261.30	0.17	0.25	0.53	0.25	0.17	0	-4.30	-4.30	-4.30	0	4.30	4.30	4.30	
263.40	0.17	0.25	0.53	0.25	0.17	0	-4.10	-4.10	-4.10	0	4.10	4.10	4.10	
265.50	0.17	0.25	0.53	0.25	0.17	0	-3.90	-3.90	-3.90	0	3.90	3.90	3.90	
267.60	0.17	0.25	0.53	0.25	0.17	0	-3.70	-3.70	-3.70	0	3.70	3.70	3.70	
269.70	0.17	0.25	0.53	0.25	0.17	0	-3.50	-3.50	-3.50	0	3.50	3.50	3.50	
271.80	0.17	0.25	0.53	0.25	0.17	0	-3.29	-3.29	-3.29	0	3.29	3.29	3.29	
	0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
z														
273.90	0.17	0.25	0.53	0.25	0.17	0	-3.09	-3.09	-3.09	0	3.09	3.09	3.09	
276.00	0.17	0.25	0.53	0.25	0.17	0	-2.89	-2.89	-2.89	0	2.89	2.89	2.89	
	0	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00
276.00	0.17	0.25	0.53	0.25	0.17	0	-2.89	-2.89	-2.89	0	2.89	2.89	2.89	
278.10	0.17	0.25	0.53	0.25	0.17	0	-2.69	-2.69	-2.69	0	2.69	2.69	2.69	
280.20	0.17	0.25	0.53	0.25	0.17	0	-2.49	-2.49	-2.49	0	2.49	2.49	2.49	
282.30	0.17	0.25	0.53	0.25	0.17	0	-2.29	-2.29	-2.29	0	2.29	2.29	2.29	
284.40	0.17	0.25	0.53	0.25	0.17	0	-2.09	-2.09	-2.09	0	2.09	2.09	2.09	
286.50	0.17	0.25	0.53	0.25	0.17	0	-1.89	-1.89	-1.89	0	1.89	1.89	1.89	
288.60	0.17	0.25	0.53	0.25	0.17	0	-1.69	-1.69	-1.69	0	1.69	1.69	1.69	
290.70	0.17	0.25	0.53	0.25	0.17	0	-1.49	-1.49	-1.49	0	1.49	1.49	1.49	
292.80	0.17	0.25	0.53	0.25	0.17	0	-1.29	-1.29	-1.29	0	1.29	1.29	1.29	
294.90	0.17	0.25	0.53	0.25	0.17	0	-1.09	-1.09	-1.09	0	1.09	1.09	1.09	
297.00	0.17	0.25	0.53	0.25	0.17	0	-0.89	-0.89	-0.89	0	0.89	0.89	0.89	
	0	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00

297.00	0.17	0.25	0.53	0.25	0.17	0	-0.89	-0.89	-0.89	0	0.89	0.89	0.89	
299.10	0.17	0.25	0.53	0.25	0.17	0	-0.69	-0.69	-0.69	0	0.69	0.69	0.69	
301.20	0.17	0.25	0.53	0.25	0.17	0	-0.49	-0.49	-0.49	0	0.49	0.49	0.49	
303.30	0.17	0.25	0.53	0.25	0.17	0	-0.28	-0.28	-0.28	0	0.28	0.28	0.28	
305.40	0.17	0.25	0.53	0.25	0.17	0	-0.08	-0.08	-0.08	0	0.08	0.08	0.08	
307.50	0.17	0.25	0.53	0.25	0.17	0	0.12	0.12	0.12	0	-0.12	-0.12	-0.12	
309.60	0.17	0.25	0.53	0.25	0.17	0	0.32	0.32	0.32	0	-0.32	-0.32	-0.32	
311.70	0.17	0.25	0.53	0.25	0.17	0	0.52	0.52	0.52	0	-0.52	-0.52	-0.52	
313.80	0.17	0.25	0.53	0.25	0.17	0	0.72	0.72	0.72	0	-0.72	-0.72	-0.72	
315.90	0.17	0.25	0.53	0.25	0.17	0	0.92	0.92	0.92	0	-0.92	-0.92	-0.92	
318.00	0.17	0.25	0.53	0.25	0.17	0	1.12	1.12	1.12	0	-1.12	-1.12	-1.12	
	0	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00	
318.00	0.17	0.25	0.53	0.25	0.17	0	1.12	1.12	1.12	0	-1.12	-1.12	-1.12	
320.10	0.17	0.25	0.53	0.25	0.17	0	1.32	1.32	1.32	0	-1.32	-1.32	-1.32	
322.20	0.17	0.25	0.53	0.25	0.17	0	1.52	1.52	1.52	0	-1.52	-1.52	-1.52	
324.30	0.17	0.25	0.53	0.25	0.17	0	1.72	1.72	1.72	0	-1.72	-1.72	-1.72	
326.40	0.17	0.25	0.53	0.25	0.17	0	1.92	1.92	1.92	0	-1.92	-1.92	-1.92	
328.50	0.17	0.25	0.53	0.25	0.17	0	2.12	2.12	2.12	0	-2.12	-2.12	-2.12	
330.60	0.17	0.25	0.53	0.25	0.17	0	2.32	2.32	2.32	0	-2.32	-2.32	-2.32	
332.70	0.17	0.25	0.53	0.25	0.17	0	2.52	2.52	2.52	0	-2.52	-2.52	-2.52	
334.80	0.17	0.25	0.53	0.25	0.17	0	2.73	2.73	2.73	0	-2.73	-2.73	-2.73	
336.90	0.17	0.25	0.53	0.25	0.17	0	2.93	2.93	2.93	0	-2.93	-2.93	-2.93	
339.00	0.17	0.25	0.53	0.25	0.17	0	3.13	3.13	3.13	0	-3.13	-3.13	-3.13	
	0	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	0	0.00	0.00	0.00	
339.00	0.17	0.25	0.53	0.25	0.17	0	3.13	3.13	3.13	0	-3.13	-3.13	-3.13	
341.10	0.17	0.25	0.53	0.25	0.17	0	3.33	3.33	3.33	0	-3.33	-3.33	-3.33	
343.20	0.17	0.25	0.53	0.25	0.17	0	3.53	3.53	3.53	0	-3.53	-3.53	-3.53	
345.30	0.17	0.25	0.53	0.25	0.17	0	3.73	3.73	3.73	0	-3.73	-3.73	-3.73	
347.40	0.17	0.25	0.53	0.25	0.17	0	3.93	3.93	3.93	0	-3.93	-3.93	-3.93	
	0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
349.50	0.17	0.25	0.53	0.25	0.17	0	4.13	4.13	4.13	0	-4.13	-4.13	-4.13	
351.60	0.17	0.25	0.53	0.25	0.17	0	4.33	4.33	4.33	0	-4.33	-4.33	-4.33	
353.70	0.17	0.25	0.53	0.25	0.17	0	4.53	4.53	4.53	0	-4.53	-4.53	-4.53	
355.80	0.17	0.25	0.53	0.25	0.17	0	4.73	4.73	4.73	0	-4.73	-4.73	-4.73	
357.90	0.17	0.25	0.53	0.25	0.17	0	4.93	4.93	4.93	0	-4.93	-4.93	-4.93	
360.00	0.17	0.25	0.53	0.25	0.17	0	5.13	5.13	5.13	0	-5.13	-5.13	-5.13	
				0.00	0.00	0					0.00	0.00	0.00	
360.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	5.13	5.13	5.13	0	-5.13	-5.13	-5.13	
362.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	5.05	5.05	5.05	0	-5.05	-5.05	-5.05	
364.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.96	4.96	4.96	0	-4.96	-4.96	-4.96	
366.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.88	4.88	4.88	0	-4.88	-4.88	-4.88	
368.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.79	4.79	4.79	0	-4.79	-4.79	-4.79	
370.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.71	4.71	4.71	0	-4.71	-4.71	-4.71	

372.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.62	4.62	4.62	0	-4.62	-4.62	-4.62	
374.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.53	4.53	4.53	0	-4.53	-4.53	-4.53	
376.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.45	4.45	4.45	0	-4.45	-4.45	-4.45	
378.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.36	4.36	4.36	0	-4.36	-4.36	-4.36	
380.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.28	4.28	4.28	0	-4.28	-4.28	-4.28	
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	
380.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.28	4.28	4.28	0	-4.28	-4.28	-4.28	
382.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.19	4.19	4.19	0	-4.19	-4.19	-4.19	
384.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.11	4.11	4.11	0	-4.11	-4.11	-4.11	
386.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	4.02	4.02	4.02	0	-4.02	-4.02	-4.02	
388.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.94	3.94	3.94	0	-3.94	-3.94	-3.94	
390.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.85	3.85	3.85	0	-3.85	-3.85	-3.85	
392.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.76	3.76	3.76	0	-3.76	-3.76	-3.76	
394.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.68	3.68	3.68	0	-3.68	-3.68	-3.68	
396.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.59	3.59	3.59	0	-3.59	-3.59	-3.59	
398.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.51	3.51	3.51	0	-3.51	-3.51	-3.51	
400.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.42	3.42	3.42	0	-3.42	-3.42	-3.42	
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	
400.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.42	3.42	3.42	0	-3.42	-3.42	-3.42	
402.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.34	3.34	3.34	0	-3.34	-3.34	-3.34	
404.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.25	3.25	3.25	0	-3.25	-3.25	-3.25	
406.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.17	3.17	3.17	0	-3.17	-3.17	-3.17	
408.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	3.08	3.08	3.08	0	-3.08	-3.08	-3.08	
410.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.99	2.99	2.99	0	-2.99	-2.99	-2.99	
412.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.91	2.91	2.91	0	-2.91	-2.91	-2.91	
414.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.82	2.82	2.82	0	-2.82	-2.82	-2.82	
416.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.74	2.74	2.74	0	-2.74	-2.74	-2.74	
418.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.65	2.65	2.65	0	-2.65	-2.65	-2.65	
420.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.57	2.57	2.57	0	-2.57	-2.57	-2.57	
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	
420.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.57	2.57	2.57	0	-2.57	-2.57	-2.57	
	0	1	2	3	2'	1'	0'	0	1	2	3	2'	1'	0'
Z														
422.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.48	2.48	2.48	0	-2.48	-2.48	-2.48	
424.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.40	2.40	2.40	0	-2.40	-2.40	-2.40	
426.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.31	2.31	2.31	0	-2.31	-2.31	-2.31	
428.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.22	2.22	2.22	0	-2.22	-2.22	-2.22	
430.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.14	2.14	2.14	0	-2.14	-2.14	-2.14	
432.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	2.05	2.05	2.05	0	-2.05	-2.05	-2.05	
434.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.97	1.97	1.97	0	-1.97	-1.97	-1.97	
436.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.88	1.88	1.88	0	-1.88	-1.88	-1.88	
438.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.80	1.80	1.80	0	-1.80	-1.80	-1.80	
440.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.71	1.71	1.71	0	-1.71	-1.71	-1.71	
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00	



440.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.71	1.71	1.71	0	-1.71	-1.71	-1.71
442.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.63	1.63	1.63	0	-1.63	-1.63	-1.63
444.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.54	1.54	1.54	0	-1.54	-1.54	-1.54
446.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.45	1.45	1.45	0	-1.45	-1.45	-1.45
448.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.37	1.37	1.37	0	-1.37	-1.37	-1.37
450.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.28	1.28	1.28	0	-1.28	-1.28	-1.28
452.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.20	1.20	1.20	0	-1.20	-1.20	-1.20
454.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.11	1.11	1.11	0	-1.11	-1.11	-1.11
456.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	1.03	1.03	1.03	0	-1.03	-1.03	-1.03
458.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.94	0.94	0.94	0	-0.94	-0.94	-0.94
460.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.86	0.86	0.86	0	-0.86	-0.86	-0.86
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0	0.00	0.00	0.00
460.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.86	0.86	0.86	0	-0.86	-0.86	-0.86
462.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.77	0.77	0.77	0	-0.77	-0.77	-0.77
464.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.68	0.68	0.68	0	-0.68	-0.68	-0.68
466.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.60	0.60	0.60	0	-0.60	-0.60	-0.60
468.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.51	0.51	0.51	0	-0.51	-0.51	-0.51
470.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.43	0.43	0.43	0	-0.43	-0.43	-0.43
472.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.34	0.34	0.34	0	-0.34	-0.34	-0.34
474.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.26	0.26	0.26	0	-0.26	-0.26	-0.26
476.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.17	0.17	0.17	0	-0.17	-0.17	-0.17
478.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.09	0.09	0.09	0	-0.09	-0.09	-0.09
480.00	-0.08	-0.11	-0.24	-0.11	-0.08	0	0.00	0.00	0.00	0	0.00	0.00	0.00

Table of Combined Normal Stresses

Z								
0.0	0.000	0.000	0.000	0.000	0.000	0.000	0.000	0.000
2.0	0.139	0.148	0.152	0.000	-0.152	-0.148	-0.139	
4.0	0.277	0.296	0.304	0.000	-0.304	-0.296	-0.277	
6.0	0.415	0.443	0.455	0.000	-0.455	-0.443	-0.415	
8.0	0.553	0.591	0.607	0.000	-0.607	-0.591	-0.553	
10.0	0.691	0.739	0.759	0.000	-0.759	-0.739	-0.691	
12.0	0.828	0.887	0.912	0.000	-0.912	-0.887	-0.828	
14.0	0.965	1.034	1.064	0.000	-1.064	-1.034	-0.965	
16.0	1.101	1.182	1.217	0.000	-1.217	-1.182	-1.101	
18.0	1.236	1.330	1.370	0.000	-1.370	-1.330	-1.236	
20.0	1.371	1.478	1.524	0.000	-1.524	-1.478	-1.371	
	0.000	0.000	0.000	0.000	0.000	0.000	0.000	
20.0	1.371	1.478	1.524	0.000	-1.524	-1.478	-1.371	
22.0	1.504	1.626	1.678	0.000	-1.678	-1.626	-1.504	
24.0	1.637	1.773	1.832	0.000	-1.832	-1.773	-1.637	
26.0	1.768	1.921	1.987	0.000	-1.987	-1.921	-1.768	
28.0	1.898	2.069	2.142	0.000	-2.142	-2.069	-1.898	
30.0	2.026	2.217	2.299	0.000	-2.299	-2.217	-2.026	

32.0	2.152	2.364	2.456	0.000	-2.456	-2.364	-2.152
34.0	2.277	2.512	2.613	0.000	-2.613	-2.512	-2.277
36.0	2.399	2.660	2.772	0.000	-2.772	-2.660	-2.399
38.0	2.519	2.808	2.932	0.000	-2.932	-2.808	-2.519
40.0	2.636	2.956	3.093	0.000	-3.093	-2.956	-2.636
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
40.0	2.636	2.956	3.093	0.000	-3.093	-2.956	-2.636
42.0	2.750	3.103	3.255	0.000	-3.255	-3.103	-2.750
44.0	2.861	3.251	3.418	0.000	-3.418	-3.251	-2.861
46.0	2.969	3.399	3.583	0.000	-3.583	-3.399	-2.969
48.0	3.073	3.547	3.750	0.000	-3.750	-3.547	-3.073
50.0	3.172	3.694	3.919	0.000	-3.919	-3.694	-3.172
52.0	3.266	3.842	4.089	0.000	-4.089	-3.842	-3.266
54.0	3.356	3.990	4.262	0.000	-4.262	-3.990	-3.356
56.0	3.439	4.138	4.438	0.000	-4.438	-4.138	-3.439
58.0	3.516	4.286	4.616	0.000	-4.616	-4.286	-3.516
60.0	3.586	4.433	4.797	0.000	-4.797	-4.433	-3.586
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
60.0	3.586	4.433	4.797	0.000	-4.797	-4.433	-3.586
62.0	3.648	4.581	4.982	0.000	-4.982	-4.581	-3.648
64.0	3.701	4.729	5.170	0.000	-5.170	-4.729	-3.701
66.0	3.746	4.877	5.362	0.000	-5.362	-4.877	-3.746
68.0	3.780	5.024	5.558	0.000	-5.558	-5.024	-3.780
70.0	3.802	5.172	5.760	0.000	-5.760	-5.172	-3.802
72.0	3.812	5.320	5.967	0.000	-5.967	-5.320	-3.812
74.0	3.808	5.468	6.180	0.000	-6.180	-5.468	-3.808
76.0	3.788	5.616	6.400	0.000	-6.400	-5.616	-3.788
78.0	3.753	5.763	6.626	0.000	-6.626	-5.763	-3.753
80.0	3.699	5.911	6.860	0.000	-6.860	-5.911	-3.699
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
80.0	3.700	5.911	6.860	0.000	-6.860	-5.911	-3.700
82.0	3.624	6.059	7.104	0.000	-7.104	-6.059	-3.624
84.0	3.527	6.207	7.357	0.000	-7.357	-6.207	-3.527
86.0	3.406	6.354	7.620	0.000	-7.620	-6.354	-3.406
88.0	3.257	6.502	7.894	0.000	-7.894	-6.502	-3.257
90.0	3.081	6.650	8.182	0.000	-8.182	-6.650	-3.081
92.0	2.869	6.798	8.484	0.000	-8.484	-6.798	-2.869
94.0	2.619	6.946	8.802	0.000	-8.802	-6.946	-2.619
96.0	2.333	7.093	9.136	0.000	-9.136	-7.093	-2.333
98.0	2.006	7.241	9.487	0.000	-9.487	-7.241	-2.006
100.0	1.628	7.389	9.861	0.000	-9.861	-7.389	-1.628
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
100.0	1.631	7.389	9.859	0.000	-9.859	-7.389	-1.631

102.0	1.195	7.537	10.258	0.000	-10.258	-7.537	-1.195
104.0	0.705	7.684	10.679	0.000	-10.679	-7.684	-0.705
106.0	0.154	7.832	11.127	0.000	-11.127	-7.832	-0.154
108.0	-0.469	7.980	11.605	0.000	-11.605	-7.980	0.469
110.0	-1.169	8.128	12.117	0.000	-12.117	-8.128	1.169
112.0	-1.954	8.276	12.665	0.000	-12.665	-8.276	1.954
114.0	-2.835	8.423	13.254	0.000	-13.254	-8.423	2.835
116.0	-3.820	8.571	13.888	0.000	-13.888	-8.571	3.820
118.0	-4.914	8.719	14.568	0.000	-14.568	-8.719	4.914
120.0	-6.130	8.867	15.301	0.000	-15.301	-8.867	6.130
					0.000	0.000	0.000
120.0	-6.134	8.867	15.303	0.000	-15.303	-8.867	6.134
121.8	-5.285	8.372	14.232	0.000	-14.232	-8.372	5.285
123.6	-4.534	7.877	13.203	0.000	-13.203	-7.877	4.534
125.4	-3.872	7.383	12.212	0.000	-12.212	-7.383	3.872
127.2	-3.294	6.888	11.257	0.000	-11.257	-6.888	3.294
129.0	-2.795	6.393	10.336	0.000	-10.336	-6.393	2.795
130.8	-2.363	5.899	9.444	0.000	-9.444	-5.899	2.363
132.6	-1.993	5.404	8.578	0.000	-8.578	-5.404	1.993
134.4	-1.676	4.909	7.735	0.000	-7.735	-4.909	1.676
136.2	-1.411	4.415	6.914	0.000	-6.914	-4.415	1.411
138.0	-1.183	3.920	6.110	0.000	-6.110	-3.920	1.183
					0.000	0.000	0.000
138.0	-1.186	3.920	6.111	0.000	-6.111	-3.920	1.186
139.8	-0.999	3.425	5.324	0.000	-5.324	-3.425	0.999
141.6	-0.846	2.931	4.551	0.000	-4.551	-2.931	0.846
143.4	-0.722	2.436	3.791	0.000	-3.791	-2.436	0.722
145.2	-0.620	1.941	3.040	0.000	-3.040	-1.941	0.620
147.0	-0.537	1.447	2.298	0.000	-2.298	-1.447	0.537
148.8	-0.469	0.952	1.562	0.000	-1.562	-0.952	0.469
150.6	-0.412	0.457	0.830	0.000	-0.830	-0.457	0.412
152.4	-0.361	-0.037	0.102	0.000	-0.102	0.037	0.361
154.2	-0.312	-0.532	-0.626	0.000	0.626	0.532	0.312
156.0	-0.261	-1.027	-1.355	0.000	1.355	1.027	0.261
					0.000	0.000	0.000
156.0	-0.263	-1.027	-1.354	0.000	1.354	1.027	0.263
157.8	-0.206	-1.521	-2.086	0.000	2.086	1.521	0.206
159.6	-0.140	-2.016	-2.821	0.000	2.821	2.016	0.140
161.4	-0.061	-2.511	-3.562	0.000	3.562	2.511	0.061
163.2	0.037	-3.005	-4.311	0.000	4.311	3.005	-0.037
165.0	0.158	-3.500	-5.069	0.000	5.069	3.500	-0.158
166.8	0.304	-3.995	-5.839	0.000	5.839	3.995	-0.304
168.6	0.485	-4.489	-6.624	0.000	6.624	4.489	-0.485

170.4	0.703	-4.984	-7.424	0.000	7.424	4.984	-0.703
172.2	0.958	-5.479	-8.240	0.000	8.240	5.479	-0.958
174.0	1.264	-5.973	-9.079	0.000	9.079	5.973	-1.264
					0.000	0.000	0.000
174.0	1.261	-5.973	-9.077	0.000	9.077	5.973	-1.261
175.8	1.628	-6.468	-9.942	0.000	9.942	6.468	-1.628
177.6	2.046	-6.963	-10.828	0.000	10.828	6.963	-2.046
179.4	2.532	-7.457	-11.743	0.000	11.743	7.457	-2.532
181.2	3.092	-7.952	-12.691	0.000	12.691	7.952	-3.092
183.0	3.734	-8.447	-13.673	0.000	13.673	8.447	-3.734
184.8	4.468	-8.941	-14.695	0.000	14.695	8.941	-4.468
186.6	5.303	-9.436	-15.760	0.000	15.760	9.436	-5.303
188.4	6.247	-9.931	-16.872	0.000	16.872	9.931	-6.247
190.2	7.306	-10.425	-18.033	0.000	18.033	10.425	-7.306
192.0	8.491	-10.920	-19.249	0.000	19.249	10.920	-8.491
					0.000	0.000	0.000
192.0	8.491	-10.920	-19.249	0.000	19.249	10.920	-8.491
194.1	6.846	-10.719	-18.256	0.000	18.256	10.719	-6.846
196.2	5.367	-10.519	-17.335	0.000	17.335	10.519	-5.367
198.3	4.048	-10.318	-16.482	0.000	16.482	10.318	-4.048
200.4	2.871	-10.117	-15.690	0.000	15.690	10.117	-2.871
202.5	1.830	-9.917	-14.957	0.000	14.957	9.917	-1.830
204.6	0.907	-9.716	-14.274	0.000	14.274	9.716	-0.907
206.7	0.094	-9.515	-13.638	0.000	13.638	9.515	-0.094
208.8	-0.622	-9.315	-13.044	0.000	13.044	9.315	0.622
210.9	-1.252	-9.114	-12.487	0.000	12.487	9.114	1.252
213.0	-1.815	-8.913	-11.959	0.000	11.959	8.913	1.815
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
213.0	-1.808	-8.913	-11.962	0.000	11.962	8.913	1.808
215.1	-2.286	-8.713	-11.470	0.000	11.470	8.713	2.286
217.2	-2.700	-8.512	-11.006	0.000	11.006	8.512	2.700
219.3	-3.056	-8.311	-10.566	0.000	10.566	8.311	3.056
221.4	-3.360	-8.111	-10.149	0.000	10.149	8.111	3.360
223.5	-3.618	-7.910	-9.752	0.000	9.752	7.910	3.618
225.6	-3.828	-7.709	-9.375	0.000	9.375	7.709	3.828
227.7	-4.000	-7.509	-9.014	0.000	9.014	7.509	4.000
229.8	-4.136	-7.308	-8.669	0.000	8.669	7.308	4.136
231.9	-4.241	-7.107	-8.337	0.000	8.337	7.107	4.241
234.0	-4.320	-6.907	-8.017	0.000	8.017	6.907	4.320
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
234.0	-4.318	-6.907	-8.017	0.000	8.017	6.907	4.318
236.1	-4.367	-6.706	-7.710	0.000	7.710	6.706	4.367
238.2	-4.393	-6.505	-7.412	0.000	7.412	6.505	4.393

240.3	-4.398	-6.305	-7.123	0.000	7.123	6.305	4.398
242.4	-4.384	-6.104	-6.842	0.000	6.842	6.104	4.384
244.5	-4.352	-5.903	-6.569	0.000	6.569	5.903	4.352
246.6	-4.305	-5.703	-6.302	0.000	6.302	5.703	4.305
248.7	-4.243	-5.502	-6.042	0.000	6.042	5.502	4.243
250.8	-4.168	-5.301	-5.788	0.000	5.788	5.301	4.168
252.9	-4.082	-5.101	-5.538	0.000	5.538	5.101	4.082
255.0	-3.988	-4.900	-5.291	0.000	5.291	4.900	3.988
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
255.0	-3.987	-4.900	-5.292	0.000	5.292	4.900	3.987
257.1	-3.881	-4.699	-5.050	0.000	5.050	4.699	3.881
259.2	-3.768	-4.499	-4.812	0.000	4.812	4.499	3.768
261.3	-3.647	-4.298	-4.577	0.000	4.577	4.298	3.647
263.4	-3.520	-4.097	-4.345	0.000	4.345	4.097	3.520
265.5	-3.386	-3.897	-4.116	0.000	4.116	3.897	3.386
267.6	-3.248	-3.696	-3.888	0.000	3.888	3.696	3.248
269.7	-3.105	-3.495	-3.663	0.000	3.663	3.495	3.105
271.8	-2.958	-3.295	-3.439	0.000	3.439	3.295	2.958
273.9	-2.807	-3.094	-3.217	0.000	3.217	3.094	2.807
276.0	-2.655	-2.893	-2.996	0.000	2.996	2.893	2.655
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
276.0	-2.654	-2.893	-2.996	0.000	2.996	2.893	2.654
278.1	-2.499	-2.693	-2.776	0.000	2.776	2.693	2.499
280.2	-2.341	-2.492	-2.557	0.000	2.557	2.492	2.341
282.3	-2.182	-2.291	-2.338	0.000	2.338	2.291	2.182
284.4	-2.022	-2.091	-2.120	0.000	2.120	2.091	2.022
286.5	-1.861	-1.890	-1.903	0.000	1.903	1.890	1.861
288.6	-1.699	-1.689	-1.685	0.000	1.685	1.689	1.699
290.7	-1.538	-1.489	-1.467	0.000	1.467	1.489	1.538
292.8	-1.378	-1.288	-1.250	0.000	1.250	1.288	1.378
294.9	-1.218	-1.087	-1.031	0.000	1.031	1.087	1.218
297.0	-1.059	-0.887	-0.813	0.000	0.813	0.887	1.059
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
297.0	-1.059	-0.887	-0.813	0.000	0.813	0.887	1.059
299.1	-0.903	-0.686	-0.593	0.000	0.593	0.686	0.903
301.2	-0.748	-0.485	-0.372	0.000	0.372	0.485	0.748
303.3	-0.596	-0.285	-0.151	0.000	0.151	0.285	0.596
305.4	-0.448	-0.084	0.072	0.000	-0.072	0.084	0.448
307.5	-0.303	0.117	0.297	0.000	-0.297	-0.117	0.303
309.6	-0.162	0.317	0.523	0.000	-0.523	-0.317	0.162
311.7	-0.026	0.518	0.751	0.000	-0.751	-0.518	0.026
313.8	0.105	0.719	0.982	0.000	-0.982	-0.719	-0.105
315.9	0.229	0.919	1.216	0.000	-1.216	-0.919	-0.229

318.0	0.346	1.120	1.452	0.000	-1.452	-1.120	-0.346
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
318.0	0.347	1.120	1.452	0.000	-1.452	-1.120	-0.347
320.1	0.455	1.321	1.692	0.000	-1.692	-1.321	-0.455
322.2	0.556	1.521	1.936	0.000	-1.936	-1.521	-0.556
324.3	0.647	1.722	2.183	0.000	-2.183	-1.722	-0.647
326.4	0.727	1.923	2.436	0.000	-2.436	-1.923	-0.727
328.5	0.796	2.123	2.693	0.000	-2.693	-2.123	-0.796
330.6	0.850	2.324	2.956	0.000	-2.956	-2.324	-0.850
332.7	0.889	2.525	3.226	0.000	-3.226	-2.525	-0.889
334.8	0.912	2.725	3.503	0.000	-3.503	-2.725	-0.912
336.9	0.917	2.926	3.788	0.000	-3.788	-2.926	-0.917
339.0	0.903	3.127	4.081	0.000	-4.081	-3.127	-0.903
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
339.0	0.904	3.127	4.080	0.000	-4.080	-3.127	-0.904
341.1	0.864	3.327	4.384	0.000	-4.384	-3.327	-0.864
343.2	0.801	3.528	4.698	0.000	-4.698	-3.528	-0.801
345.3	0.711	3.729	5.023	0.000	-5.023	-3.729	-0.711
347.4	0.591	3.929	5.362	0.000	-5.362	-3.929	-0.591
349.5	0.435	4.130	5.715	0.000	-5.715	-4.130	-0.435
351.6	0.245	4.331	6.083	0.000	-6.083	-4.331	-0.245
353.7	0.012	4.531	6.470	0.000	-6.470	-4.531	-0.012
355.8	-0.269	4.732	6.878	0.000	-6.878	-4.732	0.269
357.9	-0.598	4.933	7.306	0.000	-7.306	-4.933	0.598
360.0	-0.981	5.133	7.757	0.000	-7.757	-5.133	0.981
					0.000	0.000	0.000
360.0	-0.981	5.133	7.757	0.000	-7.757	-5.133	0.981
362.0	-0.510	5.048	7.432	0.000	-7.432	-5.048	0.510
364.0	-0.090	4.962	7.130	0.000	-7.130	-4.962	0.090
366.0	0.286	4.877	6.846	0.000	-6.846	-4.877	-0.286
368.0	0.621	4.791	6.580	0.000	-6.580	-4.791	-0.621
370.0	0.916	4.706	6.332	0.000	-6.332	-4.706	-0.916
372.0	1.176	4.620	6.098	0.000	-6.098	-4.620	-1.176
374.0	1.404	4.534	5.878	0.000	-5.878	-4.534	-1.404
376.0	1.604	4.449	5.670	0.000	-5.670	-4.449	-1.604
378.0	1.778	4.363	5.473	0.000	-5.473	-4.363	-1.778
380.0	1.931	4.278	5.285	0.000	-5.285	-4.278	-1.931
380.0	1.929	4.278	5.285	0.000	-5.285	-4.278	-1.929
382.0	2.057	4.192	5.108	0.000	-5.108	-4.192	-2.057
384.0	2.167	4.107	4.939	0.000	-4.939	-4.107	-2.167
386.0	2.258	4.021	4.777	0.000	-4.777	-4.021	-2.258
388.0	2.334	3.936	4.623	0.000	-4.623	-3.936	-2.334
390.0	2.395	3.850	4.474	0.000	-4.474	-3.850	-2.395

392.0	2.442	3.764	4.332	0.000	-4.332	-3.764	-2.442
394.0	2.477	3.679	4.195	0.000	-4.195	-3.679	-2.477
396.0	2.501	3.593	4.062	0.000	-4.062	-3.593	-2.501
398.0	2.515	3.508	3.934	0.000	-3.934	-3.508	-2.515
400.0	2.521	3.422	3.809	0.000	-3.809	-3.422	-2.521
400.0	2.521	3.422	3.809	0.000	-3.809	-3.422	-2.521
402.0	2.517	3.337	3.688	0.000	-3.688	-3.337	-2.517
404.0	2.506	3.251	3.571	0.000	-3.571	-3.251	-2.506
406.0	2.489	3.166	3.456	0.000	-3.456	-3.166	-2.489
408.0	2.465	3.080	3.344	0.000	-3.344	-3.080	-2.465
410.0	2.436	2.994	3.234	0.000	-3.234	-2.994	-2.436
412.0	2.402	2.909	3.127	0.000	-3.127	-2.909	-2.402
414.0	2.362	2.823	3.021	0.000	-3.021	-2.823	-2.362
416.0	2.319	2.738	2.918	0.000	-2.918	-2.738	-2.319
418.0	2.272	2.652	2.815	0.000	-2.815	-2.652	-2.272
420.0	2.222	2.567	2.715	0.000	-2.715	-2.567	-2.222
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
420.0	2.221	2.567	2.715	0.000	-2.715	-2.567	-2.221
422.0	2.167	2.481	2.616	0.000	-2.616	-2.481	-2.167
424.0	2.111	2.396	2.518	0.000	-2.518	-2.396	-2.111
426.0	2.051	2.310	2.421	0.000	-2.421	-2.310	-2.051
428.0	1.990	2.224	2.325	0.000	-2.325	-2.224	-1.990
430.0	1.926	2.139	2.230	0.000	-2.230	-2.139	-1.926
432.0	1.860	2.053	2.136	0.000	-2.136	-2.053	-1.860
434.0	1.793	1.968	2.043	0.000	-2.043	-1.968	-1.793
436.0	1.723	1.882	1.950	0.000	-1.950	-1.882	-1.723
438.0	1.653	1.797	1.858	0.000	-1.858	-1.797	-1.653
440.0	1.581	1.711	1.767	0.000	-1.767	-1.711	-1.581
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
440.0	1.581	1.711	1.767	0.000	-1.767	-1.711	-1.581
442.0	1.508	1.626	1.676	0.000	-1.676	-1.626	-1.508
444.0	1.434	1.540	1.586	0.000	-1.586	-1.540	-1.434
446.0	1.358	1.454	1.496	0.000	-1.496	-1.454	-1.358
448.0	1.282	1.369	1.406	0.000	-1.406	-1.369	-1.282
450.0	1.205	1.283	1.317	0.000	-1.317	-1.283	-1.205
452.0	1.128	1.198	1.228	0.000	-1.228	-1.198	-1.128
454.0	1.050	1.112	1.139	0.000	-1.139	-1.112	-1.050
456.0	0.971	1.027	1.051	0.000	-1.051	-1.027	-0.971
458.0	0.892	0.941	0.962	0.000	-0.962	-0.941	-0.892
460.0	0.812	0.856	0.874	0.000	-0.874	-0.856	-0.812
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
460.0	0.812	0.856	0.874	0.000	-0.874	-0.856	-0.812
462.0	0.732	0.770	0.786	0.000	-0.786	-0.770	-0.732

464.0	0.651	0.684	0.699	0.000	-0.699	-0.684	-0.651
466.0	0.571	0.599	0.611	0.000	-0.611	-0.599	-0.571
468.0	0.490	0.513	0.524	0.000	-0.524	-0.513	-0.490
470.0	0.408	0.428	0.436	0.000	-0.436	-0.428	-0.408
472.0	0.327	0.342	0.349	0.000	-0.349	-0.342	-0.327
474.0	0.245	0.257	0.262	0.000	-0.262	-0.257	-0.245
476.0	0.164	0.171	0.174	0.000	-0.174	-0.171	-0.164
478.0	0.082	0.086	0.087	0.000	-0.087	-0.086	-0.082
480.0	0.000	0.000	0.000	0.000	0.000	0.000	0.000

Total Shear with Positive St. Vt. Contribution

z	Combined Shear Stress with Positive St. Vt.						
0.0	-0.901	-0.771	-0.713	-0.505	-0.713	-0.771	-0.901
2.0	-0.901	-0.771	-0.712	-0.505	-0.712	-0.771	-0.901
4.0	-0.900	-0.770	-0.712	-0.504	-0.712	-0.770	-0.900
6.0	-0.900	-0.770	-0.711	-0.503	-0.711	-0.770	-0.900
8.0	-0.898	-0.769	-0.710	-0.502	-0.710	-0.769	-0.898
10.0	-0.897	-0.768	-0.709	-0.501	-0.709	-0.768	-0.897
12.0	-0.895	-0.766	-0.707	-0.498	-0.707	-0.766	-0.895
14.0	-0.893	-0.764	-0.706	-0.496	-0.706	-0.764	-0.893
16.0	-0.890	-0.762	-0.703	-0.493	-0.703	-0.762	-0.890
18.0	-0.887	-0.759	-0.700	-0.490	-0.700	-0.759	-0.887
20.0	-0.884	-0.756	-0.697	-0.487	-0.697	-0.756	-0.884
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
20.0	-0.884	-0.756	-0.697	-0.486	-0.697	-0.756	-0.884
22.0	-0.880	-0.753	-0.694	-0.483	-0.694	-0.753	-0.880
24.0	-0.876	-0.749	-0.690	-0.478	-0.690	-0.749	-0.876
26.0	-0.871	-0.745	-0.685	-0.472	-0.685	-0.745	-0.871
28.0	-0.865	-0.740	-0.681	-0.467	-0.681	-0.740	-0.865
30.0	-0.859	-0.735	-0.675	-0.460	-0.675	-0.735	-0.859
32.0	-0.852	-0.729	-0.669	-0.452	-0.669	-0.729	-0.852
34.0	-0.844	-0.722	-0.662	-0.444	-0.662	-0.722	-0.844
36.0	-0.836	-0.715	-0.654	-0.435	-0.654	-0.715	-0.836
38.0	-0.826	-0.707	-0.646	-0.425	-0.646	-0.707	-0.826
40.0	-0.816	-0.698	-0.637	-0.414	-0.637	-0.698	-0.816
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
40.0	-0.816	-0.698	-0.637	-0.413	-0.637	-0.698	-0.816
42.0	-0.805	-0.688	-0.627	-0.402	-0.627	-0.688	-0.805
44.0	-0.792	-0.677	-0.616	-0.388	-0.616	-0.677	-0.792
46.0	-0.778	-0.665	-0.603	-0.373	-0.603	-0.665	-0.778
48.0	-0.762	-0.651	-0.589	-0.356	-0.589	-0.651	-0.762



50.0	-0.746	-0.637	-0.574	-0.338	-0.574	-0.637	-0.746
52.0	-0.726	-0.620	-0.557	-0.318	-0.557	-0.620	-0.726
54.0	-0.706	-0.603	-0.539	-0.296	-0.539	-0.603	-0.706
56.0	-0.683	-0.583	-0.519	-0.271	-0.519	-0.583	-0.683
58.0	-0.658	-0.561	-0.496	-0.244	-0.496	-0.561	-0.658
60.0	-0.630	-0.537	-0.471	-0.214	-0.471	-0.537	-0.630
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
60.0	-0.630	-0.538	-0.472	-0.214	-0.472	-0.538	-0.630
62.0	-0.600	-0.511	-0.445	-0.182	-0.445	-0.511	-0.600
64.0	-0.566	-0.482	-0.415	-0.146	-0.415	-0.482	-0.566
66.0	-0.529	-0.450	-0.382	-0.106	-0.382	-0.450	-0.529
68.0	-0.489	-0.415	-0.346	-0.062	-0.346	-0.415	-0.489
70.0	-0.444	-0.377	-0.306	-0.014	-0.306	-0.377	-0.444
72.0	-0.395	-0.334	-0.263	0.039	-0.263	-0.334	-0.395
74.0	-0.340	-0.287	-0.215	0.097	-0.215	-0.287	-0.340
76.0	-0.281	-0.236	-0.162	0.161	-0.162	-0.236	-0.281
78.0	-0.215	-0.179	-0.103	0.231	-0.103	-0.179	-0.215
80.0	-0.143	-0.116	-0.038	0.308	-0.038	-0.116	-0.143
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
80.0	-0.143	-0.118	-0.040	0.309	-0.040	-0.118	-0.143
82.0	-0.063	-0.049	0.031	0.394	0.031	-0.049	-0.063
84.0	0.024	0.028	0.109	0.488	0.109	0.028	0.024
86.0	0.121	0.111	0.195	0.591	0.195	0.111	0.121
88.0	0.227	0.202	0.289	0.705	0.289	0.202	0.227
90.0	0.343	0.303	0.393	0.830	0.393	0.303	0.343
92.0	0.472	0.413	0.506	0.968	0.506	0.413	0.472
94.0	0.613	0.535	0.632	1.120	0.632	0.535	0.613
96.0	0.768	0.669	0.770	1.286	0.770	0.669	0.768
98.0	0.940	0.818	0.923	1.470	0.923	0.818	0.940
100.0	1.127	0.981	1.091	1.671	1.091	0.981	1.127
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
100.0	1.127	0.975	1.085	1.673	1.085	0.975	1.127
102.0	1.335	1.157	1.272	1.895	1.272	1.157	1.335
104.0	1.563	1.355	1.476	2.138	1.476	1.355	1.563
106.0	1.814	1.573	1.699	2.407	1.699	1.573	1.814
108.0	2.090	1.811	1.944	2.704	1.944	1.811	2.090
110.0	2.393	2.072	2.213	3.029	2.213	2.072	2.393
112.0	2.727	2.359	2.509	3.388	2.509	2.359	2.727
114.0	3.095	2.677	2.836	3.783	2.836	2.677	3.095
116.0	3.500	3.027	3.196	4.217	3.196	3.027	3.500
118.0	3.945	3.413	3.593	4.694	3.593	3.413	3.945
120.0	4.436	3.840	4.032	5.219	4.032	3.840	4.436

					0.000	0.000	0.000
120.0	4.436	4.730	4.369	2.671	4.369	4.730	4.436
121.8	4.877	5.113	4.763	3.143	4.763	5.113	4.877
123.6	5.279	5.461	5.121	3.573	5.121	5.461	5.279
125.4	5.644	5.776	5.445	3.965	5.445	5.776	5.644
127.2	5.974	6.061	5.739	4.319	5.739	6.061	5.974
129.0	6.271	6.316	6.001	4.638	6.001	6.316	6.271
130.8	6.538	6.546	6.238	4.925	6.238	6.546	6.538
132.6	6.782	6.757	6.455	5.187	6.455	6.757	6.782
134.4	6.997	6.943	6.647	5.417	6.647	6.943	6.997
136.2	7.189	7.110	6.818	5.622	6.818	7.110	7.189
138.0	7.357	7.258	6.969	5.802	6.969	7.258	7.357
					0.000	0.000	0.000
138.0	7.357	7.254	6.966	5.804	6.966	7.254	7.357
139.8	7.503	7.380	7.096	5.960	7.096	7.380	7.503
141.6	7.631	7.490	7.210	6.096	7.210	7.490	7.631
143.4	7.735	7.580	7.302	6.209	7.302	7.580	7.735
145.2	7.828	7.661	7.385	6.308	7.385	7.661	7.828
147.0	7.898	7.721	7.447	6.383	7.447	7.721	7.898
148.8	7.950	7.766	7.493	6.439	7.493	7.766	7.950
150.6	7.985	7.796	7.524	6.477	7.524	7.796	7.985
152.4	8.003	7.811	7.540	6.496	7.540	7.811	8.003
154.2	8.003	7.811	7.540	6.496	7.540	7.811	8.003
156.0	7.985	7.796	7.524	6.477	7.524	7.796	7.985
					0.000	0.000	0.000
156.0	7.985	7.797	7.525	6.477	7.525	7.797	7.985
157.8	7.956	7.772	7.499	6.446	7.499	7.772	7.956
159.6	7.904	7.726	7.452	6.390	7.452	7.726	7.904
161.4	7.840	7.671	7.395	6.321	7.395	7.671	7.840
163.2	7.753	7.595	7.317	6.228	7.317	7.595	7.753
165.0	7.654	7.511	7.230	6.121	7.230	7.511	7.654
166.8	7.532	7.406	7.122	5.990	7.122	7.406	7.532
168.6	7.387	7.280	6.993	5.835	6.993	7.280	7.387
170.4	7.224	7.140	6.848	5.660	6.848	7.140	7.224
172.2	7.038	6.978	6.683	5.461	6.683	6.978	7.038
174.0	6.823	6.791	6.490	5.231	6.490	6.791	6.823
					0.000	0.000	0.000
174.0	6.823	6.797	6.495	5.228	6.495	6.797	6.823
175.8	6.590	6.594	6.287	4.980	6.287	6.594	6.590
177.6	6.323	6.361	6.048	4.694	6.048	6.361	6.323
179.4	6.032	6.110	5.789	4.382	5.789	6.110	6.032
181.2	5.709	5.831	5.502	4.035	5.502	5.831	5.709

183.0	5.351	5.523	5.185	3.651	5.185	5.523	5.351
184.8	4.957	5.183	4.835	3.228	4.835	5.183	4.957
186.6	4.523	4.810	4.451	2.762	4.451	4.810	4.523
188.4	4.047	4.398	4.027	2.251	4.027	4.398	4.047
190.2	3.524	3.945	3.561	1.692	3.561	3.945	3.524
192.0	2.952	3.448	3.051	1.079	3.051	3.448	2.952
					0.000	0.000	0.000
192.0	2.952	2.181	2.429	3.915	2.429	2.181	2.952
194.1	2.287	1.602	1.834	3.204	1.834	1.602	2.287
196.2	1.687	1.081	1.298	2.560	1.298	1.081	1.687
198.3	1.143	0.611	0.815	1.977	0.815	0.611	1.143
200.4	0.651	0.187	0.378	1.449	0.378	0.187	0.651
202.5	0.207	-0.196	-0.015	0.972	-0.015	-0.196	0.207
204.6	-0.195	-0.542	-0.372	0.540	-0.372	-0.542	-0.195
206.7	-0.559	-0.855	-0.694	0.150	-0.694	-0.855	-0.559
208.8	-0.888	-1.140	-0.987	-0.202	-0.987	-1.140	-0.888
210.9	-1.186	-1.399	-1.254	-0.521	-1.254	-1.399	-1.186
213.0	-1.454	-1.635	-1.496	-0.807	-1.496	-1.635	-1.454
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
213.0	-1.454	-1.626	-1.488	-0.811	-1.488	-1.626	-1.454
215.1	-1.698	-1.838	-1.706	-1.071	-1.706	-1.838	-1.698
217.2	-1.917	-2.029	-1.902	-1.306	-1.902	-2.029	-1.917
219.3	-2.116	-2.201	-2.079	-1.519	-2.079	-2.201	-2.116
221.4	-2.296	-2.356	-2.238	-1.712	-2.238	-2.356	-2.296
223.5	-2.458	-2.496	-2.382	-1.887	-2.382	-2.496	-2.458
225.6	-2.605	-2.622	-2.512	-2.045	-2.512	-2.622	-2.605
227.7	-2.738	-2.736	-2.630	-2.187	-2.630	-2.736	-2.738
229.8	-2.858	-2.841	-2.737	-2.316	-2.737	-2.841	-2.858
231.9	-2.967	-2.935	-2.834	-2.432	-2.834	-2.935	-2.967
234.0	-3.064	-3.021	-2.922	-2.536	-2.922	-3.021	-3.064
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
234.0	-3.064	-3.018	-2.920	-2.538	-2.920	-3.018	-3.064
236.1	-3.153	-3.095	-2.999	-2.633	-2.999	-3.095	-3.153
238.2	-3.233	-3.164	-3.070	-2.718	-3.070	-3.164	-3.233
240.3	-3.306	-3.227	-3.135	-2.796	-3.135	-3.227	-3.306
242.4	-3.371	-3.283	-3.193	-2.866	-3.193	-3.283	-3.371
244.5	-3.429	-3.334	-3.245	-2.929	-3.245	-3.334	-3.429
246.6	-3.482	-3.379	-3.292	-2.986	-3.292	-3.379	-3.482
248.7	-3.530	-3.420	-3.334	-3.037	-3.334	-3.420	-3.530
250.8	-3.573	-3.458	-3.372	-3.083	-3.372	-3.458	-3.573
252.9	-3.612	-3.492	-3.407	-3.124	-3.407	-3.492	-3.612
255.0	-3.646	-3.522	-3.438	-3.161	-3.438	-3.522	-3.646

	0.000	0.000	0.000	0.000	0.000	0.000	0.000
255.0	-3.646	-3.520	-3.437	-3.161	-3.437	-3.520	-3.646
257.1	-3.678	-3.548	-3.465	-3.195	-3.465	-3.548	-3.678
259.2	-3.705	-3.572	-3.490	-3.225	-3.490	-3.572	-3.705
261.3	-3.730	-3.594	-3.512	-3.252	-3.512	-3.594	-3.730
263.4	-3.753	-3.613	-3.532	-3.275	-3.532	-3.613	-3.753
265.5	-3.772	-3.629	-3.549	-3.296	-3.549	-3.629	-3.772
267.6	-3.789	-3.644	-3.564	-3.315	-3.564	-3.644	-3.789
269.7	-3.804	-3.657	-3.578	-3.331	-3.578	-3.657	-3.804
271.8	-3.817	-3.668	-3.589	-3.345	-3.589	-3.668	-3.817
273.9	-3.829	-3.678	-3.599	-3.357	-3.599	-3.678	-3.829
276.0	-3.838	-3.687	-3.608	-3.367	-3.608	-3.687	-3.838
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
276.0	-3.838	-3.686	-3.607	-3.367	-3.607	-3.686	-3.838
278.1	-3.846	-3.693	-3.614	-3.375	-3.614	-3.693	-3.846
280.2	-3.852	-3.698	-3.620	-3.382	-3.620	-3.698	-3.852
282.3	-3.857	-3.702	-3.624	-3.387	-3.624	-3.702	-3.857
284.4	-3.859	-3.705	-3.627	-3.390	-3.627	-3.705	-3.859
286.5	-3.861	-3.706	-3.628	-3.392	-3.628	-3.706	-3.861
288.6	-3.862	-3.707	-3.629	-3.393	-3.629	-3.707	-3.862
290.7	-3.861	-3.706	-3.628	-3.391	-3.628	-3.706	-3.861
292.8	-3.858	-3.704	-3.626	-3.389	-3.626	-3.704	-3.858
294.9	-3.854	-3.700	-3.622	-3.385	-3.622	-3.700	-3.854
297.0	-3.849	-3.696	-3.617	-3.379	-3.617	-3.696	-3.849
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
297.0	-3.849	-3.696	-3.617	-3.379	-3.617	-3.696	-3.849
299.1	-3.842	-3.690	-3.611	-3.371	-3.611	-3.690	-3.842
301.2	-3.833	-3.682	-3.603	-3.362	-3.603	-3.682	-3.833
303.3	-3.823	-3.673	-3.594	-3.351	-3.594	-3.673	-3.823
305.4	-3.811	-3.663	-3.583	-3.338	-3.583	-3.663	-3.811
307.5	-3.797	-3.651	-3.571	-3.323	-3.571	-3.651	-3.797
309.6	-3.780	-3.637	-3.556	-3.305	-3.556	-3.637	-3.780
311.7	-3.762	-3.621	-3.540	-3.286	-3.540	-3.621	-3.762
313.8	-3.742	-3.603	-3.522	-3.264	-3.522	-3.603	-3.742
315.9	-3.718	-3.582	-3.501	-3.238	-3.501	-3.582	-3.718
318.0	-3.692	-3.560	-3.477	-3.210	-3.477	-3.560	-3.692
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
318.0	-3.692	-3.561	-3.478	-3.210	-3.478	-3.561	-3.692
320.1	-3.662	-3.535	-3.451	-3.178	-3.451	-3.535	-3.662
322.2	-3.629	-3.506	-3.422	-3.143	-3.422	-3.506	-3.629
324.3	-3.593	-3.475	-3.390	-3.104	-3.390	-3.475	-3.593
326.4	-3.552	-3.439	-3.353	-3.060	-3.353	-3.439	-3.552

328.5	-3.507	-3.400	-3.313	-3.012	-3.313	-3.400	-3.507
330.6	-3.456	-3.357	-3.268	-2.957	-3.268	-3.357	-3.456
332.7	-3.400	-3.309	-3.219	-2.898	-3.219	-3.309	-3.400
334.8	-3.338	-3.255	-3.164	-2.831	-3.164	-3.255	-3.338
336.9	-3.270	-3.196	-3.103	-2.757	-3.103	-3.196	-3.270
339.0	-3.193	-3.129	-3.034	-2.676	-3.034	-3.129	-3.193
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
339.0	-3.193	-3.132	-3.037	-2.675	-3.037	-3.132	-3.193
341.1	-3.109	-3.058	-2.961	-2.585	-2.961	-3.058	-3.109
343.2	-3.016	-2.977	-2.878	-2.485	-2.878	-2.977	-3.016
345.3	-2.913	-2.887	-2.785	-2.375	-2.785	-2.887	-2.913
347.4	-2.799	-2.789	-2.684	-2.252	-2.684	-2.789	-2.799
349.5	-2.672	-2.680	-2.572	-2.116	-2.572	-2.680	-2.672
351.6	-2.533	-2.560	-2.449	-1.967	-2.449	-2.560	-2.533
353.7	-2.378	-2.426	-2.311	-1.800	-2.311	-2.426	-2.378
355.8	-2.207	-2.279	-2.159	-1.617	-2.159	-2.279	-2.207
357.9	-2.017	-2.114	-1.990	-1.414	-1.990	-2.114	-2.017
360.0	-1.808	-1.932	-1.803	-1.190	-1.803	-1.932	-1.808
					0.000	0.000	0.000
360.0	-1.808	-1.588	-1.676	-2.197	-1.676	-1.588	-1.808
362.0	-1.609	-1.415	-1.498	-1.984	-1.498	-1.415	-1.609
364.0	-1.427	-1.257	-1.336	-1.789	-1.336	-1.257	-1.427
366.0	-1.262	-1.114	-1.189	-1.612	-1.189	-1.114	-1.262
368.0	-1.112	-0.985	-1.056	-1.451	-1.056	-0.985	-1.112
370.0	-0.976	-0.868	-0.935	-1.305	-0.935	-0.868	-0.976
372.0	-0.852	-0.761	-0.826	-1.172	-0.826	-0.761	-0.852
374.0	-0.739	-0.664	-0.725	-1.052	-0.725	-0.664	-0.739
376.0	-0.637	-0.575	-0.634	-0.942	-0.634	-0.575	-0.637
378.0	-0.544	-0.495	-0.552	-0.843	-0.552	-0.495	-0.544
380.0	-0.460	-0.421	-0.475	-0.752	-0.475	-0.421	-0.460
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
380.0	-0.460	-0.423	-0.478	-0.751	-0.478	-0.423	-0.460
382.0	-0.383	-0.356	-0.409	-0.669	-0.409	-0.356	-0.383
384.0	-0.313	-0.296	-0.347	-0.595	-0.347	-0.296	-0.313
386.0	-0.250	-0.241	-0.290	-0.527	-0.290	-0.241	-0.250
388.0	-0.192	-0.192	-0.239	-0.465	-0.239	-0.192	-0.192
390.0	-0.140	-0.146	-0.193	-0.409	-0.193	-0.146	-0.140
392.0	-0.092	-0.106	-0.151	-0.358	-0.151	-0.106	-0.092
394.0	-0.049	-0.068	-0.113	-0.311	-0.113	-0.068	-0.049
396.0	-0.010	-0.034	-0.078	-0.269	-0.078	-0.034	-0.010
398.0	0.026	-0.003	-0.046	-0.231	-0.046	-0.003	0.026
400.0	0.058	0.025	-0.017	-0.196	-0.017	0.025	0.058

	0.000	0.000	0.000	0.000	0.000	0.000	0.000
400.0	0.058	0.024	-0.017	-0.196	-0.017	0.024	0.058
402.0	0.088	0.050	0.009	-0.165	0.009	0.050	0.088
404.0	0.114	0.073	0.033	-0.136	0.033	0.073	0.114
406.0	0.139	0.094	0.055	-0.110	0.055	0.094	0.139
408.0	0.161	0.113	0.074	-0.086	0.074	0.113	0.161
410.0	0.181	0.130	0.092	-0.064	0.092	0.130	0.181
412.0	0.199	0.146	0.108	-0.045	0.108	0.146	0.199
414.0	0.216	0.160	0.123	-0.027	0.123	0.160	0.216
416.0	0.231	0.173	0.136	-0.011	0.136	0.173	0.231
418.0	0.244	0.185	0.148	0.004	0.148	0.185	0.244
420.0	0.257	0.196	0.160	0.017	0.160	0.196	0.257
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
420.0	0.257	0.196	0.159	0.017	0.159	0.196	0.257
422.0	0.268	0.206	0.169	0.029	0.169	0.206	0.268
424.0	0.278	0.215	0.179	0.040	0.179	0.215	0.278
426.0	0.288	0.223	0.187	0.050	0.187	0.223	0.288
428.0	0.296	0.230	0.194	0.059	0.194	0.230	0.296
430.0	0.304	0.236	0.201	0.067	0.201	0.236	0.304
432.0	0.311	0.242	0.207	0.075	0.207	0.242	0.311
434.0	0.317	0.248	0.213	0.082	0.213	0.248	0.317
436.0	0.323	0.253	0.218	0.088	0.218	0.253	0.323
438.0	0.328	0.257	0.223	0.093	0.223	0.257	0.328
440.0	0.333	0.262	0.227	0.098	0.227	0.262	0.333
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
440.0	0.333	0.261	0.227	0.098	0.227	0.261	0.333
442.0	0.337	0.265	0.230	0.103	0.230	0.265	0.337
444.0	0.341	0.268	0.234	0.107	0.234	0.268	0.341
446.0	0.344	0.271	0.237	0.111	0.237	0.271	0.344
448.0	0.347	0.274	0.240	0.114	0.240	0.274	0.347
450.0	0.350	0.277	0.242	0.117	0.242	0.277	0.350
452.0	0.353	0.279	0.244	0.120	0.244	0.279	0.353
454.0	0.355	0.281	0.246	0.122	0.246	0.281	0.355
456.0	0.357	0.282	0.248	0.124	0.248	0.282	0.357
458.0	0.359	0.284	0.250	0.126	0.250	0.284	0.359
460.0	0.360	0.285	0.251	0.128	0.251	0.285	0.360
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
460.0	0.360	0.285	0.251	0.128	0.251	0.285	0.360
462.0	0.362	0.286	0.253	0.130	0.253	0.286	0.362
464.0	0.363	0.288	0.254	0.131	0.254	0.288	0.363
466.0	0.364	0.288	0.255	0.132	0.255	0.288	0.364
468.0	0.365	0.289	0.255	0.133	0.255	0.289	0.365

470.0	0.366	0.290	0.256	0.134	0.256	0.290	0.366
472.0	0.366	0.290	0.257	0.134	0.257	0.290	0.366
474.0	0.367	0.291	0.257	0.135	0.257	0.291	0.367
476.0	0.367	0.291	0.257	0.135	0.257	0.291	0.367
478.0	0.367	0.291	0.257	0.135	0.257	0.291	0.367
480.0	0.367	0.291	0.258	0.136	0.258	0.291	0.367

Total Shear with Negative St. Vt. Contribution

z	Combined Shear Stress with Negative St. Vt.						
0.0	0.901	1.031	1.090	1.330	1.090	1.031	0.901
2.0	0.901	1.031	1.089	1.329	1.089	1.031	0.901
4.0	0.900	1.030	1.089	1.329	1.089	1.030	0.900
6.0	0.900	1.029	1.088	1.328	1.088	1.029	0.900
8.0	0.898	1.028	1.087	1.327	1.087	1.028	0.898
10.0	0.897	1.027	1.085	1.326	1.085	1.027	0.897
12.0	0.895	1.024	1.083	1.324	1.083	1.024	0.895
14.0	0.893	1.022	1.081	1.322	1.081	1.022	0.893
16.0	0.890	1.019	1.078	1.319	1.078	1.019	0.890
18.0	0.887	1.016	1.075	1.317	1.075	1.016	0.887
20.0	0.884	1.012	1.071	1.313	1.071	1.012	0.884
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
20.0	0.884	1.011	1.071	1.313	1.071	1.011	0.884
22.0	0.880	1.008	1.067	1.310	1.067	1.008	0.880
24.0	0.876	1.002	1.062	1.305	1.062	1.002	0.876
26.0	0.871	0.996	1.056	1.300	1.056	0.996	0.871
28.0	0.865	0.990	1.050	1.295	1.050	0.990	0.865
30.0	0.859	0.983	1.043	1.289	1.043	0.983	0.859
32.0	0.852	0.975	1.035	1.282	1.035	0.975	0.852
34.0	0.844	0.967	1.027	1.275	1.027	0.967	0.844
36.0	0.836	0.957	1.017	1.267	1.017	0.957	0.836
38.0	0.826	0.946	1.007	1.258	1.007	0.946	0.826
40.0	0.816	0.934	0.995	1.248	0.995	0.934	0.816
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
40.0	0.816	0.934	0.995	1.248	0.995	0.934	0.816
42.0	0.805	0.922	0.983	1.237	0.983	0.922	0.805
44.0	0.792	0.907	0.969	1.225	0.969	0.907	0.792
46.0	0.778	0.891	0.953	1.211	0.953	0.891	0.778
48.0	0.762	0.873	0.936	1.196	0.936	0.873	0.762
50.0	0.746	0.854	0.917	1.180	0.917	0.854	0.746
52.0	0.726	0.832	0.895	1.161	0.895	0.832	0.726
54.0	0.706	0.809	0.873	1.142	0.873	0.809	0.706
56.0	0.683	0.783	0.847	1.119	0.847	0.783	0.683
58.0	0.658	0.755	0.819	1.095	0.819	0.755	0.658

60.0	0.630	0.723	0.789	1.068	0.789	0.723	0.630
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
60.0	0.630	0.722	0.788	1.069	0.788	0.722	0.630
62.0	0.600	0.688	0.754	1.040	0.754	0.688	0.600
64.0	0.566	0.650	0.717	1.007	0.717	0.650	0.566
66.0	0.529	0.608	0.676	0.972	0.676	0.608	0.529
68.0	0.489	0.562	0.631	0.932	0.631	0.562	0.489
70.0	0.444	0.511	0.581	0.889	0.581	0.511	0.444
72.0	0.395	0.455	0.526	0.842	0.526	0.455	0.395
74.0	0.340	0.393	0.466	0.790	0.466	0.393	0.340
76.0	0.281	0.326	0.400	0.732	0.400	0.326	0.281
78.0	0.215	0.251	0.327	0.669	0.327	0.251	0.215
80.0	0.143	0.170	0.247	0.599	0.247	0.170	0.143
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
80.0	0.143	0.167	0.245	0.600	0.245	0.167	0.143
82.0	0.063	0.078	0.158	0.523	0.158	0.078	0.063
84.0	-0.024	-0.021	0.061	0.438	0.061	-0.021	-0.024
86.0	-0.121	-0.130	-0.046	0.345	-0.046	-0.130	-0.121
88.0	-0.227	-0.251	-0.164	0.243	-0.164	-0.251	-0.227
90.0	-0.343	-0.384	-0.294	0.131	-0.294	-0.384	-0.343
92.0	-0.472	-0.530	-0.437	0.008	-0.437	-0.530	-0.472
94.0	-0.613	-0.691	-0.594	-0.129	-0.594	-0.691	-0.613
96.0	-0.768	-0.867	-0.767	-0.278	-0.767	-0.867	-0.768
98.0	-0.940	-1.061	-0.957	-0.443	-0.957	-1.061	-0.940
100.0	-1.127	-1.273	-1.164	-0.625	-1.164	-1.273	-1.127
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
100.0	-1.127	-1.280	-1.170	-0.622	-1.170	-1.280	-1.127
102.0	-1.335	-1.513	-1.398	-0.823	-1.398	-1.513	-1.335
104.0	-1.563	-1.770	-1.650	-1.043	-1.650	-1.770	-1.563
106.0	-1.814	-2.055	-1.928	-1.285	-1.928	-2.055	-1.814
108.0	-2.090	-2.369	-2.236	-1.551	-2.236	-2.369	-2.090
110.0	-2.393	-2.715	-2.573	-1.843	-2.573	-2.715	-2.393
112.0	-2.727	-3.095	-2.946	-2.165	-2.946	-3.095	-2.727
114.0	-3.095	-3.514	-3.355	-2.519	-3.355	-3.514	-3.095
116.0	-3.500	-3.974	-3.805	-2.909	-3.805	-3.974	-3.500
118.0	-3.945	-4.478	-4.298	-3.339	-4.298	-4.478	-3.945
120.0	-4.436	-5.032	-4.840	-3.813	-4.840	-5.032	-4.436
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
120.0	-4.436	-4.142	-4.503	-6.361	-4.503	-4.142	-4.436
121.8	-4.877	-4.641	-4.991	-6.787	-4.991	-4.641	-4.877
123.6	-5.279	-5.097	-5.437	-7.174	-5.437	-5.097	-5.279
125.4	-5.644	-5.511	-5.842	-7.525	-5.842	-5.511	-5.644
127.2	-5.974	-5.887	-6.210	-7.844	-6.210	-5.887	-5.974
129.0	-6.271	-6.225	-6.540	-8.129	-6.540	-6.225	-6.271



130.8	-6.538	-6.530	-6.838	-8.386	-6.838	-6.530	-6.538
132.6	-6.782	-6.807	-7.109	-8.621	-7.109	-6.807	-6.782
134.4	-6.997	-7.051	-7.348	-8.829	-7.348	-7.051	-6.997
136.2	-7.189	-7.268	-7.560	-9.014	-7.560	-7.268	-7.189
138.0	-7.357	-7.457	-7.746	-9.177	-7.746	-7.457	-7.357
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
138.0	-7.357	-7.461	-7.749	-9.175	-7.749	-7.461	-7.357
139.8	-7.503	-7.626	-7.910	-9.316	-7.910	-7.626	-7.503
141.6	-7.631	-7.771	-8.052	-9.439	-8.052	-7.771	-7.631
143.4	-7.735	-7.890	-8.168	-9.540	-8.168	-7.890	-7.735
145.2	-7.828	-7.995	-8.271	-9.629	-8.271	-7.995	-7.828
147.0	-7.898	-8.075	-8.349	-9.696	-8.349	-8.075	-7.898
148.8	-7.950	-8.134	-8.407	-9.747	-8.407	-8.134	-7.950
150.6	-7.985	-8.174	-8.446	-9.780	-8.446	-8.174	-7.985
152.4	-8.003	-8.194	-8.465	-9.797	-8.465	-8.194	-8.003
154.2	-8.003	-8.194	-8.465	-9.797	-8.465	-8.194	-8.003
156.0	-7.985	-8.174	-8.446	-9.780	-8.446	-8.174	-7.985
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
156.0	-7.985	-8.173	-8.445	-9.781	-8.445	-8.173	-7.985
157.8	-7.956	-8.141	-8.413	-9.752	-8.413	-8.141	-7.956
159.6	-7.904	-8.082	-8.356	-9.702	-8.356	-8.082	-7.904
161.4	-7.840	-8.009	-8.285	-9.640	-8.285	-8.009	-7.840
163.2	-7.753	-7.910	-8.188	-9.556	-8.188	-7.910	-7.753
165.0	-7.654	-7.797	-8.078	-9.461	-8.078	-7.797	-7.654
166.8	-7.532	-7.658	-7.942	-9.344	-7.942	-7.658	-7.532
168.6	-7.387	-7.493	-7.780	-9.204	-7.780	-7.493	-7.387
170.4	-7.224	-7.308	-7.599	-9.047	-7.599	-7.308	-7.224
172.2	-7.038	-7.097	-7.393	-8.868	-7.393	-7.097	-7.038
174.0	-6.823	-6.854	-7.155	-8.660	-7.155	-6.854	-6.823
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
174.0	-6.823	-6.849	-7.150	-8.662	-7.150	-6.849	-6.823
175.8	-6.590	-6.587	-6.894	-8.437	-6.894	-6.587	-6.590
177.6	-6.323	-6.285	-6.598	-8.179	-6.598	-6.285	-6.323
179.4	-6.032	-5.955	-6.276	-7.899	-6.276	-5.955	-6.032
181.2	-5.709	-5.587	-5.916	-7.588	-5.916	-5.587	-5.709
183.0	-5.351	-5.179	-5.517	-7.243	-5.517	-5.179	-5.351
184.8	-4.957	-4.730	-5.078	-6.864	-5.078	-4.730	-4.957
186.6	-4.523	-4.237	-4.596	-6.446	-4.596	-4.237	-4.523
188.4	-4.047	-3.696	-4.066	-5.987	-4.066	-3.696	-4.047
190.2	-3.524	-3.103	-3.487	-5.483	-3.487	-3.103	-3.524
192.0	-2.952	-2.457	-2.854	-4.931	-2.854	-2.457	-2.952
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
192.0	-2.952	-3.723	-3.476	-2.096	-3.476	-3.723	-2.952
194.1	-2.287	-2.973	-2.741	-1.453	-2.741	-2.973	-2.287

196.2	-1.687	-2.292	-2.075	-0.874	-2.075	-2.292	-1.687
198.3	-1.143	-1.675	-1.471	-0.349	-1.471	-1.675	-1.143
200.4	-0.651	-1.115	-0.924	0.124	-0.924	-1.115	-0.651
202.5	-0.207	-0.609	-0.429	0.551	-0.429	-0.609	-0.207
204.6	0.195	-0.151	0.019	0.938	0.019	-0.151	0.195
206.7	0.559	0.263	0.424	1.288	0.424	0.263	0.559
208.8	0.888	0.636	0.789	1.605	0.789	0.636	0.888
210.9	1.186	0.972	1.118	1.893	1.118	0.972	1.186
213.0	1.454	1.273	1.413	2.153	1.413	1.273	1.454
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
213.0	1.454	1.282	1.420	2.150	1.420	1.282	1.454
215.1	1.698	1.557	1.689	2.385	1.689	1.557	1.698
217.2	1.917	1.806	1.933	2.597	1.933	1.806	1.917
219.3	2.116	2.031	2.153	2.789	2.153	2.031	2.116
221.4	2.296	2.236	2.353	2.961	2.353	2.236	2.296
223.5	2.458	2.421	2.534	3.118	2.534	2.421	2.458
225.6	2.605	2.588	2.698	3.260	2.698	2.588	2.605
227.7	2.738	2.739	2.846	3.387	2.846	2.739	2.738
229.8	2.858	2.876	2.979	3.503	2.979	2.876	2.858
231.9	2.967	2.998	3.099	3.608	3.099	2.998	2.967
234.0	3.064	3.108	3.206	3.703	3.206	3.108	3.064
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
234.0	3.064	3.111	3.209	3.701	3.209	3.111	3.064
236.1	3.153	3.211	3.307	3.787	3.307	3.211	3.153
238.2	3.233	3.302	3.396	3.864	3.396	3.302	3.233
240.3	3.306	3.384	3.476	3.934	3.476	3.384	3.306
242.4	3.371	3.458	3.549	3.997	3.549	3.458	3.371
244.5	3.429	3.525	3.614	4.053	3.614	3.525	3.429
246.6	3.482	3.585	3.673	4.104	3.673	3.585	3.482
248.7	3.530	3.640	3.726	4.150	3.726	3.640	3.530
250.8	3.573	3.688	3.774	4.192	3.774	3.688	3.573
252.9	3.612	3.732	3.817	4.229	3.817	3.732	3.612
255.0	3.646	3.771	3.855	4.262	3.855	3.771	3.646
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
255.0	3.646	3.772	3.856	4.262	3.856	3.772	3.646
257.1	3.678	3.807	3.890	4.292	3.890	3.807	3.678
259.2	3.705	3.839	3.921	4.319	3.921	3.839	3.705
261.3	3.730	3.867	3.949	4.343	3.949	3.867	3.730
263.4	3.753	3.892	3.973	4.365	3.973	3.892	3.753
265.5	3.772	3.914	3.995	4.383	3.995	3.914	3.772
267.6	3.789	3.934	4.014	4.400	4.014	3.934	3.789
269.7	3.804	3.951	4.031	4.414	4.031	3.951	3.804
271.8	3.817	3.966	4.045	4.427	4.045	3.966	3.817
273.9	3.829	3.979	4.058	4.438	4.058	3.979	3.829

276.0	3.838	3.989	4.068	4.447	4.068	3.989	3.838
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
276.0	3.838	3.990	4.069	4.447	4.069	3.990	3.838
278.1	3.846	3.998	4.077	4.454	4.077	3.998	3.846
280.2	3.852	4.005	4.084	4.460	4.084	4.005	3.852
282.3	3.857	4.011	4.089	4.465	4.089	4.011	3.857
284.4	3.859	4.014	4.092	4.468	4.092	4.014	3.859
286.5	3.861	4.016	4.094	4.469	4.094	4.016	3.861
288.6	3.862	4.017	4.095	4.470	4.095	4.017	3.862
290.7	3.861	4.015	4.094	4.469	4.094	4.015	3.861
292.8	3.858	4.013	4.091	4.466	4.091	4.013	3.858
294.9	3.854	4.008	4.087	4.462	4.087	4.008	3.854
297.0	3.849	4.002	4.081	4.457	4.081	4.002	3.849
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
297.0	3.849	4.002	4.081	4.458	4.081	4.002	3.849
299.1	3.842	3.994	4.073	4.451	4.073	3.994	3.842
301.2	3.833	3.984	4.063	4.442	4.063	3.984	3.833
303.3	3.823	3.973	4.052	4.432	4.052	3.973	3.823
305.4	3.811	3.959	4.038	4.420	4.038	3.959	3.811
307.5	3.797	3.943	4.023	4.407	4.023	3.943	3.797
309.6	3.780	3.924	4.005	4.391	4.005	3.924	3.780
311.7	3.762	3.904	3.984	4.374	3.984	3.904	3.762
313.8	3.742	3.880	3.961	4.354	3.961	3.880	3.742
315.9	3.718	3.853	3.935	4.331	3.935	3.853	3.718
318.0	3.692	3.823	3.906	4.306	3.906	3.823	3.692
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
318.0	3.692	3.822	3.905	4.306	3.905	3.822	3.692
320.1	3.662	3.789	3.873	4.277	3.873	3.789	3.662
322.2	3.629	3.752	3.837	4.246	3.837	3.752	3.629
324.3	3.593	3.711	3.796	4.211	3.796	3.711	3.593
326.4	3.552	3.665	3.751	4.171	3.751	3.665	3.552
328.5	3.507	3.613	3.700	4.128	3.700	3.613	3.507
330.6	3.456	3.555	3.644	4.079	3.644	3.555	3.456
332.7	3.400	3.492	3.582	4.025	3.582	3.492	3.400
334.8	3.338	3.421	3.513	3.965	3.513	3.421	3.338
336.9	3.270	3.344	3.437	3.899	3.437	3.344	3.270
339.0	3.193	3.258	3.353	3.826	3.353	3.258	3.193
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
339.0	3.193	3.255	3.350	3.827	3.350	3.255	3.193
341.1	3.109	3.160	3.258	3.745	3.258	3.160	3.109
343.2	3.016	3.055	3.155	3.655	3.155	3.055	3.016
345.3	2.913	2.938	3.040	3.556	3.040	2.938	2.913
347.4	2.799	2.808	2.913	3.446	2.913	2.808	2.799
349.5	2.672	2.664	2.772	3.324	2.772	2.664	2.672

351.6	2.533	2.505	2.617	3.190	2.617	2.505	2.533
353.7	2.378	2.329	2.444	3.040	2.444	2.329	2.378
355.8	2.207	2.135	2.254	2.876	2.254	2.135	2.207
357.9	2.017	1.920	2.045	2.693	2.045	1.920	2.017
360.0	1.808	1.684	1.813	2.491	1.813	1.684	1.808
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
360.0	1.808	2.028	1.940	1.484	1.940	2.028	1.808
362.0	1.609	1.803	1.719	1.291	1.719	1.803	1.609
364.0	1.427	1.597	1.518	1.115	1.518	1.597	1.427
366.0	1.262	1.409	1.334	0.956	1.334	1.409	1.262
368.0	1.112	1.239	1.168	0.812	1.168	1.239	1.112
370.0	0.976	1.084	1.016	0.681	1.016	1.084	0.976
372.0	0.852	0.943	0.878	0.562	0.878	0.943	0.852
374.0	0.739	0.814	0.753	0.454	0.753	0.814	0.739
376.0	0.637	0.698	0.639	0.355	0.639	0.698	0.637
378.0	0.544	0.594	0.537	0.265	0.537	0.594	0.544
380.0	0.460	0.499	0.444	0.183	0.444	0.499	0.460
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
380.0	0.460	0.496	0.442	0.184	0.442	0.496	0.460
382.0	0.383	0.410	0.357	0.110	0.357	0.410	0.383
384.0	0.313	0.330	0.280	0.043	0.280	0.330	0.313
386.0	0.250	0.258	0.209	-0.018	0.209	0.258	0.250
388.0	0.192	0.193	0.145	-0.073	0.145	0.193	0.192
390.0	0.140	0.133	0.087	-0.124	0.087	0.133	0.140
392.0	0.092	0.079	0.034	-0.170	0.034	0.079	0.092
394.0	0.049	0.030	-0.014	-0.211	-0.014	0.030	0.049
396.0	0.010	-0.014	-0.058	-0.249	-0.058	-0.014	0.010
398.0	-0.026	-0.055	-0.097	-0.283	-0.097	-0.055	-0.026
400.0	-0.058	-0.091	-0.133	-0.315	-0.133	-0.091	-0.058
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
400.0	-0.058	-0.092	-0.134	-0.314	-0.134	-0.092	-0.058
402.0	-0.088	-0.125	-0.166	-0.343	-0.166	-0.125	-0.088
404.0	-0.114	-0.156	-0.196	-0.369	-0.196	-0.156	-0.114
406.0	-0.139	-0.183	-0.223	-0.392	-0.223	-0.183	-0.139
408.0	-0.161	-0.209	-0.248	-0.414	-0.248	-0.209	-0.161
410.0	-0.181	-0.231	-0.270	-0.433	-0.270	-0.231	-0.181
412.0	-0.199	-0.252	-0.290	-0.450	-0.290	-0.252	-0.199
414.0	-0.216	-0.271	-0.309	-0.466	-0.309	-0.271	-0.216
416.0	-0.231	-0.288	-0.325	-0.481	-0.325	-0.288	-0.231
418.0	-0.244	-0.304	-0.341	-0.494	-0.341	-0.304	-0.244
420.0	-0.257	-0.318	-0.354	-0.506	-0.354	-0.318	-0.257
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
420.0	-0.257	-0.318	-0.355	-0.506	-0.355	-0.318	-0.257
422.0	-0.268	-0.331	-0.367	-0.517	-0.367	-0.331	-0.268

424.0	-0.278	-0.342	-0.378	-0.527	-0.378	-0.342	-0.278
426.0	-0.288	-0.353	-0.389	-0.536	-0.389	-0.353	-0.288
428.0	-0.296	-0.362	-0.398	-0.544	-0.398	-0.362	-0.296
430.0	-0.304	-0.371	-0.407	-0.551	-0.407	-0.371	-0.304
432.0	-0.311	-0.379	-0.414	-0.558	-0.414	-0.379	-0.311
434.0	-0.317	-0.386	-0.421	-0.564	-0.421	-0.386	-0.317
436.0	-0.323	-0.393	-0.428	-0.569	-0.428	-0.393	-0.323
438.0	-0.328	-0.399	-0.433	-0.574	-0.433	-0.399	-0.328
440.0	-0.333	-0.404	-0.439	-0.579	-0.439	-0.404	-0.333
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
440.0	-0.333	-0.404	-0.439	-0.579	-0.439	-0.404	-0.333
442.0	-0.337	-0.409	-0.443	-0.583	-0.443	-0.409	-0.337
444.0	-0.341	-0.413	-0.448	-0.587	-0.448	-0.413	-0.341
446.0	-0.344	-0.417	-0.451	-0.590	-0.451	-0.417	-0.344
448.0	-0.347	-0.421	-0.455	-0.593	-0.455	-0.421	-0.347
450.0	-0.350	-0.424	-0.458	-0.596	-0.458	-0.424	-0.350
452.0	-0.353	-0.427	-0.461	-0.598	-0.461	-0.427	-0.353
454.0	-0.355	-0.429	-0.463	-0.600	-0.463	-0.429	-0.355
456.0	-0.357	-0.432	-0.466	-0.602	-0.466	-0.432	-0.357
458.0	-0.359	-0.434	-0.468	-0.604	-0.468	-0.434	-0.359
460.0	-0.360	-0.435	-0.469	-0.606	-0.469	-0.435	-0.360
	0.000	0.000	0.000	0.000	0.000	0.000	0.000
460.0	-0.360	-0.435	-0.469	-0.606	-0.469	-0.435	-0.360
462.0	-0.362	-0.437	-0.471	-0.607	-0.471	-0.437	-0.362
464.0	-0.363	-0.438	-0.472	-0.608	-0.472	-0.438	-0.363
466.0	-0.364	-0.440	-0.474	-0.609	-0.474	-0.440	-0.364
468.0	-0.365	-0.441	-0.475	-0.610	-0.475	-0.441	-0.365
470.0	-0.366	-0.441	-0.475	-0.611	-0.475	-0.441	-0.366
472.0	-0.366	-0.442	-0.476	-0.611	-0.476	-0.442	-0.366
474.0	-0.367	-0.443	-0.477	-0.612	-0.477	-0.443	-0.367
476.0	-0.367	-0.443	-0.477	-0.612	-0.477	-0.443	-0.367
478.0	-0.367	-0.443	-0.477	-0.612	-0.477	-0.443	-0.367
480.0	-0.367	-0.443	-0.477	-0.612	-0.477	-0.443	-0.367

## APPENDIX H

### MEDWADOWSKI'S TORSION PROBLEM

In the EBC of case study four, this material corresponds to the input model, input data, input forms, figures of output data with checks in notepad version, and excel processed output data and charts.

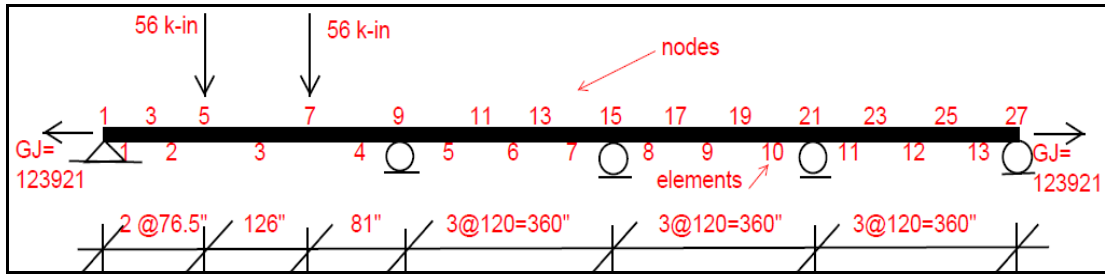
Charts are presented containing both partial and combined stresses along interest points of the beam and cross section profile. Again, the asymptotic behavior of the thin-walled beam elastic line is successfully shown in the charts, which evidences the efficiency of the high order finite element used by BMTORSWP.

Positive shear and axial stresses are assumed similar to those occurring at the cross section flanges and web when the beam undergoes bending as shown below.

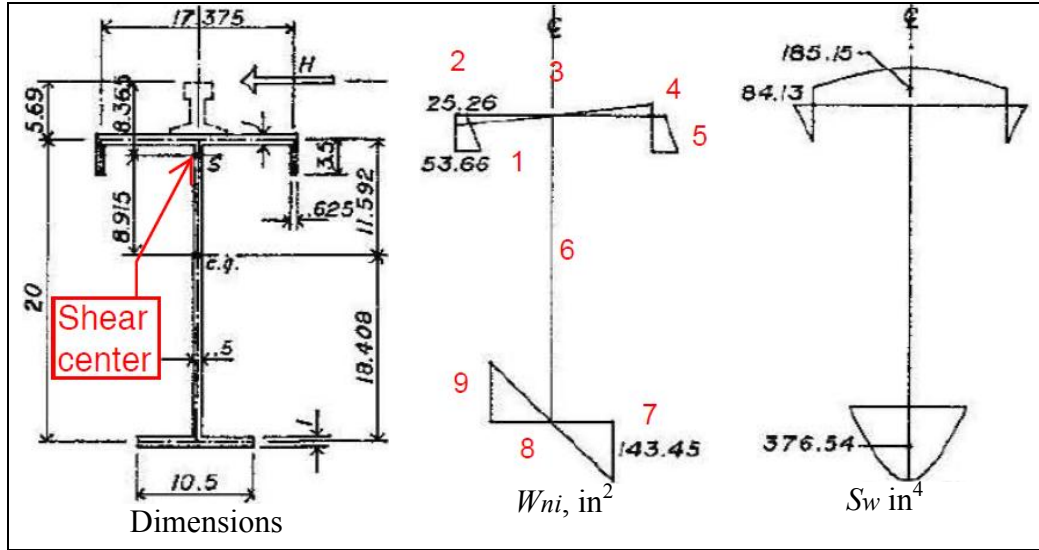
There is a match between the silhouette of BMTORSWP and Medwadowski charts. Nevertheless, Medwadowski committed a huge error in the scale of torques. He found torques 100 times larger than expected, which in terms of stresses is not acceptable at all. The reasoning to support this statement is as follows:

Given the hyperbolic nature of the functions involve in the angle of twist  $\theta(z)$ , each of its derivatives decreases an order of magnitude equal to the characteristic length "a". On the other hand, the ratio between bimoment  $B (= - ECw\theta''')$ , ksi-in<sup>2</sup>) and warping moment  $T_w (= - ECw\theta''')$ , ksi-in) should be in the range of "a" ( $\sqrt{ECw/ GJ} = 138.5$  in).

Therefore, using Medwadowski own bimoment data, the warping moment  $T_w$  is expected to be around  $(4000+3000)$ ks-in<sup>2</sup>/138.50in  $\sim 50$  k-in; as shown in the BMTORSWP chart.



EBC Model for BMTORSWP



Section Properties and Profile Interest Points "s"

Mdw

ASCE-JSD, V. 111, N. 2, Feb-85, p. 453, Warp. M. Distribution, Medwadowski

13	27	5	0	9	0	2	0	29000.	E,ksi	-123921.										
1	1	2	3	81999.	1.	76.50				-123921.										
2	3	4	5	81999.	1.	76.50				-123921.										
3	5	6	7	81999.	1.	126.0				-123921.										
4	7	8	9	81999.	1.	81.00				-123921.										
5	9	10	11	81999.	1.	120.0				-123921.										
6	11	12	13	81999.	1.	120.0				-123921.										
7	13	14	15	81999.	1.	120.0				-123921.										
8	15	16	17	81999.	1.	120.0				-123921.										
9	17	18	19	81999.	1.	120.0				-123921.										
10	19	20	21	81999.	1.	120.0				-123921.										
11	21	22	23	81999.	1.	120.0				-123921.										
12	23	24	25	81999.	1.	120.0				-123921.										
13	25	26	27	81999.	1.	120.0				-123921.										
5	1	1	0	9	0	1	0	15	0	1	0	21	0	1	0	27	0	1	0	
2	5	56.		7	56.															

applied loads, kip

GJ, kip-in<sup>2</sup>  
analog EBC  
tensile load

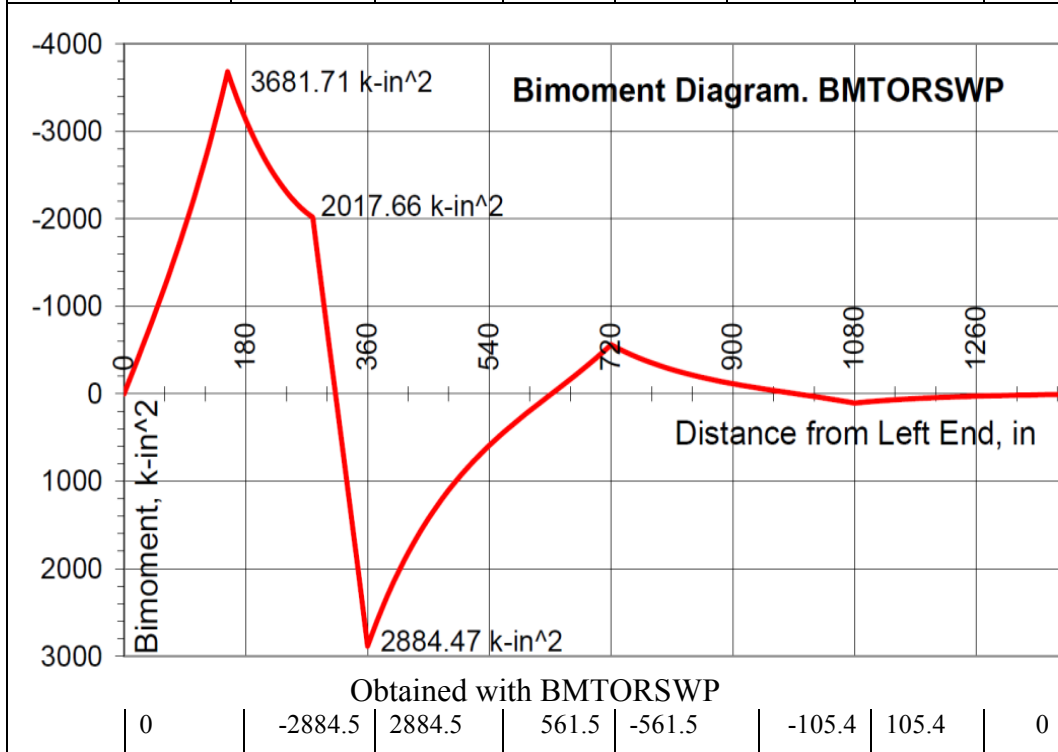
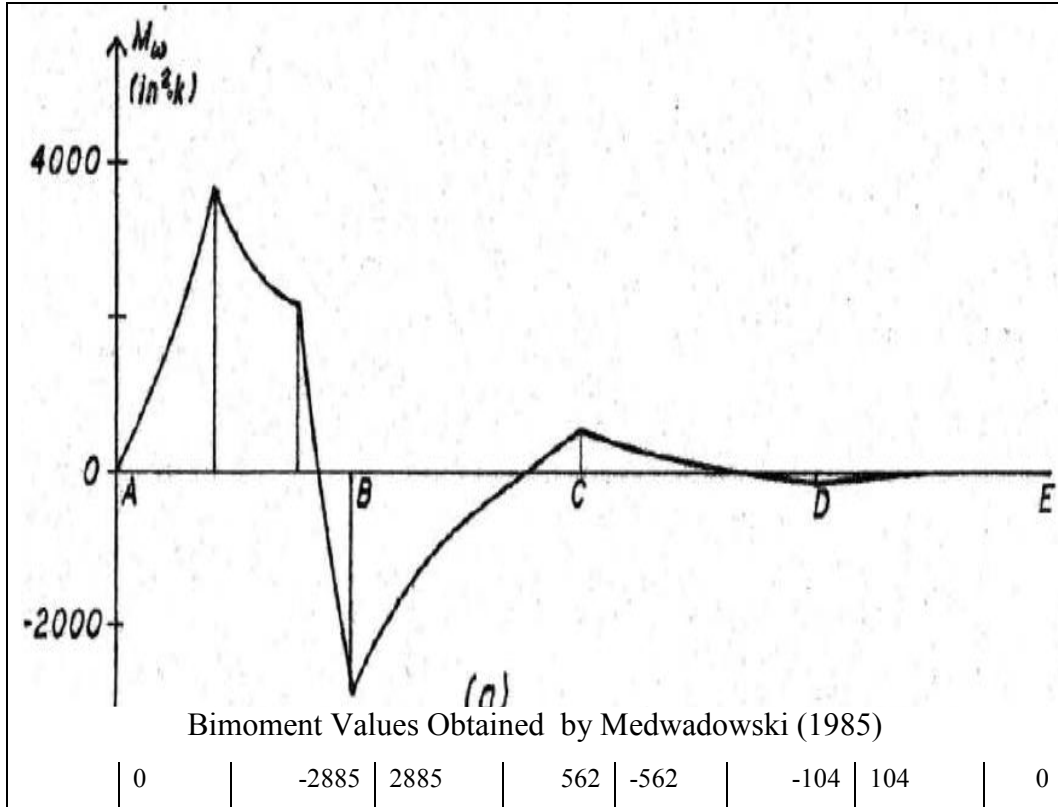
Input Notepad for Medwadowski Problem

PROGRAM BMTORSW INPUT FORM																																								
ALPHAMERIC DESCRIPTION OF THE JOB																														FILE NAME:										
ASCE JSD, Vol. 111, No. 2, Feb-85, p. 453, Warpi																																								
NEL	NOD	NSUP	NSPD	JBW	NFX	NFY	NFZ	ELASTICITY (E11.4)																																
13	27	5	0	9	0	2	0	29000.																																
NE	NI	NC	NJ	Cw			AREA	LENGTH	WTANG	SOIL	NI	SOI																												
1	1	2	3	81	98	9.	1.	76.5																																
2	3	4	5	"	"	"	"	76.5																																
3	5	6	7	"	"	"	"	126.																																
4	7	8	9	"	"	"	"	81.																																
5	9	10	11	"	"	"	"	120.																																
6	11	12	13	"	"	"	"	"																																
7	13	14	15	"	"	"	"	"																																
8	15	16	17	"	"	"	"	"																																
9	17	18	19	"	"	"	"	"																																
10	19	20	21	"	"	"	"	"																																
11	21	22	23	"	"	"	"	"																																
12	23	24	25	"	"	"	"	"																																
13	25	26	27	"	"	"	"	"																																
NF	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ	N	KX	KY	KZ																				
5	1	1	1	0	9	0	1	0	15	0	1	0	21	0	1	0	27	0	1	0																				
NF	N	KD	SPRING						N	KD	SPRING						N	KD																						
NF	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE	N	LOAD VALUE																												
2	5	56.	7	56.																																				
NF	ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATIONAL SPRING MODULUS				ELEM	END	ROTATION																									

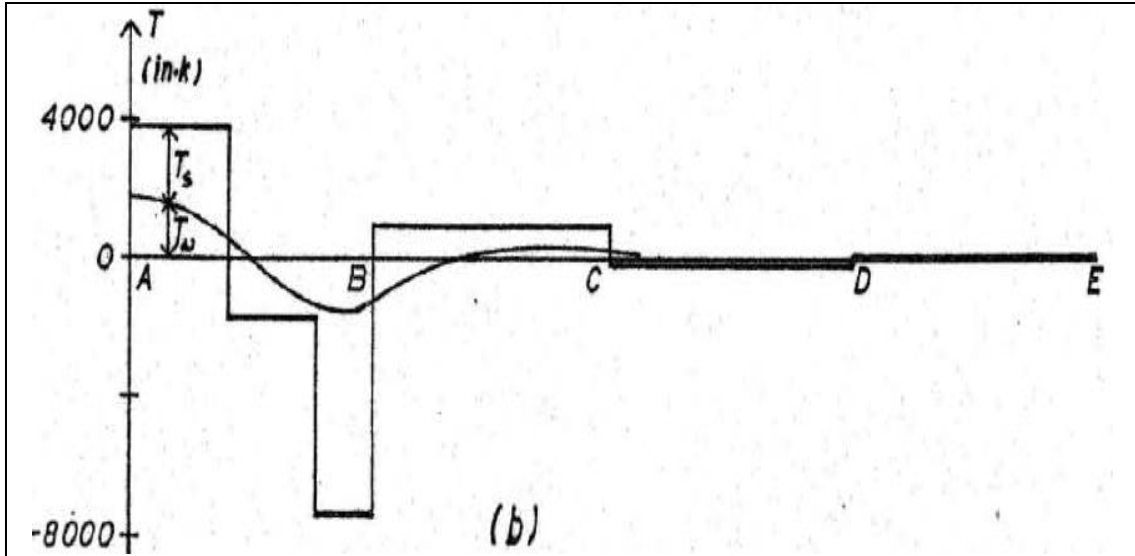
Input Form Left Side for Medwadowski Problem



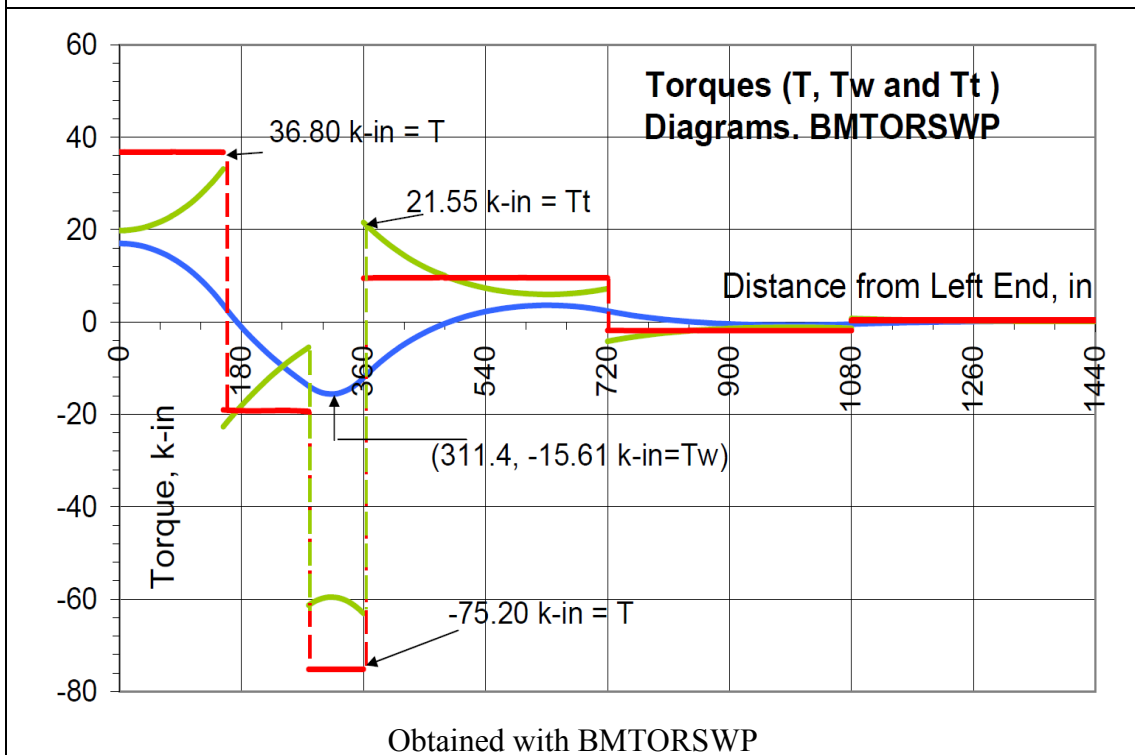




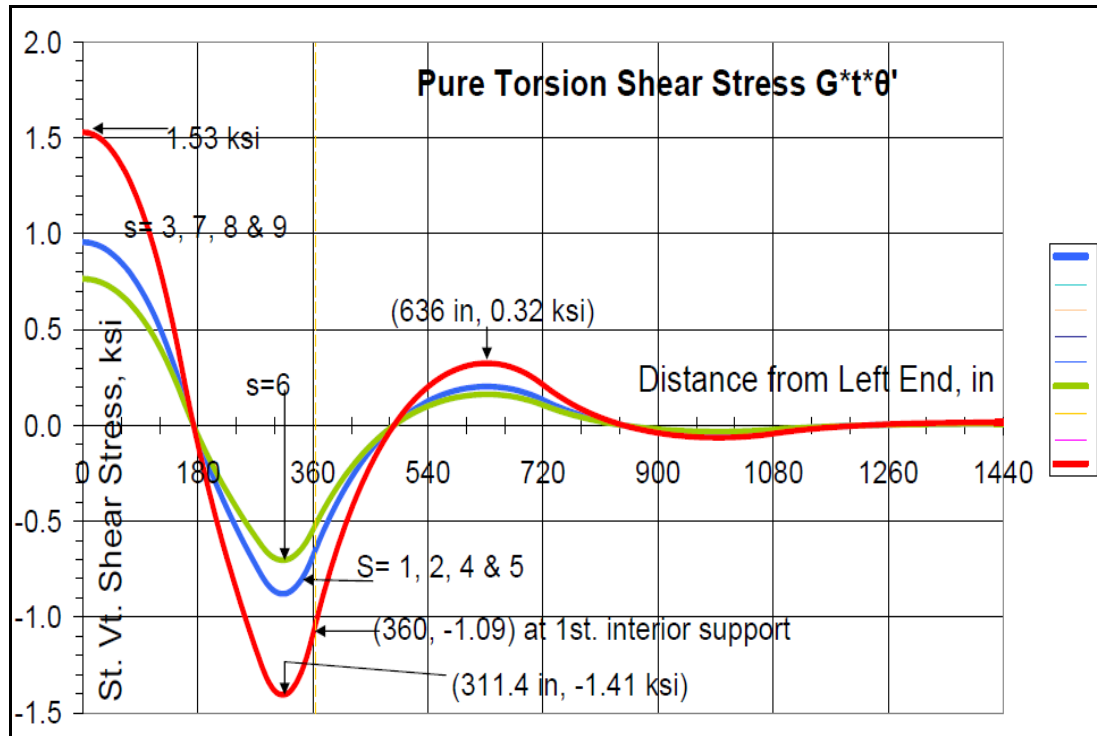
Comparison of Bimoment Charts by Medwadowski and with BMTORSWP



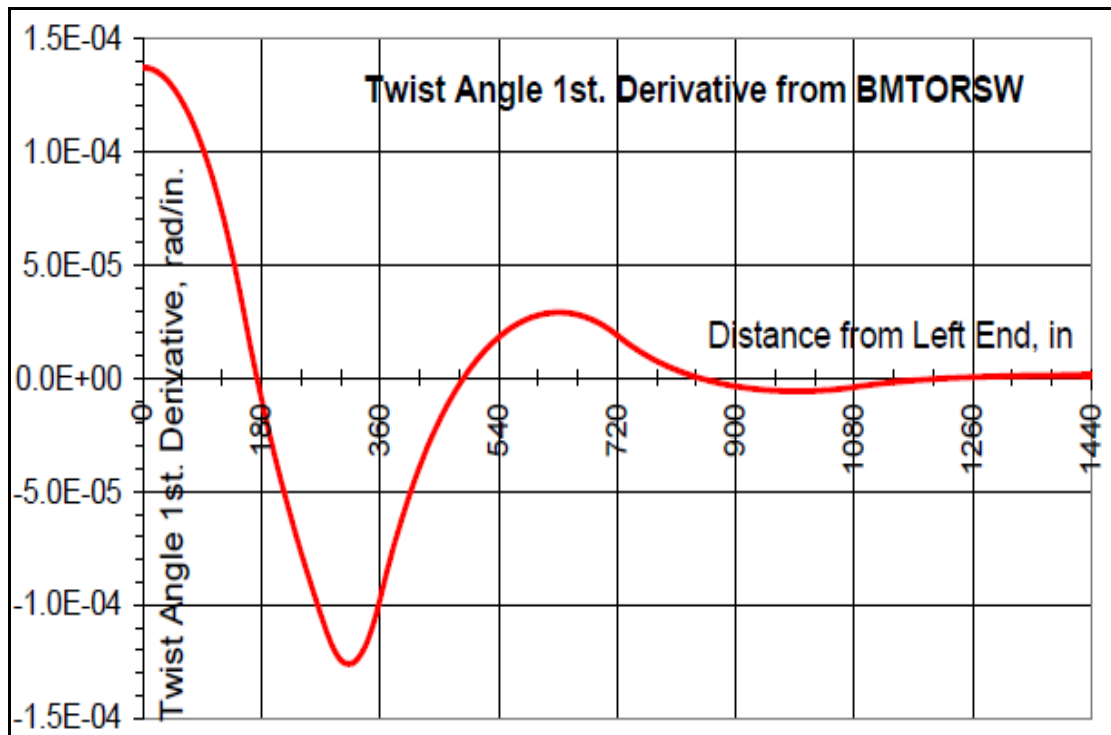
Medwadowski's with Scale Error. Given the hyperbolic nature of the functions involve in  $\theta(z)$ , each of its derivatives decreases an order of magnitude equal to the characteristic length "a". Thus  $B = -ECw\theta''$  should be around "a" ( $\sqrt{ECw/GJ} = 138.5$ ) times  $T_w = -ECw\theta''$ . From the bimoment chart,  $T_w$  could be reasonably expected to be around  $(4000+3000)/138.50 \sim 50$ ; that is, a two-figure-number as shown below



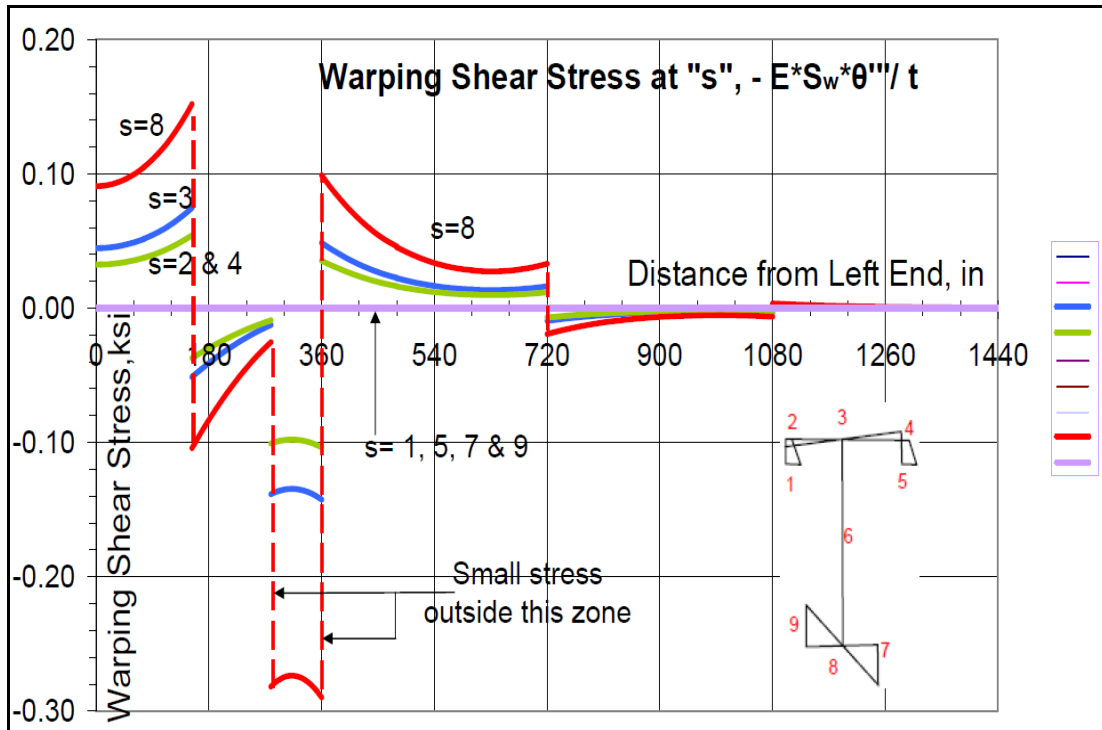
Comparison of Torque Charts by Medwadowski and BMTORSWP



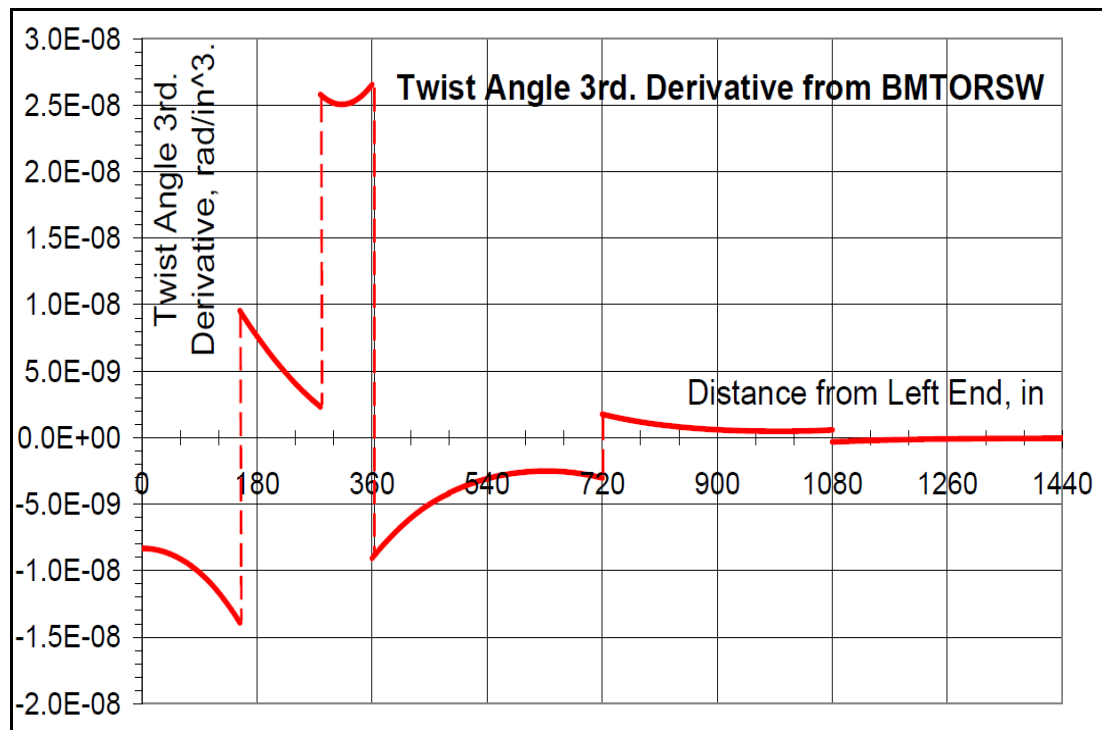
Pure Torsion Shear Stress Beam and Profile



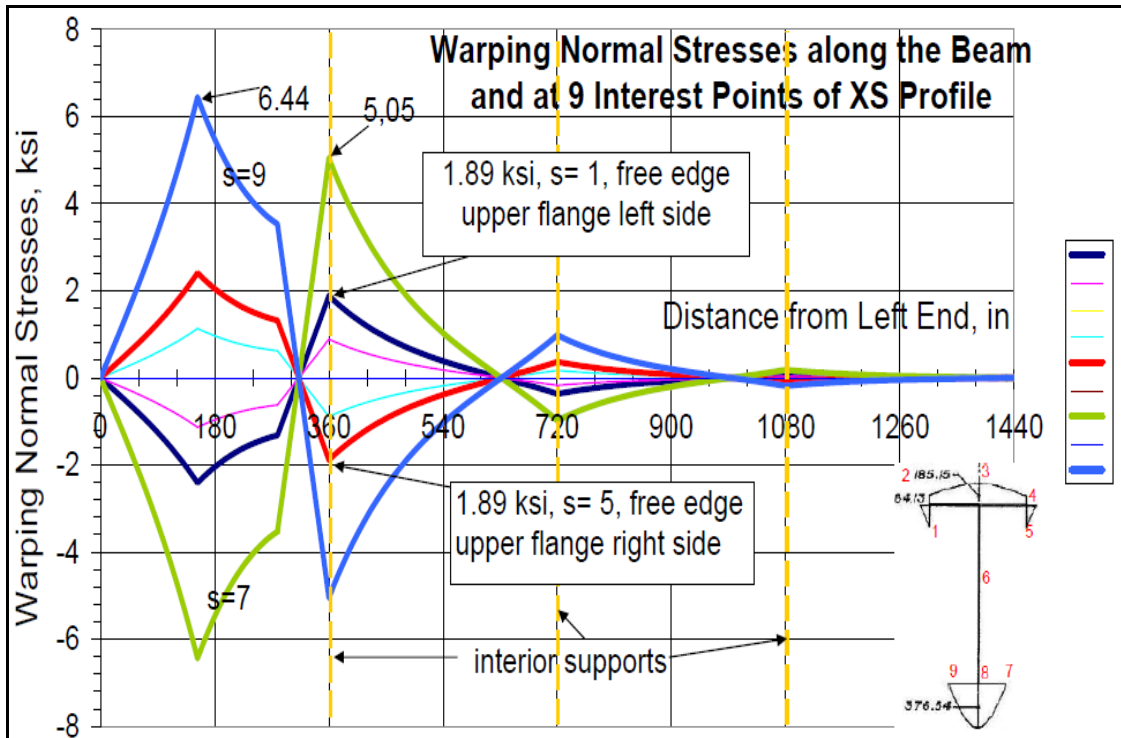
Twist Angle 1st. Derivative from BMTORSW



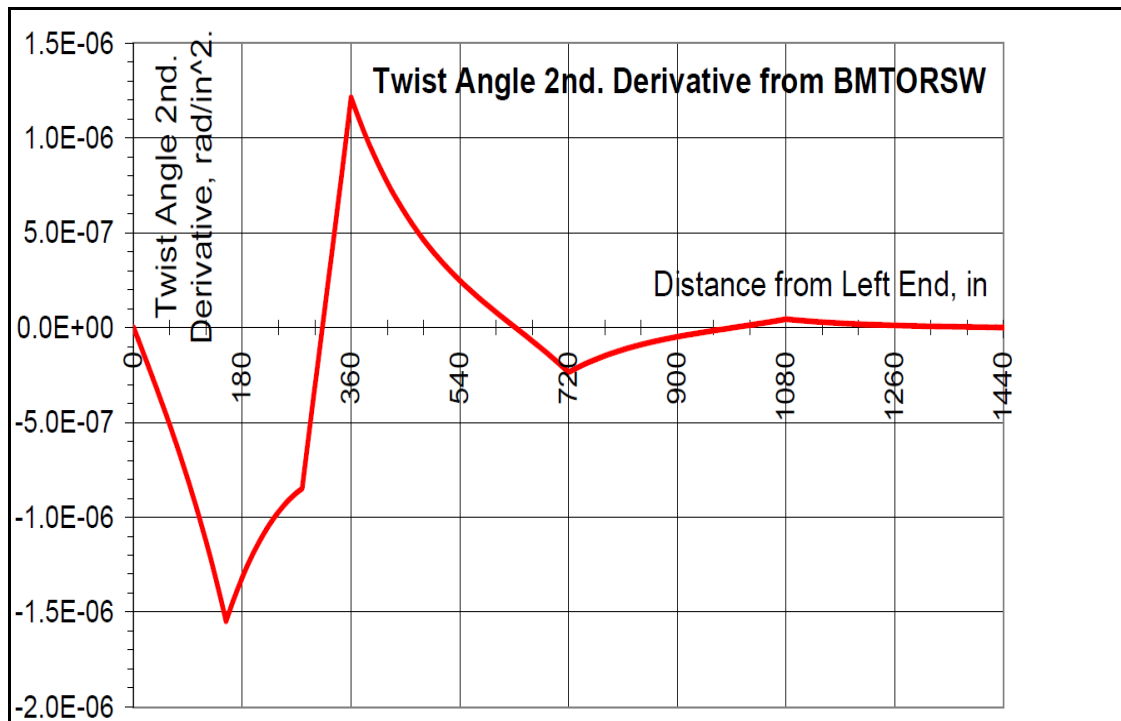
Warping Shear Stress along the Beam and Profile



Twist Angle 3rd. Derivative from BMTORSW



Warping Normal Stresses along the Beam and Profile



Twist Angle 2nd. Derivative from BMTORSW

```

-----
                                Mdwout
-----
YOU ARE USING COMPUTER PROGRAM BMTORSW,DEVELOPED BY DR. BERNARDO DESCHAPELLES
0      INPUT DATA FILE NAME IS = Mdw.txt
0      OUTPUT FILE NAME IS = Mdwout.txt
0      STORAGE FILE FOR POST-PROCESSING WITH EXCEL = Mdwgr.grf
-----
0
ASCE-JSD,V.111,N.2,Feb-85,p.453, Warp. M. Distribution, Medwadowski
Omodulus of elasticity of the material= 29000. k/ft2
OELEM nodes inertia length distrib. load AXIAL SOIL NORMAL MODULUS,Ksf angle
      i  j  ft.4    ft    at i    at j    LOAD    1st END    2nd END    rad
1  1  3***** 76.50  0.000  0.000*****  0.0    0.0    0.000 00
2  3  5***** 76.50  0.000  0.000*****  0.0    0.0    0.000 00
3  5  7*****126.00  0.000  0.000*****  0.0    0.0    0.000 00
4  7  9***** 81.00  0.000  0.000*****  0.0    0.0    0.000 00
5  9 11*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
6 11 13*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
7 13 15*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
8 15 17*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
9 17 19*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
10 19 21*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
11 21 23*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
12 23 25*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
13 25 27*****120.00  0.000  0.000*****  0.0    0.0    0.000 00
0      INPUT DATA RELATED TO THE 5 SUPPORTS
5 1 1 1 0 9 0 1 015 0 1 021 0 1 027 0 1 0
0      INPUT OF NODAL FORCES RELATED TO GLOBAL AXIS 2
      2 5 56.00 7 56.00
0      FINAL SOLUTION FOUND AFTER 1 ITERATIONS
0      Output of nodal displacements in reference to global axes
Onode displ.      displ.      displ.      node displ.      displ.      displ.
      along x      along y      around z      along x      along y      around z
      or nonn1      or nonn2      or nonn 3      or nonn1      or nonn2      or nonn 3
+ 1 0.0000E+00  0.0000E+00  0.1372E-03      2 0.0000E+00  0.5090E-02  0.8274E-03
+
+ 3 0.0000E+00  0.9864E-02  0.1122E-03      4 0.0000E+00  0.1331E-01  0.4942E-03
+
+ 5 0.0000E+00  0.1571E-01  0.2947E-04      6 0.0000E+00  0.1412E-01 -0.5240E-03
+
+ 7 0.0000E+00  0.9604E-02 -0.1119E-03      8 0.0000E+00  0.4705E-02 -0.8189E-03
+
+ 9 0.0000E+00  0.0000E+00 -0.9755E-04      10 0.0000E+00 -0.3503E-02 -0.4132E-03
+
+ 11 0.0000E+00 -0.5136E-02 -0.2877E-05      12 0.0000E+00 -0.4465E-02  0.1699E-03
+
+ 13 0.0000E+00 -0.3187E-02  0.2782E-04      14 0.0000E+00 -0.1507E-02  0.2718E-03
+
+ 15 0.0000E+00  0.0000E+00  0.1904E-04      16 0.0000E+00  0.6849E-03  0.8093E-04
+
+ 17 0.0000E+00  0.1006E-02  0.5994E-06      18 0.0000E+00  0.8765E-03 -0.3288E-04
+
+ 19 0.0000E+00  0.6284E-03 -0.5426E-05      20 0.0000E+00  0.2986E-03 -0.5356E-04
+
Page 1

```



```

Mdwout
+ 21 0.0000E+00 0.0000E+00 -0.3847E-05
+
+ 23 0.0000E+00 -0.2187E-03 -0.3150E-06
+
+ 25 0.0000E+00 -0.1591E-03 0.1078E-05
+
+ 27 0.0000E+00 0.0000E+00 0.1445E-05
0-----
      OUTPUT OF SOIL REACTIONS,STRESSES AND TRANSVERSE DISPLACEMENTS
0-----
O      ELEMENT 1  DISPLACEMENTS IN INCIDENCES  1  2  3
      NODE 1      0.00000E+00      0.00000E+00      0.13718E-03
      NODE 2      0.00000E+00      0.50904E-02      0.82741E-03
      NODE 3      0.00000E+00      0.98641E-02      0.11221E-03
O      FORCES ACTING ALONG THE 9 DOF
      NODE 1      0.00000E+00      -0.36788E+02      -0.25466E-10
      NODE 2      0.00000E+00      -0.40927E-11      0.00000E+00
      NODE 3      0.00000E+00      0.36788E+02      -0.15919E+04
OELEMENT 1, FROM NODE 1, TO NODE 3 - LENGTH = 76.50 ft
0 left half of span,at tenth points of length
      span      span      span      span      span      span      span
+      soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000 0.000
      shear,k   -36.79 -36.79 -36.79 -36.79 -36.79 -36.79 -36.79
      bmom,kft  0.00 -151.45 -303.37 -456.21 -610.44 -766.53
      tdisp,ft  0.00000 0.00105 0.00209 0.00313 0.00416 0.00517
      axial,k   0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
      span      span      span      span      span      span
+      soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000 0.000
      shear,k   -36.79 -36.79 -36.79 -36.79 -36.79 -36.79 -36.79
      bmom,kft -766.53 -924.97 -1086.22 -1250.79 -1419.17 -1591.88
      tdisp,ft  0.00517 0.00616 0.00713 0.00807 0.00899 0.00986
      axial,k   0.00 AT 1st END and 0.00 AT 2nd END
0-----
O      ELEMENT 2  DISPLACEMENTS IN INCIDENCES  3  4  5
      NODE 3      0.00000E+00      0.98641E-02      0.11221E-03
      NODE 4      0.00000E+00      0.13312E-01      0.49424E-03
      NODE 5      0.00000E+00      0.15710E-01      0.29475E-04
O      FORCES ACTING ALONG THE 9 DOF
      NODE 3      0.00000E+00      -0.36788E+02      0.15919E+04
      NODE 4      0.00000E+00      -0.26148E-11      -0.10368E-09
      NODE 5      0.00000E+00      0.36788E+02      -0.36817E+04
OELEMENT 2, FROM NODE 3, TO NODE 5 - LENGTH = 76.50 ft
0 left half of span,at tenth points of length
      span      span      span      span      span      span
+      soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000 0.000
      shear,k   -36.79 -36.79 -36.79 -36.79 -36.79 -36.79 -36.79
      bmom,kft -1591.88 -1769.45 -1952.42 -2141.34 -2336.79 -2539.38
      tdisp,ft  0.00986 0.01070 0.01150 0.01224 0.01294 0.01357
      axial,k   0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
      span      span      span      span      span      span
+      soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000 0.000
      shear,k   -36.79 -36.79 -36.79 -36.79 -36.79 -36.79 -36.79
      bmom,kft -2539.38 -2749.71 -2968.42 -3196.19 -3433.72 -3681.71
↓ (-Di + Dj)* GJ + Vj*L = Mj-Mi, where GJ=123921 Page 2
.01571*123921-36.79*(76.5+76.5)= 1956.713-5628.87= -3672.157
Mj-Mi = -3681.71 OK

```

Output Notepad, Page 2



```

Mdwout
tdisp,ft 0.01357 0.01415 0.01465 0.01509 0.01544 0.01571
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 3 DISPLACEMENTS IN INCIDENCES 5 6 7
  NODE 5 0.00000E+00 0.15710E-01 0.29475E-04
  NODE 6 0.00000E+00 0.14121E-01 -0.52400E-03
  NODE 7 0.00000E+00 0.96036E-02 -0.11190E-03
0 FORCES ACTING ALONG THE 9 DOF
  NODE 5 0.00000E+00 0.19212E+02 0.36817E+04
  NODE 6 0.00000E+00 0.90949E-12 -0.58208E-10
  NODE 7 0.00000E+00 -0.19212E+02 -0.20177E+04
0 ELEMENT 3, FROM NODE 5, TO NODE 7 - LENGTH =126.00 ft
0 left half of span,at tenth points of length
span span span span span span
+ soil,k/ft 0.0 0.1 0.2 0.3 0.4 0.5
  shear,k 0.000 0.000 0.000 0.000 0.000 0.000
  bmom,kft -3681.71 -3408.46 -3163.42 -2944.57 -2750.11 -2578.41
  tdisp,ft 0.01571 0.01596 0.01599 0.01580 0.01541 0.01485
  axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span span span span span span
+ soil,k/ft 0.5 0.6 0.7 0.8 0.9 1.0
  shear,k 19.21 19.21 19.21 19.21 19.21 19.21
  bmom,kft -2578.41 -2428.06 -2297.80 -2186.57 -2093.45 -2017.66
  tdisp,ft 0.01485 0.01411 0.01320 0.01215 0.01095 0.00960
  axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 4 DISPLACEMENTS IN INCIDENCES 7 8 9
  NODE 7 0.00000E+00 0.96036E-02 -0.11190E-03
  NODE 8 0.00000E+00 0.47054E-02 -0.81893E-03
  NODE 9 0.00000E+00 0.00000E+00 -0.97548E-04
0 FORCES ACTING ALONG THE 9 DOF
  NODE 7 0.00000E+00 0.75212E+02 0.20177E+04
  NODE 8 0.00000E+00 -0.22737E-11 0.36380E-11
  NODE 9 0.00000E+00 -0.75212E+02 0.28845E+04
0 ELEMENT 4, FROM NODE 7, TO NODE 9 - LENGTH = 81.00 ft
0 left half of span,at tenth points of length
span span span span span span
+ soil,k/ft 0.0 0.1 0.2 0.3 0.4 0.5
  shear,k 0.000 0.000 0.000 0.000 0.000 0.000
  bmom,kft -2017.66 -1523.93 -1035.41 -550.44 -67.34 415.52
  tdisp,ft 0.00960 0.00867 0.00770 0.00669 0.00568 0.00466
  axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span span span span span span
+ soil,k/ft 0.5 0.6 0.7 0.8 0.9 1.0
  shear,k 75.21 75.21 75.21 75.21 75.21 75.21
  bmom,kft -2017.66 -1523.93 -1035.41 -550.44 -67.34 415.52
  tdisp,ft 0.00466 0.00365 0.00267 0.00172 0.00083 0.00000
  axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 5 DISPLACEMENTS IN INCIDENCES 9 10 11
  NODE 9 0.00000E+00 0.00000E+00 -0.97548E-04
  NODE 10 0.00000E+00 -0.35030E-02 -0.41323E-03
  NODE 11 0.00000E+00 -0.51357E-02 -0.28765E-05
0 FORCES ACTING ALONG THE 9 DOF
  NODE 9 0.00000E+00 -0.95721E+01 -0.28845E+04
  NODE 10 0.00000E+00 0.23093E-12 -0.14779E-11
  NODE 11 0.00000E+00 0.95721E+01 0.10994E+04

```

$(-D_i + D_j) * GJ + V_j * L = M_j - M_i$ , where  $GJ=123921$  Page 3  
 $\{(-0.01571 + 0.00960) * 123921 = -757.157\} + \{19.21 * 126 = 2420.460\} = 2420.460 - 757.157 = 1663.303$   
 $M_j - M_i = -2017.66 + 3681.71 = 1664.05$  OK

Output Notepad, Page 3

$(-Di + Dj) * GJ + Vj * L = Mj - Mi$ , where  $GJ = 123921$        $-0.00514 * 123921 * 9.57 * 120 = -636.954 - 1148.4$   
 $Mdwout = -1785.354 <OK> 1099.39 - 2884.47 = -1785.08$

ELEMENT 5, FROM NODE 9, TO NODE 11 - LENGTH = 120.00 ft  
 0 left half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	2884.47	2635.05	2405.41	2193.84	1998.74	1818.65	
tdisp, ft	0.00000	-0.00109	-0.00201	-0.00279	-0.00344	-0.00397	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

0 right half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	1818.65	1652.22	1498.19	1355.41	1222.81	1099.39	
tdisp, ft	-0.00397	-0.00438	-0.00470	-0.00492	-0.00507	-0.00514	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

---

ELEMENT 6 DISPLACEMENTS IN INCIDENCES 11 12 13

NODE 11	0.00000E+00	-0.51357E-02	-0.28765E-05
NODE 12	0.00000E+00	-0.44646E-02	0.16988E-03
NODE 13	0.00000E+00	-0.31872E-02	0.27817E-04

FORCES ACTING ALONG THE 9 DOF

NODE 11	0.00000E+00	-0.95721E+01	-0.10994E+04
NODE 12	0.00000E+00	-0.17053E-12	-0.15916E-10
NODE 13	0.00000E+00	0.95721E+01	0.19221E+03

ELEMENT 6, FROM NODE 11, TO NODE 13 - LENGTH = 120.00 ft  
 0 left half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	1099.39	984.23	876.45	775.26	679.89	589.62	
tdisp, ft	-0.00514	-0.00514	-0.00508	-0.00497	-0.00481	-0.00461	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

0 right half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	589.62	503.78	421.73	342.84	266.52	192.21	
tdisp, ft	-0.00461	-0.00438	-0.00412	-0.00383	-0.00351	-0.00319	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

---

ELEMENT 7 DISPLACEMENTS IN INCIDENCES 13 14 15

NODE 13	0.00000E+00	-0.31872E-02	0.27817E-04
NODE 14	0.00000E+00	-0.15069E-02	0.27182E-03
NODE 15	0.00000E+00	0.00000E+00	0.19041E-04

FORCES ACTING ALONG THE 9 DOF

NODE 13	0.00000E+00	-0.95721E+01	-0.19221E+03
NODE 14	0.00000E+00	-0.28422E-13	0.50022E-11
NODE 15	0.00000E+00	0.95721E+01	-0.56149E+03

ELEMENT 7, FROM NODE 13, TO NODE 15 - LENGTH = 120.00 ft  
 0 left half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.0	0.1	0.2	0.3	0.4	0.5	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	192.21	119.34	47.36	-24.25	-96.05	-168.58	
tdisp, ft	-0.00319	-0.00285	-0.00250	-0.00215	-0.00181	-0.00146	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

0 right half of span, at tenth points of length

	span	span	span	span	span	span	span
	0.5	0.6	0.7	0.8	0.9	1.0	
soil, k/ft	0.000	0.000	0.000	0.000	0.000	0.000	0.000
shear, k	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57	-9.57
bmom, kft	192.21	119.34	47.36	-24.25	-96.05	-168.58	
tdisp, ft	-0.00319	-0.00285	-0.00250	-0.00215	-0.00181	-0.00146	
axial, k	0.00	AT 1st END and				0.00	AT 2nd END

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```

+                               Mdwout
+      0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft  0.000      0.000      0.000      0.000      0.000      0.000
shear,k    -9.57      -9.57      -9.57      -9.57      -9.57      -9.57
bmom,kft  -168.58    -242.36    -317.97    -395.97    -476.94    -561.49
tdisp,ft  -0.00146    -0.00113    -0.00082    -0.00052    -0.00024    0.00000
axial,k    0.00 AT 1st END and 0.00 AT 2nd END
-----
0      ELEMENT 8 DISPLACEMENTS IN INCIDENCES 15 16 17
      NODE 15 0.00000E+00 0.00000E+00 0.19041E-04
      NODE 16 0.00000E+00 0.68492E-03 0.80928E-04
      NODE 17 0.00000E+00 0.10055E-02 0.59935E-06
0      FORCES ACTING ALONG THE 9 DOF
      NODE 15 0.00000E+00 0.18526E+01 0.56149E+03
      NODE 16 0.00000E+00 0.37303E-13 0.17053E-11
      NODE 17 0.00000E+00 -0.18526E+01 -0.21457E+03
0ELEMENT 8, FROM NODE 15, TO NODE 17 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
      span      span      span      span      span      span
+      0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft  0.000      0.000      0.000      0.000      0.000      0.000
shear,k    1.85      1.85      1.85      1.85      1.85      1.85
bmom,kft  -561.49    -512.98    -468.33    -427.20    -389.28    -354.27
tdisp,ft  0.00000      0.00021      0.00039      0.00055      0.00067      0.00078
axial,k    0.00 AT 1st END and 0.00 AT 2nd END
0right half of span,at tenth points of length
      span      span      span      span      span      span
+      0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft  0.000      0.000      0.000      0.000      0.000      0.000
shear,k    1.85      1.85      1.85      1.85      1.85      1.85
bmom,kft  -354.27    -321.93    -292.01    -264.27    -238.53    -214.57
tdisp,ft  0.00078      0.00086      0.00092      0.00096      0.00099      0.00101
axial,k    0.00 AT 1st END and 0.00 AT 2nd END
-----
0      ELEMENT 9 DISPLACEMENTS IN INCIDENCES 17 18 19
      NODE 17 0.00000E+00 0.10055E-02 0.59935E-06
      NODE 18 0.00000E+00 0.87648E-03 -0.32878E-04
      NODE 19 0.00000E+00 0.62840E-03 -0.54263E-05
0      FORCES ACTING ALONG THE 9 DOF
      NODE 17 0.00000E+00 0.18526E+01 0.21457E+03
      NODE 18 0.00000E+00 0.14211E-13 0.90949E-12
      NODE 19 0.00000E+00 -0.18526E+01 -0.38994E+02
0ELEMENT 9, FROM NODE 17, TO NODE 19 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
      span      span      span      span      span      span
+      0.0      0.1      0.2      0.3      0.4      0.5
soil,k/ft  0.000      0.000      0.000      0.000      0.000      0.000
shear,k    1.85      1.85      1.85      1.85      1.85      1.85
bmom,kft  -214.57    -192.22    -171.32    -151.71    -133.23    -115.75
tdisp,ft  0.00101      0.00101      0.00100      0.00097      0.00094      0.00091
axial,k    0.00 AT 1st END and 0.00 AT 2nd END
0right half of span,at tenth points of length
      span      span      span      span      span      span
+      0.5      0.6      0.7      0.8      0.9      1.0
soil,k/ft  0.000      0.000      0.000      0.000      0.000      0.000
shear,k    1.85      1.85      1.85      1.85      1.85      1.85
bmom,kft  -115.75    -99.15     -83.28     -68.05     -53.32     -38.99
tdisp,ft  0.00091      0.00086      0.00081      0.00075      0.00069      0.00063
axial,k    0.00 AT 1st END and 0.00 AT 2nd END
-----
0      ELEMENT 10 DISPLACEMENTS IN INCIDENCES 19 20 21
      NODE 19 0.00000E+00 0.62840E-03 -0.54263E-05
      NODE 20 0.00000E+00 0.29860E-03 -0.53560E-04
      NODE 21 0.00000E+00 0.00000E+00 -0.38472E-05

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Mdwout
O FORCES ACTING ALONG THE 9 DOF
  NODE 19 0.00000E+00 0.18526E+01 0.38994E+02
  NODE 20 0.00000E+00 0.35527E-13 -0.34106E-12
  NODE 21 0.00000E+00 -0.18526E+01 0.10544E+03
ELEMENT 10, FROM NODE 19, TO NODE 21 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
  span span span span span span
+ 0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.85 1.85 1.85 1.85 1.85 1.85
bmom,kft -38.99 -24.96 -11.12 2.65 16.43 30.33
tdisp,ft 0.00063 0.00056 0.00049 0.00043 0.00036 0.00029
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
  span span span span span span
+ 0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k 1.85 1.85 1.85 1.85 1.85 1.85
bmom,kft 30.33 44.47 58.93 73.84 89.31 105.44
tdisp,ft 0.00029 0.00023 0.00016 0.00010 0.00005 0.00000
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
O ELEMENT 11 DISPLACEMENTS IN INCIDENCES 21 22 23
  NODE 21 0.00000E+00 0.00000E+00 -0.38472E-05
  NODE 22 0.00000E+00 -0.14424E-03 -0.17711E-04
  NODE 23 0.00000E+00 -0.21871E-03 -0.31503E-06
O FORCES ACTING ALONG THE 9 DOF
  NODE 21 0.00000E+00 -0.29290E+00 -0.10544E+03
  NODE 22 0.00000E+00 0.22649E-13 0.26290E-12
  NODE 23 0.00000E+00 0.29290E+00 0.43193E+02
ELEMENT 11, FROM NODE 21, TO NODE 23 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
  span span span span span span
+ 0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 105.44 96.59 88.47 81.00 74.15 67.85
tdisp,ft 0.00000 -0.00004 -0.00008 -0.00011 -0.00014 -0.00016
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
  span span span span span span
+ 0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 67.85 62.07 56.74 51.85 47.34 43.19
tdisp,ft -0.00016 -0.00018 -0.00019 -0.00021 -0.00021 -0.00022
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
O ELEMENT 12 DISPLACEMENTS IN INCIDENCES 23 24 25
  NODE 23 0.00000E+00 -0.21871E-03 -0.31503E-06
  NODE 24 0.00000E+00 -0.20265E-03 0.51981E-05
  NODE 25 0.00000E+00 -0.15908E-03 0.10782E-05
O FORCES ACTING ALONG THE 9 DOF
  NODE 23 0.00000E+00 -0.29290E+00 -0.43193E+02
  NODE 24 0.00000E+00 -0.21316E-13 0.51159E-12
  NODE 25 0.00000E+00 0.29290E+00 0.15434E+02
ELEMENT 12, FROM NODE 23, TO NODE 25 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
  span span span span span span
+ 0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 43.19 39.37 35.84 32.58 29.56 26.76

```

```

Mdwout
tdisp,ft -0.00022 -0.00022 -0.00022 -0.00022 -0.00022 -0.00021
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span span span span span span
+ 0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 26.76 24.17 21.76 19.51 17.41 15.43
tdisp,ft -0.00021 -0.00020 -0.00019 -0.00018 -0.00017 -0.00016
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
-----
0 ELEMENT 13 DISPLACEMENTS IN INCIDENCES 25 26 27
NODE 25 0.00000E+00 -0.15908E-03 0.10782E-05
NODE 26 0.00000E+00 -0.83163E-04 0.13384E-04
NODE 27 0.00000E+00 0.00000E+00 0.14450E-05
0 FORCES ACTING ALONG THE 9 DOF
NODE 25 0.00000E+00 -0.29290E+00 -0.15434E+02
NODE 26 0.00000E+00 0.88818E-14 0.56843E-13
NODE 27 0.00000E+00 0.29290E+00 0.17053E-12
0 ELEMENT 13, FROM NODE 25, TO NODE 27 - LENGTH =120.00 ft
0 left half of span,at tenth points of length
span span span span span span
+ 0.0 0.1 0.2 0.3 0.4 0.5
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 15.43 13.58 11.82 10.16 8.57 7.05
tdisp,ft -0.00016 -0.00015 -0.00013 -0.00012 -0.00010 -0.00008
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
Oright half of span,at tenth points of length
span span span span span span
+ 0.5 0.6 0.7 0.8 0.9 1.0
soil,k/ft 0.000 0.000 0.000 0.000 0.000 0.000
shear,k -0.29 -0.29 -0.29 -0.29 -0.29 -0.29
bmom,kft 7.05 5.57 4.14 2.75 1.37 0.00
tdisp,ft -0.00008 -0.00007 -0.00005 -0.00003 -0.00002 0.00000
axial,k 0.00 AT 1st END and 0.00 AT 2nd END
1
(-Di + Dj)* GJ + Vj*L = Mj-Mi, where GJ=123921
-(-0.00016)*123921 - 0.29* 120 = 19.827 - 3.48 = 16.347
bmom= 15.43 ~ 16.347 OK

```

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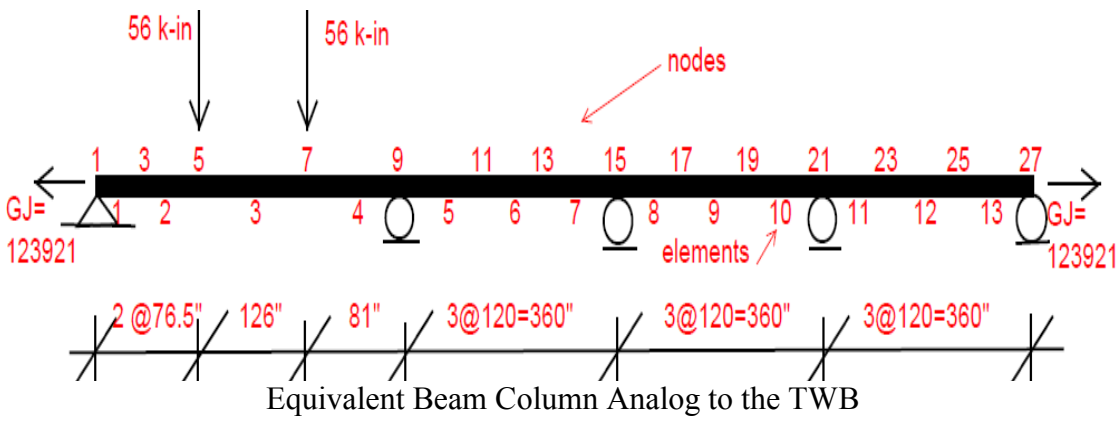


Table of Twist angle and Derivatives

	z	$\theta$	T(z)	B(z)	$\theta'$	$\theta''$	$\theta'''$
1							
	0	0.000E+00	-36.7876	0	1.372E-04	1.590E-11	-8.325E-09
	7.65	1.000E-03	-36.7876	-151.4528	1.369E-04	-6.369E-08	-8.335E-09
	15.3	2.100E-03	-36.7876	-303.3677	1.362E-04	-1.276E-07	-8.372E-09
	22.95	3.100E-03	-36.7876	-456.2081	1.350E-04	-1.919E-07	-8.435E-09
	30.6	4.200E-03	-36.7876	-610.4401	1.333E-04	-2.567E-07	-8.524E-09
	38.25	5.200E-03	-36.7876	-766.5342	1.311E-04	-3.223E-07	-8.640E-09
	45.9	6.200E-03	-36.7876	-924.9667	1.283E-04	-3.890E-07	-8.783E-09
	53.55	7.100E-03	-36.7876	-1086.221	1.251E-04	-4.568E-07	-8.952E-09
	61.2	8.100E-03	-36.7876	-1250.788	1.213E-04	-5.260E-07	-9.147E-09
	68.85	9.000E-03	-36.7876	-1419.171	1.171E-04	-5.968E-07	-9.369E-09
	76.5	9.900E-03	-36.7876	-1591.884	1.122E-04	-6.694E-07	-9.617E-09
2							
	76.5	9.900E-03	-36.7876	-1591.884	1.122E-04	-6.694E-07	-9.637E-09
	84.15	1.070E-02	-36.7876	-1769.452	1.068E-04	-7.441E-07	-9.909E-09
	91.8	1.150E-02	-36.7876	-1952.418	1.008E-04	-8.211E-07	-1.022E-08
	99.45	1.220E-02	-36.7876	-2141.34	9.424E-05	-9.005E-07	-1.056E-08
	107.1	1.290E-02	-36.7876	-2336.794	8.703E-05	-9.827E-07	-1.093E-08
	114.75	1.360E-02	-36.7876	-2539.377	7.919E-05	-1.068E-06	-1.134E-08
	122.4	1.410E-02	-36.7876	-2749.706	7.069E-05	-1.156E-06	-1.179E-08
	130.05	1.470E-02	-36.7876	-2968.422	6.149E-05	-1.248E-06	-1.227E-08
	137.7	1.510E-02	-36.7876	-3196.194	5.158E-05	-1.344E-06	-1.278E-08
	145.35	1.540E-02	-36.7876	-3433.716	4.092E-05	-1.444E-06	-1.333E-08
	153	1.570E-02	-36.7876	-3681.713	2.947E-05	-1.548E-06	-1.392E-08
3							
	153	1.570E-02	19.2124	-3681.713	2.947E-05	-1.548E-06	9.540E-09
	165.6	1.600E-02	19.2124	-3408.456	1.070E-05	-1.433E-06	8.616E-09
	178.2	1.600E-02	19.2124	-3163.419	-6.696E-06	-1.330E-06	7.736E-09
	190.8	1.580E-02	19.2124	-2944.575	-2.287E-05	-1.238E-06	6.901E-09
	203.4	1.540E-02	19.2124	-2750.11	-3.794E-05	-1.156E-06	6.111E-09
	216	1.480E-02	19.2124	-2578.411	-5.205E-05	-1.084E-06	5.366E-09
	228.6	1.410E-02	19.2124	-2428.056	-6.530E-05	-1.021E-06	4.666E-09
	241.2	1.320E-02	19.2124	-2297.802	-7.781E-05	-9.663E-07	4.011E-09
	253.8	1.210E-02	19.2124	-2186.571	-8.969E-05	-9.197E-07	3.401E-09
	266.4	1.090E-02	19.2124	-2093.447	-1.010E-04	-8.804E-07	2.836E-09
	279	9.600E-03	19.2124	-2017.656	-1.119E-04	-8.480E-07	2.315E-09
4							
	279	9.600E-03	75.2124	-2017.656	-1.119E-04	-8.485E-07	2.580E-08
	287.1	8.700E-03	75.2124	-1523.927	-1.179E-04	-6.409E-07	2.548E-08
	295.2	7.700E-03	75.2124	-1035.411	-1.223E-04	-4.354E-07	2.526E-08
	303.3	6.700E-03	75.2124	-550.4354	-1.250E-04	-2.315E-07	2.511E-08
	311.4	5.700E-03	75.2124	-67.3426	-1.260E-04	-2.832E-08	2.506E-08
	319.5	4.700E-03	75.2124	415.5199	-1.255E-04	1.747E-07	2.509E-08

	327.6	3.600E-03	75.2124	899.8037	-1.232E-04	3.784E-07	2.521E-08
	z	$\theta$	T(z)	B(z)	$\theta'$	$\theta''$	$\theta'''$
	335.7	2.700E-03	75.2124	1387.164	-1.193E-04	5.833E-07	2.541E-08
	343.8	1.700E-03	75.2124	1879.27	-1.138E-04	7.903E-07	2.570E-08
	351.9	8.000E-04	75.2124	2377.802	-1.065E-04	9.999E-07	2.608E-08
	360	0.000E+00	75.2124	2884.467	-9.755E-05	1.213E-06	2.654E-08
5							
	360	0.000E+00	-9.5721	2884.467	-9.755E-05	1.213E-06	-9.062E-09
	372	-1.100E-03	-9.5721	2635.046	-8.363E-05	1.108E-06	-8.370E-09
	384	-2.000E-03	-9.5721	2405.411	-7.092E-05	1.012E-06	-7.725E-09
	396	-2.800E-03	-9.5721	2193.841	-5.932E-05	9.226E-07	-7.125E-09
	408	-3.400E-03	-9.5721	1998.744	-4.875E-05	8.405E-07	-6.571E-09
	420	-4.000E-03	-9.5721	1818.654	-3.913E-05	7.647E-07	-6.064E-09
	432	-4.400E-03	-9.5721	1652.219	-3.037E-05	6.947E-07	-5.602E-09
	444	-4.700E-03	-9.5721	1498.189	-2.243E-05	6.301E-07	-5.186E-09
	456	-4.900E-03	-9.5721	1355.409	-1.523E-05	5.701E-07	-4.817E-09
	468	-5.100E-03	-9.5721	1222.808	-8.731E-06	5.143E-07	-4.493E-09
	480	-5.100E-03	-9.5721	1099.39	-2.877E-06	4.621E-07	-4.215E-09
6							
	480	-5.100E-03	-9.5721	1099.39	-2.877E-06	4.622E-07	-4.159E-09
	492	-5.100E-03	-9.5721	984.2255	2.377E-06	4.139E-07	-3.897E-09
	504	-5.100E-03	-9.5721	876.4516	7.069E-06	3.686E-07	-3.658E-09
	516	-5.000E-03	-9.5721	775.2594	1.123E-05	3.260E-07	-3.443E-09
	528	-4.800E-03	-9.5721	679.8887	1.490E-05	2.859E-07	-3.250E-09
	540	-4.600E-03	-9.5721	589.6228	1.810E-05	2.479E-07	-3.082E-09
	552	-4.400E-03	-9.5721	503.7838	2.086E-05	2.118E-07	-2.936E-09
	564	-4.100E-03	-9.5721	421.7273	2.320E-05	1.774E-07	-2.814E-09
	576	-3.800E-03	-9.5721	342.8376	2.512E-05	1.442E-07	-2.715E-09
	588	-3.500E-03	-9.5721	266.5228	2.666E-05	1.121E-07	-2.640E-09
	600	-3.200E-03	-9.5721	192.2095	2.782E-05	8.075E-08	-2.588E-09
7							
	600	-3.200E-03	-9.5721	192.2095	2.782E-05	8.085E-08	-2.578E-09
	612	-2.800E-03	-9.5721	119.3392	2.860E-05	5.018E-08	-2.536E-09
	624	-2.500E-03	-9.5721	47.365	2.902E-05	1.991E-08	-2.513E-09
	636	-2.200E-03	-9.5721	-24.2538	2.908E-05	-1.020E-08	-2.509E-09
	648	-1.800E-03	-9.5721	-96.0547	2.878E-05	-4.039E-08	-2.525E-09
	660	-1.500E-03	-9.5721	-168.5768	2.811E-05	-7.088E-08	-2.560E-09
	672	-1.100E-03	-9.5721	-242.3646	2.707E-05	-1.019E-07	-2.615E-09
	684	-8.000E-04	-9.5721	-317.9721	2.566E-05	-1.337E-07	-2.689E-09
	696	-5.000E-04	-9.5721	-395.9673	2.386E-05	-1.665E-07	-2.782E-09
	708	-2.000E-04	-9.5721	-476.9359	2.166E-05	-2.006E-07	-2.895E-09
	720	0.000E+00	-9.5721	-561.4858	1.904E-05	-2.361E-07	-3.028E-09
8							
	720	0.000E+00	1.8526	-561.4858	1.904E-05	-2.361E-07	1.762E-09
	732	2.000E-04	1.8526	-512.9839	1.633E-05	-2.157E-07	1.628E-09

	744	4.000E-04	1.8526	-468.334	1.386E-05	-1.970E-07	1.502E-09
	756	5.000E-04	1.8526	-427.2012	1.160E-05	-1.797E-07	1.385E-09
	$z$	$\theta$	$T(z)$	$B(z)$	$\theta'$	$\theta''$	$\theta'''$
	768	7.000E-04	1.8526	-389.2762	9.540E-06	-1.637E-07	1.277E-09
	780	8.000E-04	1.8526	-354.274	7.666E-06	-1.490E-07	1.178E-09
	792	9.000E-04	1.8526	-321.9316	5.961E-06	-1.354E-07	1.088E-09
	804	9.000E-04	1.8526	-292.0065	4.413E-06	-1.228E-07	1.007E-09
	816	1.000E-03	1.8526	-264.274	3.010E-06	-1.112E-07	9.354E-10
	828	1.000E-03	1.8526	-238.5263	1.742E-06	-1.003E-07	8.723E-10
	840	1.000E-03	1.8526	-214.5698	5.994E-07	-9.018E-08	8.181E-10
9							
	840	1.000E-03	1.8526	-214.5698	5.994E-07	-9.021E-08	8.072E-10
	852	1.000E-03	1.8526	-192.2241	-4.264E-07	-8.084E-08	7.560E-10
	864	1.000E-03	1.8526	-171.3219	-1.343E-06	-7.205E-08	7.093E-10
	876	1.000E-03	1.8526	-151.7062	-2.158E-06	-6.380E-08	6.672E-10
	888	9.000E-04	1.8526	-133.2296	-2.876E-06	-5.602E-08	6.295E-10
	900	9.000E-04	1.8526	-115.7534	-3.504E-06	-4.867E-08	5.964E-10
	912	9.000E-04	1.8526	-99.1463	-4.046E-06	-4.169E-08	5.678E-10
	924	8.000E-04	1.8526	-83.2836	-4.506E-06	-3.503E-08	5.438E-10
	936	8.000E-04	1.8526	-68.0463	-4.887E-06	-2.862E-08	5.242E-10
	948	7.000E-04	1.8526	-53.32	-5.194E-06	-2.243E-08	5.092E-10
	960	6.000E-04	1.8526	-38.9942	-5.426E-06	-1.638E-08	4.987E-10
10							
	960	6.000E-04	1.8526	-38.9942	-5.426E-06	-1.640E-08	4.967E-10
	972	6.000E-04	1.8526	-24.9611	-5.588E-06	-1.050E-08	4.881E-10
	984	5.000E-04	1.8526	-11.1154	-5.679E-06	-4.673E-09	4.831E-10
	996	4.000E-04	1.8526	2.6469	-5.700E-06	1.114E-09	4.819E-10
	1008	4.000E-04	1.8526	16.429	-5.652E-06	6.908E-09	4.844E-10
	1020	3.000E-04	1.8526	30.3345	-5.534E-06	1.275E-08	4.906E-10
	1032	2.000E-04	1.8526	44.4677	-5.345E-06	1.870E-08	5.006E-10
	1044	2.000E-04	1.8526	58.9348	-5.085E-06	2.478E-08	5.143E-10
	1056	1.000E-04	1.8526	73.8445	-4.750E-06	3.106E-08	5.316E-10
	1068	0.000E+00	1.8526	89.3086	-4.338E-06	3.756E-08	5.527E-10
	1080	0.000E+00	1.8526	105.4434	-3.847E-06	4.434E-08	5.776E-10
11							
	1080	0.000E+00	-0.2929	105.4434	-3.847E-06	4.433E-08	-3.219E-10
	1092	0.000E+00	-0.2929	96.5919	-3.338E-06	4.062E-08	-2.966E-10
	1104	-1.000E-04	-0.2929	88.4658	-2.871E-06	3.721E-08	-2.729E-10
	1116	-1.000E-04	-0.2929	81.0039	-2.444E-06	3.407E-08	-2.508E-10
	1128	-1.000E-04	-0.2929	74.1504	-2.053E-06	3.118E-08	-2.303E-10
	1140	-2.000E-04	-0.2929	67.8535	-1.694E-06	2.853E-08	-2.115E-10
	1152	-2.000E-04	-0.2929	62.0661	-1.367E-06	2.610E-08	-1.942E-10
	1164	-2.000E-04	-0.2929	56.7448	-1.067E-06	2.386E-08	-1.785E-10
	1176	-2.000E-04	-0.2929	51.8495	-7.935E-07	2.181E-08	-1.644E-10
	1188	-2.000E-04	-0.2929	47.3436	-5.433E-07	1.991E-08	-1.520E-10



	1200	-2.000E-04	-0.2929	43.1933	-3.150E-07	1.815E-08	-1.411E-10
12							
	1200	-2.000E-04	-0.2929	43.1933	-3.150E-07	1.816E-08	-1.389E-10
	z	$\theta$	T(z)	B(z)	$\theta'$	$\theta''$	$\theta'''$
	1212	-2.000E-04	-0.2929	39.3672	-1.069E-07	1.656E-08	-1.285E-10
	1224	-2.000E-04	-0.2929	35.8367	8.279E-08	1.507E-08	-1.189E-10
	1236	-2.000E-04	-0.2929	32.5754	2.553E-07	1.370E-08	-1.100E-10
	1248	-2.000E-04	-0.2929	29.5586	4.120E-07	1.243E-08	-1.018E-10
	1260	-2.000E-04	-0.2929	26.7638	5.540E-07	1.125E-08	-9.430E-11
	1272	-2.000E-04	-0.2929	24.17	6.824E-07	1.016E-08	-8.752E-11
	1284	-2.000E-04	-0.2929	21.7576	7.982E-07	9.150E-09	-8.146E-11
	1296	-2.000E-04	-0.2929	19.5086	9.023E-07	8.205E-09	-7.611E-11
	1308	-2.000E-04	-0.2929	17.4061	9.954E-07	7.321E-09	-7.148E-11
	1320	-2.000E-04	-0.2929	15.4343	1.078E-06	6.487E-09	-6.757E-11
13							
	1320	-2.000E-04	-0.2929	15.4343	1.078E-06	6.489E-09	-6.679E-11
	1332	-1.000E-04	-0.2929	13.5784	1.151E-06	5.710E-09	-6.312E-11
	1344	-1.000E-04	-0.2929	11.8244	1.215E-06	4.973E-09	-5.985E-11
	1356	-1.000E-04	-0.2929	10.1592	1.271E-06	4.272E-09	-5.698E-11
	1368	-1.000E-04	-0.2929	8.5704	1.318E-06	3.604E-09	-5.451E-11
	1380	-1.000E-04	-0.2929	7.0458	1.357E-06	2.963E-09	-5.243E-11
	1392	-1.000E-04	-0.2929	5.5742	1.389E-06	2.344E-09	-5.075E-11
	1404	-1.000E-04	-0.2929	4.1444	1.414E-06	1.743E-09	-4.947E-11
	1416	0.000E+00	-0.2929	2.7458	1.431E-06	1.155E-09	-4.858E-11
	1428	0.000E+00	-0.2929	1.3678	1.442E-06	5.754E-10	-4.810E-11
	1440	0.000E+00	-0.2929	0	1.445E-06	-8.881E-13	-4.801E-11

Table of Pure Torsion Shear Stresses along “z” and “s”

Pure Torsion Shear Stresses $G \cdot t \cdot \theta'$										
“t”=	0.625	0.625	1	0.625	0.625	0.5	1	1	1	
“s”=	1	2	3	4	5	6	7	8	9	
z										
	0	0.96	0.96	1.53	0.96	0.96	0.77	1.53	1.53	1.53
	7.65	0.95	0.95	1.53	0.95	0.95	0.76	1.53	1.53	1.53
	15.3	0.95	0.95	1.52	0.95	0.95	0.76	1.52	1.52	1.52
	22.95	0.94	0.94	1.51	0.94	0.94	0.75	1.51	1.51	1.51
	30.6	0.93	0.93	1.49	0.93	0.93	0.74	1.49	1.49	1.49
	38.25	0.91	0.91	1.46	0.91	0.91	0.73	1.46	1.46	1.46
	45.9	0.89	0.89	1.43	0.89	0.89	0.72	1.43	1.43	1.43
	53.55	0.87	0.87	1.40	0.87	0.87	0.70	1.40	1.40	1.40
	61.2	0.85	0.85	1.35	0.85	0.85	0.68	1.35	1.35	1.35
	68.85	0.82	0.82	1.31	0.82	0.82	0.65	1.31	1.31	1.31
	76.5	0.78	0.78	1.25	0.78	0.78	0.63	1.25	1.25	1.25

76.5	0.78	0.78	1.25	0.78	0.78	0.63	1.25	1.25	1.25
84.15	0.74	0.74	1.19	0.74	0.74	0.60	1.19	1.19	1.19
91.8	0.70	0.70	1.12	0.70	0.70	0.56	1.12	1.12	1.12
99.45	0.66	0.66	1.05	0.66	0.66	0.53	1.05	1.05	1.05
"s"=	1	2	3	4	5	6	7	8	9
z									
107.1	0.61	0.61	0.97	0.61	0.61	0.49	0.97	0.97	0.97
114.75	0.55	0.55	0.88	0.55	0.55	0.44	0.88	0.88	0.88
122.4	0.49	0.49	0.79	0.49	0.49	0.39	0.79	0.79	0.79
130.05	0.43	0.43	0.69	0.43	0.43	0.34	0.69	0.69	0.69
137.7	0.36	0.36	0.58	0.36	0.36	0.29	0.58	0.58	0.58
145.35	0.29	0.29	0.46	0.29	0.29	0.23	0.46	0.46	0.46
153	0.21	0.21	0.33	0.21	0.21	0.16	0.33	0.33	0.33
153	0.21	0.21	0.33	0.21	0.21	0.16	0.33	0.33	0.33
165.6	0.07	0.07	0.12	0.07	0.07	0.06	0.12	0.12	0.12
178.2	-0.05	-0.05	-0.07	-0.05	-0.05	-0.04	-0.07	-0.07	-0.07
190.8	-0.16	-0.16	-0.26	-0.16	-0.16	-0.13	-0.26	-0.26	-0.26
203.4	-0.26	-0.26	-0.42	-0.26	-0.26	-0.21	-0.42	-0.42	-0.42
216	-0.36	-0.36	-0.58	-0.36	-0.36	-0.29	-0.58	-0.58	-0.58
228.6	-0.46	-0.46	-0.73	-0.46	-0.46	-0.36	-0.73	-0.73	-0.73
241.2	-0.54	-0.54	-0.87	-0.54	-0.54	-0.43	-0.87	-0.87	-0.87
253.8	-0.63	-0.63	-1.00	-0.63	-0.63	-0.50	-1.00	-1.00	-1.00
266.4	-0.70	-0.70	-1.13	-0.70	-0.70	-0.56	-1.13	-1.13	-1.13
279	-0.78	-0.78	-1.25	-0.78	-0.78	-0.62	-1.25	-1.25	-1.25
279	-0.78	-0.78	-1.25	-0.78	-0.78	-0.62	-1.25	-1.25	-1.25
287.1	-0.82	-0.82	-1.32	-0.82	-0.82	-0.66	-1.32	-1.32	-1.32
295.2	-0.85	-0.85	-1.36	-0.85	-0.85	-0.68	-1.36	-1.36	-1.36
303.3	-0.87	-0.87	-1.39	-0.87	-0.87	-0.70	-1.39	-1.39	-1.39
311.4	-0.88	-0.88	-1.41	-0.88	-0.88	-0.70	-1.41	-1.41	-1.41
319.5	-0.87	-0.87	-1.40	-0.87	-0.87	-0.70	-1.40	-1.40	-1.40
327.6	-0.86	-0.86	-1.37	-0.86	-0.86	-0.69	-1.37	-1.37	-1.37
335.7	-0.83	-0.83	-1.33	-0.83	-0.83	-0.67	-1.33	-1.33	-1.33
343.8	-0.79	-0.79	-1.27	-0.79	-0.79	-0.63	-1.27	-1.27	-1.27
351.9	-0.74	-0.74	-1.19	-0.74	-0.74	-0.59	-1.19	-1.19	-1.19
360	-0.68	-0.68	-1.09	-0.68	-0.68	-0.54	-1.09	-1.09	-1.09
360	-0.68	-0.68	-1.09	-0.68	-0.68	-0.54	-1.09	-1.09	-1.09
372	-0.58	-0.58	-0.93	-0.58	-0.58	-0.47	-0.93	-0.93	-0.93
384	-0.49	-0.49	-0.79	-0.49	-0.49	-0.40	-0.79	-0.79	-0.79
396	-0.41	-0.41	-0.66	-0.41	-0.41	-0.33	-0.66	-0.66	-0.66
408	-0.34	-0.34	-0.54	-0.34	-0.34	-0.27	-0.54	-0.54	-0.54
420	-0.27	-0.27	-0.44	-0.27	-0.27	-0.22	-0.44	-0.44	-0.44
432	-0.21	-0.21	-0.34	-0.21	-0.21	-0.17	-0.34	-0.34	-0.34

444	-0.16	-0.16	-0.25	-0.16	-0.16	-0.13	-0.25	-0.25	-0.25
456	-0.11	-0.11	-0.17	-0.11	-0.11	-0.08	-0.17	-0.17	-0.17
468	-0.06	-0.06	-0.10	-0.06	-0.06	-0.05	-0.10	-0.10	-0.10
480	-0.02	-0.02	-0.03	-0.02	-0.02	-0.02	-0.03	-0.03	-0.03
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
"s"=	1	2	3	4	5	6	7	8	9
z									
480	-0.02	-0.02	-0.03	-0.02	-0.02	-0.02	-0.03	-0.03	-0.03
492	0.02	0.02	0.03	0.02	0.02	0.01	0.03	0.03	0.03
504	0.05	0.05	0.08	0.05	0.05	0.04	0.08	0.08	0.08
516	0.08	0.08	0.13	0.08	0.08	0.06	0.13	0.13	0.13
528	0.10	0.10	0.17	0.10	0.10	0.08	0.17	0.17	0.17
540	0.13	0.13	0.20	0.13	0.13	0.10	0.20	0.20	0.20
552	0.15	0.15	0.23	0.15	0.15	0.12	0.23	0.23	0.23
564	0.16	0.16	0.26	0.16	0.16	0.13	0.26	0.26	0.26
576	0.18	0.18	0.28	0.18	0.18	0.14	0.28	0.28	0.28
588	0.19	0.19	0.30	0.19	0.19	0.15	0.30	0.30	0.30
600	0.19	0.19	0.31	0.19	0.19	0.16	0.31	0.31	0.31
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
600	0.19	0.19	0.31	0.19	0.19	0.16	0.31	0.31	0.31
612	0.20	0.20	0.32	0.20	0.20	0.16	0.32	0.32	0.32
624	0.20	0.20	0.32	0.20	0.20	0.16	0.32	0.32	0.32
636	0.20	0.20	0.32	0.20	0.20	0.16	0.32	0.32	0.32
648	0.20	0.20	0.32	0.20	0.20	0.16	0.32	0.32	0.32
660	0.20	0.20	0.31	0.20	0.20	0.16	0.31	0.31	0.31
672	0.19	0.19	0.30	0.19	0.19	0.15	0.30	0.30	0.30
684	0.18	0.18	0.29	0.18	0.18	0.14	0.29	0.29	0.29
696	0.17	0.17	0.27	0.17	0.17	0.13	0.27	0.27	0.27
708	0.15	0.15	0.24	0.15	0.15	0.12	0.24	0.24	0.24
720	0.13	0.13	0.21	0.13	0.13	0.11	0.21	0.21	0.21
720	0.13	0.13	0.21	0.13	0.13	0.11	0.21	0.21	0.21
732	0.11	0.11	0.18	0.11	0.11	0.09	0.18	0.18	0.18
744	0.10	0.10	0.15	0.10	0.10	0.08	0.15	0.15	0.15
756	0.08	0.08	0.13	0.08	0.08	0.06	0.13	0.13	0.13
768	0.07	0.07	0.11	0.07	0.07	0.05	0.11	0.11	0.11
780	0.05	0.05	0.09	0.05	0.05	0.04	0.09	0.09	0.09
792	0.04	0.04	0.07	0.04	0.04	0.03	0.07	0.07	0.07
804	0.03	0.03	0.05	0.03	0.03	0.02	0.05	0.05	0.05
816	0.02	0.02	0.03	0.02	0.02	0.02	0.03	0.03	0.03
828	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
840	0.00	0.00	0.01	0.00	0.00	0.00	0.01	0.01	0.01
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
840	0.00	0.00	0.01	0.00	0.00	0.00	0.01	0.01	0.01
852	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

864	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01
876	-0.02	-0.02	-0.02	-0.02	-0.02	-0.01	-0.02	-0.02	-0.02
888	-0.02	-0.02	-0.03	-0.02	-0.02	-0.02	-0.03	-0.03	-0.03
900	-0.02	-0.02	-0.04	-0.02	-0.02	-0.02	-0.04	-0.04	-0.04
912	-0.03	-0.03	-0.05	-0.03	-0.03	-0.02	-0.05	-0.05	-0.05
924	-0.03	-0.03	-0.05	-0.03	-0.03	-0.03	-0.05	-0.05	-0.05
"s"=	1	2	3	4	5	6	7	8	9
z									
936	-0.03	-0.03	-0.05	-0.03	-0.03	-0.03	-0.05	-0.05	-0.05
948	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
960	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
960	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
972	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
984	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
996	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
1008	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
1020	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
1032	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
1044	-0.04	-0.04	-0.06	-0.04	-0.04	-0.03	-0.06	-0.06	-0.06
1056	-0.03	-0.03	-0.05	-0.03	-0.03	-0.03	-0.05	-0.05	-0.05
1068	-0.03	-0.03	-0.05	-0.03	-0.03	-0.02	-0.05	-0.05	-0.05
1080	-0.03	-0.03	-0.04	-0.03	-0.03	-0.02	-0.04	-0.04	-0.04
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1080	-0.03	-0.03	-0.04	-0.03	-0.03	-0.02	-0.04	-0.04	-0.04
1092	-0.02	-0.02	-0.04	-0.02	-0.02	-0.02	-0.04	-0.04	-0.04
1104	-0.02	-0.02	-0.03	-0.02	-0.02	-0.02	-0.03	-0.03	-0.03
1116	-0.02	-0.02	-0.03	-0.02	-0.02	-0.01	-0.03	-0.03	-0.03
1128	-0.01	-0.01	-0.02	-0.01	-0.01	-0.01	-0.02	-0.02	-0.02
1140	-0.01	-0.01	-0.02	-0.01	-0.01	-0.01	-0.02	-0.02	-0.02
1152	-0.01	-0.01	-0.02	-0.01	-0.01	-0.01	-0.02	-0.02	-0.02
1164	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01	-0.01
1176	-0.01	-0.01	-0.01	-0.01	-0.01	0.00	-0.01	-0.01	-0.01
1188	0.00	0.00	-0.01	0.00	0.00	0.00	-0.01	-0.01	-0.01
1200	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1200	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1212	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1224	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1236	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1248	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1260	0.00	0.00	0.01	0.00	0.00	0.00	0.01	0.01	0.01
1272	0.00	0.00	0.01	0.00	0.00	0.00	0.01	0.01	0.01
1284	0.01	0.01	0.01	0.01	0.01	0.00	0.01	0.01	0.01
1296	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01

1308	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
1320	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1320	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
1332	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
1344	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
1356	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
"s"=	1	2	3	4	5	6	7	8	9
z									
1368	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01	0.01
1380	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
1392	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
1404	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
1416	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
1428	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02
1440	0.01	0.01	0.02	0.01	0.01	0.01	0.02	0.02	0.02

Table of Warping Shear Stress along "z" and "s"

Warping Shear Stress at "s" (- E*Sw*θ'''/ t)										
Sw	0.0	84.13	185.15	84.13	0.0	0.0	0.0	376.54	0.0	
"s"=	1	2	3	4	5	6	7	8	9	
z										
	0	0	0.03	0.04	0.03	0	0	0	0.09	0
7.65	0	0.03	0.04	0.03	0	0	0	0.09	0	
15.3	0	0.03	0.04	0.03	0	0	0	0.09	0	
22.95	0	0.03	0.05	0.03	0	0	0	0.09	0	
30.6	0	0.03	0.05	0.03	0	0	0	0.09	0	
38.25	0	0.03	0.05	0.03	0	0	0	0.09	0	
45.9	0	0.03	0.05	0.03	0	0	0	0.10	0	
53.55	0	0.03	0.05	0.03	0	0	0	0.10	0	
61.2	0	0.04	0.05	0.04	0	0	0	0.10	0	
68.85	0	0.04	0.05	0.04	0	0	0	0.10	0	
76.5	0	0.04	0.05	0.04	0	0	0	0.11	0	
76.5	0	0.04	0.05	0.04	0	0	0	0.11	0	
84.15	0	0.04	0.05	0.04	0	0	0	0.11	0	
91.8	0	0.04	0.05	0.04	0	0	0	0.11	0	
99.45	0	0.04	0.06	0.04	0	0	0	0.12	0	
107.1	0	0.04	0.06	0.04	0	0	0	0.12	0	
114.75	0	0.04	0.06	0.04	0	0	0	0.12	0	
122.4	0	0.05	0.06	0.05	0	0	0	0.13	0	
130.05	0	0.05	0.07	0.05	0	0	0	0.13	0	
137.7	0	0.05	0.07	0.05	0	0	0	0.14	0	
145.35	0	0.05	0.07	0.05	0	0	0	0.15	0	
153	0	0.05	0.07	0.05	0	0	0	0.15	0	
153	0	-0.04	-0.05	-0.04	0	0	0	-0.10	0	
165.6	0	-0.03	-0.05	-0.03	0	0	0	-0.09	0	
178.2	0	-0.03	-0.04	-0.03	0	0	0	-0.08	0	
190.8	0	-0.03	-0.04	-0.03	0	0	0	-0.08	0	
203.4	0	-0.02	-0.03	-0.02	0	0	0	-0.07	0	
216	0	-0.02	-0.03	-0.02	0	0	0	-0.06	0	
228.6	0	-0.02	-0.03	-0.02	0	0	0	-0.05	0	
228.6	0	-0.02	-0.03	-0.02	0	0	0	-0.05	0	
"s"=	1	2	3	4	5	6	7	8	9	
z										
241.2	0	-0.02	-0.02	-0.02	0	0	0	-0.04	0	
253.8	0	-0.01	-0.02	-0.01	0	0	0	-0.04	0	
266.4	0	-0.01	-0.02	-0.01	0	0	0	-0.03	0	
279	0	-0.01	-0.01	-0.01	0	0	0	-0.03	0	
279	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0	
287.1	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0	

295.2	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0
303.3	0	-0.10	-0.13	-0.10	0	0	0	-0.27	0
311.4	0	-0.10	-0.13	-0.10	0	0	0	-0.27	0
319.5	0	-0.10	-0.13	-0.10	0	0	0	-0.27	0
327.6	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0
335.7	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0
343.8	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0
351.9	0	-0.10	-0.14	-0.10	0	0	0	-0.28	0
360	0	-0.10	-0.14	-0.10	0	0	0	-0.29	0
360	0	0.04	0.05	0.04	0	0	0	0.10	0
372	0	0.03	0.04	0.03	0	0	0	0.09	0
384	0	0.03	0.04	0.03	0	0	0	0.08	0
396	0	0.03	0.04	0.03	0	0	0	0.08	0
408	0	0.03	0.04	0.03	0	0	0	0.07	0
420	0	0.02	0.03	0.02	0	0	0	0.07	0
432	0	0.02	0.03	0.02	0	0	0	0.06	0
444	0	0.02	0.03	0.02	0	0	0	0.06	0
456	0	0.02	0.03	0.02	0	0	0	0.05	0
468	0	0.02	0.02	0.02	0	0	0	0.05	0
480	0	0.02	0.02	0.02	0	0	0	0.05	0
480	0	0.02	0.02	0.02	0	0	0	0.05	0
492	0	0.02	0.02	0.02	0	0	0	0.04	0
504	0	0.01	0.02	0.01	0	0	0	0.04	0
516	0	0.01	0.02	0.01	0	0	0	0.04	0
528	0	0.01	0.02	0.01	0	0	0	0.04	0
540	0	0.01	0.02	0.01	0	0	0	0.03	0
552	0	0.01	0.02	0.01	0	0	0	0.03	0
564	0	0.01	0.02	0.01	0	0	0	0.03	0
576	0	0.01	0.01	0.01	0	0	0	0.03	0
588	0	0.01	0.01	0.01	0	0	0	0.03	0
600	0	0.01	0.01	0.01	0	0	0	0.03	0
600	0	0.01	0.01	0.01	0	0	0	0.03	0
612	0	0.01	0.01	0.01	0	0	0	0.03	0
624	0	0.01	0.01	0.01	0	0	0	0.03	0
636	0	0.01	0.01	0.01	0	0	0	0.03	0
636	0	0.01	0.01	0.01	0	0	0	0.03	0
“s”=	1	2	3	4	5	6	7	8	9
z									
648	0	0.01	0.01	0.01	0	0	0	0.03	0
660	0	0.01	0.01	0.01	0	0	0	0.03	0
672	0	0.01	0.01	0.01	0	0	0	0.03	0
684	0	0.01	0.01	0.01	0	0	0	0.03	0

696	0	0.01	0.01	0.01	0	0	0	0.03	0
708	0	0.01	0.02	0.01	0	0	0	0.03	0
720	0	0.01	0.02	0.01	0	0	0	0.03	0
720	0	-0.01	-0.01	-0.01	0	0	0	-0.02	0
732	0	-0.01	-0.01	-0.01	0	0	0	-0.02	0
744	0	-0.01	-0.01	-0.01	0	0	0	-0.02	0
756	0	-0.01	-0.01	-0.01	0	0	0	-0.02	0
768	0	0.00	-0.01	0.00	0	0	0	-0.01	0
780	0	0.00	-0.01	0.00	0	0	0	-0.01	0
792	0	0.00	-0.01	0.00	0	0	0	-0.01	0
804	0	0.00	-0.01	0.00	0	0	0	-0.01	0
816	0	0.00	-0.01	0.00	0	0	0	-0.01	0
828	0	0.00	0.00	0.00	0	0	0	-0.01	0
840	0	0.00	0.00	0.00	0	0	0	-0.01	0
840	0	0.00	0.00	0.00	0	0	0	-0.01	0
852	0	0.00	0.00	0.00	0	0	0	-0.01	0
864	0	0.00	0.00	0.00	0	0	0	-0.01	0
876	0	0.00	0.00	0.00	0	0	0	-0.01	0
888	0	0.00	0.00	0.00	0	0	0	-0.01	0
900	0	0.00	0.00	0.00	0	0	0	-0.01	0
912	0	0.00	0.00	0.00	0	0	0	-0.01	0
924	0	0.00	0.00	0.00	0	0	0	-0.01	0
936	0	0.00	0.00	0.00	0	0	0	-0.01	0
948	0	0.00	0.00	0.00	0	0	0	-0.01	0
960	0	0.00	0.00	0.00	0	0	0	-0.01	0
960	0	0.00	0.00	0.00	0	0	0	-0.01	0
972	0	0.00	0.00	0.00	0	0	0	-0.01	0
984	0	0.00	0.00	0.00	0	0	0	-0.01	0
996	0	0.00	0.00	0.00	0	0	0	-0.01	0
1008	0	0.00	0.00	0.00	0	0	0	-0.01	0
1020	0	0.00	0.00	0.00	0	0	0	-0.01	0
1032	0	0.00	0.00	0.00	0	0	0	-0.01	0
1044	0	0.00	0.00	0.00	0	0	0	-0.01	0
1056	0	0.00	0.00	0.00	0	0	0	-0.01	0
1068	0	0.00	0.00	0.00	0	0	0	-0.01	0
1080	0	0.00	0.00	0.00	0	0	0	-0.01	0
1080	0	0.00	0.00	0.00	0	0	0	0.00	0
1092	0	0.00	0.00	0.00	0	0	0	0.00	0
“s”=	1	2	3	4	5	6	7	8	9
z									
1104	0	0.00	0.00	0.00	0	0	0	0.00	0



1116	0	0.00	0.00	0.00	0	0	0	0.00	0
1128	0	0.00	0.00	0.00	0	0	0	0.00	0
1140	0	0.00	0.00	0.00	0	0	0	0.00	0
1152	0	0.00	0.00	0.00	0	0	0	0.00	0
1164	0	0.00	0.00	0.00	0	0	0	0.00	0
1176	0	0.00	0.00	0.00	0	0	0	0.00	0
1188	0	0.00	0.00	0.00	0	0	0	0.00	0
1200	0	0.00	0.00	0.00	0	0	0	0.00	0
	0	0.00	0.00	0.00	0	0	0	0.00	0
1200	0	0.00	0.00	0.00	0	0	0	0.00	0
1212	0	0.00	0.00	0.00	0	0	0	0.00	0
1224	0	0.00	0.00	0.00	0	0	0	0.00	0
1236	0	0.00	0.00	0.00	0	0	0	0.00	0
1248	0	0.00	0.00	0.00	0	0	0	0.00	0
1260	0	0.00	0.00	0.00	0	0	0	0.00	0
1272	0	0.00	0.00	0.00	0	0	0	0.00	0
1284	0	0.00	0.00	0.00	0	0	0	0.00	0
1296	0	0.00	0.00	0.00	0	0	0	0.00	0
1308	0	0.00	0.00	0.00	0	0	0	0.00	0
1320	0	0.00	0.00	0.00	0	0	0	0.00	0
1320	0	0.00	0.00	0.00	0	0	0	0.00	0
1332	0	0.00	0.00	0.00	0	0	0	0.00	0
1344	0	0.00	0.00	0.00	0	0	0	0.00	0
1356	0	0.00	0.00	0.00	0	0	0	0.00	0
1368	0	0.00	0.00	0.00	0	0	0	0.00	0
1380	0	0.00	0.00	0.00	0	0	0	0.00	0
1392	0	0.00	0.00	0.00	0	0	0	0.00	0
1404	0	0.00	0.00	0.00	0	0	0	0.00	0
1416	0	0.00	0.00	0.00	0	0	0	0.00	0
1428	0	0.00	0.00	0.00	0	0	0	0.00	0
1440	0	0.00	0.00	0.00	0	0	0	0.00	0

Table of Warping Normal Stresses along "z" and "s"

Warping Normal Stresses, $E \cdot W_n \cdot \theta''$									
Wn	53.66	25.26	0.00	- 25.26	- 53.66	0.00	143.45	0	-144
"s"=	1	2	3	4	5	6	7	8	9
z									
	0	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
7.65	-0.10	-0.05	0.00	0.05	0.10	0.00	-0.26	0.00	0.27
15.3	-0.20	-0.09	0.00	0.09	0.20	0.00	-0.53	0.00	0.53
22.95	-0.30	-0.14	0.00	0.14	0.30	0.00	-0.80	0.00	0.80
30.6	-0.40	-0.19	0.00	0.19	0.40	0.00	-1.07	0.00	1.07
38.25	-0.50	-0.24	0.00	0.24	0.50	0.00	-1.34	0.00	1.34
"s"=	1	2	3	4	5	6	7	8	9

z									
45.9	-0.61	-0.28	0.00	0.28	0.61	0.00	-1.62	0.00	1.62
53.55	-0.71	-0.33	0.00	0.33	0.71	0.00	-1.90	0.00	1.90
61.2	-0.82	-0.39	0.00	0.39	0.82	0.00	-2.19	0.00	2.19
68.85	-0.93	-0.44	0.00	0.44	0.93	0.00	-2.48	0.00	2.48
76.5	-1.04	-0.49	0.00	0.49	1.04	0.00	-2.78	0.00	2.79
76.5	-1.04	-0.49	0.00	0.49	1.04	0.00	-2.78	0.00	2.79
84.15	-1.16	-0.55	0.00	0.55	1.16	0.00	-3.10	0.00	3.10
91.8	-1.28	-0.60	0.00	0.60	1.28	0.00	-3.42	0.00	3.42
99.45	-1.40	-0.66	0.00	0.66	1.40	0.00	-3.75	0.00	3.75
107.1	-1.53	-0.72	0.00	0.72	1.53	0.00	-4.09	0.00	4.09
114.75	-1.66	-0.78	0.00	0.78	1.66	0.00	-4.44	0.00	4.45
122.4	-1.80	-0.85	0.00	0.85	1.80	0.00	-4.81	0.00	4.81
130.05	-1.94	-0.91	0.00	0.91	1.94	0.00	-5.19	0.00	5.20
137.7	-2.09	-0.98	0.00	0.98	2.09	0.00	-5.59	0.00	5.60
145.35	-2.25	-1.06	0.00	1.06	2.25	0.00	-6.01	0.00	6.01
153	-2.41	-1.13	0.00	1.13	2.41	0.00	-6.44	0.00	6.44
153	-2.41	-1.13	0.00	1.13	2.41	0.00	-6.44	0.00	6.44
165.6	-2.23	-1.05	0.00	1.05	2.23	0.00	-5.96	0.00	5.97
178.2	-2.07	-0.97	0.00	0.97	2.07	0.00	-5.53	0.00	5.54
190.8	-1.93	-0.91	0.00	0.91	1.93	0.00	-5.15	0.00	5.15
203.4	-1.80	-0.85	0.00	0.85	1.80	0.00	-4.81	0.00	4.81
216	-1.69	-0.79	0.00	0.79	1.69	0.00	-4.51	0.00	4.51
228.6	-1.59	-0.75	0.00	0.75	1.59	0.00	-4.25	0.00	4.25
241.2	-1.50	-0.71	0.00	0.71	1.50	0.00	-4.02	0.00	4.02
253.8	-1.43	-0.67	0.00	0.67	1.43	0.00	-3.83	0.00	3.83
266.4	-1.37	-0.64	0.00	0.64	1.37	0.00	-3.66	0.00	3.67
279	-1.32	-0.62	0.00	0.62	1.32	0.00	-3.53	0.00	3.53
279	-1.32	-0.62	0.00	0.62	1.32	0.00	-3.53	0.00	3.53
287.1	-1.00	-0.47	0.00	0.47	1.00	0.00	-2.67	0.00	2.67
295.2	-0.68	-0.32	0.00	0.32	0.68	0.00	-1.81	0.00	1.81
303.3	-0.36	-0.17	0.00	0.17	0.36	0.00	-0.96	0.00	0.96
311.4	-0.04	-0.02	0.00	0.02	0.04	0.00	-0.12	0.00	0.12
319.5	0.27	0.13	0.00	-0.13	-0.27	0.00	0.73	0.00	-0.73
327.6	0.59	0.28	0.00	-0.28	-0.59	0.00	1.57	0.00	-1.58
335.7	0.91	0.43	0.00	-0.43	-0.91	0.00	2.43	0.00	-2.43
343.8	1.23	0.58	0.00	-0.58	-1.23	0.00	3.29	0.00	-3.29
351.9	1.56	0.73	0.00	-0.73	-1.56	0.00	4.16	0.00	-4.16
360	1.89	0.89	0.00	-0.89	-1.89	0.00	5.05	0.00	-5.05
360	1.89	0.89	0.00	-0.89	-1.89	0.00	5.05	0.00	-5.05
372	1.72	0.81	0.00	-0.81	-1.72	0.00	4.61	0.00	-4.61

384	1.57	0.74	0.00	-0.74	-1.57	0.00	4.21	0.00	-4.21
“s”=	1	2	3	4	5	6	7	8	9
z									
396	1.44	0.68	0.00	-0.68	-1.44	0.00	3.84	0.00	-3.84
408	1.31	0.62	0.00	-0.62	-1.31	0.00	3.50	0.00	-3.50
420	1.19	0.56	0.00	-0.56	-1.19	0.00	3.18	0.00	-3.18
432	1.08	0.51	0.00	-0.51	-1.08	0.00	2.89	0.00	-2.89
444	0.98	0.46	0.00	-0.46	-0.98	0.00	2.62	0.00	-2.62
456	0.89	0.42	0.00	-0.42	-0.89	0.00	2.37	0.00	-2.37
468	0.80	0.38	0.00	-0.38	-0.80	0.00	2.14	0.00	-2.14
480	0.72	0.34	0.00	-0.34	-0.72	0.00	1.92	0.00	-1.92
480	0.72	0.34	0.00	-0.34	-0.72	0.00	1.92	0.00	-1.92
492	0.64	0.30	0.00	-0.30	-0.64	0.00	1.72	0.00	-1.72
504	0.57	0.27	0.00	-0.27	-0.57	0.00	1.53	0.00	-1.53
516	0.51	0.24	0.00	-0.24	-0.51	0.00	1.36	0.00	-1.36
528	0.44	0.21	0.00	-0.21	-0.44	0.00	1.19	0.00	-1.19
540	0.39	0.18	0.00	-0.18	-0.39	0.00	1.03	0.00	-1.03
552	0.33	0.16	0.00	-0.16	-0.33	0.00	0.88	0.00	-0.88
564	0.28	0.13	0.00	-0.13	-0.28	0.00	0.74	0.00	-0.74
576	0.22	0.11	0.00	-0.11	-0.22	0.00	0.60	0.00	-0.60
588	0.17	0.08	0.00	-0.08	-0.17	0.00	0.47	0.00	-0.47
600	0.13	0.06	0.00	-0.06	-0.13	0.00	0.34	0.00	-0.34
600	0.13	0.06	0.00	-0.06	-0.13	0.00	0.34	0.00	-0.34
612	0.08	0.04	0.00	-0.04	-0.08	0.00	0.21	0.00	-0.21
624	0.03	0.01	0.00	-0.01	-0.03	0.00	0.08	0.00	-0.08
636	-0.02	-0.01	0.00	0.01	0.02	0.00	-0.04	0.00	0.04
648	-0.06	-0.03	0.00	0.03	0.06	0.00	-0.17	0.00	0.17
660	-0.11	-0.05	0.00	0.05	0.11	0.00	-0.29	0.00	0.30
672	-0.16	-0.07	0.00	0.07	0.16	0.00	-0.42	0.00	0.42
684	-0.21	-0.10	0.00	0.10	0.21	0.00	-0.56	0.00	0.56
696	-0.26	-0.12	0.00	0.12	0.26	0.00	-0.69	0.00	0.69
708	-0.31	-0.15	0.00	0.15	0.31	0.00	-0.83	0.00	0.84
720	-0.37	-0.17	0.00	0.17	0.37	0.00	-0.98	0.00	0.98
720	-0.37	-0.17	0.00	0.17	0.37	0.00	-0.98	0.00	0.98
732	-0.34	-0.16	0.00	0.16	0.34	0.00	-0.90	0.00	0.90
744	-0.31	-0.14	0.00	0.14	0.31	0.00	-0.82	0.00	0.82
756	-0.28	-0.13	0.00	0.13	0.28	0.00	-0.75	0.00	0.75
768	-0.25	-0.12	0.00	0.12	0.25	0.00	-0.68	0.00	0.68
780	-0.23	-0.11	0.00	0.11	0.23	0.00	-0.62	0.00	0.62
792	-0.21	-0.10	0.00	0.10	0.21	0.00	-0.56	0.00	0.56
804	-0.19	-0.09	0.00	0.09	0.19	0.00	-0.51	0.00	0.51
816	-0.17	-0.08	0.00	0.08	0.17	0.00	-0.46	0.00	0.46

828	-0.16	-0.07	0.00	0.07	0.16	0.00	-0.42	0.00	0.42
840	-0.14	-0.07	0.00	0.07	0.14	0.00	-0.38	0.00	0.38
840	-0.14	-0.07	0.00	0.07	0.14	0.00	-0.38	0.00	0.38
“s”=	1	2	3	4	5	6	7	8	9
z									
852	-0.13	-0.06	0.00	0.06	0.13	0.00	-0.34	0.00	0.34
864	-0.11	-0.05	0.00	0.05	0.11	0.00	-0.30	0.00	0.30
876	-0.10	-0.05	0.00	0.05	0.10	0.00	-0.27	0.00	0.27
888	-0.09	-0.04	0.00	0.04	0.09	0.00	-0.23	0.00	0.23
900	-0.08	-0.04	0.00	0.04	0.08	0.00	-0.20	0.00	0.20
912	-0.06	-0.03	0.00	0.03	0.06	0.00	-0.17	0.00	0.17
924	-0.05	-0.03	0.00	0.03	0.05	0.00	-0.15	0.00	0.15
936	-0.04	-0.02	0.00	0.02	0.04	0.00	-0.12	0.00	0.12
948	-0.03	-0.02	0.00	0.02	0.03	0.00	-0.09	0.00	0.09
960	-0.03	-0.01	0.00	0.01	0.03	0.00	-0.07	0.00	0.07
960	-0.03	-0.01	0.00	0.01	0.03	0.00	-0.07	0.00	0.07
972	-0.02	-0.01	0.00	0.01	0.02	0.00	-0.04	0.00	0.04
984	-0.01	0.00	0.00	0.00	0.01	0.00	-0.02	0.00	0.02
996	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1008	0.01	0.01	0.00	-0.01	-0.01	0.00	0.03	0.00	-0.03
1020	0.02	0.01	0.00	-0.01	-0.02	0.00	0.05	0.00	-0.05
1032	0.03	0.01	0.00	-0.01	-0.03	0.00	0.08	0.00	-0.08
1044	0.04	0.02	0.00	-0.02	-0.04	0.00	0.10	0.00	-0.10
1056	0.05	0.02	0.00	-0.02	-0.05	0.00	0.13	0.00	-0.13
1068	0.06	0.03	0.00	-0.03	-0.06	0.00	0.16	0.00	-0.16
1080	0.07	0.03	0.00	-0.03	-0.07	0.00	0.18	0.00	-0.18
1080	0.07	0.03	0.00	-0.03	-0.07	0.00	0.18	0.00	-0.18
1092	0.06	0.03	0.00	-0.03	-0.06	0.00	0.17	0.00	-0.17
1104	0.06	0.03	0.00	-0.03	-0.06	0.00	0.15	0.00	-0.15
1116	0.05	0.02	0.00	-0.02	-0.05	0.00	0.14	0.00	-0.14
1128	0.05	0.02	0.00	-0.02	-0.05	0.00	0.13	0.00	-0.13
1140	0.04	0.02	0.00	-0.02	-0.04	0.00	0.12	0.00	-0.12
1152	0.04	0.02	0.00	-0.02	-0.04	0.00	0.11	0.00	-0.11
1164	0.04	0.02	0.00	-0.02	-0.04	0.00	0.10	0.00	-0.10
1176	0.03	0.02	0.00	-0.02	-0.03	0.00	0.09	0.00	-0.09
1188	0.03	0.01	0.00	-0.01	-0.03	0.00	0.08	0.00	-0.08
1200	0.03	0.01	0.00	-0.01	-0.03	0.00	0.08	0.00	-0.08
1200	0.03	0.01	0.00	-0.01	-0.03	0.00	0.08	0.00	-0.08
1212	0.03	0.01	0.00	-0.01	-0.03	0.00	0.07	0.00	-0.07
1224	0.02	0.01	0.00	-0.01	-0.02	0.00	0.06	0.00	-0.06
1236	0.02	0.01	0.00	-0.01	-0.02	0.00	0.06	0.00	-0.06
1248	0.02	0.01	0.00	-0.01	-0.02	0.00	0.05	0.00	-0.05

1260	0.02	0.01	0.00	-0.01	-0.02	0.00	0.05	0.00	-0.05
1272	0.02	0.01	0.00	-0.01	-0.02	0.00	0.04	0.00	-0.04
1284	0.01	0.01	0.00	-0.01	-0.01	0.00	0.04	0.00	-0.04
1296	0.01	0.01	0.00	-0.01	-0.01	0.00	0.03	0.00	-0.03
1308	0.01	0.01	0.00	-0.01	-0.01	0.00	0.03	0.00	-0.03
“s”=	1	2	3	4	5	6	7	8	9
z									
1320	0.01	0.00	0.00	0.00	-0.01	0.00	0.03	0.00	-0.03
1320	0.01	0.00	0.00	0.00	-0.01	0.00	0.03	0.00	-0.03
1332	0.01	0.00	0.00	0.00	-0.01	0.00	0.02	0.00	-0.02
1344	0.01	0.00	0.00	0.00	-0.01	0.00	0.02	0.00	-0.02
1356	0.01	0.00	0.00	0.00	-0.01	0.00	0.02	0.00	-0.02
1368	0.01	0.00	0.00	0.00	-0.01	0.00	0.01	0.00	-0.02
1380	0.00	0.00	0.00	0.00	0.00	0.00	0.01	0.00	-0.01
1392	0.00	0.00	0.00	0.00	0.00	0.00	0.01	0.00	-0.01
1404	0.00	0.00	0.00	0.00	0.00	0.00	0.01	0.00	-0.01
1416	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1428	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00
1440	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00

## BIBLIOGRAPHY

- Åkesson, B. Å., Discussion to Warping Moment Distributions by J. S. Medwadowski. Journal of Structural Engineering, ASCE 113(1); 169-70, 1987.
- Bleich, F., Buckling Strength of Metal Structures, Second Edition. McGraw-Hill Book Co., Inc.: New York, 1952.
- Boothby, T. E., The Applications of Flexural Methods to Torsional Analysis of Thin-Walled Open Sections. Engineering Journal, AISC, 21 (4), Chicago, IL, 1984.
- Bresler, B., Lin, T. Y., Design of Steel Structures. Second Edition. John Wiley & Sons, Inc.: New York, London, 1963.
- Carlson, A., Hannauer, G., Carey, T., Holsberg, P. J., Handbook of Analog Computation, Second Edition. Electronic Associates, Inc.: Princeton, New Jersey, 1967.
- Chaudbary, A. B., Generalized Stiffness Matrix for Thin Walled Beams. Journal of the Structural Division, ASCE Proceedings, 108 (3); 1982.
- Cisek, P., Beyond the Computer Metaphor: Behaviour as Interaction. Journal of Consciousness Studies, 6 (11-12); 125-42, 1999.
- Cross H., Analysis of Continuous Frames by Distributing Fixed Moments. ASCE Proceedings, 57; 919-928, 1930.
- Cross H., Analysis of Continuous Frames by Distributing Fixed Moments. ASCE Transactions, 96; 1-10, 1932.
- Cross H., Continuous Frames of Reinforced Concrete. Thirteenth Printing. John Wiley & Sons, Inc.: New York, 1954.
- Dabrowski, R., Equations of Bending and Torsion of a Curved Thin-Walled Bar with Asymmetric Cross-Section. Archives of Mechanics, 12 (5-6), 1960.
- Den Hartog, J. P., Advanced Strength of Materials. McGraw-Hill Book Company, Inc.: USA, 1952.
- Deschapelles, B., Discussion to Frame Analysis of Shear Wall Cores by Iain A. MacLeod. Journal of Structural Division, ASCE, 104 (ST9), 1978.
- Deschapelles, B., A Hybrid 8 DOF Plane Stress Element for the Analysis of Non-Planar Shear Walls. paper presented at the Tall Building Council Meeting: New Orleans, La, 1984.

- Deschappelles, B., Discussion to The Applications of Flexural Methods to Torsional Analysis of Thin-Walled Open Sections by T.E. Boothby. Engineering Journal, AISC, 22(4); 176-77, 1985.
- Deschappelles, B., Discussion to Warping Moment Distributions by J.S. Medwadowski. Journal of Structural Engineering, ASCE 113(1), 170-72, 1987.
- Deschappelles, B., Refined Beam Finite Element with a Non Nodal Degree of Freedom. 15<sup>th</sup> ASCE Engineering Mechanics Conference, Columbia University: New York, 2002.
- Deschappelles, B., Class Notes and Files. Puerto Rico, 2009-2011.
- Dvorkin, E. N., Celentano, D. et al., A Vlasov Beam Element. Instituto de Materiales y Estructuras, Facultad de Ingeniería, Universidad de Buenos Aires, 1988.
- Feodosiev, V. I., Resistencia de Materiales. Editorial MIR: URSS, 1972.
- Gere, J. M., Moment Distribution. D. Van Nostrand Company, Inc.: Princeton, N.J., 1963.
- Gjelsvik, A., Guide to Stability Design Criteria for Metal Structures. Fourth Edition. John Wiley & Sons, Inc.: New York, N. Y., 1981.
- Goodier, J. N., Torsional and Flexural Buckling of Bars of Thin-Walled Open Section under Compressive and Bending Loads. Journal of Applied Mechanics, ASME 9 (9); A103-A107, 1942.
- Grinter L. E., Discussion to paper by H. Cross. ASCE Transactions, 96:11–20, 1932.
- Grinter L. E., Wind stress analysis simplified. ASCE Proceedings, 59:3–27, 1933
- Heins, C. P., Seaburg, P. A., Steel Design File-Torsion Analysis of Rolled Steel Sections. Bethlehem Steel Corporation: Bethlehem, Pennsylvania, 1963.
- Heins, C. P., Bending and Torsional Design in Structural Members. Lexington Books: Lexington, Massachusetts, 1975.
- Heins, C. P., Hall, D. H., Designer's Guide to Steel Box Girder Bridges. First Edition. Bethlehem Steel Corporation, Bethlehem, Pennsylvania, 1981.
- Herbsleb, J. D., Metaphorical Representation in Collaborative Software Engineering. Bell Laboratories, Lucent Technologies: San Francisco, CA, USA, 1999.
- Hetényi, M., Beams on Elastic Foundation. Six Printing. Ann Harbor, The University of Michigan Press: Michigan, 1961.

- Hetényi, M., Beams and Plates on Elastic Foundations and Related Problems. Applied Mechanics Reviews. American Society of Mechanical Engineers, 19 (2); 95-102, 1966.
- Hetényi, M., Handbook of Experimental Stress Analysis. Chapter 16 on Analogies by Mindlin, R. D. and Salvadori M. G., Six Printing. John Wiley and Sons: New York, London, Sydney, 1966.
- Hoogenboom P.C.J., Borgart A., Method for Including Restrained Warping in Traditional Frame Analyses. HERON, 50 (1) 55-68, 2005. Erratum in HERON, 50 (3); 185, 2005.
- Hrennikoff, A., Solutions of Problems in Elasticity by the Framework Method. Journal of Applied Mechanics, 8 (4) 169-175, 1941.
- Hsu, Y. T., et al., EBEF Method for Distortional Analysis of Steel Box Girder Bridges. Journal of Structural Engineering, 121 (3), 1995.
- James, M. L., Smith, G. M., Wolforf, J. C., Analog Computer Simulation of Engineering Systems. Second Edition. Intext Educational Publishers, College division of Intext: Scranton, San Francisco, Toronto, London, 1971.
- Kron, G., Tensor Analysis of Networks. John Wiley & Sons, Inc., London, 1939.
- Lighthfoot, E., Moment Distribution, a Rapid Method of analysis for Rigid-Jointed Structures. John Wiley and Sons: New York, 1961.
- Lundquist E. E., Fligg C. M., A Theory for Primary Failure of Straight Centrally Loaded Columns. N.A.C.A. Technical Report No. 582, 1937.
- McHenry, D., A Lattice Analogy for the Solution of Stress Problems. Technical Memorandum No. 624, U. S. Bureau of Reclamation, 1942.
- McHenry, D., The Numerical Solution of Two-Dimensional Problems in Elasticity by a Lattice Analogy. Journal of the Institution of Civil Engineers, No. 2. 1943-1944, 1943.
- Medwadowski, J. S., Warping Moment Distribution. Journal of Structural Engineering, Vol. 111 No.2, ASCE, 1985.
- Mindlin, R. D., and Salvadori, M. G., Analogies, Chapter 16, Handbook on Experimental Stress Analysis edited by M. Hetényi, Sixth printing. John Wiley and Sons: New York, London, Sydney, 1966.



- Miranda, C., Nair, K., Finite Beams on Elastic Foundation. Journal of the Structural Division, ASCE Proceedings, 92 (ST2), 1966.
- Obrébski, J. B., Some Own Approaches to Computer Aided Design of Complicated Bar Structures. Faculty of Civil Engineering, Warsaw University of Technology, Al. Armii Ludowej 16, r. 143: Warsaw, Poland, 2005.
- Ojalvo, M., Wagner Hypothesis in Beam and Column Theory. Journal of the Engineering Mechanics Division, ASCE Proceedings, 107 (EM4), 1981.
- Ojalvo, M., Closure of Wagner Hypothesis in Beam and Column Theory by Ojalvo. Journal of the Engineering Mechanics Division, ASCE Proceedings, 108 (3), 1982
- Ojalvo, M., Thin-Walled bars with Open Profiles. The Olive Press: Columbus, Ohio, 1990.
- Pettersson, O., Method of successive approximations for design of continuous I beams submitted to torsion. Publications of the International Association for Bridge and Structural Engineering, 15; 167-186, 1955.
- Rekach, V. G., Problemas de la Teoría de Elasticidad. Editorial MIR: URSS, 1978.
- Roark, R., Young, W.C., Roark's Formulas of Stress & Strain, Fifth Edition. McGraw-Hill: New York, USA, 1982.
- Rhodes, J., Walker, A. C., Developments in Thin Walled Structures-2. Elsevier Applied Science Publishers: London and New York, 1984.
- Rhodes, J., Walker, A. C., Editors, Developments in Thin Walled Structures-3. Elsevier Applied Science Publishers: London and New York, 1987.
- Saadeé, K., Finite Element Modeling of Shear in Thin Walled Beams with a Single Warping Function. Université Libre De Bruxelles, Faculté des Sciences Appliquées, Services des Milieux Continus & Génie Civil, Dissertation originale présentée en vue de l'obtention du grade de docteur en Sciences Appliquées, 2004-2005.
- Seaburg, P. A., Carter, C. J., Torsional Analysis of Structural Steel Members. Second Printing. AISC: USA, 1997.
- Samuelsson, A., and Zienkiewicz, O. C., History of the Stiffness Method. International Journal for Numerical Methods in Engineering; 67:149–157, 2006.
- Southwell R. V., Stress calculation in frameworks by the method of 'systematic relaxation of constraints'. Part I & II. Proceedings of the Royal Society of London, Series A, 151:56–95, 1935.

- Terzaghi, K., Peck, R.B., Mecánica de suelos en la ingeniería práctica. Segunda edición. Librería “El Ateneo” Editorial: Argentina, Buenos Aires, 1973.
- Trahair, N. S., et al., The behavior and Design of Steel Structures to EC3. Fourth Edition. Taylor & Francis: London and New York, 2008.
- Timoshenko, S., Theory of Elasticity, First Edition, Tenth Impression. McGraw-Hill Book Company, Inc.: New York and London, 1934.
- Timoshenko, S., Theory of Bending, Torsion and Buckling of Thin-Walled Members of Open Cross-Section. Journal of the Franklin Institute, March/April/May, Philadelphia, PA, 1945.
- Timoshenko, S., History of Strength of Materials. McGraw-Hill Book Company, Inc.: N.Y., 1953.
- Timoshenko, S., Resistencia de Materiales, Segunda Parte. Espasa-Calpe, S.A.: Madrid, España, 1978.
- Timoshenko, S., Gere, J., Theory of Elastic Stability. Unabridged republication of the second edition. Dover Edition: New York, 2009.
- Timoshenko, S., and Goodier, J. N., Theory of Elasticity, Second Edition. Engineering Society Monographs. McGraw-Hill Book Company, Inc.: New York, Toronto, London, 1951.
- Ugural, A.C., Fenster, S.K., Advanced Strength and Applied Elasticity. Second SI Edition. Prentice-Hall Inc.: New Jersey, 1987.
- Vlasov, V. Z., Thin-Walled Elastic Beams. Second edition, NSF, Washington D. C. and the Department of Commerce, USA, by the Israel Program for Scientific Translations: Jerusalem, 1961.
- Wagner, H., Torsion and Buckling of Open Sections. National Advisory Committee for Aeronautics. Technical Memorandum No. 807, Washington, 1936, Translated from Technische Hochschule, 1929.
- Wilkinson, J. H., Rounding Errors in Algebraic Processes. Prentice Hall, Inc., Englewood Cliffs.: NJ, 1963.
- Wright, R. N., Abdel-Samad, S.R., BEF Analogy for Analysis of Box Girders. Journal of the Structural Division, ASCE, 94 (ST 7), Proc. Paper 6025, 1968.